MAN 51/60DF

Project Guide – Marine Four-stroke dual fuel engine compliant with IMO Tier III

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MAN 51/60DF IMO Tier III Project Guide – Marine

Four-stroke dual fuel engine

MAN Energy Solutions SE
86224 Augsburg
GERMANY

Phone +49 (0) 821 322-0
Fax +49 (0) 821 322-3382
primeserv-aug@man-es.com
https://primeserv.man-es.com/

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1 Introduction

1.1 Medium-speed propulsion engine programme

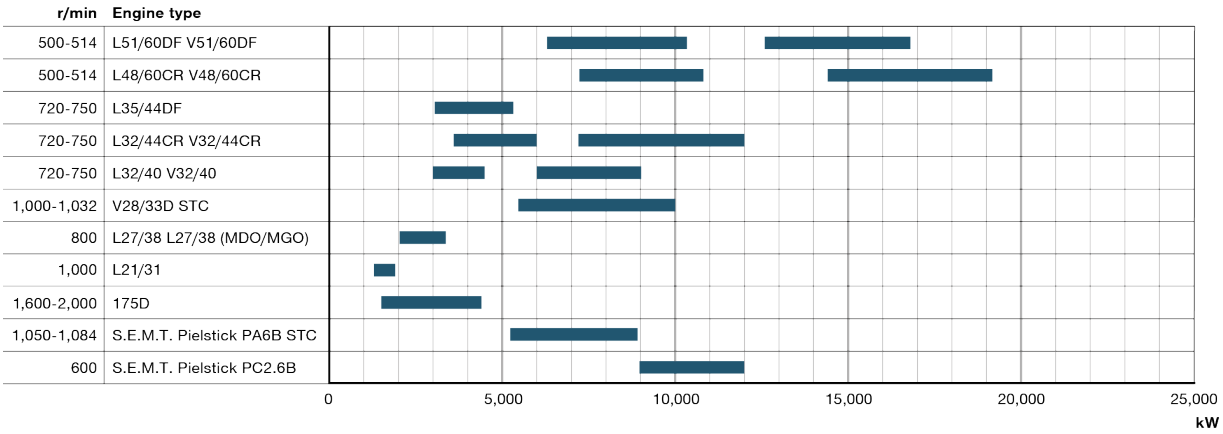

Figure 1: MAN Energy Solutions engine programme

1.2 Engine description MAN 51/60DF IMO Tier III

General

The MAN 51/60DF engine from MAN Energy Solutions is a dual fuel marine engine that converts liquid fuel or natural gas into electrical or mechanical propulsion power efficiently and with low emissions. In combination with a safety concept designed by MAN Energy Solutions for applications on LNG carriers, the multi-fuel capability of the engine represents an appropriate drive solution for this type of vessel, as well as for other marine applications. The capability to changeover from gas to liquid fuel operation without interruption rounds off the flexible field of application of this engine.

Wide range of start capability

For utmost usability, following possibilities are given:

- Engine start in liquid fuel mode
- Engine start in gas mode
- Stand-by: The engine is capable of starting from stand-by either in gas or liquid fuel mode

MAN 51/60DF for electrical and mechanical propulsion

The first type approval for constant speed application was passed successfully in year 2007. As a result of continuous development MAN Energy Solutions has opened the application range of the MAN 51/60DF engine and passed successfully the type approval for mechanical propulsion with Controllable Pitch Propeller (CPP) in year 2016 the output and efficiency could be even further improved.

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Fuels

The MAN 51/60DF engine is designed for operation with liquid and gaseous fuels. The gaseous fuel must match the latest applicable MAN Energy Solutions directives for natural gas. In liquid fuel mode the MAN 51/60DF engine can be operated with MGO (DMA, DFA), MDO (DMB, DFB) and with HFO up to a viscosity of 700 mm^2/s (cSt) at 50 °C. It is designed for fuels up to and including the specification CIMAC 2003 H/K700/DIN ISO 8217.

Fuels for operation with SCR catalyst

The MAN SCR components have been specially designed for operation with heavy fuel oil (HFO) in accordance with specification DIN ISO 8217 up to sulphur content of max. 3.5 % and optimised for our engine portfolio.

Con-rods and con-rod bearings

Optimised marine head version with split joint in upper shaft area, thus no release of the con-rod bearing necessary during piston extraction; low piston extension height. Optimised shells for con-rod bearings increase operating safety.

Cylinder head

With its combustion chamber geometry, the cylinder head assures optimum combustion of gaseous and liquid fuels. Atomisation of the fuel spray in both operating modes is unimpeded – thus leading to very good air: Fuel mixture formation and an optimum combustion process, i.e. reduction in fuel consumption in both operating modes.

Engine frame

Rigid housing in cast monoblock waterless design construction with tie bolts running from the suspended main bearing through the top edge of the engine frame and from the cylinder head through the intermediate plate.

Cylinder liner

The cylinder liner, mounted in individual cylinder jacket, is free of deformations arising from the engine frame and thus assures optimum piston running, i.e. high service life and long service intervals.

Stepped piston

Forged dimensionally stable steel crown (with shaker cooling) made from high grade materials and skirt in spheroidal graphite cast iron (skirt also available in steel upon request). The stepped piston and the fire ring together prevent “bore polishing” of the cylinder liner, thereby reducing operating costs by keeping lubricating oil consumption consistently low. Chromium ceramic coating of the first piston ring with wear resistant ceramic particles in the ring surface results in minimal wear and tear, ensuring extremely long periods between maintenance.

Valves

The exhaust valves have water-cooled, armoured exhaust valve seat rings and thereby low valve temperatures. Propellers on the exhaust valve shaft cause rotation of the valve due to the gas flow with resultant cleaning effect of the sealing surfaces.

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The inlet valves are equipped with Rotocaps. This results in a low rate of wear, i.e. long service intervals.

Main injection by Sealed Plunger injection pumps (SP injection pumps)

The MAN 51/60DF conventional injection system is equipped with Sealed Plunger injection pumps. SP injection pumps have been designed for an operation with all specified fuels.

Benefit:

+ The fuel and the lube oil within the injection pumps are completely separated and cannot get in contact with each other, so that the leakage fuel of the SP injection pumps can be completely reused again.

+ For the same reason, there is no need for sealing oil anymore in the case of continuous MGO-operation.

Liquid boost

To achieve an optimised load response in gas mode operation, liquid boost is activated automatically if necessary. Hereby by the main injection system, a small amount of liquid fuel (the currently used one, even HFO) will be injected for a further improved load application behavior. After reaching the aimed engine load the engine is operated in gas mode without main injection system again.

Common rail pilot fuel injection system

The MAN 51/60DF employs advanced pilot fuel injection strategy, creating new degrees of freedom in terms of engine adaptation flexibility. The gaseous fuel is ignited by injection of a distillate pilot fuel.

The MAN 51/60DF pilot injection system allows flexible setting of injection timing, duration and pressure for each cylinder. Likewise, MAN Energy Solutions common rail technology also allows the MAN 51/60DF to respond rapidly to combustion knocking and misfiring on a cylinder-by-cylinder basis.

To ensure nozzle cooling pilot fuel injection stays in operation during liquid fuel operation.

Rocker housing

Modified, weight-reduced rocker arm casing allows quick replacement of injectors for gas and liquid fuel modes. The components required for gas operation are completely integrated into the rocker housing. High design strength, good heat dissipation and a configuration for the highest ignition pressures ensure that the unit has a very high level of component safety, i.e. long service life.

SCR technology

MAN Energy Solutions has developed its own SCR technology to be able to optimise the emissions technology and the engine performance in addition with the MAN Energy Solutions own SCR control programme for increased customer benefit.

Common SCR systems require constantly high exhaust gas temperatures. The MAN Energy Solutions SCR system however is an integrated system (engine plus SCR) that is automatically adjusting the exhaust gas temperature in an optimal way to ensure ideal operation of both engine plus SCR.

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For example, the engine is operating at optimum condition, however the system is registering an increasing back pressure over the SCR reactor. To resolve this, the regeneration feature of the integrated SCR system is activated and the waste gate engaged to increase exhaust gas temperature. After a short time, the SCR system is regenerated and the engine can continue operation in the design point area. Thus the SCR assures ideal engine operation by regenerating the SCR system whenever necessary to achieve minimum fuel oil consumption. Nevertheless, the SCR system complies with the IMO Tier III regulations on NO_x emissions at any time.

For the SCR reactor two designs are available to provide more flexibility to fit the reactor into the available space. The two reactor variants are named LPC and LPS. LPC is the standard reactor with a Low Pressure Compact design. The LPC reactor is equipped with a single layer of honeycombs resulting in a very low height.

The LPS is a Low Pressure Slim variant of the reactor with smallest footprint. Therefore the honeycombs are arranged in two layers. The honeycombs have different cell density per layer to keep the backpressure of the reactor on a low level.

MAN Energy Solutions turbocharging system with VTA

Optimally adapted charging system (constant pressure) with modern MAN Energy Solutions turbochargers from the TCA series having long bearing overhaul intervals and high efficiency. Good part load operation thanks to very high turbocharger efficiency even under low pressure conditions. The MAN 51/60DF engines are charged by just one TCA turbocharger, which means that only one common exhaust gas collector pipe is required for all cylinders.

By using VTA (Variable Turbine Area, achieved by adjustable turbine nozzle ring) lambda control is realised. VTA-turbochargers allow precise, stepless and continuous control of charge air pressure and air-flow according to the respective engine operating conditions.

Service-friendly design

Hydraulic tools for tightening and loosening cylinder head nuts; quick locks and/or clamp and stub connections on pipes/lines; generously sized crankcase cover; hydraulic tools for crankshaft bearings and lower connecting rod bearings; very low maintenance Geislinger sleeve spring vibration dampers.

SaCoSone

The MAN 51/60DF is equipped with the Classification Society compliant safety and control system SaCoSone.

The SaCoSone control system allows safe engine operation in liquid fuel and gas modes with optimum consumption and low emissions. In gas mode, the SaCoSone control system guarantees safe operation between the knock and misfire boundaries. All cylinders are controlled individually in this instance. For operation with liquid fuel, control is based on the standard SaCoSone control system for diesel engines. The complete system is subject to a test-run in the factory with the engine so that fine tuning and functional testing during commissioning in the vessel only involve a minimum of effort.

Special functionalities have been implemented to cover the requirements on the LNG carrier business. Exemplary can be named:

- Fuel quality manager

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During a round trip of an LNG carrier the fuel gas composition is changing in a big range. After bunkering the natural boil-off gas (NBOG) contains a high amount of nitrogen. Contents of 20 % and higher are quite common. This lowers the heat value of the fuel gas and leads to longer gas injection. In the SaCoSone system after comparison of an external engine output signal with actual engine parameters an adjustment of parameters in the control is done, to feed the engine with sufficient gas fuel amount according to the required load.

- Adaptive air fuel control

  Additionaly the air fuel ratio will be adjusted according to the change in fuel gas and the corresponding changed heat value and knocking characteristic.

- Cleaning cycle for change over

  During HFO operation the combustion chamber will be contaminated with deposits formed by the combustion of HFO. The cleaning cycle function will be activated in case of recognised HFO operation and knocking events during change over to gas operation. So for this cleaning cycle no intermediate fuel like MDO is required and heavy knocking events will be avoided.

- Crankcase monitoring system plus oil mist detection

  As a standard for all our four-stroke medium-speed engines manufactured in Augsburg, these engines will be equipped with a crankcase monitoring system (CCM = splash oil & main bearing temperature) plus OMD (oil mist detection). OMD and CCM are integral part of the MAN Energy Solutions´ safety philosophy and the combination of both will increase the possibility to early detect a possible engine failure and prevent subsequent component damage.

Soot

Soot emissions during operation on liquid fuel are on very low level by means of optimised combustion and turbocharging. For increased demands the engine might be equipped with VVT instead of VIT. In gas mode soot emissions are in the whole load range well below the limit of visibility.

Miller valve timing

To reduce the temperature peaks which promote the formation of NO_x, early closure of the inlet valve causes the charge air to expand and cool before start of compression. The resulting reduction in combustion temperature reduces NO_x emissions.

NO_x emission with gaseous fuels

On natural gas, the MAN 51/60DF undercuts IMO Tier II levels by extremely wide margin – indeed, in gaseous fuel mode, the MAN 51/60DF already fulfils the strict IMO Tier III NO_x limitations prescribed for Emissions Control Zones (ECA's).

NO_x emission with liquid fuels

The MAN 51/60DF complies with IMO Tier II NO_x emission limits and in connection with an appropriate MAN Energy Solutions´ SCR exhaust gas aftertreatment system also with IMO Tier III NO_x limitations prescribed for ECA´s.

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Knocking detection

The individual knocking levels from each cylinder are collected by the knocking detection unit. In combination with the cylinder individual control of the ignition begin, the SaCoSone control ensures a stable operation with a sufficient margin to the knocking limit.

Adaptive combustion control (ACC)

The MAN 51/60DF equipped with cylinder pressure based combustion control, which individually adjusts the combustion of each cylinder. Special applications may not require this feature.

Additional notes/brief summary

Dual fuel engines offers fuel flexibility. If the gas supply fails once, also a full load running engine is automatically switched over to liquid fuel mode without interruption in power supply. DF engines can run in:

- Liquid fuel mode
- Gas mode (for ignition a small amount of diesel oil is injected by separate pilot fuel injection nozzles)
- Back up mode operation (in case the pilot fuel injection should fail, the engine can still be operated. For details see section General introduction of liquid fuel oil system for dual fuel engines (designed to burn HFO, MDO and MGO), Page 365)

Starting and stopping of the engine can be performed in liquid fuel mode as well as in gas mode. The engine power in gas mode is generally equal to the generated power in liquid fuel mode.

Pilot fuel injection is also activated during liquid fuel mode. The injected pilot fuel quantity depends on the engine load.

MAN Energy Solutions total system competence

As the leading engine builder in the marine sector, MAN Energy Solutions has unrestricted access to the know-how required to design and execute highly efficient SCR systems for both new engines and retrofit applications on engines already in the field.

In MAN Energy Solutions´ case, this clear “Advantage” over other supplier of SCR systems is further multiplied by our status as a global leader in the design and manufacture of turbochargers and fuel injection systems for large engines. MAN Energy Solutions and its PrimeServ after-sales organisation is ideally placed to supply and service the optimum SCR system for your engine.

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1.3 Engine overview and SCR system components

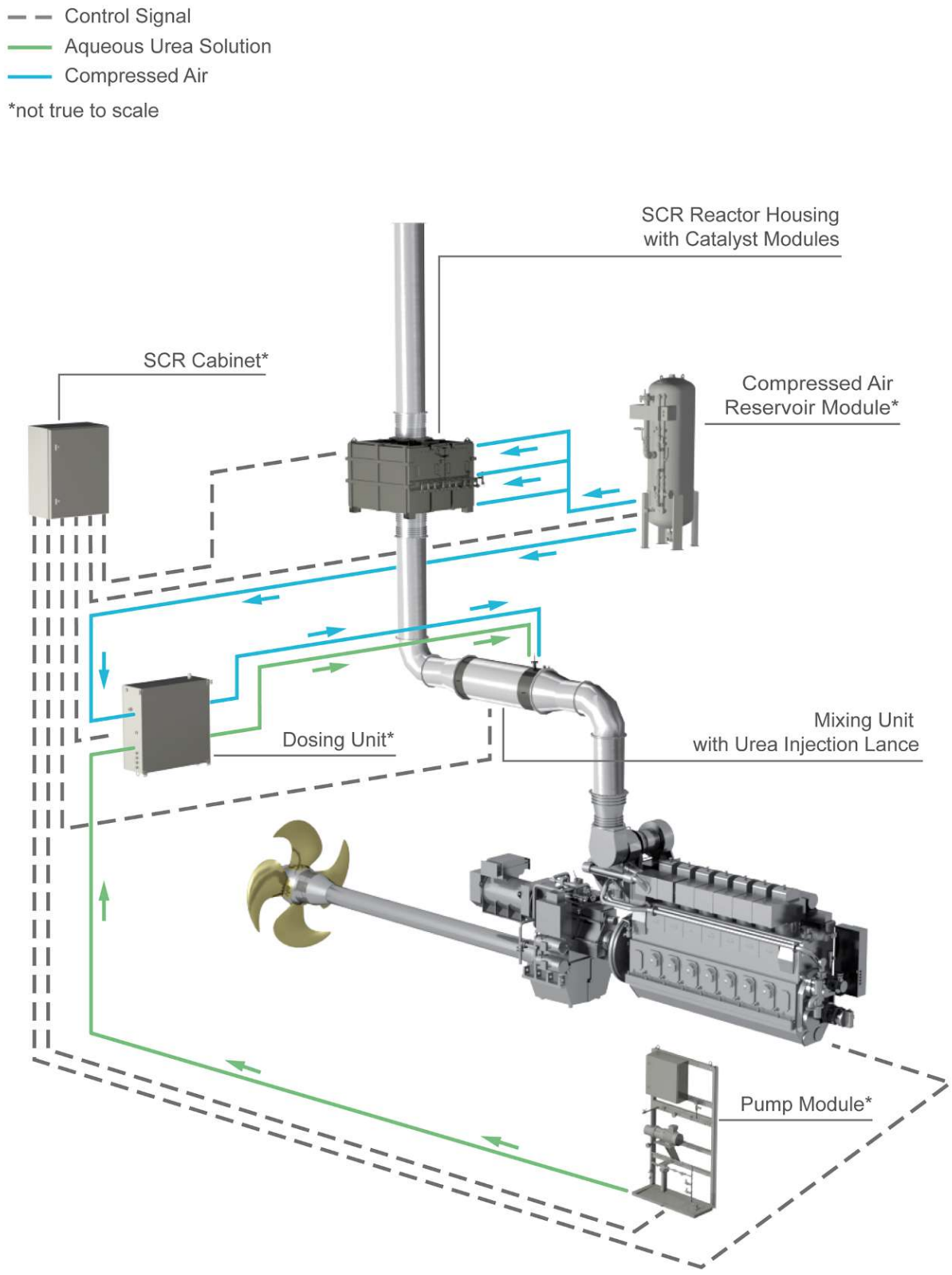

Figure 2: SCR system components overview (example)

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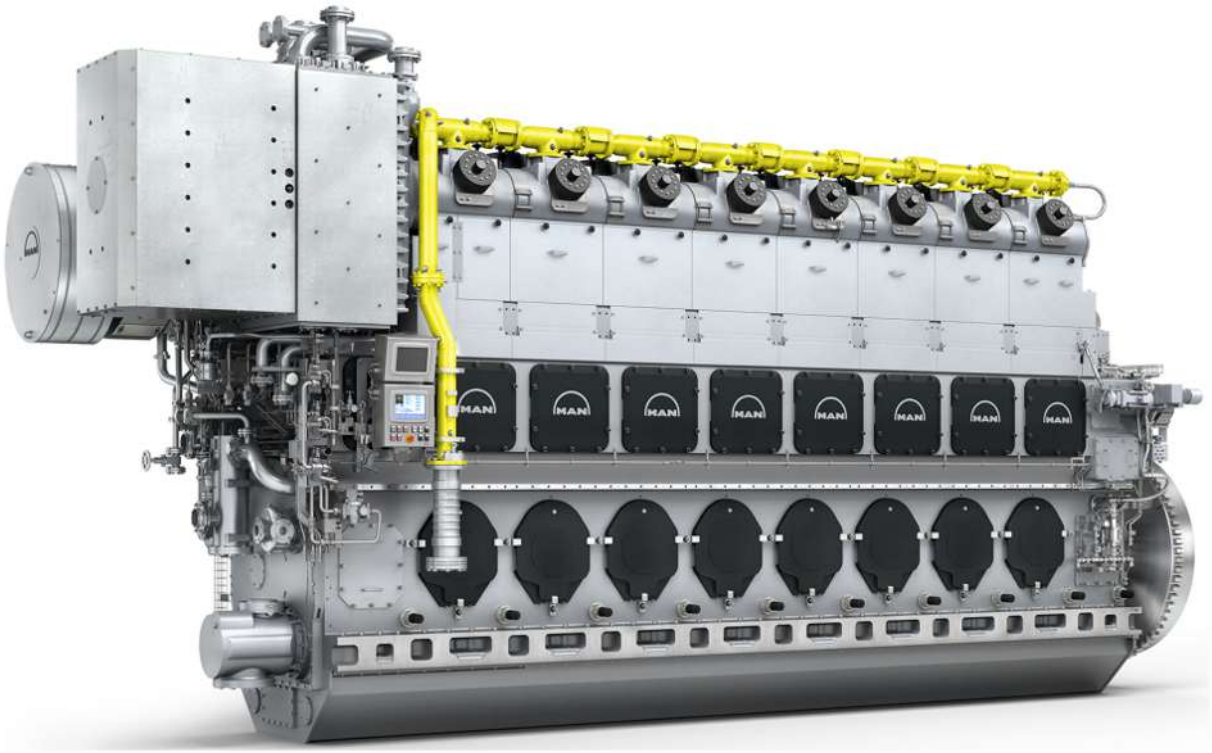

Figure 3: Engine overview, MAN L51/60DF view on counter coupling side

Figure 4: Engine overview, MAN L51/60DF view on coupling side

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1.4 Turbocharger overview

1.4.1 View of a TCA type turbocharger

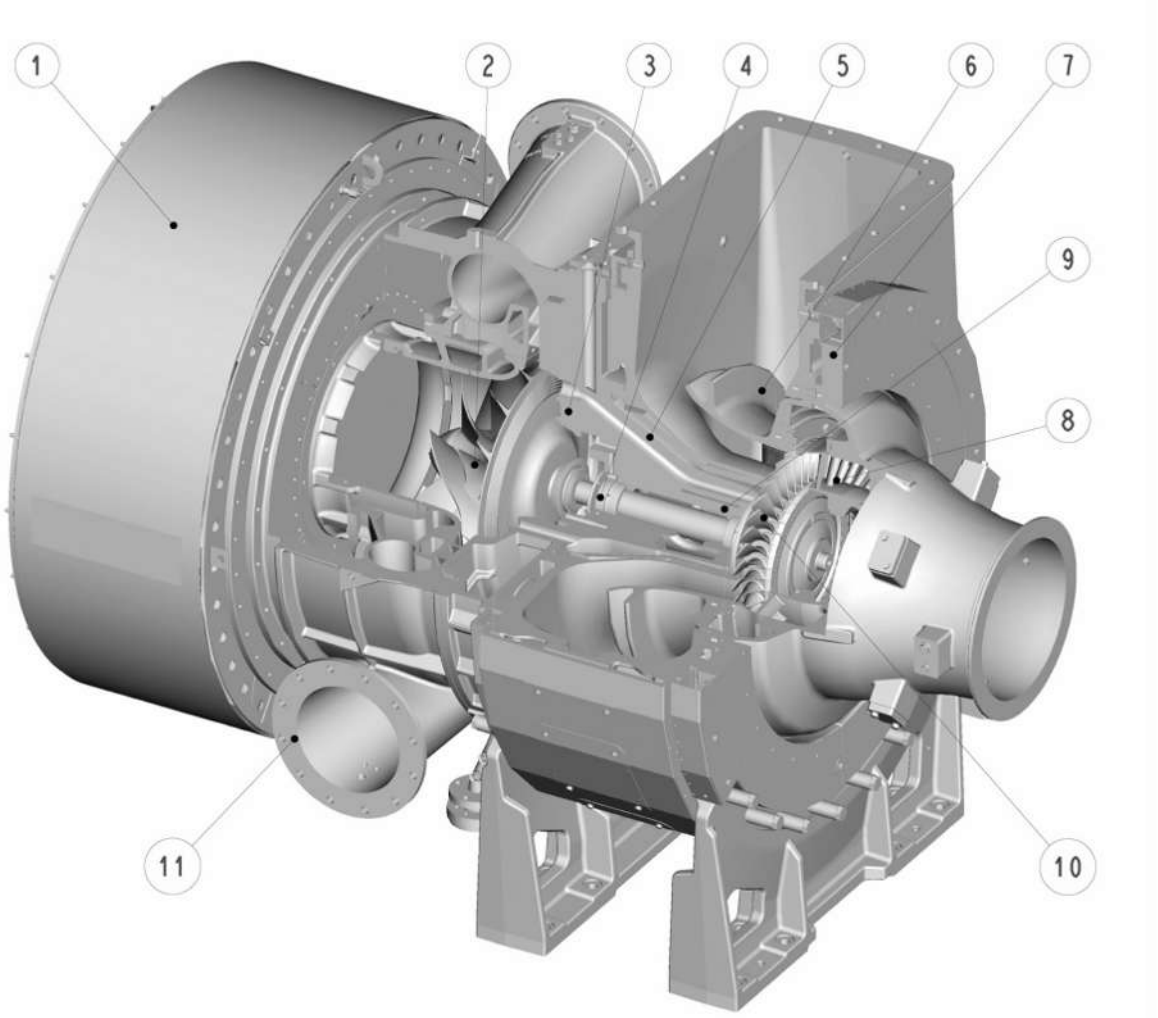

Figure 5: TCA type turbocharger

1	Silencer	7	Burst-proof casing
2	Compressor wheel	8	Nozzle ring
3	Bearing casing	9	Plain bearing
4	Thrust bearing	10	Turbine blades
5	Integrated sealing air system	11	Compressor casing
6	Exhaust diffuser		

The following view indicates the modern design principle of the TCA series:

- Whispering silencer
- Easy-to-service, low-noise compressor wheel
- Optional water-cooled compressor wheel

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- Uncooled bearing casing
- Easy access to thrust bearing
- Integrated sealing air, oil pipe and venting systems

1.4.2 Compensator between turbine outlet (engine) and exhaust gas pipe (plant)

All turbocharger casing flanges, with the exception of the turbine outlet, may only be subjected to loads generated by the gas forces, and not to additional external forces or torques.

This necessitates the use of compensators directly at the turbine outlet.

The compensators must be pre-loaded in such a manner that thermal expansion of the pipes and casings does not exert forces or torques in addition to those generated by the air and gas.

- Forces and torques according to API standard 617.
- Operating direction implemented according to MAN Energy Solutions standard.
- Minimising the load as far as possible.
- Characteristic values include gas forces, masses and compensator.

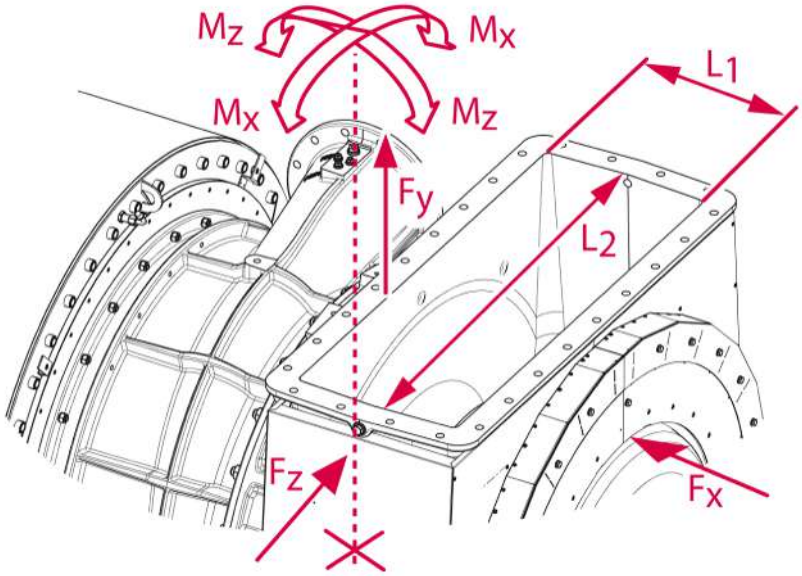

Figure 6: Maximum connection loads on gas outlet casing

Type	F_x [N]	F_y [N]	F_z [N]	M_x [Nm]	M_z [Nm]	L_1 [mm]	L_2 [mm]
TCA33	3,900	7,900	7,900	6,000	3,000	400	690
TCA44	4,200	8,500	8,500	6,400	3,200	340	949
TCA55	4,500	9,100	9,100	6,900	3,400	390	1,080
TCA66	4,900	9,900	9,900	7,500	3,700	463	1,283
TCA77	5,400	10,900	10,900	8,200	4,100	550	1,524
TCA88	5,900	12,000	12,000	9,100	4,500	653	1,810
TCA88-25	6,300	12,700	12,700	9,600	4,800	653	1,812

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Note:

- The layout of the compensator has to consider the possible movement of the engine according its foundation and engine mounting and the movement/extension of the exhaust gas pipe of the plant.
- Recommendations for the layout of the exhaust gas pipe in the section Exhaust gas system, Page 427 have to be considered.

1.4.3 No additional masses allowed

In general no masses are to be fixed to any part of the engine/turbocharger, as these have an impact to the general vibration behaviour of the engine.

Additional loads and changed vibration behaviour can endanger the operational safety of the engine.

Consequently, for any questions in this regard consult MAN Energy Solutions in advance.

1.5 Safety concept of MAN Energy Solutions dual fuel engine – Short overview

This section serves to describe in a short form the safety philosophy of MAN Energy Solutions´ dual fuel engines and the necessary safety installations and engine room arrangements. The engines serve as diesel-mechanical prime movers as well as power generation unit in diesel electric applications onboard of LNG carriers or other gas fueled ships.

Possible operation modes are pure gas mode or pure diesel mode.

This safety concept deals only with the necessary gas related safety installations.

The MAN Energy Solutions dual fuel engines are four-stroke engines with either liquid fuel or gas as main fuel.

The operating principle in gas-mode is the lean-burn concept. A lean-mixture of gas and air is provided to the combustion chamber of each cylinder by individually controlled gas admission valves. The mixture is ignited by a small amount of pilot fuel. In liquid fuel mode the fuel is injected in the combustion chamber by the main injection system.

The safety concept of MAN Energy Solutions´ dual fuel engines is designed to operate in gas mode with the same safety level as present in liquid fuel mode. The concept is based on an early detection of critical situations, which are related to different components of the gas supply system, the combustion and the exhaust system. If necessary the safety system triggers different actions, leading to alarm or automatically switching to liquid fuel mode, without interruption of shaft power or a shutdown of engines and gas supply systems.

The safety philosophy is to create along the gas supply and gas reaction chain an atmosphere in the engine room, which under normal operation conditions is never loaded with gas. The gas supply piping is double walled. Negative pressure prevails in the interspace between the inner and the outer pipe. Engine rooms, gas valve unit room and additonal necessary rooms are monitored and controlled, and are always sufficient ventilated, in the way that a (small) negative pressure is set. Gas detection is required in the gas valve unit compartment, in the interspace between the inner and the outer pipe of the double walled pipes and the engine rooms.

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The exhaust system can be purged by an explosion proofed fan installed in the exhaust gas system. The purged air is always led through the exhaust gas duct outside the engine room. Rupture discs or explosion relief valves are installed in the exhaust gas duct.

All system requirements and descriptions have to be in accordance with international rules and normatives, the IMO (International Marine Organisation) and the IGC (International Gas Carrier Code) and classification societies rules. Note that all systems have to be built in accordance with the above mentioned requirements.

For further information, refer to our separate brochures "Safety Concept – Marine dual fuel engines".

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2 Engine and operation

2.1 Approved applications and destination/suitability of the engine

Approved applications

The MAN 51/60DF is designed as multi-purpose drive. It has been approved by type approval as marine main engine by all main classification societies (ABS, BV, CCS, ClassNK, DNV, KR, LR, RINA, RS).

As marine main engine[1] it may be applied for mechanical or electric propulsion[2] for applications as:

- Bulker, container vessel and general cargo vessel
- Ferry and cruise liner
- Tanker
- Others – to fulfill all customers needs the project requirements have to be defined at an early stage

For the applications named above the MAN 51/60DF can be applied for single- and for multi-engine plants.

For single-engine plants a speed governor with mechanical backup is mandatory to fulfil the requirements of the classification societies.

The MAN 51/60DF as marine auxiliary engine may be applied for electric power generation[2] for auxiliary duties for applications as:

- Auxiliary GenSet[2]

Note:
The engine is not designed for operation in hazardous areas. It has to be ensured by the ship's own systems, that the atmosphere of the engine room is monitored and in case of detecting a gas-containing atmosphere the engine will be stopped immediately.

[1] In line with rules of classifications societies each engine whose driving force may be used for propulsion purpose is stated as main engine.

[2] See section Engine ratings (output) for different applications, Page 43.

Destination/suitability of the engine

Note:
Regardless of their technical capabilities, engines of our design and the respective vessels in which they are installed must at all times be operated in line with the legal requirements, as applicable, including such requirements that may apply in the respective geographical areas in which such engines are actually being operated.

Operation of the engine outside the specified operated range, not in line with the media specifications or under specific emergency situations (e.g. suppressed load reduction or engine stop by active "Override", triggered firefighting system, crash of the vessel, fire or water ingress inside engine room) is declared as not intended use of the engine (for details see engine specific operating manuals). If an operation of the engine occurs outside of the scope of supply of the intended use a thorough check of the engine and its compon-

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ents needs to be performed by supervision of the MAN Energy Solutions service department. These events, the checks and measures need to be documented.

Electric and electronic components attached to the engine – Required engine room temperature

In general our engine components meet the high requirements of the marine classification societies.

Relevant design criteria for the engine room air temperature:

Minimum air temperature in the area of the engine and its components ≥ 5 °C.

Maximum air temperature in the area of the engine and its components ≤ 45 °C.

The electronic components are suitable for proper operation within an air temperature range from 5 °C to 55 °C.

For further information see also section Engine automation, Technical data, envirionmental conditions, Page 239.

Note:
Condensation of the air at engine components must be prevented.

Note:
It can be assumed that the air temperature in the area of the engine and attached components will be 5 – 10 K above the ambient air temperature outside the engine room. If the temperature range is not observed, this can affect or reduce the lifetime of electrical/electronic components at the engine or the functional capability of engine components. Air temperatures at the engine > 55 °C are not permissible.

2.2 Certification IMO Tier III

The engine's certification for compliance with NO_x limits according to NO_x technical code will be done according scheme B, meaning engine plus SCR will be handled as separate parts. Certification has to be in line with IMO Resolution MEPC 198(62), adopted 15 July 2011.

Liquid fuel mode — Emission level engine: IMO Tier II

Emission level engine plus SCR catalyst: IMO Tier III

Gas mode — Emission level engine: IMO Tier III

Certification of engine

Liquid fuel mode — Engine will be tested as specified in section Programme for Factory Acceptance Test (FAT), Page 514 according to relevant classification rules. It will also certified as member or parent engine according NO_x technical code for emission category IMO Tier II.

Gas mode — In gas mode the SCR system will be deactivated. Therefore IMO NO_x certification will be carried out while factory acceptance test for gas operation for compliance with IMO Tier III.

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Certification of complete system (engine plus SCR system)

Liquid fuel mode

Certification of SCR catalyst and components will be done in accordance to MEPC 198(62) for a scaled, standardised SCR reactor and SCR components based on product features and following scaled parameters:

- Exhaust gas mass flow
- Exhaust gas composition (NO_x, O_2, CO_2, H_2O, SO_2)
- Exhaust gas temperature
- Catalyst modules (AV, SV or LV value)
- Reducing agent
- Desired NO_x conversion rate

The on-board confirmation test required for a scheme B certification will be done for the parent engine plus SCR system for a group according to IMO resolution MEPC 198(62).

2.3 SCR – Special notes

2.3.1 Principle of SCR technology

The selective catalytic reduction SCR uses ammonia (NH_3) to convert nitrogen oxides in the exhaust gas to harmless nitrogen and water within a catalyst. However, ammonia is a hazardous substance which has to be handled carefully to avoid any dangers for crews, passengers and the environment. Therefore urea as a possible ammonia source is used. Urea is harmless and, solved in water, it is easy to transport and to handle. Today, aqueous urea solutions of 40 % is the choice for SCR operation in mobile applications on land and at sea.

Using urea, the reaction within the exhaust gas pipe and the catalyst consists of two steps. In the beginning, the urea decomposes in the hot exhaust gas to ammonia and carbon dioxide using the available water in the injected solution and the heat of the exhaust gas:

$(NH_2)_2CO + H_2O \rightarrow 2NH_3 + CO_2$ [1]

The literal NO_x-reduction takes place supported by the catalyst, where ammonia reduces nitrogen oxides to nitrogen and water:

$4NO + 4NH_3 + O_2 \rightarrow 4N_2 + 6H_2O$ [2]

2.3.2 System overview

The MAN Energy Solutions SCR system is available in different sizes to cover the whole medium-speed engine portfolio. The SCR system consists of the reactor, the mixing unit, the urea supply system, the pump module, the dosing unit, the SCR cabinet and the soot-blowing system.

After initial start-up of the engine, the SCR system operates continuously in automatic mode. The amount of urea injection into the SCR system depends on the operating conditions of the engine. Since the SCR cabinet of the SCR system is connected to the engine control system all engine related information are continuously and currently available. This is one of the important benefits of the MAN Energy Solutions SCR system.
The urea is sprayed into the mixing unit which is part of the exhaust gas duct. Entering the reactor the reducing agent starts to react with NO_x coming from

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the combustion. The amount of reducing agent is controlled by the dosing unit, which is supported by a pump connected to an urea tank. It furthermore regulates the compressed air flow for the injector.

Each reactor is equipped with a soot blowing system to prevent blocking of the SCR catalyst by ashes and soot.

2.3.3 Scope of supply

- Engine in standard configuration according stated emission level (see above).
- Engine attached equipment for control of the temperature after turbine.
- Engine SaCoS software including functions for control of temperature after turbine and for optimising engine plus SCR performance.
- IMO Tier III Certificate.
- MAN Energy Solutions will act as "Applicant" within the meaning of the IMO.

Main components of SCR system in the standard scope of supply

- SCR reactor
- Catalyst modules
- Soot blowing system
- Dosing unit
- Mixing unit
- Urea injection lance
- SCR cabinet
- Pump module
- Compressed air reservoir module

Not included in the standard scope of supply, among others

- Urea storage tank
- Urea storage tank minimum level switch
- Piping
- Insulation

2.3.4 Operation

Standard operation

Common SCR systems provided by third parties require constantly high exhaust gas temperatures. The MAN Energy Solutions SCR system on the other hand is an integrated engine plus SCR system, that allows operation on lower exhaust gas temperature levels.
The MAN Energy Solutions SCR system automatically adjusts the engine exhaust gas temperature to ensure both optimum engine plus SCR operation. For a maximum on safety the surveillance mode is always activated.

Enhanced operation

The MAN Energy Solutions SCR system assures ideal engine operation, regenerating the SCR system whenever necessary to account for minimum fuel oil consumption while complying with IMO Tier III emission limits at all times. The regeneration will be started automatically. One specific regeneration trig-

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ger is the back pressure increase of the catalysts. It varies depending on operation, fuel oil, engine type, ambient conditions etc. Especially long-term low-load operation might cause a significant back pressure increase.

See section Low-load operation, Page 65.

2.3.5 Boundary conditions for SCR operation

Consider following boundary conditions for the SCR operation:

- Temperature control of temperature turbine outlet:
  - By adjustable waste gate (attached to engine) or for individual engine types by VTA.
  - Set point 320 °C as minimum temperature for active SCR.
  - Set point 290 °C as minimum temperature for deactivated SCR.
- Fuel:
  - In line with MAN Energy Solutions specification, maximum **3.5 %** sulfur content.
- SCR active in following range:
  - **–10 °C** (arctic) up to **45 °C** (tropic) intake air temperature.
  - In the range of **25 % to 100 % engine load**.
- IMO requirements for handling of SCR operation disturbances:
  - In case of SCR malfunction IMO regulations allow that the system will be turned off and the ship's journey will be continued to the port of destination. There, the ship needs to be repaired, if the emission limits of the harbor/sea area would be exceeded.
- Differential pressure Δp SCR (normal operation):
  - Max. 20 mbar for the MAN SCR-LPC.
  - Max. 37 mbar for the MAN SCR-LPS.

For the design of the complete exhaust gas line, consider:

- Maximum permissible exhaust gas back pressure (to be calculated from engine turbocharger outlet to end of complete exhaust gas line):
  - Max. 50 mbar (at 100 % engine load).
- Maximum permissible temperature drop of exhaust gas line (to be calculated as difference of exhaust gas temperature turbine outlet and temperature SCR inlet):
  - Max. 5 K in the range of 25 % to 100 % engine load (calculated at 5 °C air temperature in the engine room).
- Recommended for exhaust gas line:
  - Insulation according to SOLAS standard.

Note:
The SCR system requires high exhaust gas temperatures for an effective operation. MAN Energy Solutions therefore recommends to arrange the SCR as the first device in the exhaust gas line, followed by other auxiliaries like boiler, silencer etc.

For the design of the ship automation, consider:

- In order to ensure reliable regeneration of the SCR, it is necessary in some applications to bring individual engines or engine groups into a load range suitable for regeneration. This must be considered in the design phase of

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the load distribution control. In the case of single-engine plants and exceptional case, a temporary adjustment of the cruise profile may be necessary.

Contact MAN Energy Solutions for further details.

2.3.6 Performance coverage for SCR system

- Performance guarantee for engine plus SCR within defined in section Boundary conditions for SCR operation, Page 27.
- Guarantee for engine plus SCR for marine applications to meet IMO Tier III level as defined by IMO within defined in section Boundary conditions for SCR operation, Page 27 (details will be handled within the relevant contracts).

Be aware:
All statements in this document refer to MAN Energy Solutions SCR systems only.

MAN Energy Solutions can only deliver an IMO Tier III certificate and act as “Applicant” (within the meaning of the IMO) if the engine plus SCR system is supplied by MAN Energy Solutions.

If the engine is supplied without MAN Energy Solutions SCR system, only a standard warranty for a single engine will be given. No guarantee regarding minimum exhaust gas temperature after turbine or emissions after third party SCR or suitability of the engine in conjunction with a third party SCR system can be given.

If the engine is supplied without MAN Energy Solutions SCR system, no optimisation function within SaCoS can be applied and as maximum exhaust gas temperature after turbine only will be possible:

- 320 °C (25 % load – 100 % load).

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2.4 Engine design

2.4.1 Engine cross section

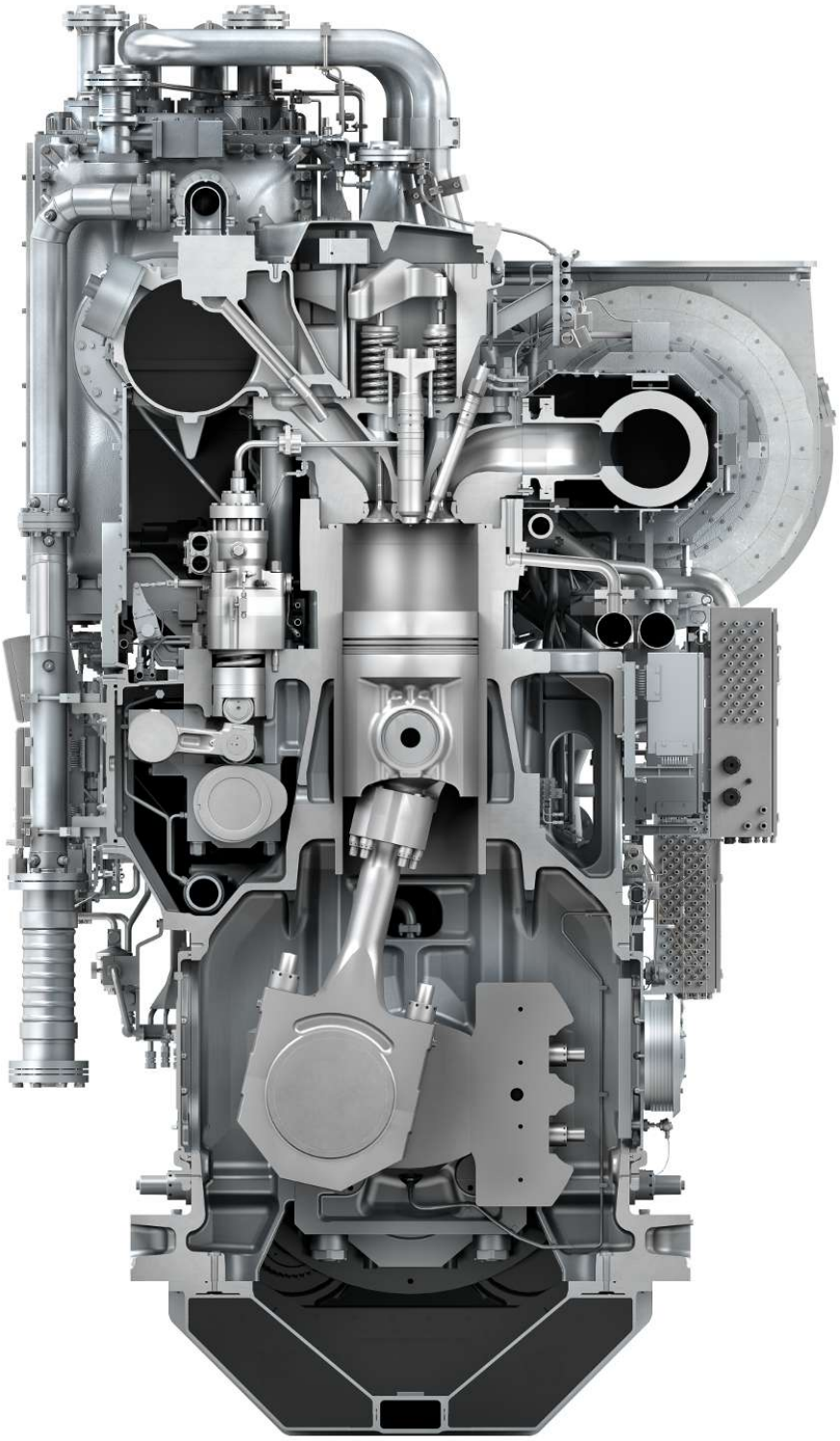

Figure 7: Engine cross section MAN L51/60DF

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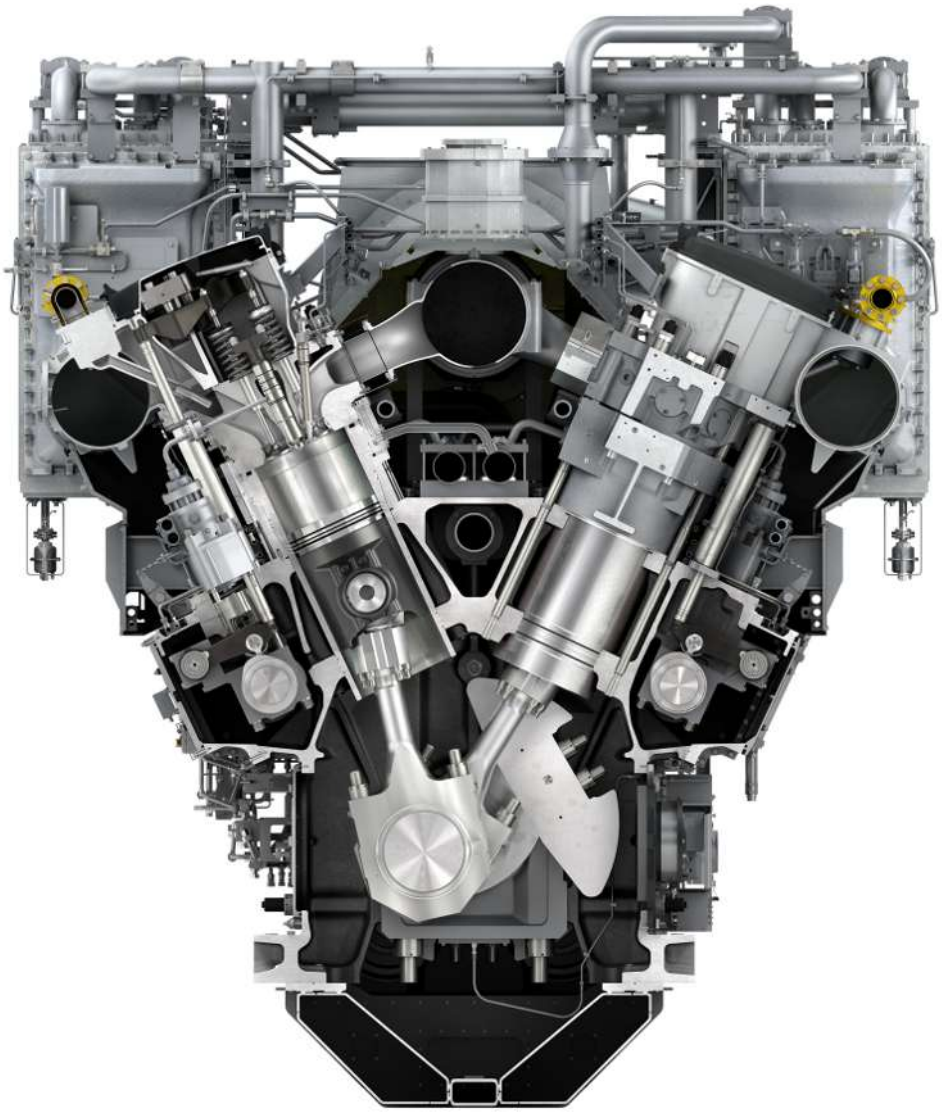

Figure 8: Engine cross section MAN V51/60DF

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2.4.2 Engine designations – Design parameters

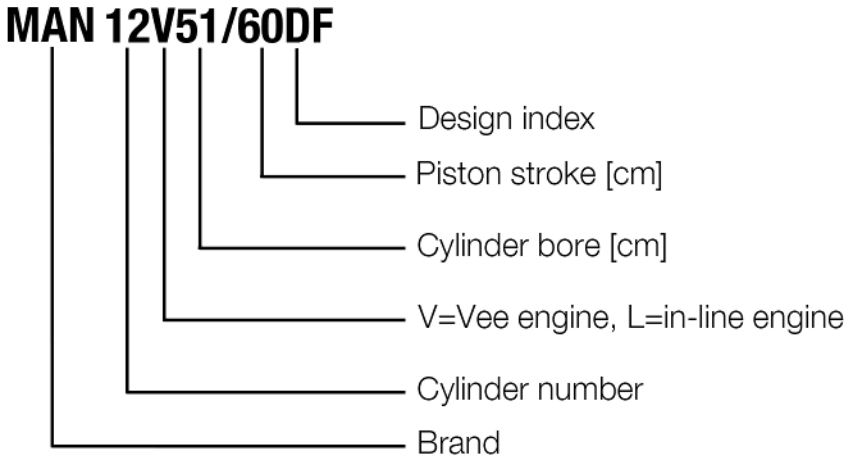

Figure 9: Example to declare engine designations

Parameter	Value	Unit
Number of cylinders	6, 7, 8, 9, 12, 14, 16	-
Cylinder bore	510	mm
Piston stroke	600	
Displacement per cylinder	122.5	litre
Distance between cylinder centres, in-line engine	820	mm
Distance between cylinder centres, vee engine	1,000	
Vee engine, vee angle	50	°
Crankshaft diameter at journal, in-line engine	415	mm
Crankshaft diameter at journal, vee engine	480	
Crankshaft diameter at crank pin	415	

Table 1: Design parameters

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2.4.3 Turbocharger assignments

MAN 51/60DF		
No. of cylinders, config.	Mechanical propulsion with CPP/electric propulsion	
	1,050 kW/cyl., 500 or 514 rpm	1,150 kW/cyl., 500 or 514 rpm
6L	TCA55-42V	TCA55-42V
7L	TCA55-42V	TCA55-42V
8L	TCA55-42V	TCA66-42V

MAN 51/60DF		
No. of cylinders, config.	Mechanical propulsion with CPP/electric propulsion	
	1,050 kW/cyl., 500 or 514 rpm	1,150 kW/cyl., 500 or 514 rpm
9L	TCA66-42V	TCA66-42V
12V	TCA66-42V	TCA77-42V
14V	TCA77-42V	TCA77-42V
16V	TCA77-42V	-

Table 2: Turbocharger assignments

Turbocharger assignments mentioned above are for guidance only and may vary due to project-specific reasons. Consider the relevant turbocharger Project Guides for additional information.

2.4.4 Engine main dimensions, weights and views

L engine

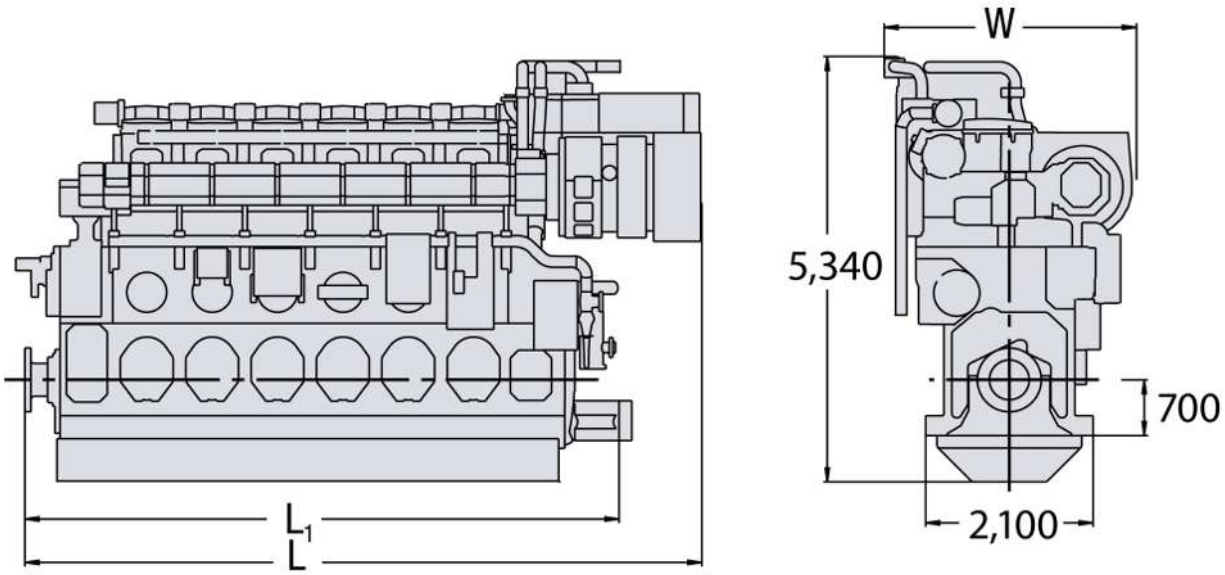

Figure 10: Main dimensions and weights – L engine

No. of cylinders, config.	L	L_1	W	Weight without flywheel
	mm			tons
6L	8,494	7,455	3,165	121
7L	9,314	8,275		138
8L	10,134	9,095		153
9L	11,160	9,915	3,283	168
The dimensions and weights are given for guidance only (weight given without media filling of engine).				

Minimum centreline distance for multi-engine installation, see section Installation drawings, Page 448.

Flywheel data, see section Moments of inertia – Crankshaft, damper, flywheel, Page 174.

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V engine

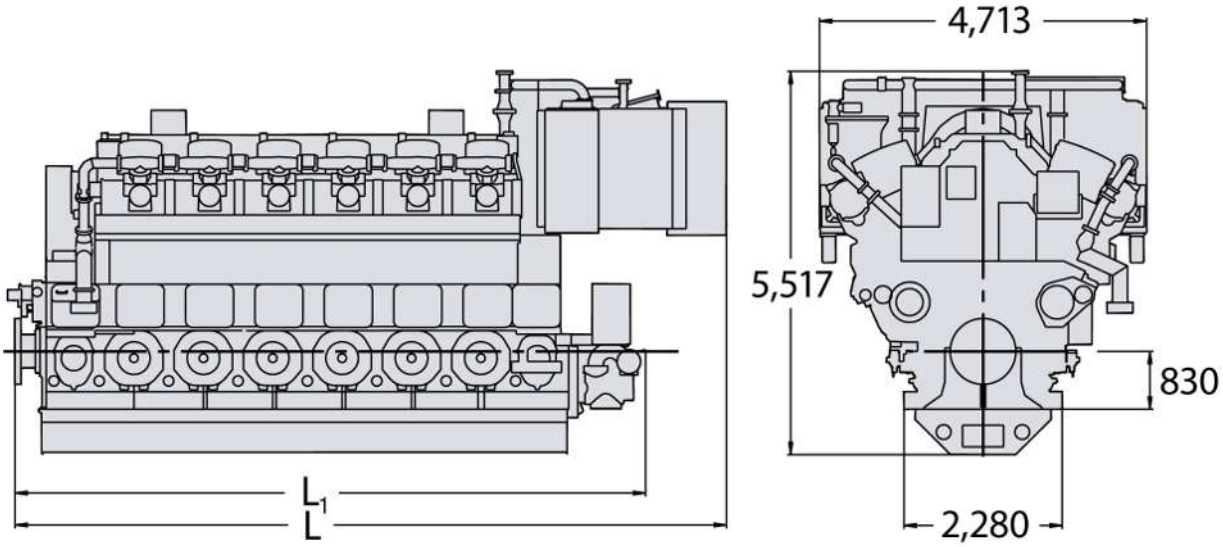

Figure 11: Main dimensions and weights – V engine

No. of cylinders, config.	L	L_1	Weight without flywheel
	mm		tons
12V	10,254	9,088	191
14V	11,254	10,088	222
16V	12,254	11,088	248
The dimensions and weights are given for guidance only (weight given without media filling of engine).			

Minimum centreline distance for multi-engine installation, see section Installation drawings, Page 448.

Flywheel data, see section Moments of inertia – Crankshaft, damper, flywheel, Page 174.

2.4.5 Main dimensions, weights and views of SCR components

Depending on the individual projects SCR properties may vary. The following dimensions and weights are for guidance only.

For the SCR reactor two designs are available, SCR reactor LPC and SCR reactor LPS. LPC is the standard reactor with a compact design. LPS is a specialized reactor with smallest footprint.

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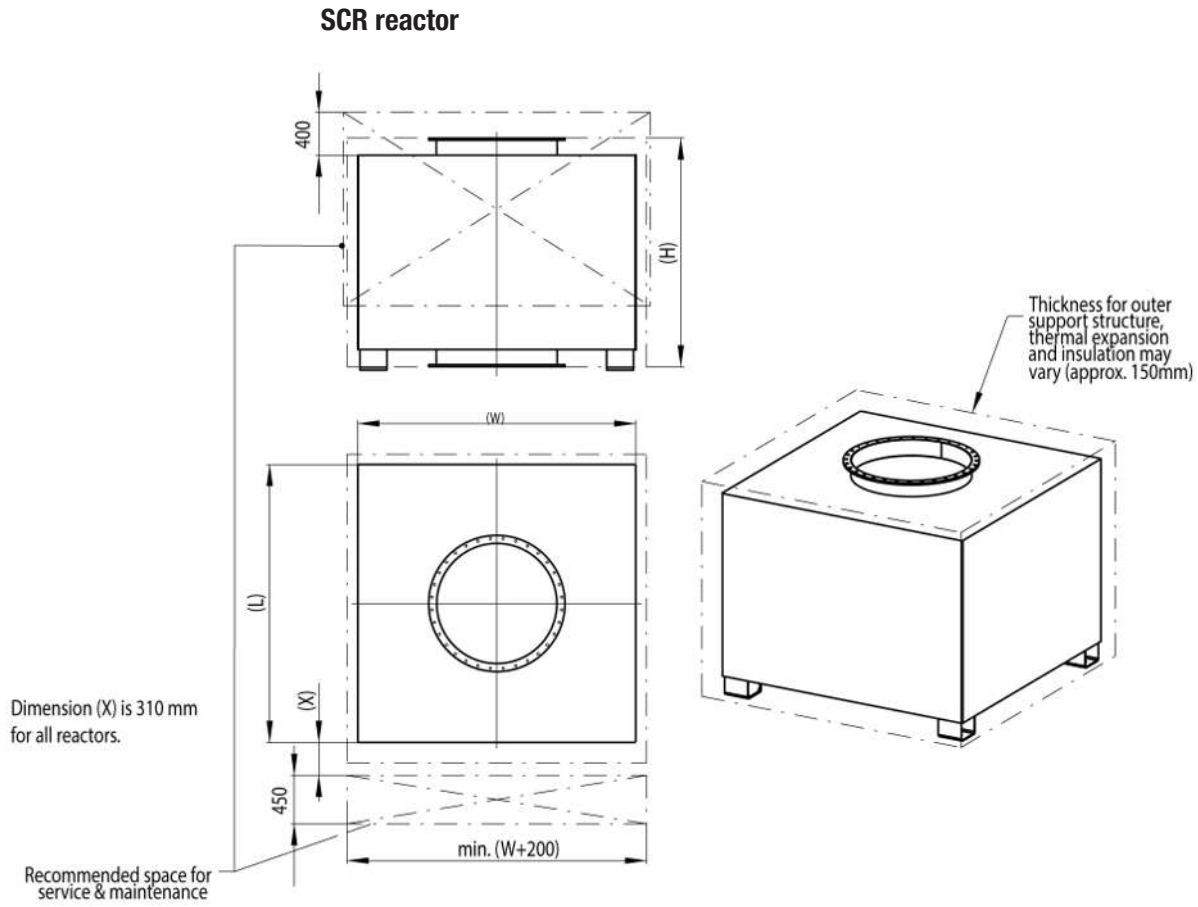

Figure 12: SCR reactor

SCR reactor - LPC	Engine power approximately	Duct	H (total length)	W (without insulation)	L (without insulation)	Handling weight	Operational weight
No.	[kW]	DN	[mm]	[mm]	[mm]	[kg]	[kg]
1	0 – 1,400	500	1,892	1,252	1,252	1,020	1,484
2	1,401 – 2,400	600	1,938	1,502	1,502	1,190	1,886
3	2,401 – 3,600	800	1,985	1,752	1,752	1,530	2,574
4	3,601 – 4,900	900	2,032	2,084	2,084	1,900	3,321
5	4,901 – 6,000	1,000	2,079	2,334	2,334	2,460	4,316
6	6,001 – 7,800	1,100	2,126	2,584	2,584	2,680	5,029
7	7,801 – 9,000	1,200	2,228	2,916	2,916	3,310	6,210
8	9,001 – 12,000	1,400	2,330	3,248	3,248	4,035	7,544
9	12,001 – 13,700	1,500	2,432	3,416	3,416	4,255	8,431
10	13,701 – 15,000	1,600	2,534	3,748	3,748	4,780	9,681
11	15,001 – 17,000	1,700	2,635	3,990	3,990	5,410	11,094
12	17,001 – 20,000	1,800	2,737	4,248	4,248	6,040	12,565
13	20,001 – 21,600	2,000	2,839	4,580	4,580	6,985	14,409

Table 3: SCR reactor - LPC

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SCR reactor - LPS	Engine power approximately	Duct	H (total length)	W (without insulation)	L (without insulation)	Handling weight	Operational weight
No.	[kW]	DN	[mm]	[mm]	[mm]	[kg]	[kg]
1	0 – 2,400	500	3,072	1,252	1,252	1,500	2,428
2	2,401 – 3,600	600	3,118	1,502	1,502	1,800	3,192
3	3,601 – 4,900	800	3,165	1,752	1,752	2,200	4,228
4	4,901 – 6,000	900	3,212	2,084	2,084	2,800	5,642
5	6,001 – 9,000	1,000	3,259	2,334	2,334	3,790	7,502
6	9,001 – 12,000	1,100	3,306	2,584	2,584	4,410	9,108
7	12,001 – 13,700	1,200	3,408	2,916	2,916	5,300	11,100
8	13,701 – 15,000	1,400	3,510	3,248	3,248	6,200	13,218
9	15,001 – 20,000	1,500	3,612	3,416	3,416	6,930	15,282
10	20,001 – 21,600	1,600	3,714	3,748	3,748	7,980	17,782

Table 4: SCR reactor - LPS

Note:
In accordance with applicable security policies there must be provided adequate maintenance space, which permits the safe execution of all necessary maintenance work.

Mixing unit with urea injection lance

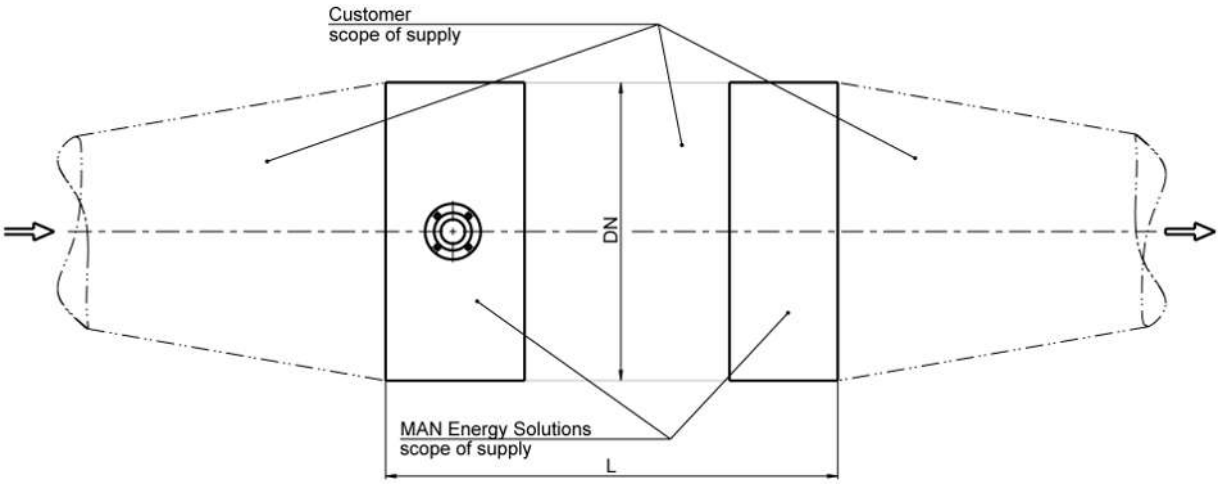

Figure 13: Mixing unit with urea injection lance

Mixing unit	Engine power approximately	Mixing pipe[1)]	Length straight mixing pipe (L)
No.	[kW]	DN	[mm]
1	0 – 1,000	500	1,160 – 3,450
2	1,001 – 2,000	600	2,650 – 3,500
3	2,001 – 3,000	800	2,400 – 3,610
4	3,001 – 4,200	1,000	2,740 – 3,740
5	4,201 – 5,400	1,100	3,330 – 3,810

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Mixing unit	Engine power approximately	Mixing pipe[1)]	Length straight mixing pipe (L)
No.	[kW]	DN	[mm]
6	5,401 – 6,800	1,200	3,420 – 3,890
7	6,801 – 8,500	1,400	3,260 – 3,980
8	8,501 – 10,500	1,500	3,760 – 4,110
9	10,501 – 13,000	1,600	3,840 – 4,480
10	13,001 – 20,000	2,100	3,490 – 4,480
11	20,001 – 21,600	2,300	4,130 – 4,690

[1)] Diameter mixing pipe differs from exhaust pipe diameter.

Table 5: Mixing unit with urea injection lance

Dosing unit

Dosing unit	Height	Width	Depth	Weight
No.	[mm]	[mm]	[mm]	[kg]
1	800	800	300	80

Table 6: Dosing unit

SCR cabinet

SCR cabinet	Height	Width	Depth	Weight
No.	[mm]	[mm]	[mm]	[kg]
1	800	400	300	30

Table 7: SCR cabinet

Pump module

Pump module	Height	Width	Depth	Weight
No.	[mm]	[mm]	[mm]	[kg]
1	1,800	1,000	400	210

Table 8: Pump module

Compressed air reservoir module

Air module	Height	Width	Depth	Weight
No.	[mm]	[mm]	[mm]	[kg]
1	1,730	810	620	195

Table 9: Compressed air reservoir module

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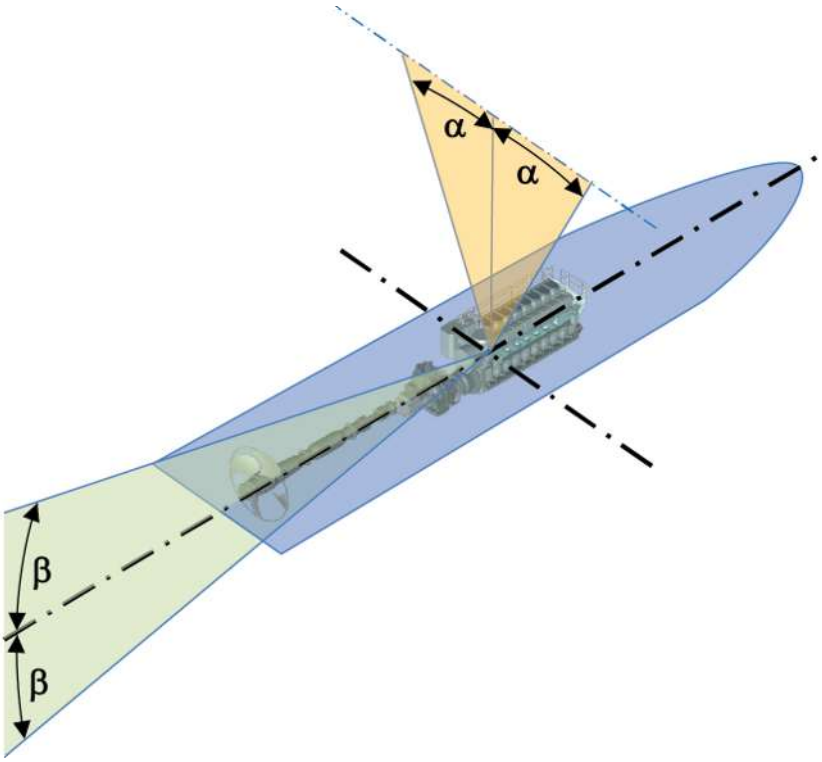
2.4.6 Engine inclination



α Athwartships β Fore and aft

Figure 14: Angle of inclination

Max. permissible angle of inclination [°][1]					
Application	**Athwartships α**		**Fore and aft β**		
	Heel to each side (static)	**Rolling to each side (dynamic)**	**Trim (static)[2]**		**Pitching (dynamic)**
			L < 100 m	**L > 100 m**	
Main engines	15	22.5	5	500/L	7.5

[1] Athwartships and fore and aft inclinations may occur simultaneously.

[2] Depending on length L of the ship.

Table 10: Inclinations

Note:
For higher requirements contact MAN Energy Solutions. Arrange engines always lengthwise of the ship.

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2.4.7 Engine equipment for various applications

Device/measure, (figure pos.)	Ship	
	Mechanical propulsion	Electric propulsion
Charge air by-pass ("hot compressor by-pass")	O	O
VTA for Lambda control	X	X
Compressor wheel cooling	X	X
Turbocharger – Compressor cleaning device (wet)	X	X
Turbocharger – Turbine cleaning device (dry)	X	X
SCR system	X	X
Jet assist	O/X[1)]	O/X[2)]
Two-stage charge air cooler	X	X
Charge air preheating by HT-LT switching	O	O
Adaptive Combustion Control (ACC)	X	X
VIT	X[3)]	X[3)]
VVT (optional available for 1,050 kW/cyl. engine variant)	O[3)]	O[3)]
Oil mist detector	X	X
Splash oil monitoring	X	X
Main bearing temperature monitoring	X	X
Starting system – Starting air valves within cylinder head	X	X
Slow Turn	X	X
Attached HT cooling water pump	X	X
Attached LT cooling water pump	O	O
Attached lube oil pump	X	X
Sealed plunger main injection pumps	X	X
Fuel oil duplex filter (only pilot fuel system)	X	X
Lube oil pressure control valve (only V engine)	X	X
Torque measurement flange	X	–

X = required, O = optional, – = not required

[1)] Jet assist is required, if a shaft generator with an output higher than 25 % of the nominal engine output is attached to the gear/engine.

[2)] Jet assist is required, if highly dynamical load application (0 % to 100 % within 3 load steps) is desired. Check section Start-up and load application, Page 54 and the plant requirements for the additionally necessity of jet assist.

[3)] If optional VVT for low soot emission is required, it is instead of VIT.

Table 11: Engine equipment

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Engine equipment for various applications – General description

Charge air by-pass ("hot compressor by-pass", see figure Overview flaps, Page 39)

For gas and DF engines it is used at cold ambient conditions to blow by a part of the hot charge air downstream of the compressor into the intake air duct. This serves for preheating the intake air and thereby expands the engine-specific "temperature compensation range". This feature is only available in connection with an external intake air system. It can not be applied to an engine with TC silencer.

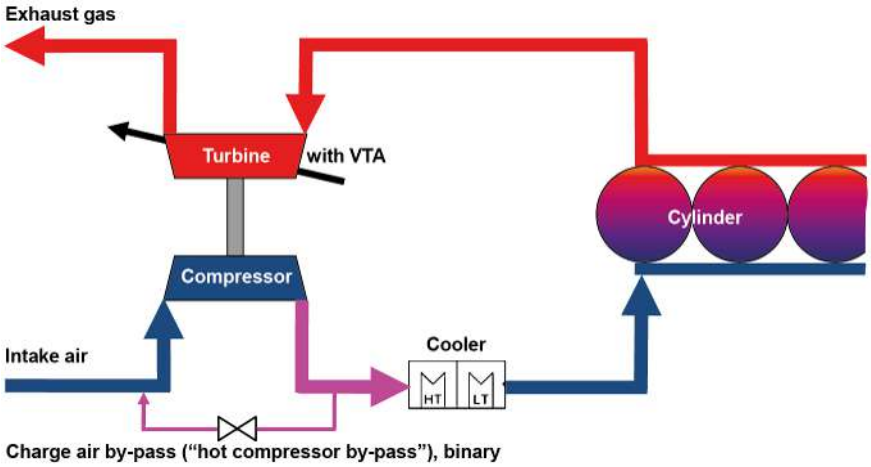

Figure 15: Overview flaps

VTA for lambda control and temperature after turbine control

VTA turbochargers (Variable Turbine Area) allow precise, stepless and continuous control of charge air pressure and air-flow according to the respective engine operating conditions.

For plants with an SCR catalyst, downstream of the turbine, a minimum exhaust gas temperature upstream the SCR catalyst is necessary in order to ensure its proper performance.

In case the temperature downstream the turbine falls below the set minimum exhaust gas temperature value, the VTA is also used to regulate the temperature upstream of the SCR catalyst.

Compressor wheel cooling

The compressor wheel cooling is integrated in the bearing casing of the turbocharger and lowers the temperature in the relevant areas of the compressor. It depends on the application or project-specific requirements if the compressor wheel cooling is required. As a rule of thumb a pressure ratio of approximately 1:4.5 and higher demands this cooling feature.

Turbocharger – Compressor cleaning device (wet)

Depending on the quality of the intake air, deposits may be formed on the blades of the compressor wheel and diffuser. This contamination reduces the efficiency of the compressor. Cleaning of the compressor is carried out with water during operation at full load with a special compressor cleaning device.

Turbocharger – Turbine cleaning device (dry)

The turbochargers of engines operated with heavy fuel oil (HFO), marine diesel oil (MDO) or marine gas oil (MGO) must be cleaned prior to initial operation and at regular intervals to remove combustion residue from the blades of the turbine rotor and nozzle ring.

Dry cleaning of the turbine should also be applied, in case of operation with inferior gas quality. Dry cleaning of the turbine is particularly suitable for cleaning the turbine rotor (turbine blades). Herefore a special cleaning device to be used.

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SCR system	The SCR system uses a reduction agent to transform the pollutant NO_x into environmentally friendly nitrogen and water vapour. The MAN Energy Solutions SCR system is capable of complying with the IMO Tier III limits over the entire range of applications.
Jet assist	Jet assist is used to improve the dynamic behavior of the engine to load steps and ramps. By means of nozzles in the turbocharger, compressed air is directed onto the compressor wheel to accelerate it. This results in a quicker adaption of the turbocharger to the new load condition. Jet assist is generally working with a compressed air pressure between 18 bar and 30 bar at the engine connection. However, it works most efficiently with a pressure of 27 bar. Therefore, the pressure difference from the compressed air tanks to the engine connections should, in any case, not be higher than 3 bar. Jet assist activating time: 3 seconds to 10 seconds (5 seconds on average).
Two-stage charge air cooler	The two-stage charge air cooler consists of two stages which differ in the temperature level of the connected water circuits. The charge air is first cooled by the HT circuit (high temperature stage of the charge air cooler, engine) and then further cooled down by the LT circuit (low temperature stage of the charge air cooler, lube oil cooler).
Charge air preheating by HT-LT switching	Charge air preheating by HT-LT switching is used in the load range from 0 % up to 20 % to achieve high charge air temperatures during part-load operation. It contributes to improved combustion and, consequently, reduced exhaust gas discoloration. Unlike the charge air preheating by means of the CHATCO control valve, there is no time delay in this case. The charge air is preheated immediately after the switching process by HT cooling water, which is routed through both stages of the two-stage charge air cooler.
Adaptive combustion control (ACC)	ACC evaluates the combustion and adapts each cylinder individually for an optimum overall performance of the engine, based on cylinder pressure sensors implemented in the cylinder head.
VIT	For some engine types with conventional injection a VIT (Variable Injection Timing) is available allowing a shifting of injection start. A shifting in the direction of “advanced injection” is supposed to increase the ignition pressure and thus reduces fuel consumption. Shifting in the direction of “retarded injection” helps to reduce NO_x emissions.
VVT	VVT (Variable Valve Timing) enables variations in the opening and closing timing of the inlet valves. At low-load operation it is used to attain higher combustion temperatures and thus lower soot emissions. At higher loads it is used to attain low combustion temperatures and thus lower NO_x emissions (Miller Valve Timing).
Oil mist detector	Bearing damage, piston seizure and blow-by in combustion chamber leads to increased oil mist formation. As a part of the safety system the oil mist detector monitors the oil mist concentration in crankcase to indicate these failures at an early stage.
Splash oil monitoring	The splash oil monitoring system is a constituent part of the safety system. Sensors are used to monitor the temperature of each individual drive unit (or pair of drive at V engines) indirectly via splash oil.
Main bearing temperature monitoring	As an important part of the safety system the temperatures of the crankshaft main bearings are measured just underneath the bearing shells in the bearing caps. This is carried out using oil-tight resistance temperature sensors.

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Starting system – Starting air valves within cylinder head

The engine is equipped with starting air valves within some of the cylinder heads. On starting command, compressed air will be led in a special sequence into the cylinder and will push down the piston and turn thereby the crankshaft untill a defined speed is reached.

Slow Turn

Engines, which are equipped with Slow Turn, are automatically turned prior to engine start with the turning process being monitored by the engine control. If the engine does not reach the expected number of crankshaft revolutions (2.5 revolutions) within a specified period of time, or in case the Slow Turn time is shorter than the programmed minimum Slow Turn time, an error message is issued. This error message serves as an indication that there is liquid (oil, water, fuel) in the combustion chamber. If the Slow Turn manoeuvre is completed successfully, the engine is started automatically.

Slow Turn is always required for plants with power management system (PMS) demanding automatic engine start.

Sealed plunger main injection pumps (SP injection pumps)

These conventional injection pumps have been designed for an operation with all specified fuels and especially for a complete separation of the fuel and the lube oil circuits within the injection pumps. And as fuel and lube oil are completely separated and cannot get in contact with each other, accordingly the leakage fuel of the SP injection pumps can be completely reused again. For the same reason, there is no need for sealing oil anymore in the case of continuous MGO-operation.

Torque measurement flange

For a mechanical CP (controllable pitch) propeller driven by a dual fuel engine, a torque measurement flange has to be provided. The torque measurement flange gives an accurate power output signal to the engine control, thus enabling exact lambda control and rapid switchover operations (liquid fuel/gas and vice versa).

2.5 Ratings (output) and speeds

2.5.1 General remark

The engine power which is stated on the type plate derives from the following sections and corresponds to $P_{Operating}$ as described in section Derating, definition of P Operating, Page 44.

2.5.2 Standard engine ratings

$P_{ISO, standard}$: ISO standard output (as specified in DIN ISO 3046-1)

No. of cylinders, config.	Engine rating, $P_{ISO, standard}$ [1) 2)]			
	1,050 kW/cyl., 500 or 514 rpm		1,150 kW/cyl., 500 or 514 rpm	
	Available turning direction CW/CCW[3)]	kW	Available turning direction CW/CCW[3)]	kW
6L	Yes/Yes	6,300	Yes/Yes	6,900
7L	Yes/Yes	7,350	Yes/Yes	8,050
8L	Yes/Yes	8,400	Yes/Yes	9,200
9L	Yes/Yes	9,450	Yes/Yes	10,350
12V	Yes/Yes	12,600	Yes/Yes	13,800

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No. of cylinders, config.	Engine rating, $P_{ISO, standard}$ [1) 2)]			
	1,050 kW/cyl., 500 or 514 rpm		1,150 kW/cyl., 500 or 514 rpm	
	Available turning direction CW/CCW[3)]	kW	Available turning direction CW/CCW[3)]	kW
14V	Yes/Yes	14,700	Yes/Yes	16,100
16V	Yes/Yes	16,800	-	-

Note: Nm^3 corresponds to one cubic meter of gas at 0 °C and 101.32 kPa abs.

1) $P_{ISO, standard}$ as specified in DIN ISO 3046-1, see paragraph Reference conditions for engine rating, Page 42.

2) Engine fuel: Liquid fuel mode = distillate according to ISO 8217 or RM-grade fuel, fulfilling the stated quality requirements. Gas mode = natural gas with a methane number ≥ 80, NCV ≥ 28,000 kJ/Nm^3 and fulfilling the stated quality requirements.

3) CW = clockwise; CCW = counter clockwise.

Table 12: Engine ratings

Reference conditions for engine rating

According to ISO 15550; ISO 3046-1

Air temperature before turbocharger t_r	K/°C	298/25
Total atmospheric pressure p_r	kPa	100
Relative humidity Φ_r	%	30
Cooling water temperature inlet charge air cooler (LT stage)	K/°C	298/25

Table 13: Reference conditions for engine rating

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2.5.3 Engine ratings (output) for different applications

$P_{Application, ISO}$: Available rating (output) under ISO conditions dependent on application

	$P_{Application}$ Available output in percentage of ISO standard output	Max. fuel admission (blocking)	Max. permissible speed reduction at maximum torque[1)]	Tropic conditions (t_r/t_{cr}/ p_r=100kP_a)[2)]	Notes	Optional power take-off in percentage of ISO standard output
Kind of application	**%**	**%**	**%**	**°C**		**%**
Marine main engines with mechanical or electric propulsion						
Main drive with electric propulsion	100	110	-	45/38	[3)]	Up to 100
Main drive with controllable pitch propeller	100	100	-	45/38	-	Up to 100

[1)] Maximum torque given by available output and nominal speed.

[2)] t_r = air temperature at compressor inlet of turbocharger,

t_{cr} = cooling water temperature before charge air cooler,

p_r = atmospheric pressure.

[3)] In accordance with DIN ISO 3046-1 and for further clarification of relevant sections within DIN ISO 8528-1, the following is specified:
- The maximum output (MCR) has to be observed by the power management system of the plant.
- The range of 100 % up to 110 % fuel admission may only be used for a short time for governing purposes (e.g. transient load conditions and suddenly applied load).

Table 14: Available outputs/related reference conditions

Note:
Power fluctuations in the electrical grid.
It is an intrinsic property of the powertrain of a generating set that it acts as a torsional vibration system. This complex system consists of the engine, coupling and generator (within or outside MAN Energy Solutions scope of supply) and the electric plant. Such electric power plant can be consisting of further power sources as well as consumers (such as electric motors), transformers, frequency converters, energy storage systems, bus bars or circuit breakers and the entire distribution system (within or outside MAN Energy Solutions scope of supply). The reciprocating engine, as well as the electric power distribution or the other consumers and power sources excite the system. As a consequence, the active power at the generator terminals is not completely constant over time and some additional power oscillations so-called power fluctuations occur. These power fluctuations do not affect the operational safety of the generating set, as long as the stability requirements of the electric system in regards to frequency and voltage meet the class requirements. In addition, this behavior is in accordance with ISO 8528-5 [1] and does as per MAN Energy Solutions experience not affect power system stability in an unacceptable range. Gas operated engines tend to show greater power fluctuations than engines operated with liquid fuels. This belongs to the typical cyclical fluctuations of the Otto combustion process that uses a premixed combustion. Note that MAN Energy Solutions quotations do not consider any spe-

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cific limitations regarding power fluctuations. On request by the purchaser, MAN Energy Solutions provides support or further analysis of the overall system behavior, where the GenSets as well as the electric distribution and consumers should be included in the analysis.

[1] Reciprocating internal combustion engine driven alternating current generating sets – Part 5: Generating sets.

2.5.4 Derating, definition of P Operating

$P_{Operating}$ – Liquid fuel mode relevant derating factors

Available rating (output) under local conditions and dependent on application.

Dependent on local conditions or special application demands a further load reduction of $P_{Application, ISO}$ might be required.

1. No derating

No derating necessary, provided that the conditions listed are met:

	No derating up to stated reference conditions (tropic), see 1.
Air temperature before turbocharger T_x	≤ 318 K (45 °C)
Ambient pressure	≥ 100 kPa (1 bar)
Cooling water temperature inlet charge air cooler (LT stage)	≤ 311 K (38 °C)
Intake air pressure before compressor	≥ –2 kPa[1)]
Exhaust gas back pressure after turbocharger	≤ 5 kPa[1)]
Relative humidity Φ_r	≤ 60 %
[1)] Below/above atmospheric pressure.	

Table 15: Derating – Limits of ambient conditions

2. Derating

Contact MAN Energy Solutions:

- If limits of ambient conditions mentioned in the upper table Derating – Limits of ambient conditions, Page 44 are exceeded. A special calculation is necessary.
- If higher requirements for the emission level exist. For the permissible requirements see section Exhaust gas emission, Page 157.
- If special requirements of the plant for heat recovery exist.
- If special requirements on media temperatures of the engine exist.
- If any requirements of MAN Energy Solutions mentioned in the Project Guide cannot be met.

$P_{Operating}$ – Gas mode relevant derating factors

Dependent on local conditions or special application a load reduction of $P_{Application, ISO}$ might be required. Accordingly the resulting output is called $P_{Operating}$.

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Relevant for a derating in gas mode are the methane number, the charge air temperature before cylinder, the N_2-content of the fuel gas and the ambient air temperature range, that needs to be compensated.

1. Derating if methane number is below minimum value

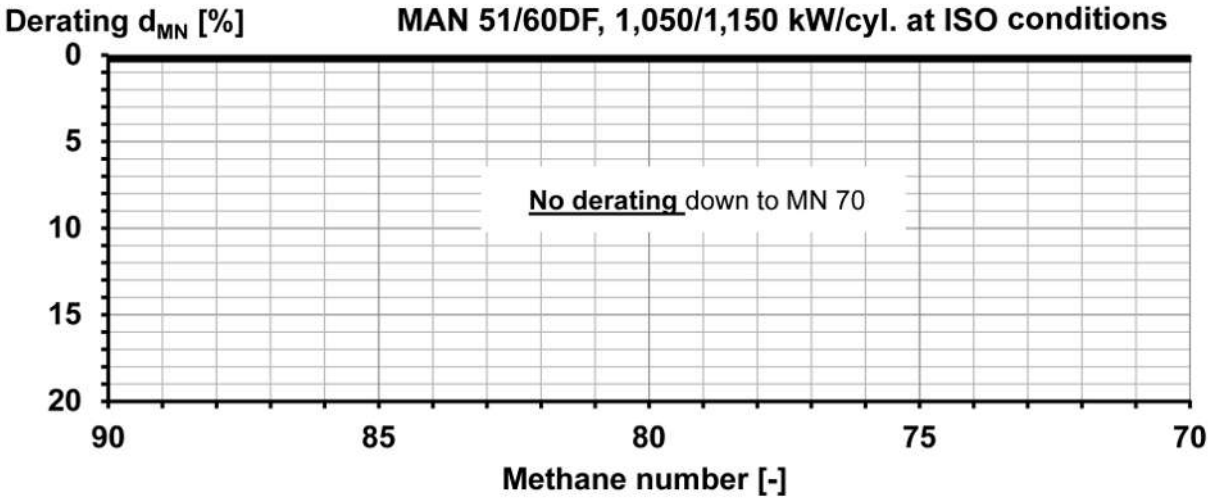

Figure 16: Derating d_{MN} as a function of methane number at ISO conditions

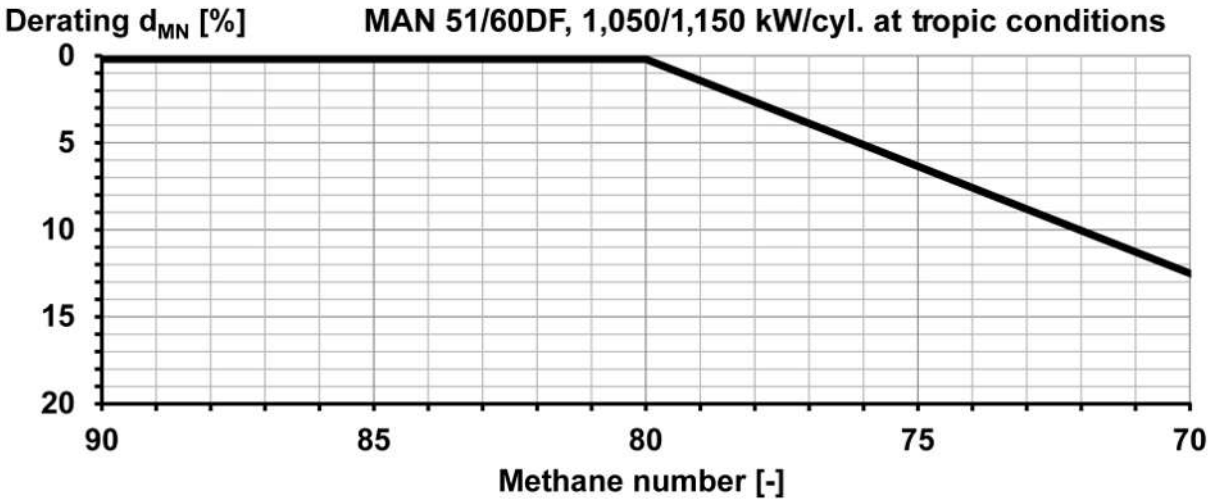

Figure 17: Derating d_{MN} as a function of methane number at tropic conditions

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2. Derating if maximum charge air temperature before cylinder is exceeded

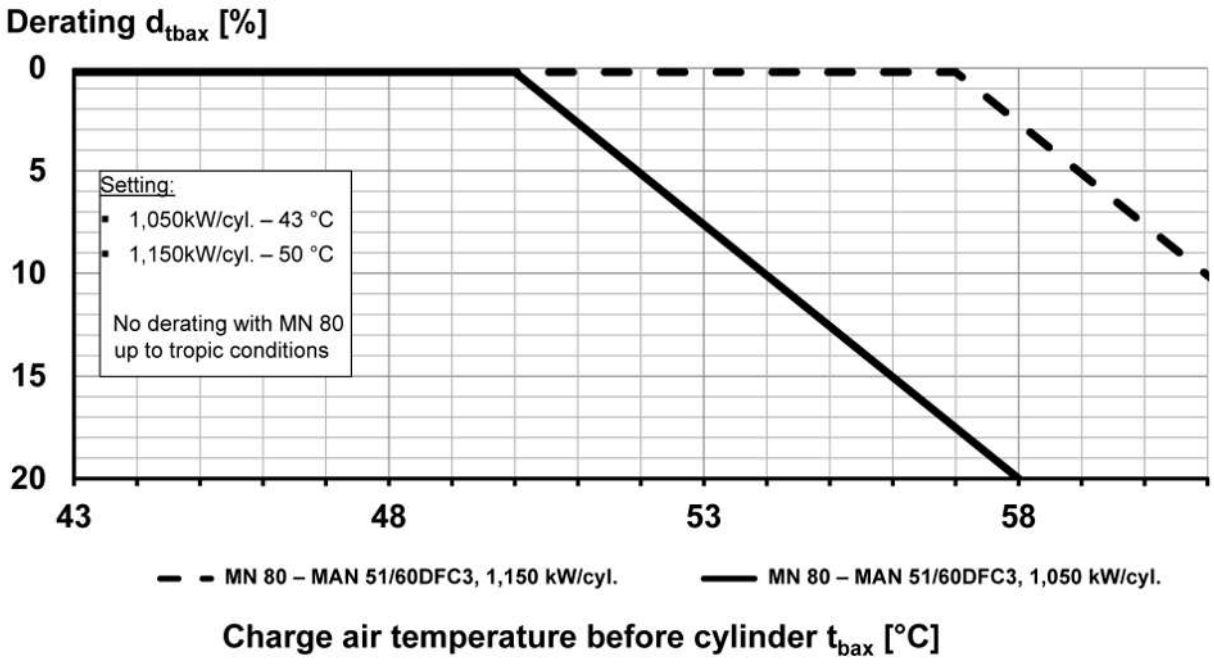

Figure 18: Derating d_{tbax} as a function of charge air temperature before cylinder

3. Derating if minimum NCV due to high N_2-content can not be kept

The NCV (Net caloric value) from the gas is influenced by the N_2-content. Up to 22 % of N_2-content no derating is necessary. Above 22 % to 30 % N_2-content derating is required.

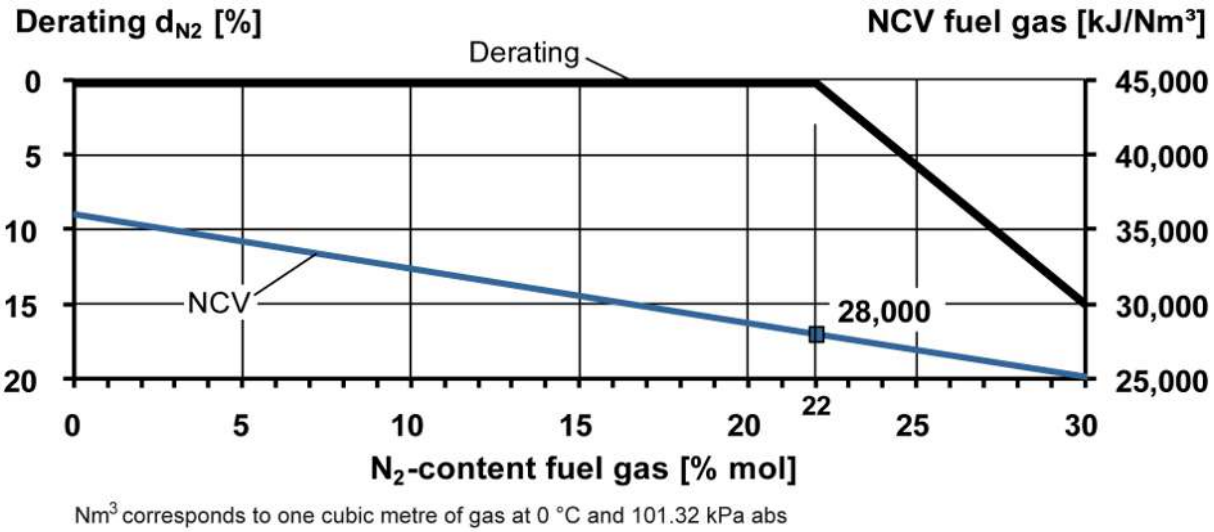

Figure 19: Derating d_{N2} as a function of N_2-content in the fuel gas

4. Derating if range of ambient air temperature compensation is exceeded

The main control device for air volume ratio adjustment (lambda control) of gas and DF engines is capable to compensate a wide range of changes of the ambient pressure and air temperature. For ambient air temperatures < 5 °C the intake air must be preheated to a minimum temperature of 5 °C

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before turbocharger. If the ambient air temperature exceeds the engine type relevant limit, the fuel air ratio adjustment is outside of its range and a derating of the engine output is required.

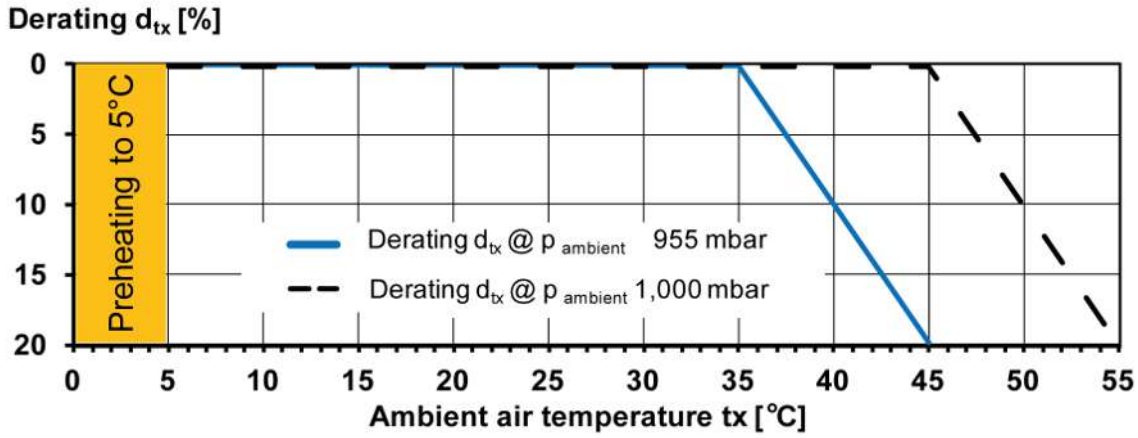

Figure 20: Derating d_{tx} if range of ambient temperature compensation is exceeded

5. Calculation of the total derating factor and $P_{Operating}$

The derating due to methane number d_{MN} and charge air temperature before cylinder d_{tbax} have to be considered additive (d_{MN} + d_{tbax}).

Beside this the derating due ambient air temperature d_{tx} and N_2-content d_{N2} have to be considered separately.

The highest element of (d_{MN} + d_{tbax}) or d_{tx} or d_{N2} has to be considered in the formula below.

$$\alpha = [100 - \max\{(d_{MN} + d_{tbax}), d_{tx}, d_{N_2}\}] / 100$$

$$P_{Operating} = P_{Application,ISO} \times \alpha$$

2.5.5 Engine speeds and related main data

Rated speed	rpm	500	514
Permissible range of speed		[1]	
Alarm overspeed (110 % of nominal speed)		550	566
Auto shutdown overspeed (115 % of nominal speed)		575	591
Speed adjusting range		See section Speed adjusting range, Page 48	
Alternator frequency	Hz	50	60
Number of pole pairs	-	6	7

[1] According to section Operating range for generator operation/electric propulsion, Page 74, Operating range for EPROX-AC, Page 75, Operating range for controllable pitch propeller (CPP), Page 82 and figure Permissible frequency deviations and corresponding max. output, Page 77.

Table 16: Engine speeds and related main data

2.5.6 Speed adjusting range

The following specification represents the standard settings. For special applications, deviating settings may be necessary.

	Drive	Speed droop	Maximum speed at full load	Maximum speed at idle running	Minimum speed
Electronic speed control	1 main engine with controllable pitch propeller and without PTO	0 %	100 % (+0.5 %)	100 % (+0.5 %)	60 %
	1 main engine with controllable pitch propeller and with PTO	0 %	100 % (+0.5 %)	100 % (+0.5 %)	60 %
	Parallel operation of 2 engines driving 1 shaft with/ without PTO: Load sharing via speed droop or master/slave operation	 5 % 0 %	 100 % (+0.5 %) 100 % (+0.5 %)	 105 % (+0.5 %) 100 % (+0.5 %)	 60 % 60 %
	GenSets/electric propulsion plants: With load sharing via speed droop or isochronous operation	 5 % 0 %	 100 % (+0.5 %) 100 % (+0.5 %)	 105 % (+0.5 %) 100 % (+0.5 %)	 100 %[1)] 100 %[1)]

[1)] Speed after start of the engine, before synchronisation.

Table 17: Electronic speed control

2.6 Increased exhaust gas pressure due to exhaust gas after treatment installations

Resulting installation demands

If the recommended exhaust gas back pressure as stated in section Operating/service temperatures and pressures, Page 147 cannot be met due to exhaust gas after treatment installations following limit values need to be considered.

Exhaust gas back pressure after turbocharger	
Operating pressure Δp_{exh}, maximum specified	0 – 50 mbar
Operating pressure Δp_{exh}, range with increase of fuel consumption or possible derating	50 – 80 mbar
Operating pressure Δp_{exh}, where agreement and feedback of MAN Energy Solutions is required	> 80 mbar

Table 18: Exhaust gas back pressure after turbocharger

Intake air pressure before turbocharger	
Operating pressure Δp_{intake}, standard	0 – –20 mbar
Operating pressure Δp_{intake}, range with increase of fuel consumption or possible derating	–20 – –40 mbar

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Intake air pressure before turbocharger	
Operating pressure Δp_{intake}, where agreement and feedback of MAN Energy Solutions is required	< –40 mbar

Table 19: Intake air pressure before turbocharger

Sum of the exhaust gas back pressure after turbocharger and the absolute value of the intake air pressure before turbocharger	
Operating pressure Δp_{exh} + Abs(Δp_{intake}), standard	0 – 70 mbar
Operating pressure Δp_{exh} + Abs(Δp_{intake}), range with increase of fuel consumption or possible derating	70 – 120 mbar
Operating pressure Δp_{exh} + Abs(Δp_{intake}), where agreement and feedback of MAN Energy Solutions is required	> 120 mbar

Table 20: Sum of the exhaust gas back pressure after turbocharger and the absolute value of the intake air pressure before turbocharger

Maximum exhaust gas pressure drop – Layout

- Supplier of equipment in exhaust gas line have to ensure that pressure drop Δp_{exh} over entire exhaust gas piping incl. pipe work, scrubber, boiler, silencer, etc. must stay below stated standard operating pressure at all operating conditions.
- It is recommended to consider an additional 10 mbar for consideration of aging and possible fouling/staining of the components over lifetime.
- A proper dimensioning of the entire flow path including all installed components is advised or even the installation of an exhaust gas blower if necessary.
- At the same time the pressure drop Δp_{intake} in the intake air path must be kept below stated standard operating pressure at all operating conditions and including aging over lifetime.
- For significant overruns in pressure losses even a reduction in the rated power output may become necessary.
- On plant side it must be prepared, that pressure sensors directly after turbine outlet and directly before compressor inlet may be installed to verify above stated figures.

By-pass for emergency operation

- Evaluate if the chosen exhaust gas after treatment installation demands a by-pass for emergency operation.
- For scrubber application, a by-pass is recommended to ensure emergency operation in case that the exhaust gas cannot flow through the scrubber freely.
- The by-pass needs to be dimensioned for the same pressure drop as the main installation that is by-passed – otherwise the engine would operated on a differing operating point with negative influence on the performance, e.g. a lower value of the pressure drop may result in too high turbocharger speeds.

Single streaming per engine recommended/multi-streaming to be evaluated project-specific

- In general each engine must be equipped with a separate exhaust gas line as single streaming installation. This will prevent reciprocal influencing of the engine as e.g. exhaust gas backflow into an engine out of operation or within an engine running at very low load (negative pressure drop over the cylinder can cause exhaust gas back flow into intake manifold during valve overlap).

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- In case a multi-streaming solution is realised (i.e. only one combined scrubber for multiple engines) this needs to be stated on early project stage. Hereby air/exhaust gas tight flaps need to be provided to safeguard engines out of operation. A specific layout of e.g. sealing air mass flow will be necessary and also a power management may become necessary in order to prevent operation of several engines at very high loads while others are running on extremely low load. A detailed analysis as HAZOP study and risk analysis by the yard becomes mandatory.

Engine to be protected from backflow of media out of exhaust gas after treatment installation

- A backflow of e.g. urea, scrubbing water, condensate or even rain from the exhaust gas after treatment installation towards the engine must be prevented under all operating conditions and circumstances, including engine or equipment shutdown and maintenance/repair work.

Turbine cleaning

- Both wet and dry turbine cleaning must be possible without causing malfunctions or performance deterioration of the exhaust system incl. any installed components such as boiler, scrubber, silencer, etc.

White exhaust plume by water condensation

- When a wet scrubber is in operation, a visible exhaust plume has to be expected under certain conditions. This is not harmful for the environment. However, countermeasures like reheating and/or a demister should be considered to prevent condensed water droplets from leaving the funnel, which would increase visibility of the plume.
- The design of the exhaust system including exhaust gas after treatment installation has to make sure that the exhaust flow has sufficient velocity in order not to sink down directly onboard the vessel or near to the plant. At the same time the exhaust pressure drop must not exceed the limit value.

Vibrations

- There must be a sufficient decoupling of vibrations between engine and exhaust gas system incl. exhaust gas after treatment installation, e.g. by compensators.

2.7 Starting

2.7.1 General remarks

Engine and plant installation need to be in accordance to the below stated requirements and the required starting procedure.

Note:
Statements are relevant for non arctic conditions.
For arctic conditions consider relevant sections and clarify undefined details with MAN Energy Solutions.

2.7.2 Type of engine start

Normal start

The standard procedure of a monitored engine start in accordance to MAN Energy Solutions guidelines.

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Stand-by start

Shortened starting up procedure of a monitored engine start: Several preconditions and additional plant installations required.

This kind of engine start has to be triggered by an external signal: "Stand-by start required".

Exceptional start (e.g. blackout start)

A monitored engine start (without monitoring of lube oil pressure) within one hour after stop of an engine that has been faultless in operation or of an engine in stand-by mode.

This kind of engine start has to be triggered by an external signal “Black Start” and may only be used in exceptional cases.

Emergency start

Manual start of the engine at emergency start valve at the engine (if applied), without supervision by the SaCoS engine control. These engine starts will be applied only in emergency cases, in which the customer accepts, that the engine might be harmed.

2.7.3 Requirements on engine and plant installation

General requirements on engine and plant installation

As a standard and for the start-up in normal starting mode (preheated engine) following installations are required:

Engine
- Lube oil service pump (attached).

Plant
- Prelubrication pump (free-standing).
- Preheating HT cooling water system (60 – 90 °C).
- Preheating lube oil system (> 40 °C). For maximum admissible value see table Lube oil, Page 149.

Requirements on engine and plant installation for "Stand-by operation" capability

To enable in addition to the normal starting mode also an engine start from PMS (power management system) from stand-by mode with thereby shortened start-up time following installations are required:

Engine
- Lube oil service pump (attached).

Plant
- Prelubrication pump (free-standing) with low pressure before engine (0.3 bar < $p_{Oil\ before\ engine}$ < 0.6 bar).
- Preheating HT cooling water system (60 – 90 °C).
- Preheating lube oil system (> 40 °C). For maximum admissible value see table Lube oil, Page 149.
- Power management system with supervision of stand-by times engines.

Additional requirements on engine and plant installation for "Blackout start" capability

Following additional installations to the above stated ones are required to enable in addition a "Blackout start":

Engine
- HT CW service pump (attached) recommended.

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Plant

- LT CW service pump (attached) recommended.
- Attached fuel oil supply pump recommended (if applicable).
- Regarding "Blackout start" fuel oil conditions, see table Fuel, Page 150.

If fuel oil supply pump is not attached to the engine:

- Air driven fuel oil supply pump or fuel oil service tanks at sufficient height or pressurised fuel oil tank.

2.7.4 Starting conditions

Type of engine start:	Blackout start	Stand-by start	Normal start
Explanation:	After blackout	From stand-by mode	After stand-still
Start-up time until load application:	< 1 minute	< 1 minute	> 2 minutes
General notes			
-	Engine start-up only within 1 h after stop of engine that has been faultless in operation or within 1 h after end of stand-by mode.	Maximum stand-by time 7 days[1]. Supervised by power management system plant. Stand-by mode is only possible after engine has been faultless in operation and has been faultless stopped.	Standard
Additional external signal:	Blackout start	Stand-by request	-

[1] If an engine has been in total for 7 days in stand-by mode, no extension of stand-by mode is allowed. The engine needs to be started and operated faultless before the next stand-by mode can be applied.

Table 21: Starting conditions – General notes

Type of engine start:	Blackout start	Stand-by start	Normal start
General engine status	No start-blocking active	Engine in proper condition No start-blocking active Note: Start-blocking of engine leads to withdraw of "stand-by mode".	Engine in proper condition No start-blocking active
Engine to be turned before start?	No	No	Yes[1]
Engine to be preheated and prelubricated?	No[2]	Yes	Yes

[1] It is recommended to install Slow Turn. Otherwise the engine has to be turned by turning gear.

[2] Valid only, if mentioned above conditions (see table Starting conditions – General notes, Page 52) have been considered. Non-observance endangers the engine or its components.

Table 22: Starting conditions – Required engine conditions

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Additional remark regarding "Blackout start"

If additional requirements on engine and plant installation for "Blackout start" capability are fullfilled, it is possible to start up the engine in shorter time. But untill all media systems are back in normal operation the engine can only be operated according to the settings of alarm and safety system.

Type of engine start:	Blackout start	Stand-by start	Normal start
Lube oil system			
Prelubrication period	No[1)]	Permanent	Yes, previous to engine start
Prelubrication pressure before engine	-	See section Operating/service temperatures and pressures, Page 147 limits according figure "Prelubrication/postlubrication lube oil pressure (duration > 10 min)"	See section Operating/service temperatures and pressures, Page 147 limits according figure "Prelubrication/postlubrication lube oil pressure (duration ≤ 10 min)"
Lube oil to be preheated?	No[1)]	Yes	Yes
HT cooling water			
HT cooling water to be preheated?	No[1)]	Yes	Yes
Fuel system			
For MGO/MDO operation	Sufficient fuel oil pressure at engine inlet required.	Supply pumps in operation or with starting command to engine.	
For HFO operation	Sufficient fuel oil pressure at engine inlet required. Emergency fuel supply pumps in MGO/MDO mode always.	Supply and booster pumps in operation, fuel preheated to operating viscosity. In case of permanent stand-by of liquid fuel engines or during operation of an DF engine in gas mode a periodical exchange of the circulating HFO has to be ensured to avoid cracking of the fuel. This can be done by releasing a certain amount of circulating HFO into the day tank and substituting it with "fresh" fuel from the tank.	
For gas operation	Blackout start only in liquid fuel operation.	Conditions for MGO/MDO respectively HFO to be fulfilled + fuel gas supply line in operation or goes in operation with starting command to engine.	

[1)] Valid only, if mentioned above conditions (see table Starting conditions – General notes, Page 52) have been considered. Non-observance endangers the engine or its components.

Table 23: Starting conditions – Required system conditions

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2.8 Start-up and load application

2.8.1 General remarks

In the case of highly-supercharged engines, load application is limited. This is due to the fact that the charge air pressure build-up is delayed by the turbocharger run-up. Besides, a low-load application promotes uniform heating of the engine.

In general, requirements of the International Association of Classification Societies (IACS) and of ISO 8528-5 are valid.

According to performance grade G2 concerning:

- Dynamic speed drop in % of the nominal speed ≤ 10 %.
- Remaining speed variation in % of the nominal speed ≤ 5 %.
- Recovery time until reaching the tolerance band ±1 % of nominal speed ≤ 5 seconds.

Clarify any higher project-specific requirements at an early project stage with MAN Energy Solutions. They must be part of the contract.

In a load drop of 100 % nominal engine power, the dynamic speed variation must not exceed:

- 12 % of the nominal speed.
- The remaining speed variation must not surpass 5 % of the nominal speed.

To limit the effort regarding regulating the media circuits, also to ensure an uniform heat input it always should be aimed for longer load application times by taking into account the realistic requirements of the specific plant.

All questions regarding the dynamic behaviour should be clarified in close cooperation between the customer and MAN Energy Solutions at an early project stage.

Requirements for plant design:

- The load application behaviour must be considered in the electrical system design of the plant.
- The system operation must be safe in case of graduated load application.
- The load application conditions (E-balance) must be approved during the planning and examination phase.
- The possible failure of one engine must be considered, see section Generator operation/electric propulsion – Power management, Page 77.

2.8.2 Definitions and requirements

General remark

Prior to the start-up of the engine it must be ensured that the emergency stop of the engine is working properly. Additionally all required supply systems must be in operation or in stand-by operation.

Start-up – Cold engine

If an engine start has to be activated under cold engine conditions, following requirements have to be fullfilled as a minimum:

- Lube oil temperature > 20 °C, HT cooling water temperature > 20 °C.
- Distillate fuel must be used until warming up phase is completed.
- The engine is prelubricated. Due to the higher viscosity of the lube oil of a cold engine **the prelubrication phase needs to be increased**.

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Before further use of the engine a warming-up phase is required to reach at least the level of the regular preheating temperatures (lube oil temperature > 40 °C, cooling water temperature > 60 °C). See diagrams in section Load application – Continuous loading, Page 56.

Note:

- It needs to be proven within plant layout, that lube oil circuit is capable to be operated at stated low lube oil temperature with accordingly high viscosity and high pressures.
- If engine cold start is frequently performed, wear could increase in a long-term perspective.
- "Start-Reliability" under stated cold start conditions is reduced and cannot be guaranteed, as the probability of a false start is increased.
- If applicable, warming-up phase can be shortened if engine is operated at lower speed.

Start-up – Preheated engine (Normal start)

For the start-up of the engine it needs to be preheated:

- Lube oil temperature ≥ 40 °C.
- HT cooling water temperature ≥ 60 °C.

The required start-up time in normal starting mode (preheated engine), with the required time for starting-up the lube oil system and prelubrication of the engine is shown in the diagrams in section Load application – Continuous loading, Page 56 in connection with the information in figure(s) Duration of the load application – Continuous loading, Page 58.

Start-up – Engine in stand-by mode (Stand-by start)

For engines in stand-by mode no start preparation is needed and accordingly the engine start will be done just after the start request (if preconditions are fulfilled).

Required conditions media system:

- 0.3 bar < prelubrication pressure before engine < 0.6 bar.
- Lube oil temperature ≥ 40 °C, see accordingly section External lube oil system, Page 313
- HT cooling water temperature ≥ 60 °C

Start-up (Exceptional start)

The engine start will be done just after the start request – but as previously stated without monitoring of lube oil pressure, and therefore this may only be used in exceptional cases.

Speed ramp-up

The standard speed ramp-up serves for all engine conditions and ensures a low opacity level of the exhaust gas.

A "fast speed ramp-up", that is near to the maximum capability of the engine, may be used in exceptional cases.

For liquid fuel engines:

- Exhaust gas will be visible (opacity > 60 %).
- Engine must be equipped with jet assist.
- Sufficient air pressure for jet assist activation must be available.
- External signal from plant to be provided for request to SaCoSone.

Load ramp-up

The time needed for load ramp-up is in high extent dependent on the engine conditions:

- Cold
  - Lube oil temperature > 20 °C.
  - HT cooling water temperature > 20 °C.

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- Warm (= preheated)
  - Lube oil temperature ≥ 40 °C.
  - HT cooling water temperature ≥ 60 °C.
- Hot (= previously been in operation)
  - Lube oil temperature ≥ 40 °C.
  - HT cooling water temperature ≥ 60 °C.
  - Exhaust gas pipe engine and turbocharger > 320 °C (within 1 h after engine stop).

Note:
Load application handled within plant automation:
The compliance of the load application with the specifications of MAN Energy Solutions has to be handled within the plant automation.
The SaCoS engine control will not interfere in the load ramp-up or load ramp-down initiated by the plant control.

2.8.3 Load application – Continuous loading

The start procedure after "Starting request" is structured into following phases:

Start preparation (consists of phase 1, 2 and 3)

- Phase 1:
  Pressure formation in the lube oil system to gain specified lube oil pressure before engine.
- Phase 2:
  Prelubrication and pre-processing of the engine for specified time.
- Phase 3:
  "Slow Turn" function activated, if required.

Start process and running-up to minimum speed (consists of phase 4 and 5)

- Phase 4:
  Activation of starting system engine till engine is running.
- Phase 5:
  Increase of engine speed till specified minimum speed is reached.

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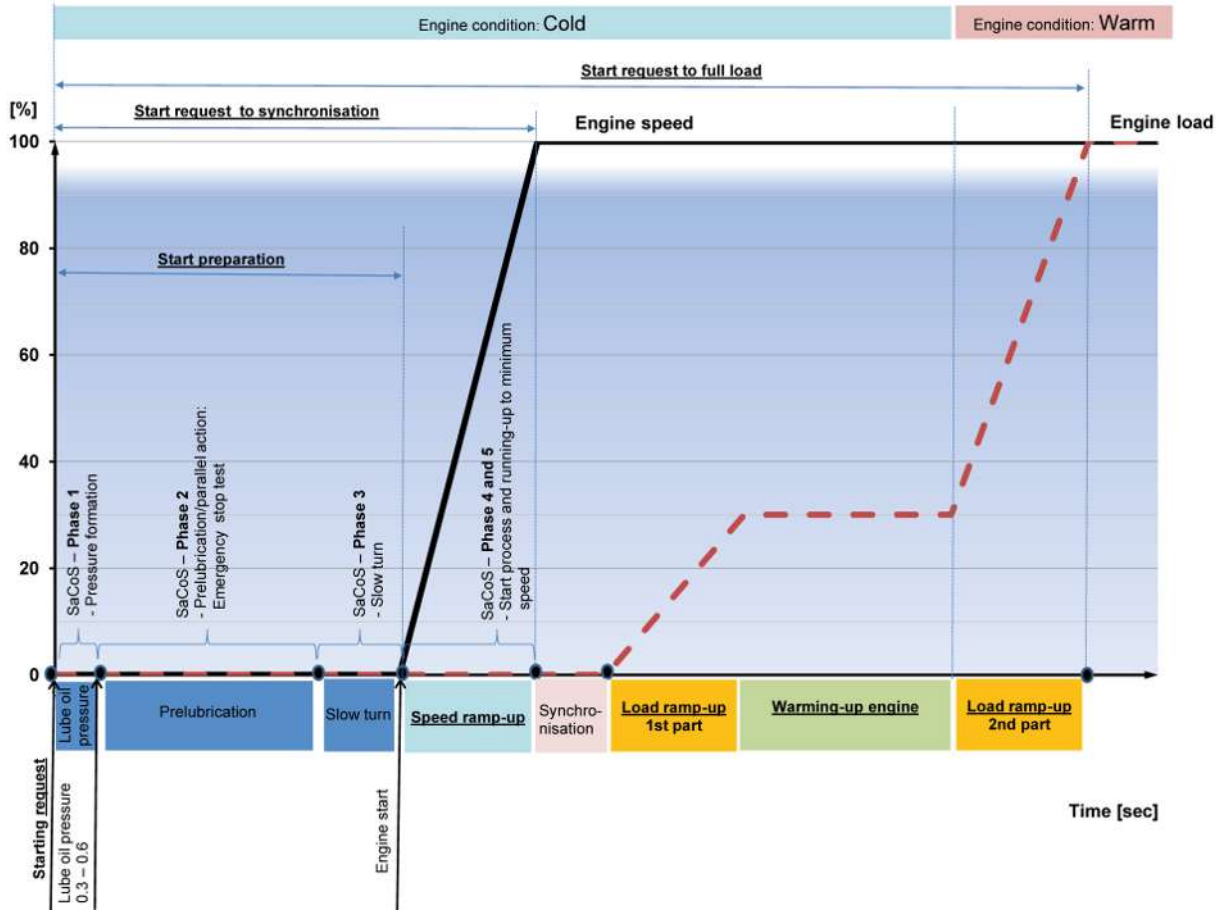

Figure 21: Start-up and load ramp-up for cold engine condition (emergency case)

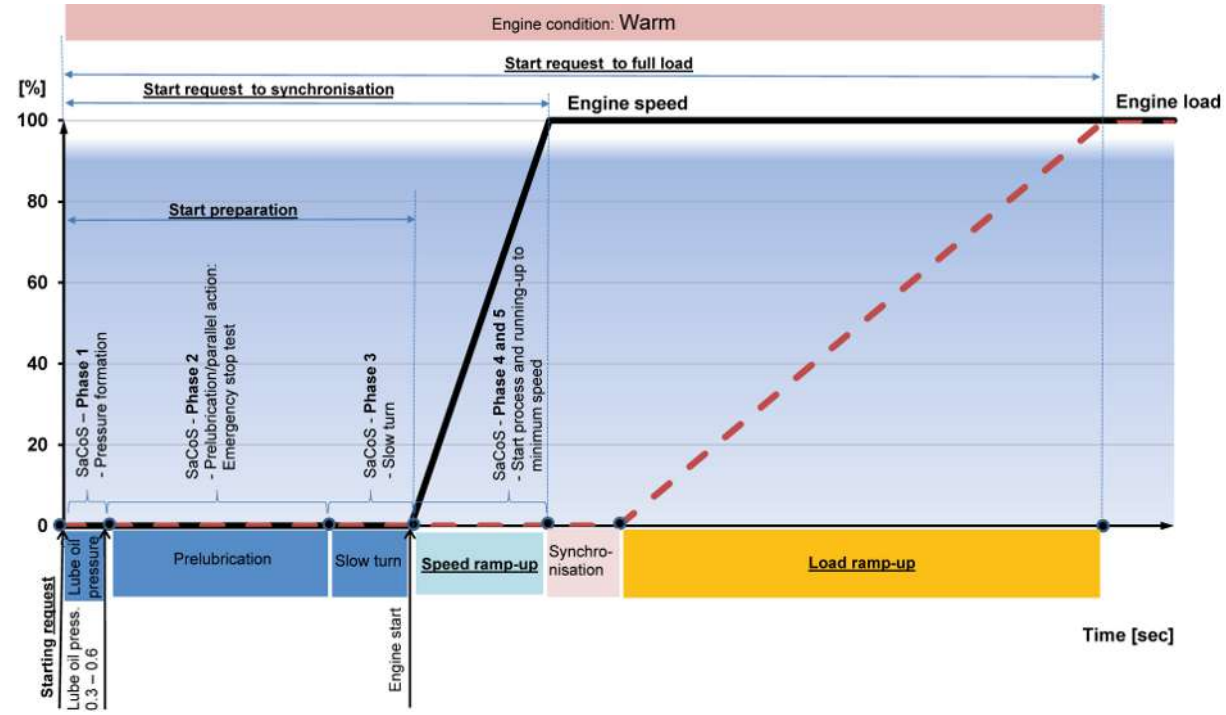

Figure 22: Start-up and load ramp-up for warm/hot engine condition

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Find in the table below the relevant durations for the phases in above given diagrams.

Note:

- For "Phase 3" the engine needs to be equipped with "Slow Turn".
- Jet assist as engine equipment is recommended.
- If "fast speed ramp-up" is needed, the possibility of this has to be clarified on a project-specific basis.
- For "stand-by" special plant equipment is required.

Special note for EPROX-AC:

- Load ramps are valid for speed range 80 % up to 100 %.
- Speed has to be ramped up with 1 %/s of nominal speed in parallel to load ramp.

		Engine condition / (recommended time)	Start preparation [sec]			Speed ramp-up [sec]	Synchronisation [sec]	Load ramp-up [sec]	Total [sec] 0 - 100% load
			Phase 1	Phase 2	Phase 3	Phase 4 and 5			
						() in liquid fuel operation: exceptional speed ramp-up () in gas operation: fast speed ramp-up		() in liquid fuel operation: exceptional load ramp-up	() see remarks in columns
L 51/60DF, 1,050 KW/cyl.[1]	Liquid fuel	Warm and stand-by	0	0	0	50 (15)	10	30 (20)	90 (45)
	MN 80		0	75	0	50 (19)		0 - 40%: 40 40% - 100%: 60 ∑ 100	235 (204)
12V/14V/16V 51/60DF, 1,050 KW/cyl.[1]	Liquid fuel	Warm and stand-by	0	0	0	50 (16)	10	35 (20)	95 (46)
	MN 80		0	75	0	50 (20)		0 - 40%: 46 40% - 100%: 69 ∑ 115	250 (220)
L 51/60DF, 1,150 KW/cyl.[1]	Liquid fuel	Warm and stand-by	0	0	0	50 (15)	10	30 (20)	90 (45)
	MN 80		0	75	0	50 (19)		0 - 40%: 22 40% - 100%: 33 ∑ 55	190 (159)
12V/14V/16V 51/60DF, 1,150 KW/cyl.[1]	Liquid fuel	Warm and stand-by	0	0	0	50 (16)	10	35 (20)	95 (46)
	MN 80		0	75	0	50 (20)		0 - 40%: 26 40% - 100%: 39 ∑ 65	200 (170)

[1] Engine equipped with jet assist.

Figure 23: Duration of the load application – Continuous loading (extract)

For further information or deviating engine condition/equipment contact MAN Energy Solutions.

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2.8.4 Load application – Load steps (for electric propulsion)

Minimum requirements of classification societies and ISO rule

The specification of the IACS (Unified Requirement M3) contains first of all guidelines for suddenly applied load steps. Originally two load steps, each 50 %, were described. In view of the technical progress regarding increasing mean effective pressures, the requirements were adapted. According to IACS and ISO 8528-5 a diagram is used to define – based on the mean effective pressure of the respective engine – the number of load steps for a load application from 0 % load to 100 % load. This diagram serves as a guideline for four stroke engines in general and is reflected in the rules of the classification societies.

Be aware, that for marine engines load application requirements must be clarified with the respective classification society as well as with the shipyard and the owner.

Accordingly MAN Energy Solutions has specified the following table.

Declared power mean effective pressure of the engine (p_{me})	Number of load steps
> 18 bar up to 22.5 bar	4
> 22.5 bar up to 27 bar	5
> 27 bar	6
The size of each load step to be calculated as: 100 % load divided by "Number of load steps". For example: 100 % load / "4" = 25 % load increase per load step.	

Table 24: Number of load steps dependent on the p_{me} of the engine

Exemplary requirements

Minimum requirements concerning dynamic speed drop, remaining speed variation and recovery time during load application are listed below.

Classification society	Dynamic speed drop in % of the nominal speed	Remaining speed variation in % of the nominal speed	Recovery time until reaching the tolerance band ±1 % of nominal speed
DNV	≤ 10 %	≤ 5 %	≤ 5 sec
RINA			
Lloyd´s Register			≤ 5 sec, max. 8 sec
American Bureau of Shipping			≤ 5 sec
Bureau Veritas			
ISO 8528-5			

Table 25: Minimum requirements of some classification societies plus ISO rule

In case of a load drop of 100 % nominal engine power, the dynamic speed variation must not exceed 10 % of the nominal speed and the remaining speed variation must not surpass 5 % of the nominal speed.

For DF engines regarding allowable load steps it must be distinguished between liquid fuel operation and gas operation.

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SCR regeneration phase

Dependent on the ambient conditions during the regeneration phase of the SCR the load application capability may be limited and may not reach the level shown below.

Engine specific load steps – Maximum load step dependent on base load

If the engine has reached the engine condition hot, the maximum load step which can be applied as a function of the currently driven base load can be derived out of the below stated diagram(s).

Before an additional load step will be applied, at least 20 seconds waiting time after initiation of the previous load step needs to be considered.

EPROX-AC

After synchronisation within the operating range of EPROX-AC as setting for a change in speed, it is recommended: 1% of nominal speed per sec.

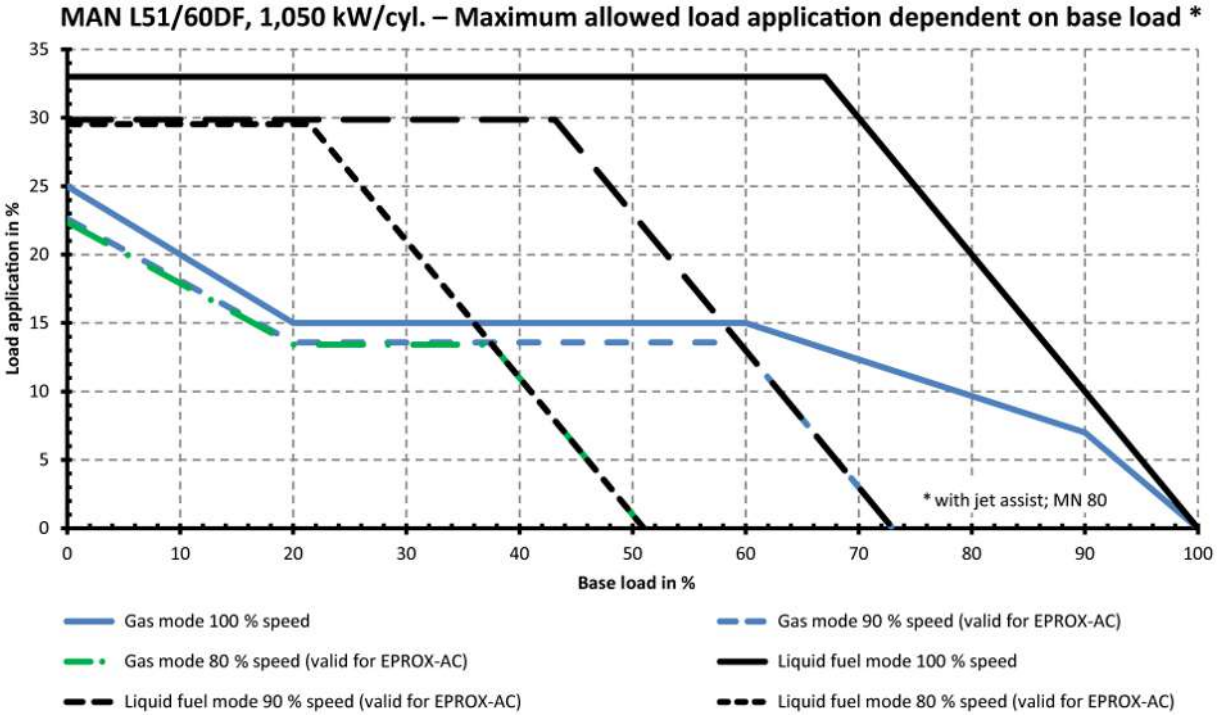

Figure 24: Load application dependent on base load (engine condition hot) – MAN L51/60DF, 1,050 kW/cyl.

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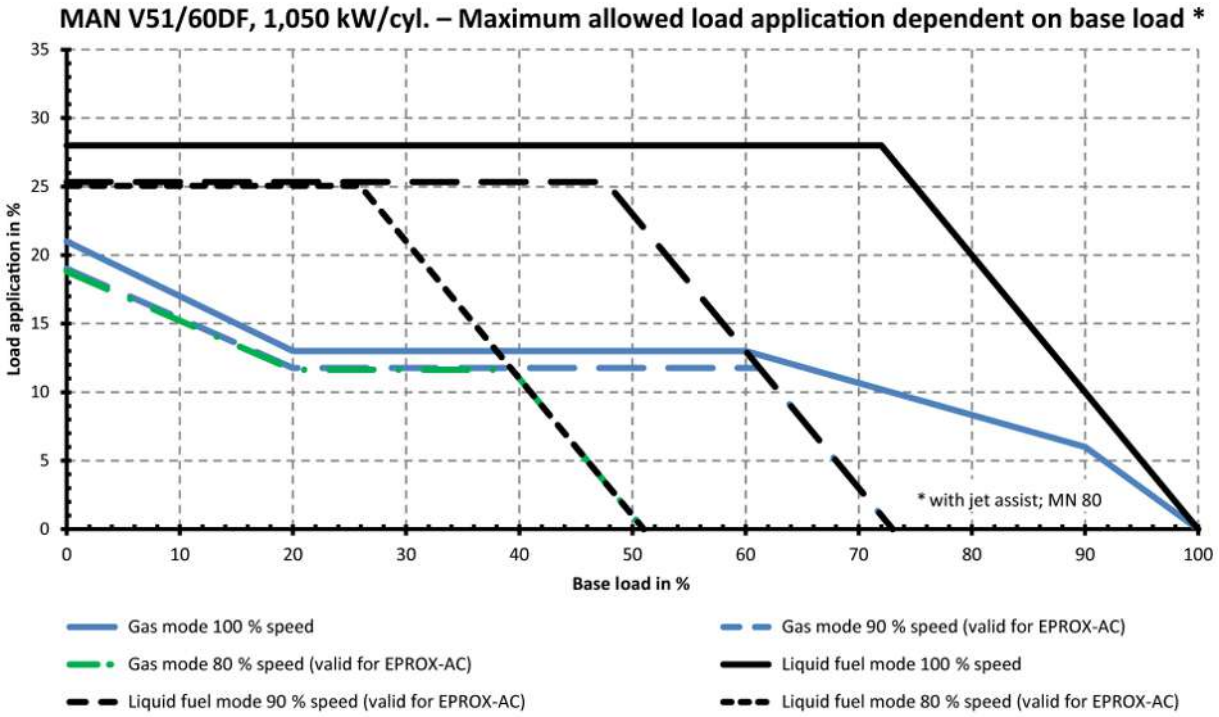

Figure 25: Load application dependent on base load (engine condition hot) – MAN V51/60DF, 1,050 kW/cyl.[1)]

[1)] Values apply to 12V, 14V, 16V engine.

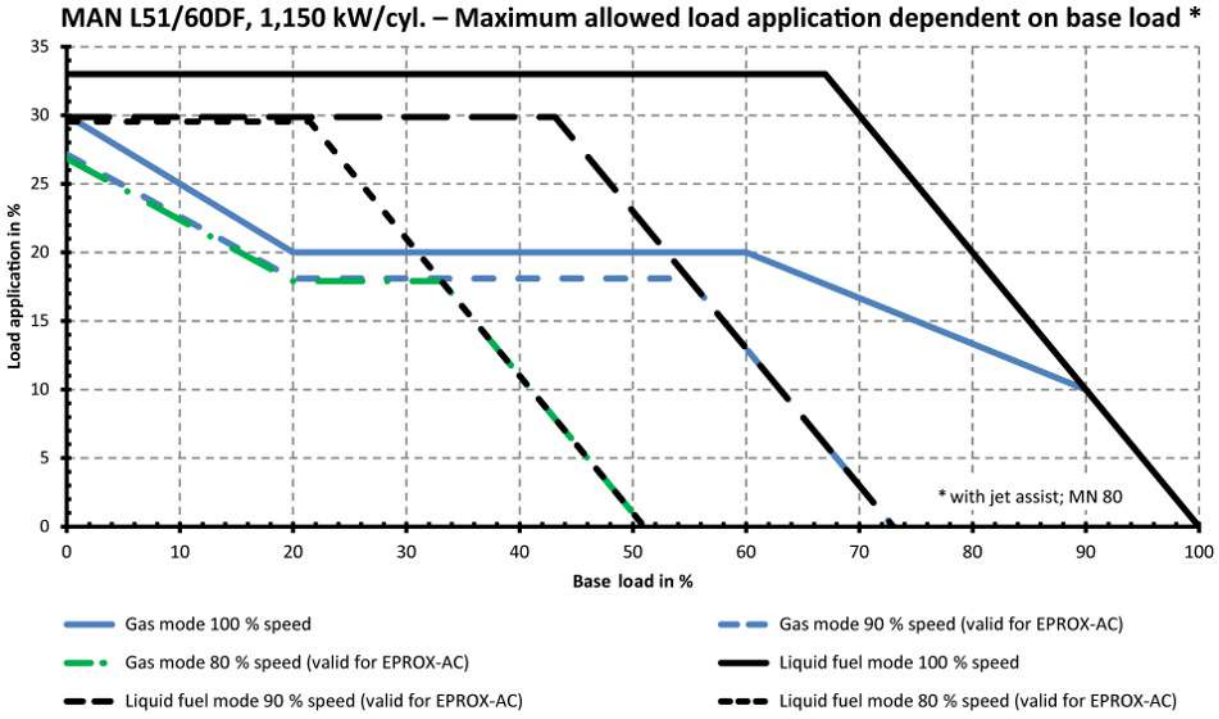

Figure 26: Load application dependent on base load (engine condition hot) – MAN L51/60DF, 1,150 kW/cyl.

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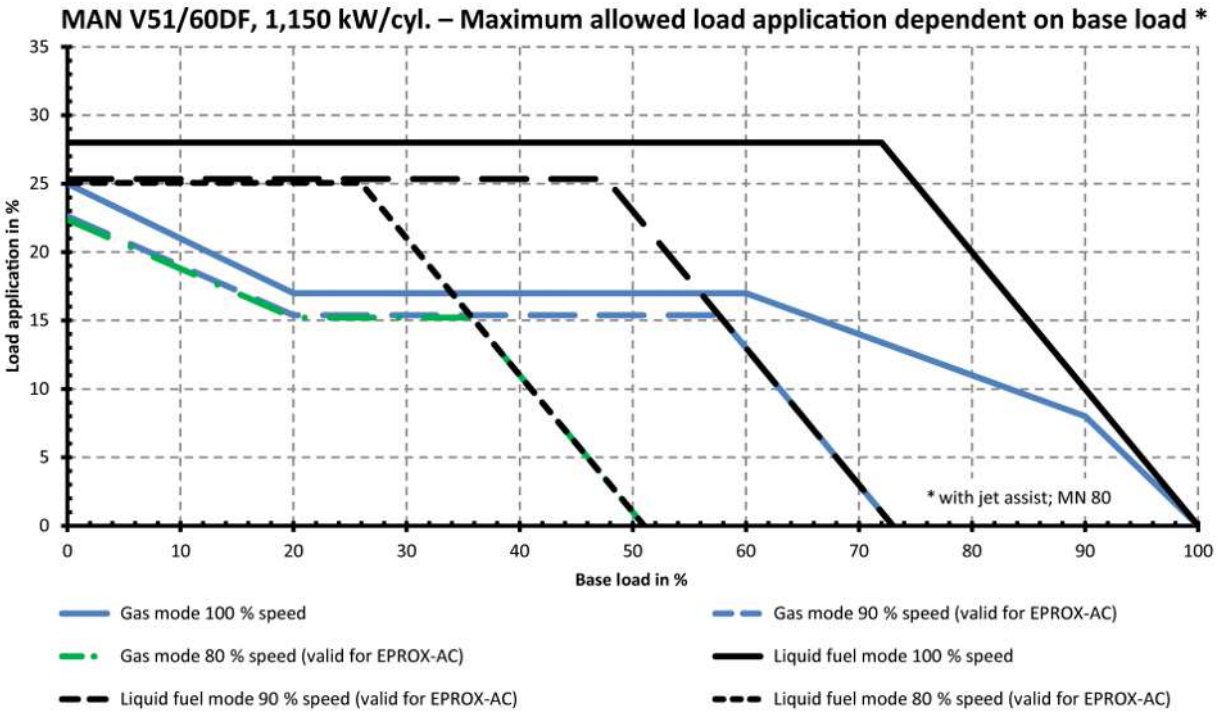

Figure 27: Load application dependent on base load (engine condition hot) – MAN V51/60DF, 1,150 kW/cyl.[1)]

[1)] Values apply to 12V, 14V engine.

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2.8.5 Load application for mechanical propulsion (CPP)

Acceleration times for controllable pitch propeller plants

General remark

Stated acceleration times in the following figure are valid for the engine itself. Depending on the project-specific propulsion train (moments of inertia, vibration calculation etc.) project-specific this may differ. Of course, the acceleration times are not valid for the ship itself, due to the fact, that the time constants for the dynamic behavior of the engine and the vessel may have a ratio of up to 1:100, or even higher (dependent on the type of vessel). The effect on the vessel must be calculated separately.

Propeller control

For remote controlled propeller drives for ships with unmanned or centrally monitored engine room operation in accordance to IACS “Requirements concerning MACHINERY INSTALLATIONS”, M43, a single control device for each independent propeller has to be provided, with automatic performance preventing overload and prolonged running in critical speed ranges of the propelling machinery. Operation of the engine according to the relevant and specific operating range (e.g. Operating range for controllable pitch propeller (CPP)) has to be ensured. In case of a manned engine room and manual operation of the propulsion drive, the engine room personnel are responsible for the soft loading sequence, before control is handed over to the bridge.

Load control programme

The lower time limits for normal and emergency manoeuvres are given in our diagrams for application and shedding of load. We strongly recommend that the limits for normal manoeuvring are observed during normal operation. An automatic change-over to a shortened load programme is required for emergency manoeuvres. The final design of the programme should be jointly determined by all the parties involved, considering the demands for manoeuvring and the actual service capacity.

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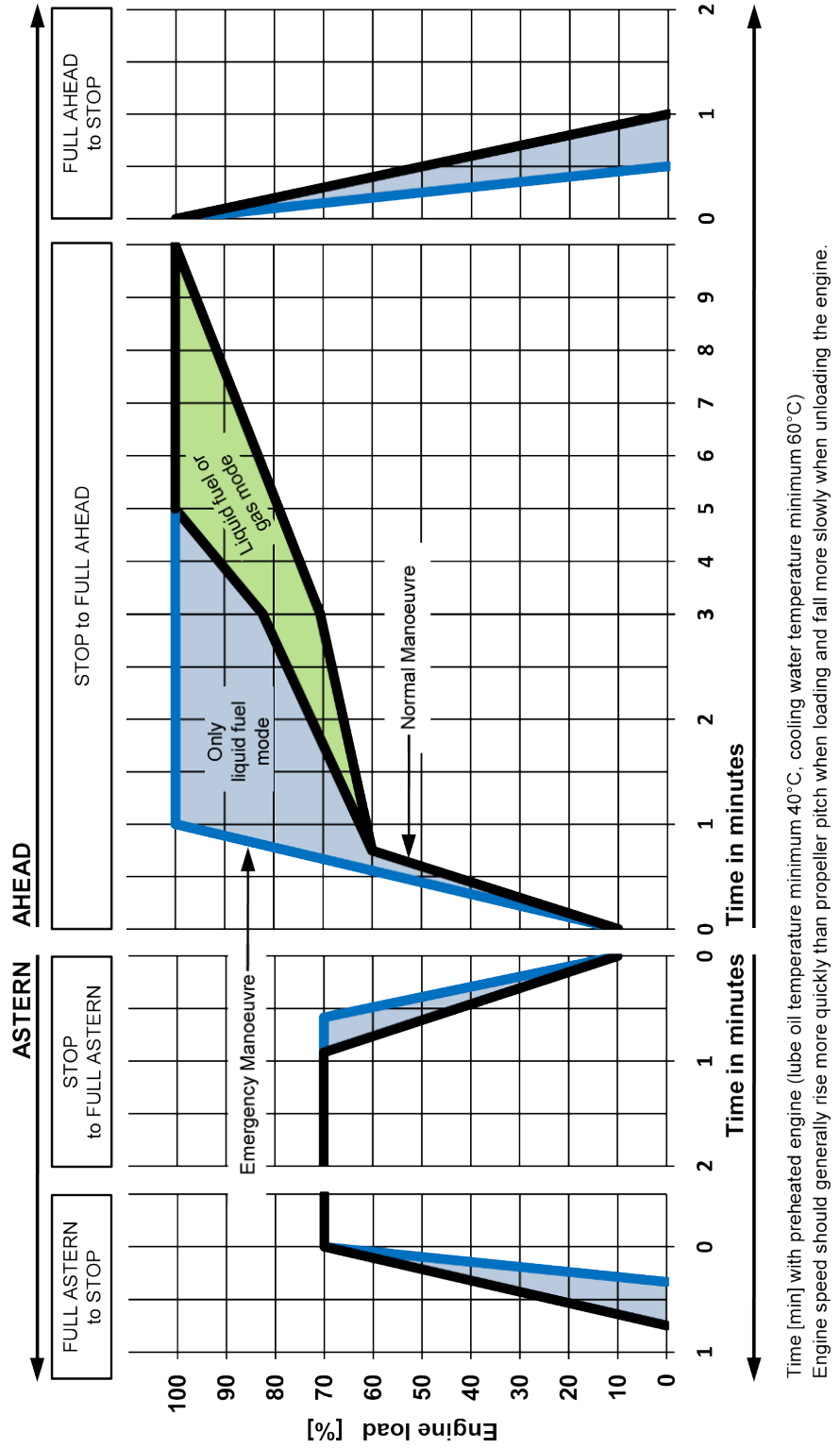

Figure 28: Control lever setting and corresponding engine specific acceleration times (for guidance)

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2.9 Low-load operation

Definition

Basically, the following load conditions are distinguished:

Overload:	> 100 % (MCR) of the engine output (not admitted, see section Engine ratings (output) for different applications, Page 43)
Full load (MCR):	100 % (MCR) of the engine output
Part load:	< 100 % (MCR) of the engine output
Low load:	< 25 % of the engine output

Correlations

The best operating conditions for the engine prevail under even loading in the range of 60 % to 90 % of full load.

During idling or engine operation at a low load, combustion in the combustion chamber is incomplete.

This may result in the forming of deposits in the combustion chamber, which will lead to increased soot emission and to increasing cylinder contamination.

This process is more acute in low-load operation and during manoeuvring when the cooling water temperatures are not kept at the required level, and are decreasing too rapidly. This may result in too low charge air and combustion chamber temperatures, deteriorating the combustion at low loads especially in heavy fuel operation.

Operation with heavy fuel oil (fuel of RM quality) or with MGO (DMA, DFA) or MDO (DMB, DFB)

Based on the above, the low-load operation in the range of < 25 % of the full load is subjected to specific limitations. According to figure Time limitation for low-load operation (left), duration of "relieving operation" (right), Page 65 immediately after a phase of low-load operation the engine must be operated at > 70 % of the full load for some time in order to reduce the deposits in the cylinders and the exhaust gas turbocharger again.

- Provided that the specified engine operating values are observed, there are no restrictions at loads > 25 % of the full load.
- Continuous operation at < 25 % of the full load should be avoided whenever possible.
- No-load operation, particularly at nominal speed (alternator operation) is only permissible for one hour maximum.

After 500 hours of continuous operation with liquid fuel, at a low load in the range of 20 % to 25 % of the full load, the engine must be run-in again.

See section Engine running in, Page 519.

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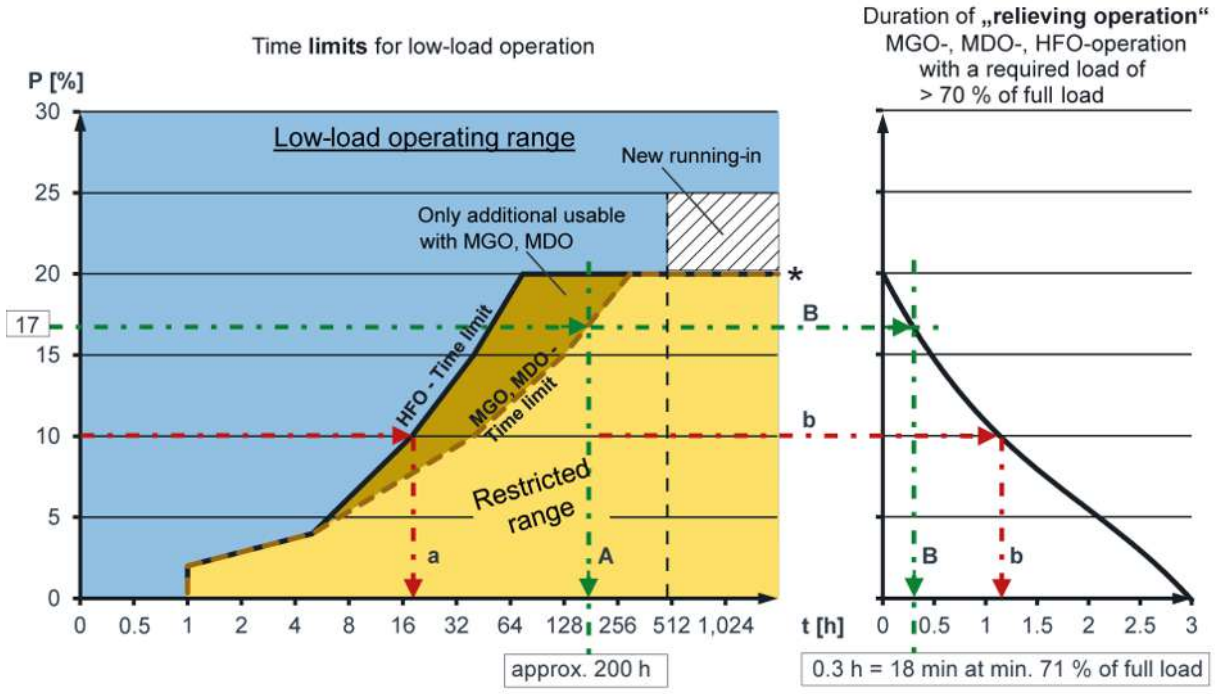

* Generally, the time limits in heavy fuel oil operation apply to all HFO grades according to the designated fuel specification. In certain rare cases, when HFO grades with a high ignition delay together with a high coke residues content are used, it may be necessary to raise the total level of the limiting curve for HFO from 20 % up to 30 %.

P % of the full load

Figure 29: Time limitation for low-load operation (left), duration of "relieving operation" (right)

Example for heavy fuel oil (HFO)

Line a — Time limits for low-load operation with heavy fuel oil:

At 10 % of the full load, operation on heavy fuel oil is allowable for 19 hours maximum.

Line b — Duration of "relieving operation":

Let the engine run at a load > 70 % of the full load appr. within 1.2 hours to burn the deposits formed.

Note:
The acceleration time from the actual load up to 70 % of the full load must be at least 15 minutes.

Example for MGO (DMA, DFA), MDO (DMB, DFB)

Line A — Time limits for low-load operation with MGO/MDO:

At 17 % of the full load, operation on MGO/MDO is allowable appr. for 200 hours maximum.

Line B — Duration of "relieving operation":

Let the engine run at a load > 70 % of the full load appr. within 18 minutes to burn the deposits formed.

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Note:
The acceleration time from the actual load up to 70 % of the full load must be at least 15 minutes.

Operation with gas

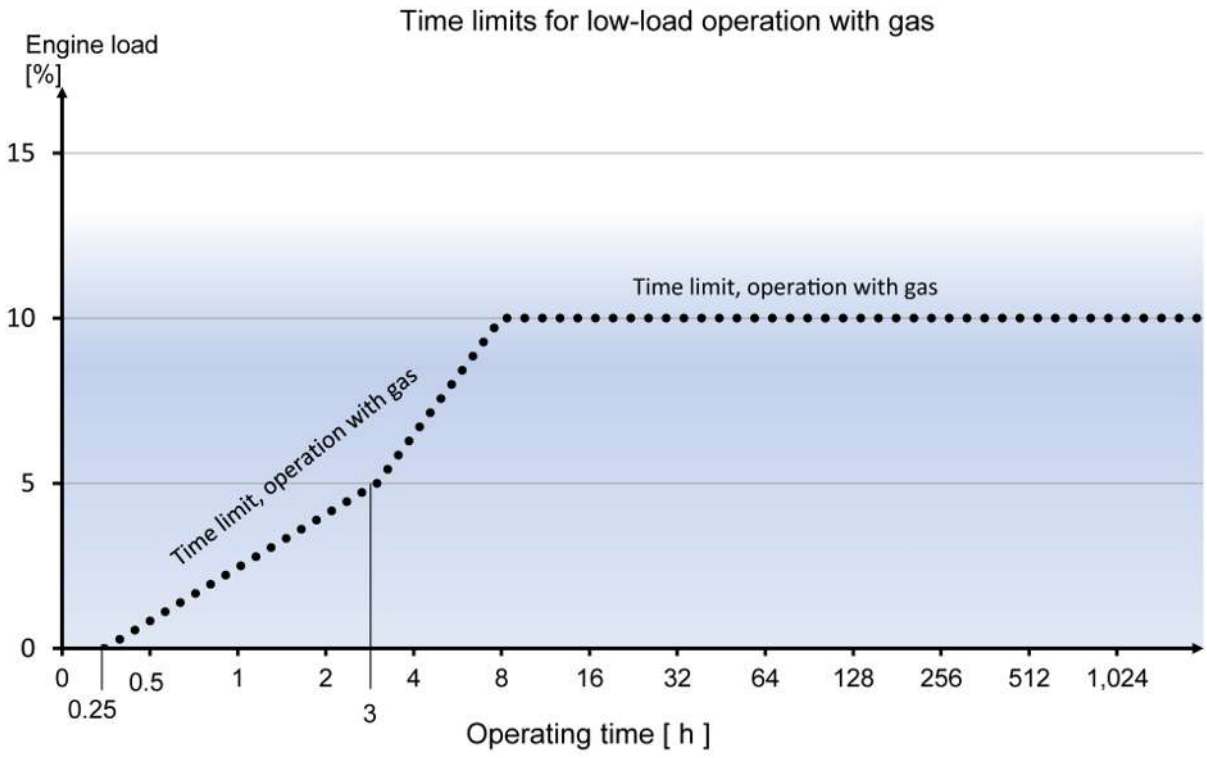
For low-load operation with gas, the following applies:

- The constantly required minimum load in gas operation is 10 % of the full load.

Provided that it can be ensured that the charge air temperature before cylinder is minimum 50 °C in low-load operation, the following conditions are applicable for gas operation at < 10 % of the full load:

- Idling operation is allowed within 15 minutes maximum.
- Continuous operation at 5 % of full load is allowed within 3 hours maximum.
- For information on further load points see figure Time limitation for low-load operation with gas, Page 67.

Provided that the specified charge air temperature can be achieved and the given engine operating values are observed, there are no restrictions at loads ≥ 10 % of the full load.



P % of the full load

t Operating time in hours (h)

Figure 30: Time limitation for low-load operation with gas

Important remark for SCR operation

Long-term low-load operation may lead to an increase of the back pressure of the SCR system and initiate the regeneration. It may be required to adapt the engine load during the regeneration phase.

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If the automatic regeneration occurs more than once per day, the engine load must be adapted manually to reach sufficient temperatures for the regeneration process.

Optionally a change of operating mode can be necessary.

2.10 Liquid fuel and gas operation and switch over

DF engines can run in:

- Liquid fuel mode
- Gas mode
  For ignition a small amount of diesel oil is injected by separate pilot fuel injection nozzles.
- Back up mode operation
  In case the pilot fuel injection should fail, the engine can still be operated, but back up mode should only be used at emergency conditions and as short as possible. For details see section General introduction of liquid fuel oil system for dual fuel engines (designed to burn HFO, MDO and MGO), Page 365.

The engine power in gas mode is generally equal to the generated power in liquid fuel mode. Pilot fuel injection is also activated during liquid fuel mode. The injected pilot fuel quantity depends on the engine load.

Starting and stopping of the engine can be performed in liquid fuel mode as well as in gas mode.

Switch over:

- If the gas supply fails once, also a full load running engine is automatically switched over to liquid fuel mode without interruption in power supply.
- Switch over from liquid fuel mode to gas mode can be initiated, if the actual output is higher than the minimum output (see accordingly the operating ranges).
- Switch over from gas mode to liquid fuel mode can be initiated in any case.

Note:
If it is intended to operate the dual fuel engine >3 month only in liquid fuel operation, without any gas operation within this time span, than contact MAN Energy Solutions for clarification of needed measures on the gas components.

2.11 Engine load reduction

Sudden load shedding

For the sudden load shedding from 100 % to 0 % engine load, several requirements of the classification societies regarding the dynamic and permanent change of engine speed have to be fulfilled.

In case of a sudden load shedding and related compressor surging, check the proper function of the turbocharger silencer filter mat.

Recommended load reduction/stopping the engine

Figure Engine ramping down, generally, Page 69 shows the shortest possible times for continuously ramping down the engine in liquid fuel operation.

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Even with the stated shortest possible times for continuously ramping down (Phase 1 and Phase 2) the requirements of ISO 8528-5 G2 will be fulfilled.

To limit the effort regarding regulating the media circuits and also to ensure an uniform heat dissipation it always should be aimed for longer ramping down times by taking into account the realistic requirements of the specific plant.

Before final engine stop, the engine has to be operated for a minimum of 1 minute at idling speed.

Run-down cooling

In order to dissipate the residual engine heat, the system circuits should be kept in operation after final engine stop for a minimum of 15 minutes.

If for any reason the HT cooling water stand-by pump is not in function, the engine has to be operated for 15 minutes at 0 % – 10 % load before final stop, so that with the engine driven HT cooling water pump the heat will be dissipated.

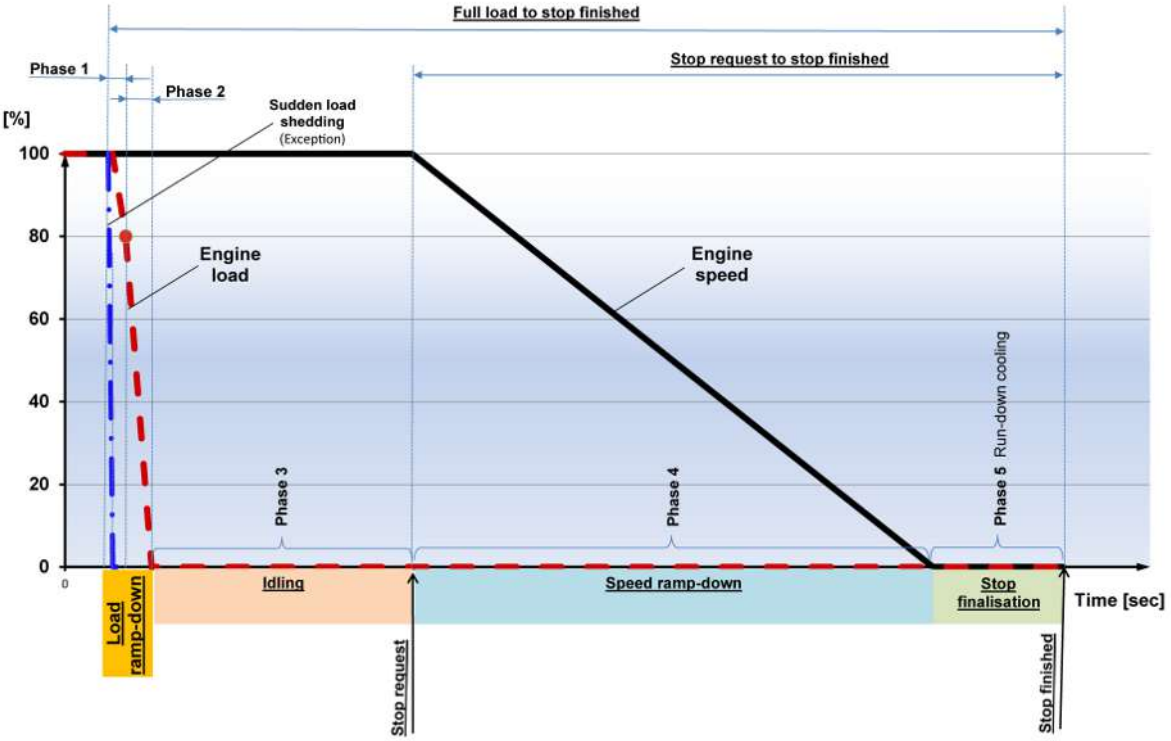

Figure 31: Engine ramping down, generally

Duration "Full load to stop finished"						
Phase	1	2	3	4	5	-
Designation	Load ramp-down	Load ramp-down	Idling	Speed ramp-down	Stop finalisation, run-down cooling	Total duration ≥
Speed [%]	100	100	100	100 – 0	0	-

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Duration "Full load to stop finished"						
Load [%]	100 – 80	80 – 0	0	0	0	-
Time [sec]	4	6	60	90	900	1,030

2.12 Engine load reduction as a protective safety measure

Requirements for the power management system/propeller control

In case of a load reduction request due to predefined abnormal engine parameter (e.g. high exhaust gas temperature, high turbine speed, high lube oil temperature) the power output (load) must be ramped down as fast as possible to ≤ 60 % load.

Therefore the power management system/propeller control has to meet the following requirements:

- After a maximum of 5 seconds after occurrence of the load reduction signal, the engine load must be reduced by at least 5 %.
- Then, within the next time period of maximum 30 sec an additional reduction of engine load by at least 35 % needs to be applied.
- The “Prohibited range” shown in figure Engine load reduction as a protective safety measure, Page 70 has to be avoided.

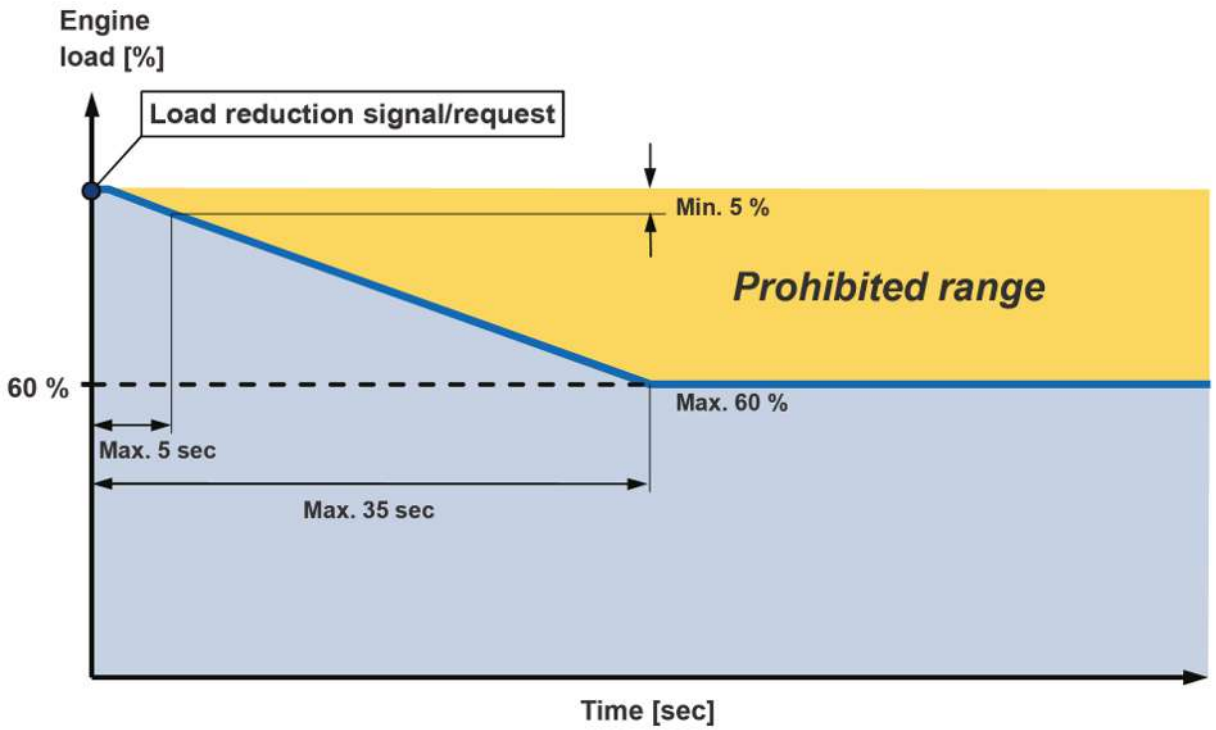

Figure 32: Engine load reduction as a protective safety measure

2.13 Engine operation under arctic conditions

Arctic condition is defined as:

Air intake temperatures of the engine below +5 °C.

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If engines operate under arctic conditions (intermittently or permanently), the engine equipment and plant installation have to hold certain design features and meet special requirements. They depend on the possible minimum air intake temperature of the engine and the specification of the fuel used.

Minimum air intake temperature of the engine, t_x:

- Category 1
  $+5\ °C > t_x > -15\ °C$
- Category 2
  $-15\ °C \geq t_x > -50\ °C$

Special engine design requirements

Special engine equipment required for arctic conditions category 1 and category 2, see section Engine equipment for various applications, Page 38.

Engine equipment

SaCoSone

- SaCoSone equipment is suitable to be stored at minimum ambient temperatures of –15 °C.
- In case these conditions cannot be met, protective measures against climatic influences have to be taken for the following electronic components:
  - EDS Databox APC620
  - TFT-touchscreen
  - Emergency switch module BD5937

  These components have to be stored at places, where the temperature is above –15 °C.
- A minimum operating temperature of ≥ –10 °C has to be ensured. The use of an optional electric heating is recommended.

Alternators

Alternator operation is possible according to suppliers specification.

Plant installation

Engine intake air conditioning

- Cooling down of engine room due to cold ambient air can be avoided by supplying the engine directly from outside with combustion air. For this the combustion air must be filtered (see quality requirements in section Specification of intake air (combustion air), Page 292). Moreover a droplet separator and air intake silencer become necessary, see section Intake air ducting in case of arctic conditions, Page 422. According to classification rules it may be required to install two air inlets from the exterior, one at starboard and one at portside.
- Cold intake air from outside is preheated in front of the cylinders in the charge air cooler. HT water serves as heat source. Depending on load and air temperature additional heat has then to be transferred to the HT circuit by a HT preheating module.
- It is necessary to ensure that the charge air cooler cannot freeze when the engine is out of operation (and the cold air is at the air inlet side). HT cooling water preheating will prevent this. Additionally it is recommended to prepare the combustion air duct upstream of the engine for the installation of a blanking plate, necessary to be installed in case of malfunction on the HT cooling water preheating system.

Category 1

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- Charge air blow-off is activated at high engine load with low combustion air temperature. With a blow-off air duct installed in the plant, it can be re-circulated in the combustion air duct upstream of the engine. Alternatively, only if blow-off air is deviated downstream of the charge air coolers and is cold (depending on engine type), blow-off air can be directly released in the engine room. Then a blow-off air silencer installed in the plant becomes necessary.
- Alternatively engine combustion air and engine room ventilation air can be supplied together in the engine room, if heated adequately and if accepted by the classification company.

Category 2

- Contact MAN Energy Solutions.

Instruction for minimum admissible fuel temperature

- In general the minimum viscosity before engine of 1.9 cSt must not be undershoot.
- The fuel specific characteristic values “pour point” and “cold filter plugging point” have to be observed to ensure pumpability respectively filterability of the fuel oil.
- Fuel temperatures of ≤ –10 °C are to be avoided, due to temporarily embrittlement of seals used in the engines fuel oil system. As a result they may suffer a loss of function.

Preheater before GVU (Gas Valve Unit)

- Be aware that the gas needs to be heated up to the minimum temperature before Gas Valve Unit.

Place of installation of the GVU

- The GVU itself needs to be installed protected from the weather, at ambient temperatures ≥ 5 °C. For lower ambient air temperatures design modifications of the GVU are required.

Minimum engine room temperature

- Ventilation of engine room.

  The air of the engine room ventilation must not be too cold (preheating is necessary) to avoid the freezing of the liquids in the engine room systems.
- Minimum power house/engine room temperature for design ≥ +5 °C.

Coolant and lube oil systems

- Coolant and lube oil system have to be preheated for each individual engine, see section Starting conditions, Page 52.

  See also the specific information regarding special arrangements for arctic conditions, see section External lube oil system, Page 313 and Cooling water system, Page 335.
- Design requirements for the external preheater of HT cooling water systems according to stated preheater sizes, see figure Required preheater size to avoid heat extraction from HT system, Page 73.
- Maximum permissible antifreeze concentration (ethylene glycol) in the engine cooling water.

  An increasing proportion of antifreeze decreases the specific heat capacity of the engine cooling water, which worsens the heat dissipation from the engine and will lead to higher component temperatures.

  Therefore, the antifreeze concentration of the engine cooling water systems (HT and LT) within the engine room, respectively power house, should be below a concentration of 40 % glycol. Any concentration of > 55 % glycol is forbidden.
- If a concentration of anti-freezing agents of > 50 % in the cooling water systems is required, contact MAN Energy Solutions for approval.
- For information regarding engine cooling water see section Specification for engine supplies, Page 255.

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Insulation

The design of the insulation of the piping systems and other plant parts (tanks, heat exchanger, external intake air duct etc.) has to be modified and designed for the special requirements of arctic conditions.

Heat tracing

To support the restart procedures in cold condition (e.g. after unmanned survival mode during winter), it is recommended to install a heat tracing system in the pipelines to the engine.

Note:
A preheating of the lube oil has to be ensured. For plants taken out of operation and cooled down below temperatures of +5 °C additional special measures are required – in this case contact MAN Energy Solutions.

Heat extraction HT system and preheater sizes

After engine start, it is necessary to ramp up the engine to the below specified Range II to prevent too high heat loss and resulting risk of engine damage.

Thereby Range I must be passed as quick as possible to reach Range II. Be aware that within Range II low-load operation restrictions may apply.

If operation within Range I is required, the preheater size within the plant must be capable to preheat the intake air to the level, where heat extraction from the HT system is not longer possible.

Example 1:

- Operation at 20 % engine load and –45 °C intake air temperature wanted.
- Preheating of intake air from –45 °C up to minimum –16.5 °C required. => According diagram preheater size of 21.5 kW/cyl. required.
- Ensure that this preheater size is installed, otherwise this operation point is not permissible.

All preheaters need to be operated in parallel to engine operation until minimum engine load is reached.

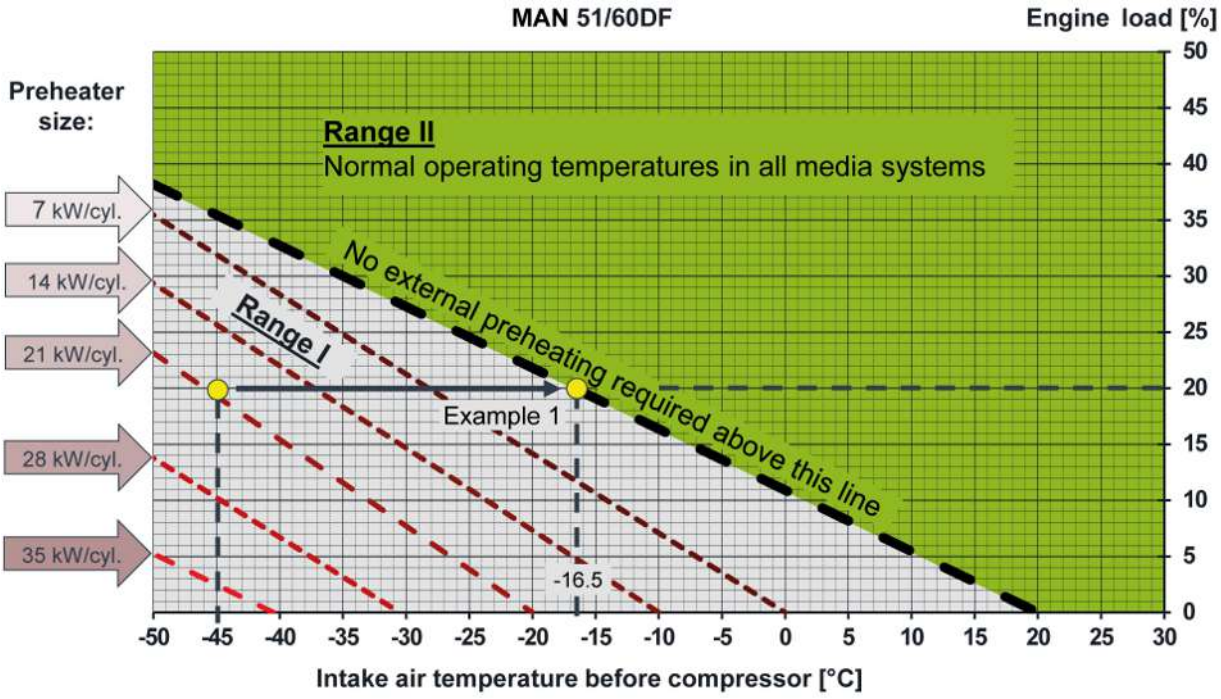

Figure 33: Required preheater size to avoid heat extraction from HT system, MAN 51/60DF

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Important remark for SCR operation under arctic conditions

- Note that the SCR will be deactivated below –10 °C intake air temperature.

2.14 Generator operation

2.14.1 Operating range for generator operation/electric propulsion

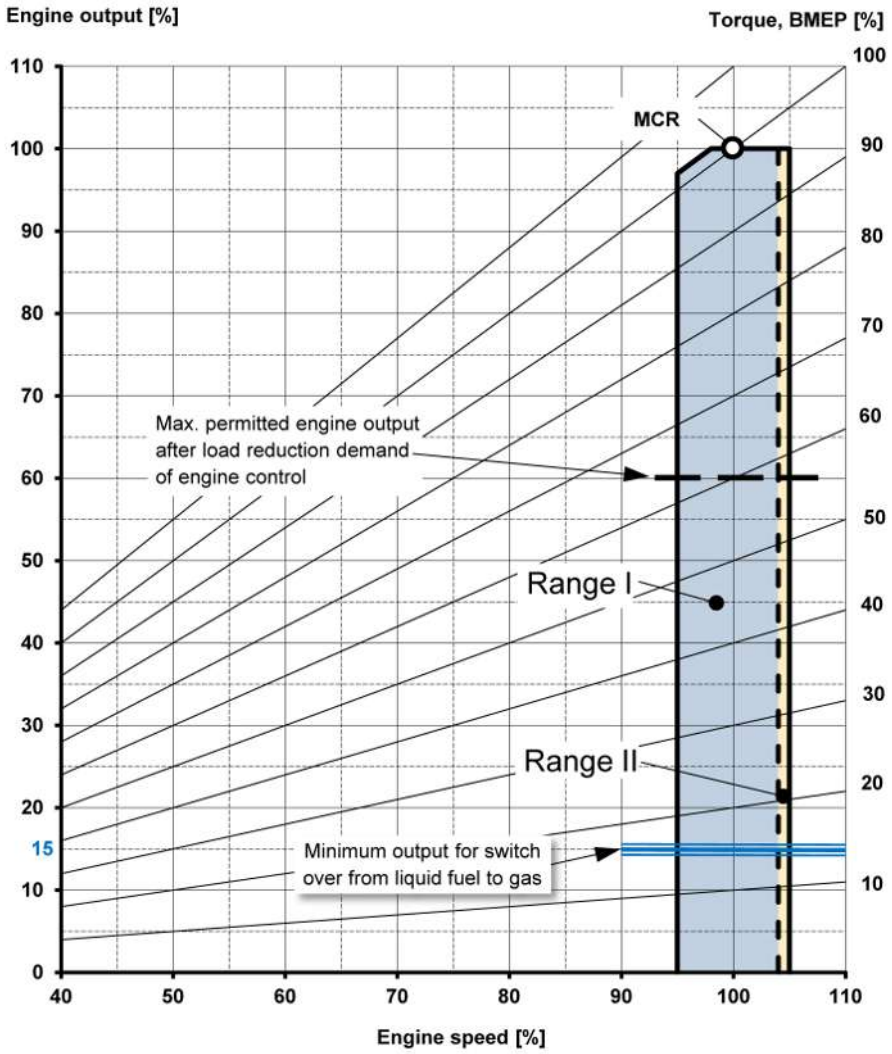

Figure 34: Operating range for generator operation/electric propulsion

- MCR[1]
  Maximum continuous rating.
- Range I
  Operating range for continuous service.
- Range II
  No continuous operation permissible.
  Maximum operating time less than 2 minutes.

[1] In accordance with DIN ISO 3046-1 and for further clarification of relevant sections within DIN ISO 8528-1, the following is specified:

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- The maximum output (MCR) has to be observed by the power management system of the plant.
- The range of 100 % up to 110 % fuel admission may only be used for a short time for governing purposes (e.g. transient load conditions and suddenly applied load).

IMO certification for engines with operating range for electric propulsion

Test cycle type E2 will be applied for the engine´s certification for compliance with the NO_x limits according to NO_x technical code.

2.14.2 Operating range for EPROX-AC

EPROX-AC is an electric propulsion system based on an AC net and allowing the system frequency to vary. The generators operate with variable speed between 80 % and 100 % of nominal speed.

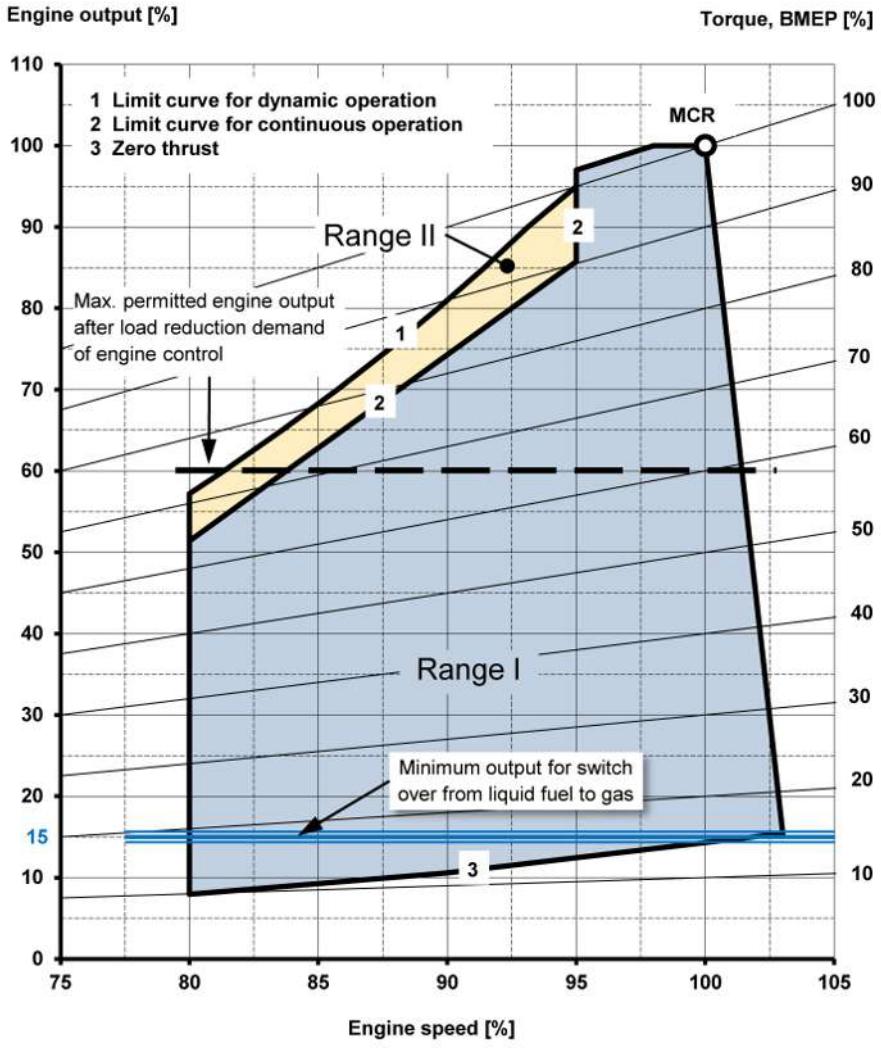

Figure 35: Operating range for EPROX-AC

Note:
In rare occasions it might be necessary that certain engine speed intervals have to be barred for continuous operation.
For applications using resiliently mounted engines, the admissible engine

speed range has to be confirmed (preferably at an early project phase) by a torsional vibration calculation, by a dimensioning of the resilient mounting, and, if necessary, by an engine operational vibration calculation.

MCR = Maximum continuous rating

Range I: Operating range for continuous operation.

Note:

Operation at higher power or lower speed than limited by the "limit curve for continuous operation" is only permitted for less than 1 minute.

Range II: Operating range which is temporarily (less than 1 minute) admissible e.g. during acceleration and manoeuvring.

IMO certification for engines with operating range for EPROX-AC

Test cycle type E2 and D2 will be applied for the engine´s certification for compliance with the NO_x limits according to NO_x technical code.

2.14.3 Available outputs and permissible frequency deviations

General

Generating sets, which are integrated in an electricity supply system, are subjected to the frequency fluctuations of the mains. Depending on the severity of the frequency fluctuations, output and operation respectively have to be restricted.

Frequency adjustment range

According to DIN ISO 8528-5, operating limits of > 2.5 % are specified for the lower and upper frequency adjustment range.

Operating range

Depending on the prevailing local ambient conditions, a certain maximum continuous rating will be available.

In the output/speed and frequency diagrams, a range has specifically been marked with “No continuous operation permissible in this area”. Operation in this range is only permissible for a short period of time, i.e. for less than 2 minutes. In special cases, a continuous rating is permissible if the standard frequency is exceeded by more than 4 %.

Limiting parameters

Max. torque

In case the frequency decreases, the available output is limited by the maximum permissible torque of the generating set.

Max. speed for continuous rating

An increase in frequency, resulting in a speed that is higher than the maximum speed admissible for continuous operation, is only permissible for a short period of time, i.e. for less than 2 minutes.

For engine-specific information see section Ratings (output) and speeds, Page 41 of the specific engine.

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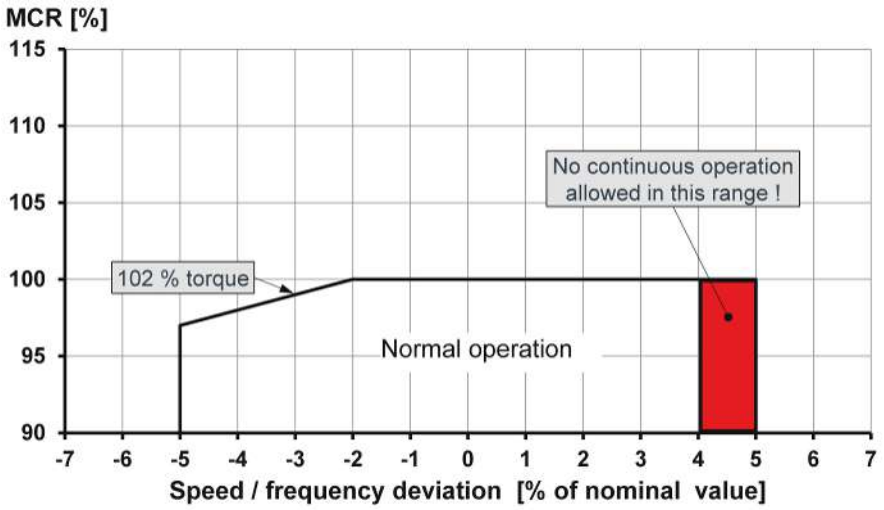

Figure 36: Permissible frequency deviations and corresponding max. output

2.14.4 Generator operation/electric propulsion – Power management

Operation of vessels with electric propulsion is defined as parallel operation of main engines with generators forming a closed system.

The power supply of the plant as a standard is done by auxilliary GenSets also forming a closed system.

In the design/layout of the plant a possible failure of one engine has to be considered in order to avoid overloading and under-frequency of the remaining engines with the risk of an electrical blackout.

Therefore we recommend to install a power management system. This ensures uninterrupted operation in the maximum output range and in case one engine fails the power management system reduces the propulsive output or switches off less important energy consumers in order to avoid under-frequency.

According to the operating conditions it is the responsibility of the ship's operator to set priorities and to decide which energy consumer has to be switched off.

The base load should be chosen as high as possible to achieve an optimum engine operation and lowest soot emissions.

The optimum operating range and the permissible part loads are to be observed (see section Low-load operation, Page 65).

Load application in case one engine fails

In case one engine fails, its output has to be made up for by the remaining engines in the system and/or the load has to be decreased by reducing the propulsive output and/or by switching off electrical consumers.

The immediate load transfer to one engine does not always correspond with the load reserve that the particular engine has available at the respective moment. That depends on the engine's base load.

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Be aware that the following section only serves as an example and is definitely not valid for this engine type. For the engine specific capability see figure(s) Load application dependent on base load (engine condition hot), Page 60.

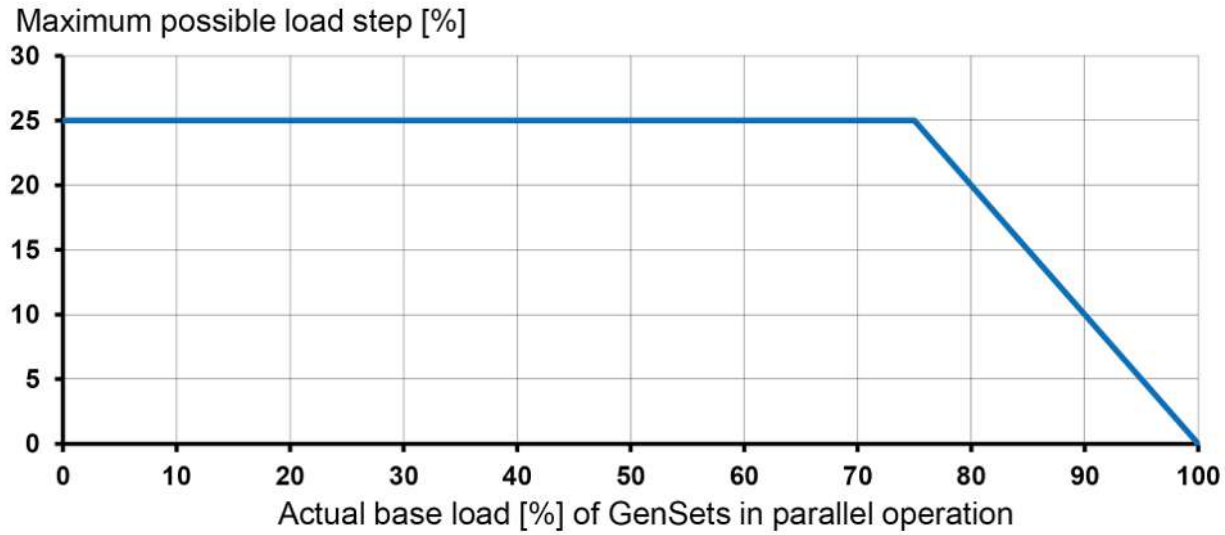

Figure 37: Maximum load step depending on base load (example may not be valid for this engine type)

Based on the above stated exemplary figure and on the total number of engines in operation the recommended maxium load of these engines can be derived. Observing this limiting maximum load ensures that the load from one failed engine can be transferred to the remaining engines in operation without power reduction.

Number of engines in parallel operation	3	4	5	6	7	8	9	10
Recommended maximum load in (%) of P_{max}	50	75	80	83	86	87.5	89	90

Table 26: Exemplary – Recommended maximum load in (%) of P_{max} dependend on number of engines in parallel operation

2.14.5 Alternator – Reverse power protection

Definition of reverse power

If an alternator, coupled to a combustion engine, is no longer driven by this engine, but is supplied with propulsive power by the connected electric grid and operates as an electric motor instead of working as an alternator, this is called reverse power. The speed of a reverse power driven engine is accordingly to the grid frequency and the rated engine speed.

Demand for reverse power protection

For each alternator (arranged for parallel operation) a reverse power protection device has to be provided because if a stopped combustion engine (fuel admission at zero) is being turned it can cause, due to poor lubrication, excessive wear on the engine´s bearings. This is also a classifications` requirement.

Examples for possible reverse power occurences

- Due to lack of fuel the combustion engine no longer drives the alternator, which is still connected to the mains.

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- Stopping of the combustion engine while the driven alternator is still connected to the electric grid.
- On ships with electric drive the propeller can also drive the electric traction motor and this in turn drives the alternator and the alternator drives the connected combustion engine.
- Sudden frequency increase, e.g. because of a load decrease in an isolated electrical system -> if the combustion engine is operated at low load (e.g. just after synchronising).

Adjusting the reverse power protection relay

The necessary power to drive an unfired diesel or gas engine at nominal speed cannot exceed the power which is necessary to overcome the internal friction of the engine. This power is called motoring power. The setting of the reverse-power relay should be, as stated in the classification rules, 50 % of the motoring power. To avoid false tripping of the alternator circuit breaker a time delay has to be implemented. A reverse power >> 6 % mostly indicates serious disturbances in the generator operation.

The following table Adjusting the reverse power relay, Page 79 provides a summary.

Admissible reverse power P_{el} [%]	Time delay for tripping the alternator circuit breaker [sec]
$P_{el} < 3$	30
$3 \leq P_{el} < 8$	3 to 10
$P_{el} \geq 8$	No delay

Table 27: Adjusting the reverse power relay

2.14.6 Earthing measures of diesel engines and bearing insulation on alternators

General

The use of electrical equipment on diesel engines requires precautions to be taken for protection against shock current and for equipotential bonding. These measures not only serve as shock protection but also for functional protection of electric and electronic devices (EMC protection, device protection in case of welding, etc.).

Earthing connections on the engine

Threaded bores M12, 20 mm deep, marked with the earthing symbol are provided in the engine foot on both ends of the engine.

It has to be ensured that earthing is carried out immediately after engine set-up. If this cannot be accomplished any other way, at least provisional earthing is to be effected right after engine set-up.

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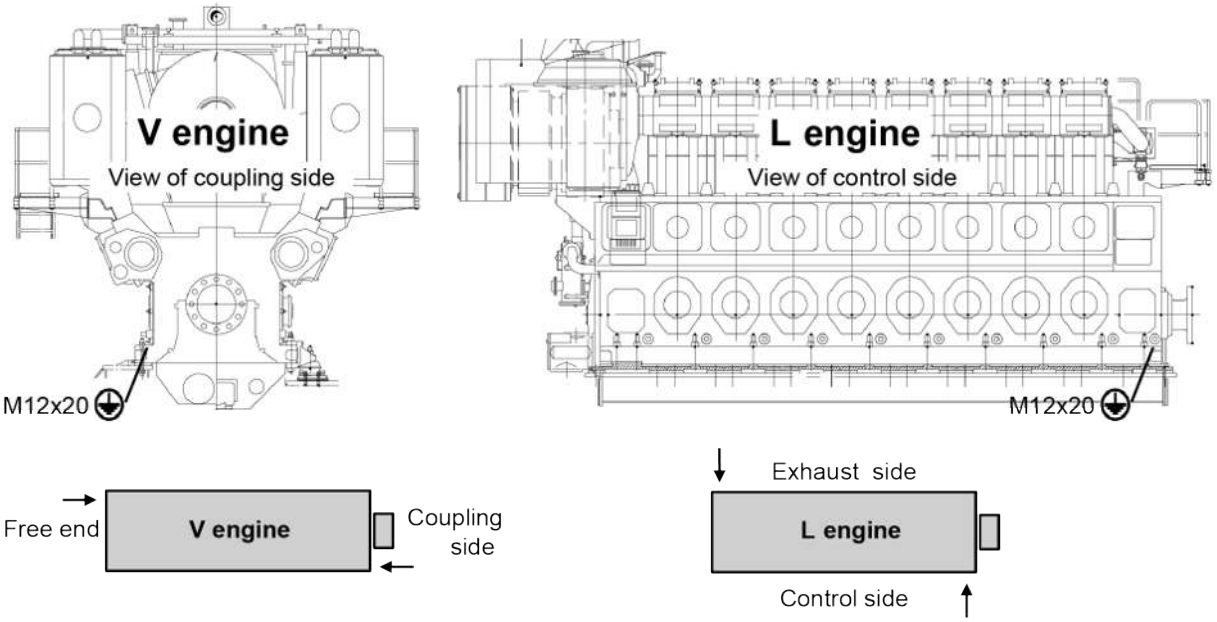

Figure 38: Earthing connection on engine (are arranged diagonally opposite each other) – Exemplary

Measures to be taken on the alternator

Shaft voltages, i.e. voltages between the two shaft ends, are generated in electrical machines because of slight magnetic unbalances and ring excitations. In the case of considerable shaft voltages (e.g. > 0.3 V), there is the risk that bearing damage occurs due to current transfers. For this reason, at least the bearing that is not located on the drive end is insulated (valid for alternators > 1 MW output). For verification, the voltage available at the shaft (shaft voltage) is measured while the alternator is running and excited. With proper insulation, a voltage can be measured. In order to protect the prime mover and to divert electrostatic charging, an earthing brush is often fitted on the coupling side.

Observation of the required measures is the alternator manufacturer's responsibility.

Consequences of inadequate bearing insulation on the alternator and insulation check

In case the bearing insulation is inadequate, e.g., if the bearing insulation was short-circuited by a measuring lead (PT100, vibration sensor), leakage currents may occur, which result in the destruction of the bearings. One possibility to check the insulation with the alternator at standstill (prior to coupling the alternator to the engine; this, however, is only possible in the case of single-bearing alternators) would be:

- Raise the alternator rotor (insulated, in the crane) on the coupling side.
- Measure the insulation by means of the megger test against earth.

Note:
Hereby the max. voltage permitted by the alternator manufacturer is to be observed.

If the shaft voltage of the alternator at rated speed and rated voltage is known (e.g. from the test record of the alternator acceptance test), it is also possible to carry out a comparative measurement.

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If the measured shaft voltage is lower than the result of the “earlier measurement” (test record), the alternator manufacturer should be consulted.

Earthing conductor

The nominal cross section of the earthing conductor (equipotential bonding conductor) has to be selected in accordance with DIN VDE 0100, part 540 (up to 1 kV) or DIN VDE 0141 (in excess of 1 kV).

Generally, the following applies:

The protective conductor to be assigned to the largest main conductor is to be taken as a basis for sizing the cross sections of the equipotential bonding conductors.

Flexible conductors have to be used for the connection of resiliently mounted engines.

Execution of earthing

The earthing must be executed by the shipyard, since generally it is not scope of supply of MAN Energy Solutions.

Earthing strips are also not included in the MAN Energy Solutions scope of supply.

Additional information regarding the use of welding equipment

In order to prevent damage on electrical components, it is imperative to earth welding equipment close to the welding area, i.e., the distance between the welding electrode and the earthing connection should not exceed 10 m.

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2.15 Propeller operation

2.15.1 Operating range for controllable pitch propeller (CPP)

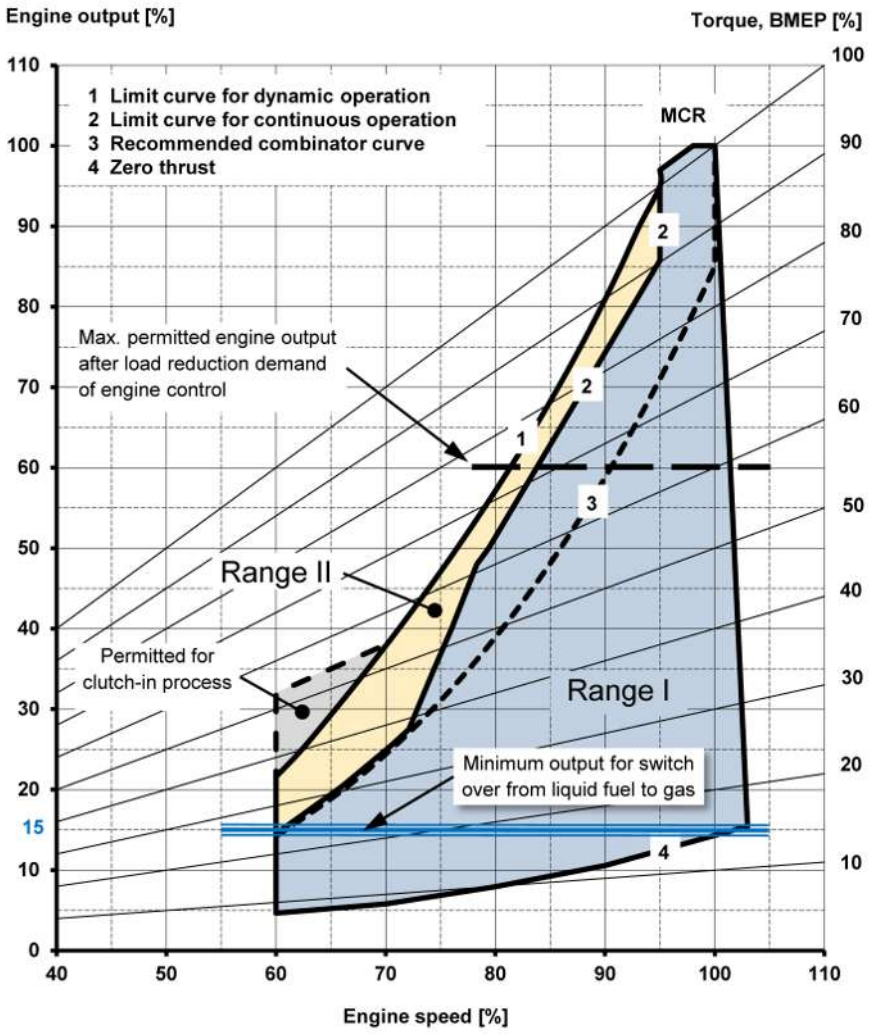

Figure 39: Operating range for controllable pitch propeller

Note:
In rare occasions it might be necessary that certain engine speed intervals have to be barred for continuous operation.
For applications using resiliently mounted engines, the admissible engine speed range has to be confirmed (preferably at an early project phase) by a torsional vibration calculation, by a dimensioning of the resilient mounting, and, if necessary, by an engine operational vibration calculation.

MCR = Maximum continuous rating

Range I: Operating range for continuous operation.

Note:

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Operation at higher power or lower speed than limited by the "limit curve for continuous operation" is only permitted for less than 1 minute.

Range II: Operating range which is temporarily (less than 1 minute) admissible e.g. during acceleration and manoeuvring.

The combinator curve must be placed at a sufficient distance to the load limit curve. For overload protection, a load control has to be provided.

Transmission losses (e.g. by gearboxes and shaft power) and additional power requirements (e.g. by PTO) must be taken into account.

IMO certification for engines with operating range for controllable pitch propeller (CPP)

Test cycle type E2 will be applied for the engine´s certification for compliance with the NO_x limits according to NO_x technical code.

2.15.2 General requirements for the CPP propulsion control

Pitch control of the propeller plant

General

A distinction between constant-speed operation and combinator-curve operation has to be ensured.

Failure of propeller pitch control:

In order to avoid overloading of the engine upon failure of the propeller pitch control, the propeller pitch must be adjusted to a value < 60 % of the maximum possible pitch.

4 – 20 mA load indication from engine control

As a load indication a 4 – 20 mA signal from the engine control is supplied to the propeller control.

Combinator-curve operation:

The 4 – 20 mA signal has to be used for the assignment of the propeller pitch to the respective engine speed. The operation curve of engine speed and propeller pitch (for power range, see section Operating range for controllable pitch propeller (CPP), Page 82) has to be observed also during acceleration/ load increase and unloading.

Acceleration/load increase

The engine speed has to be increased prior to increasing the propeller pitch (see figure Example to illustrate the change from one load step to another, Page 84).

When increasing propeller pitch and engine speed synchronously, the speed has to be increased faster than the propeller pitch.

The engine should not be operated in the area above the combinator curve (Range II in figure Operating range for controllable pitch propeller, Page 82).

Automatic limitation of the rate of load increase must be implemented in the propulsion control.

Deceleration/unloading the engine

The engine speed has to be reduced later than the propeller pitch (see figure Example to illustrate the change from one load step to another, Page 84).

When decreasing propeller pitch and engine speed synchronously, the propeller pitch has to be decreased faster than the speed.

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Example to illustrate the change from one load step to another

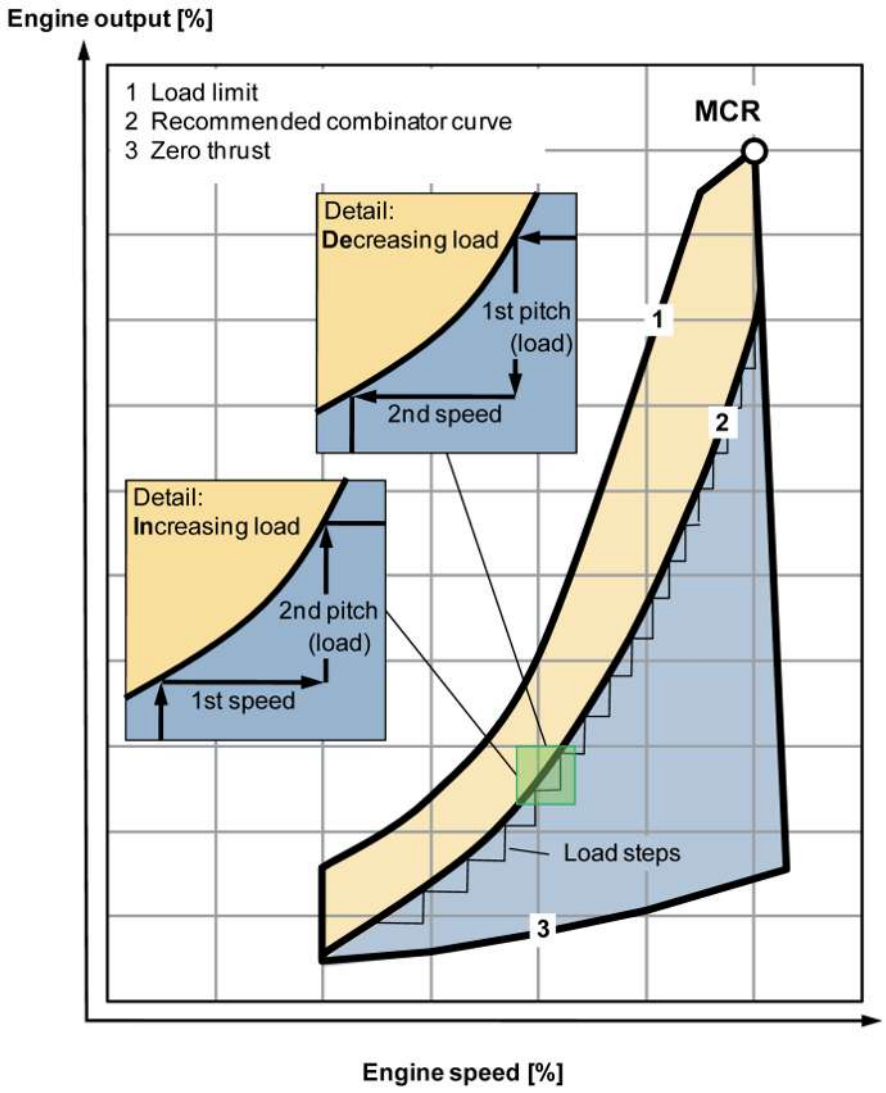

Figure 40: Example to illustrate the change from one load step to another

Windmilling protection

If a stopped engine (fuel admission at zero) is being turned by the propeller, this is called “windmilling”. The permissible period for windmilling is short, because windmilling can cause excessive wear of the engine bearings, due to poor lubrication at low propeller speed.

Single-screw ship

The propeller control has to ensure that the windmilling time is less than 40 seconds.

Multiple-screw ship

The propeller control has to ensure that the windmilling time is less than 40 seconds. In case of plants without shifting clutch, it has to be ensured that a stopped engine cannot be turned by the propeller.

For maintenance work a shaft interlock has to be provided for each propeller shaft.

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Binary signals from engine control

Overload contact

The overload contact will be activated when the engine's fuel admission reaches the maximum position. At this position, the control system has to stop the increase of the propeller pitch. If this signal remains longer than the predetermined time limit, the propeller pitch has to be decreased.

Contact "Operation close to the limit curve"

This contact is activated when the engine is operated close to a limit curve (torque limiter, charge air pressure limiter, etc.). When the contact is activated, the control system has to stop the increase of the propeller pitch. If this signal remains longer than the predetermined time limit, the propeller pitch has to be decreased.

Propeller pitch reduction contact

This contact is activated when disturbances in engine operation occur, for example too high exhaust gas mean-value deviation. When the contact is activated, the propeller control system has to reduce the propeller pitch to 60 % of the rated engine output, without change in engine speed.

In section Engine load reduction as a protective safety measure, Page 70 the requirements for the response time are stated.

Distinction between normal manoeuvre and emergency manoeuvre

The propeller control system has to be able to distinguish between normal manoeuvre and emergency manoeuvre (i.e., two different acceleration curves are necessary).

MAN Energy Solutions' guidelines concerning acceleration times and power range have to be observed

The power range (see section Operating range for controllable pitch propeller (CPP), Page 82) and the acceleration times (see paragraph Acceleration times, Page 63) have to be observed. In section Engine load reduction as a protective safety measure, Page 70 the requirements for the response time are stated.

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2.15.3 Torque measurement flange

As the fuel gas composition supplied to the dual fuel engine may change during a voyage in a wide range, it is required to adapt the engine control accordingly. This will be done in the SaCoSone system after comparison of an external engine output signal with actual engine parameters. Therefore a torque measurement flange needs to be provided for each engine separately.

Note:
Please be aware that this will influence the installation layout.

Requirements for torque measurement flange:

- For each engine its own torque measurement flange needs to be provided.
- Torque measurement flange must be certified and must be calibrated according to recommendation of manufacturer.
- Torque measurement flange must be proofed for reliability and durability.
- Torque measurement flange must be capable of operation under the specific condition of the application, e.g.:
  - Vibration
  - Wide temperature range
  - High humidity and spray water
  - Oil vapors
- Torque measurement flange must withstand torque fluctuations and torsional vibrations.
- Torque measurement flange must be accessible for check.
- Implementation of torque measurement flange between engine and gear box.
- Specific signal quality:
  - Specified for highest possible torque according to engines operating range.
  - High accuracy:

    Total deviation (inclusive non linearity, drift, hysteresis) of < 5 % of nominal (rated) signal in whole operating range of the engine.
  - Signal 4 – 20 mA.
  - Low pass filter 1 Hz to remove torque ripple.

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2.16 Fuel oil, urea, lube oil, starting air and control air consumption

2.16.1 Fuel oil consumption for emission standard: IMO Tier III

Note:
The engine's certification for compliance with the NO_x limits according to NO_x technical code will be done within the scope of supply of the factory acceptance test as member or parent engine for IMO Tier II without SCR installation. Accordingly the stated figures for the fuel oil consumption are without SCR installation.

The impact of the SCR installation on the fuel oil consumption depends on the plant layout and ambient conditions (see paragraph Additions to fuel consumption, Page 92).

MAN 51/60DF (1,050 kW/cyl.) – Electric propulsion (speed = constant)

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Spec. fuel consumption in gas mode without attached pumps[1) 2) 3)]						
a) Natural gas	kJ/kWh	7,090	7,075	7,190	7,600	8,565
b) Pilot fuel	g/kWh	2.3	2.9	3.3	5.1	10.2
	kJ/kWh	100	125	140	220	435
c) Total = a + b[4)]	kJ/kWh	7,190	7,200 [5)]	7,330	7,820	9,000

[1)] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2)] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3)] Relevant for engine´s certification for compliance with the NO_x limits according E2 test cycle.

[4)] Gas operation (including pilot fuel).

[5)] Warranted fuel consumption at 85 % MCR.

Table 28: Fuel consumption MAN 51/60DF (1,050 kW/cyl.) in gas mode – Electric propulsion (speed = constant)

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Standard Spec. fuel oil consumption with HFO, MDO (DMB, DFB), MGO (DMA, DFA) without attached pumps[1) 2) 3)]						
a) Main fuel	g/kWh	177.7	175.0	180.6	181.5	193.0
b) Pilot fuel	g/kWh	1.8	2.0	2.4	3.5	7.0
	kJ/kWh	77	85	102	149	299
c) Total = a + b[4)]	g/kWh	179.5	177.0 [5)]	183.0	185.0	200.0
	kJ/kWh	7,665	7,558	7,814	7,900	8,540
Optional no VIT, but VVT for low soot emission Spec. fuel oil consumption with HFO, MDO (DMB, DFB), MGO (DMA, DFA) without attached pumps[1) 2) 3)]						

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% Output		100	85	75	50	25
a) Main fuel	g/kWh	182.7	181.0	179.6	181.0	202.0
b) Pilot fuel	g/kWh	1.8	2.0	2.4	3.5	7.0
	kJ/kWh	77	85	102	149	299
c) Total = a + b[4]	g/kWh	184.5	183.0 [5]	182.0	184.5	209.0
	kJ/kWh	7,878	7,814	7,771	7,878	8,924

[1] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3] Relevant for engine`s certification for compliance with the NO_x limits according E2 test cycle.

[4] Liquid fuel operation (including pilot fuel). For consideration of fuel leakage amount, consider table(s) Leakage rate, Page 153 for conventional injection.

[5] Warranted fuel consumption at 85 % MCR.

Table 29: Fuel consumption MAN 51/60DF (1,050 kW/cyl.) in liquid fuel mode – Electric propulsion (speed = constant)

Engine MAN 51/60DF (1,150 kW/cyl.) – Electric propulsion (speed = constant)

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Spec. fuel consumption in gas mode without attached pumps[1) 2) 3)]						
a) Natural gas	kJ/kWh	6L: 7,340 7 – 9L: 7,320 V: 7,300	6 – 9L: 7,295 V: 7,275	L/V: 7,340	L/V: 7,580	L/V: 8,565
b) Pilot fuel	g/kWh	2.3	2.9	3.3	5.2	10.2
	kJ/kWh	100	125	140	220	435
c) Total = a + b[4]	kJ/kWh	6L: 7,440 7 – 9L: 7,420 V: 7,400	6 – 9L: 7,420 [5] V: 7,400 [5]	L/V: 7,480	L/V: 7,800	L/V: 9,000

[1] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3] Relevant for engine´s certification for compliance with the NO_x limits according E2 test cycle.

[4] Gas operation (including pilot fuel).

[5] Warranted fuel consumption at 85 % MCR.

Table 30: Fuel consumption MAN 51/60DF (1,150 kW/cyl.) in gas mode – Electric propulsion (speed = constant)

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Standard Spec. fuel oil consumption with HFO, MDO (DMB, DFB), MGO (DMA, DFA) without attached pumps[1) 2) 3)]						
a) Main fuel	g/kWh	184.7	180.5	186.1	185	191.5

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% Output		100	85	75	50	25
b) Pilot fuel	g/kWh	1.8	2.0	2.4	3.5	7.0
	kJ/kWh	77	85	102	149	299
c) Total = a + b[4]	g/kWh	186.5	182.5 [5]	188.5	188.5	198.5
	kJ/kWh	7,964	7,793	8,049	8,049	8,476

[1] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3] Relevant for engine`s certification for compliance with the NO_x limits according E2 test cycle.

[4] Liquid fuel operation (including pilot fuel). For consideration of fuel leakage amount, consider table(s) Leakage rate, Page 153 for conventional injection.

[5] Warranted fuel consumption at 85 % MCR.

Table 31: Fuel consumption MAN 51/60DF (1,150 kW/cyl.) in liquid fuel mode – Electric propulsion (speed = constant)

Engine MAN 51/60DF (1,050 kW/cyl.) – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Spec. fuel consumption in gas mode without attached pumps[1) 2) 3)]						
Speed		constant = 514 rpm or 500 rpm				
a) Natural gas	kJ/kWh	7,090	7,075	7,190	7,600	8,565
b) Pilot fuel	g/kWh	2.3	2.9	3.3	5.1	10.2
	kJ/kWh	100	125	140	220	435
c) Total = a + b[4]	kJ/kWh	7,190	7,200 [5]	7,330	7,820	9,000
Speeds on recommended combinator curve (±5 rpm)		514 (500)	514 (500)	496 (482)	442 (430)	362 (353)
a) Natural gas	kJ/kWh	7,090	7,075	7,150	7,380	7,815
b) Pilot fuel	g/kWh	2.3	2.9	3.3	5.1	10.2
	kJ/kWh	100	125	140	220	435
c) Total = a + b[4]	kJ/kWh	7,190	7,200 [5]	7,290	7,600	8,250

[1] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3] Due to engine´s certification for compliance with the NO_x limits according E2 (Test cycle for "constant-speed main propulsion application" including electric propulsion and all controllable-pitch propeller installations) factory acceptance test will be done with constant speed only.

[4] Gas operation (including pilot fuel).

[5] Warranted fuel consumption at 85 % MCR.

Table 32: Fuel consumption MAN 51/60DF (1,050 kW/cyl.) in gas mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm

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% Output		100	85	75	50	25
Speed		constant = 514 rpm or 500 rpm				
Standard Spec. fuel oil consumption with HFO, MDO (DMB, DFB), MGO (DMA, DFA) without attached pumps[1) 2) 3)]						
a) Main fuel	g/kWh	177.7	175.0	180.6	181.5	193.0
b) Pilot fuel	g/kWh	1.8	2.0	2.4	3.5	7.0
	kJ/kWh	77	85	102	149	299
c) Total = a + b[4)]	g/kWh	179.5	177.0 [5)]	183.0	185.0	200.0
	kJ/kWh	7,665	7,558	7,814	7,900	8,540

[1)] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2)] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3)] Due to engine´s certification for compliance with the NO_x limits according E2 (Test cycle for "constant-speed main propulsion application" including electric propulsion and all controllable-pitch propeller installations) factory acceptance test will be done with constant speed only.

[4)] Liquid fuel operation (including pilot fuel). For consideration of fuel leakage amount, consider table(s) Leakage rate, Page 153 for conventional injection.

[5)] Warranted fuel consumption at 85 % MCR.

Table 33: Fuel consumption MAN 51/60DF (1,050 kW/cyl.) in liquid fuel mode – Mechanical propulsion with CPP (speed = constant)

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Speeds on recommended combinator curve (±5 rpm)		514 (500)	514 (500)	496 (482)	442 (430)	362 (353)
Standard Spec. fuel oil consumption with HFO, MDO (DMB, DFB), MGO (DMA, DFA) without attached pumps[1) 2) 3)]						
a) Main fuel	g/kWh	177.7	175.0	179.1	177.0	183.0
b) Pilot fuel	g/kWh	1.8	2.0	2.4	3.5	7.0
	kJ/kWh	77	85	102	149	299
c) Total = a + b[4)]	g/kWh	179.5	177.0 [5)]	181.5	180.5	190.0
	kJ/kWh	7,665	7,558	7,750	7,707	8,113

[1)] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2)] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3)] Due to engine´s certification for compliance with the NO_x limits according E2 (Test cycle for "constant-speed main propulsion application" including electric propulsion and all controllable-pitch propeller installations) factory acceptance test will be done with constant speed only.

[4)] Liquid fuel operation (including pilot fuel). For consideration of fuel leakage amount, consider table(s) Leakage rate, Page 153 for conventional injection.

[5)] Warranted fuel consumption at 85 % MCR.

Table 34: Fuel consumption MAN 51/60DF (1,050 kW/cyl.) in liquid fuel mode – Mechanical propulsion with CPP, speeds according to recommended combinator curve

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Engine 6 – 9L/12 – 14V, MAN 51/60DF (1,150 kW/cyl.) – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Spec. fuel consumption in gas mode without attached pumps[1) 2) 3)]						
Speed		constant = 514 rpm or 500 rpm				
a) Natural gas	kJ/kWh	6L: 7,340 7 – 9L: 7,320 V: 7,300	6 – 9L: 7,295 V: 7,275	L/V: 7,340	L/V: 7,580	L/V: 8,565
b) Pilot fuel	g/kWh kJ/kWh	2.3 100	2.9 125	3.3 140	5.1 220	10.2 435
c) Total = a + b[4)]	kJ/kWh	6L: 7,440 7 – 9L: 7,420 V: 7,400	6 – 9L: 7,420 [5)] V: 7,400 [5)]	L/V: 7,480	L/V: 7,800	L/V: 9,000
Speeds (±5 rpm)		514 (500)	514 (500)	496 (482)	442 (430)	362 (353)
a) Natural gas	kJ/kWh	6L: 7,340 7 – 9L: 7,320 V: 7,300	6 – 9L: 7,295 V: 7,275	L/V: 7,270	L/V: 7,360	L/V: 7,765
b) Pilot fuel	g/kWh kJ/kWh	2.3 100	2.9 125	3.3 140	5.1 220	10.2 435
c) Total = a + b[4)]	kJ/kWh	6L: 7,440 7 – 9L: 7,420 V: 7,400	6 – 9L: 7,420 [5)] V: 7,400 [5)]	L/V: 7,410	L/V: 7,580	L/V: 8,200

[1)] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2)] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3)] Due to engine´s certification for compliance with the NO_x limits according E2 (Test cycle for "constant-speed main propulsion application" including electric propulsion and all controllable-pitch propeller installations) factory acceptance test will be done with constant speed only.

[4)] Gas operation (including pilot fuel).

[5)] Warranted fuel consumption at 85 % MCR.

Table 35: Fuel consumption MAN 51/60DF (1,150 kW/cyl.) in gas mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Speed		**constant = 514 rpm or 500 rpm**				
Standard Spec. fuel oil consumption with HFO, MDO (DMB, DFB), MGO (DMA, DFA) without attached pumps[1) 2) 3)]						
a) Main fuel	g/kWh	184.7	180.5	186.1	185.0	191.5
b) Pilot fuel	g/kWh kJ/kWh	1.8 77	2.0 85	2.4 102	3.5 149	7.0 299
c) Total = a + b[4)]	g/kWh kJ/kWh	186.5 7,964	182.5 [5)] 7,793	188.5 8,049	188.5 8,049	198.5 8,476

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% Output	100	85	75	50	25
Speed	constant = 514 rpm or 500 rpm				

[1] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3] Due to engine´s certification for compliance with the NO_x limits according E2 (Test cycle for "constant-speed main propulsion application" including electric propulsion and all controllable-pitch propeller installations) factory acceptance test will be done with constant speed only.

[4] Liquid fuel operation (including pilot fuel). For consideration of fuel leakage amount, consider table(s) Leakage rate, Page 153 for conventional injection.

[5] Warranted fuel consumption at 85 % MCR.

Table 36: Fuel consumption MAN 51/60DF (1,150 kW/cyl.) in liquid fuel mode – Mechanical propulsion with CPP (speed = constant)

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm

% Output		100	85	75	50	25
Speed		514 (500)	514 (500)	501 (488)	462 (450)	402 (391)
Standard Spec. fuel oil consumption with HFO, MDO (DMB, DFB), MGO (DMA, DFA) without attached pumps[1) 2) 3)]						
a) Main fuel	g/kWh	184.7	180.5	186.1	181.5	183.5
b) Pilot fuel	g/kWh	1.8	2.0	2.4	3.5	7.0
	kJ/kWh	77	85	102	149	299
c) Total = a + b[4)]	g/kWh	186.5	182.5 [5)]	187.5	185.0	190.5
	kJ/kWh	7,964	7,793	8,006	7,900	8,134

[1] Based on reference conditions, see table Reference conditions for fuel consumption, Page 94.

[2] Tolerance +5 %.

Note: The additions to fuel consumption must be considered before the tolerance for warranty is taken into account.

[3] Due to engine´s certification for compliance with the NO_x limits according E2 (Test cycle for "constant-speed main propulsion application" including electric propulsion and all controllable-pitch propeller installations) factory acceptance test will be done with constant speed only.

[4] Liquid fuel operation (including pilot fuel). For consideration of fuel leakage amount, consider table(s) Leakage rate, Page 153 for conventional injection.

[5] Warranted fuel consumption at 85 % MCR.

Table 37: Fuel consumption MAN 51/60DF (1,150 kW/cyl.) in liquid fuel mode – Mechanical propulsion with CPP, speeds according to recommended combinator curve

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Additions to fuel consumption

1. Engine driven pumps increase the fuel consumption by:

$$be_{ISO\ with\ pumps} = be_{ISO\ without\ pumps} \times (1 + f_{pumps})$$

$$f_{pumps} = (f_{HT\ pumps} + f_{LT\ pumps} + f_{LO\ pumps}) \times \frac{1{,}050\ kW/cyl.}{Nominal\ output\ per\ cyl.}$$

For HT CW service pump (attached)

$$f_{HT\ pumps} = i_{HT\ pumps} \times 0.0018 \times \frac{100\%}{load\%} \times \left(\frac{n_x}{n_n}\right)^3$$

For LT CW service pump (attached)

$$f_{LT\ pumps} = i_{LT\ pumps} \times 0.0021 \times a \times \frac{100\%}{load\%} \times \left(\frac{n_x}{n_n}\right)^3$$

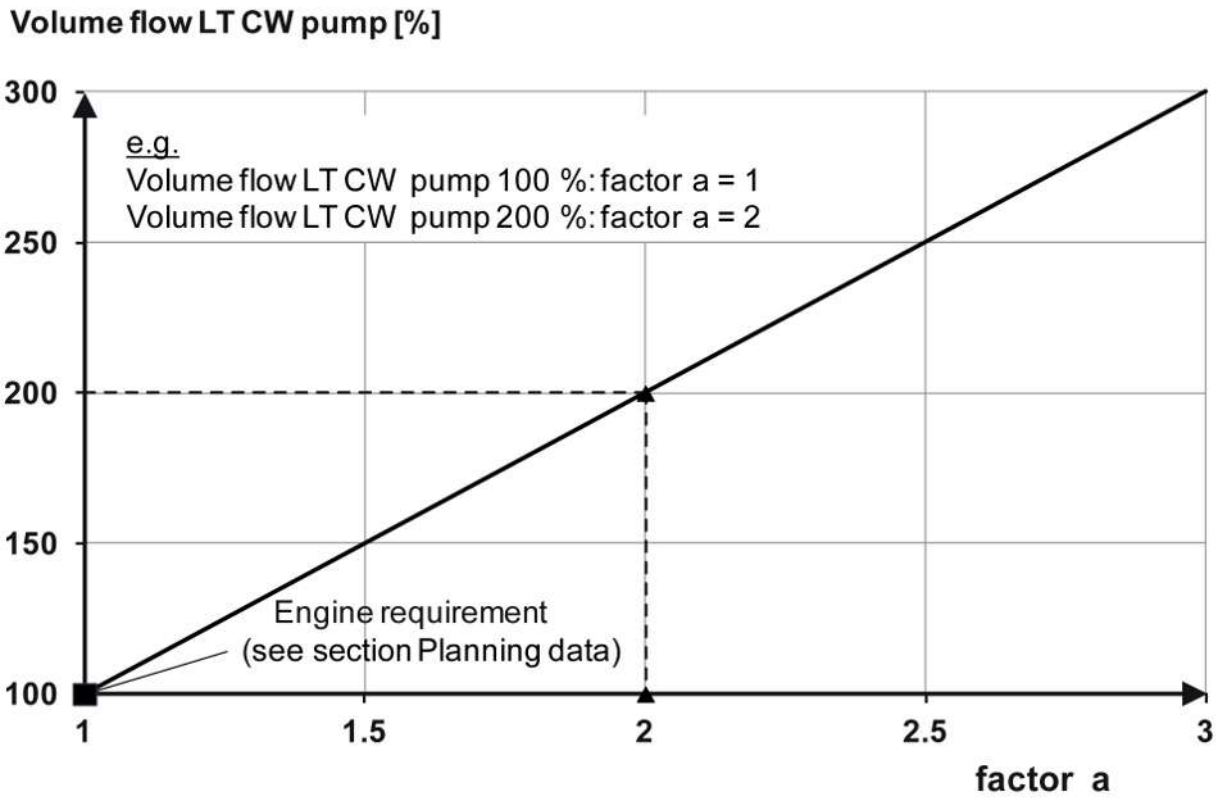

Figure 41: Derivation of factor a

Note:
Formula is relevant for centrifugal pumps and is valid for the nominal flow rate. Due to linear influence of engine speed on flow capacity of attached water pump and quadratic influence of engine speed on water pressure of attached water pump, the required drive power is influenced by the engine speed to the power of three.

For all lube oil service pumps (attached)

GenSet, electric propulsion:

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$$f_{LO\ pumps} = 0.0097 \times \frac{100\%}{load\%} \times \left(\frac{n_x}{n_n}\right)^1$$

Mechanical propulsion CPP:

$$f_{LO\ pumps} = 0.0097 \times \frac{100\%}{load\%} \times \left(\frac{n_x}{n_n}\right)^1$$

f_{pumps}	Actual factor for impact of attached pumps	[-]
$i_{HT\ pumps}$	Number of attached HT cooling water service pumps	[-]
$i_{LT\ pumps}$	Number of attached LT cooling water service pumps	[-]
n_x	Actual engine speed	[rpm]
n_n	Nominal engine speed	[rpm]
load%	Actual engine load	[%]
Nominal output per cylinder	Insert the nominal output per cylinder	[kW/cyl.]

2. For exhaust gas back pressure after turbine > 50 mbar

Every additional 1 mbar (0.1 kPa) back pressure addition of 0.025 g/kWh to be calculated.

3. For exhaust gas temperature control by VTA (SCR) – Only liquid fuel mode

For every increase of the exhaust gas temperature by 1 °C, due to activation of VTA, an addition of 0.035 g/kWh to be calculated.

Fuel oil consumption at idle running

No. of cylinders, config.		6L	7L	8L	9L	12V	14V	16V
Liquid fuel operation at 500/514 rpm, based on DMA (42,700 kj/kg)	kg/h	100	120	140	160	200	230	265
Gas operation at 500/514 rpm, based on gas (48,000 kj/kg) and DMA (42,700 kj/kg)	kg/h	Gas: 113 Pilot fuel: 24	Gas: 132 Pilot fuel: 28	Gas: 150 Pilot fuel: 32	Gas: 170 Pilot fuel: 36	Gas: 226 Pilot fuel: 48	Gas: 264 Pilot fuel: 56	Gas: 300 Pilot fuel: 64

Table 38: Liquid fuel and gaseous fuel consumption at idle running

Reference conditions for fuel consumption

According to ISO 15550; ISO 3046-1

Air temperature before turbocharger t_r	K/°C	298/25
Total atmospheric pressure p_r	kPa	100
Relative humidity Φ_r	%	30
Exhaust gas back pressure after turbocharger[1)]	kPa	5
Engine type specific reference charge air temperature before cylinder t_{bar}[2)]	K/°C	316/43

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Temperature control of temperature turbine outlet by VTA (only liquid fuel mode): Setpoint as minimum temperature for deactivated SCR	K/°C	563/290
Fuel gas, methane number[3)]	-	≥ 80
Liquid fuel: Main fuel, pilot fuel[4)] NCV	kJ/kg	42,700
Pilot fuel[5)] cetane number (CN)	-	55 ≤ CN ≤ 60

[1)] Measured at 100 % load, accordingly lower for loads < 100 %.

[2)] Regulated temperature for dual fuel and gas engines at engine loads ≥ 85 %.

[3)] Methane number (MN) calculated according to EN 16726 using GasCalc at least version 2.5.2.

[4)] Pilot fuel: Only DMA, DFA or DMB, DFB.

[5)] A lower cetane number might result in a delayed combustion and accordingly increased consumption figures in gas operation.

Table 39: Reference conditions for fuel consumption MAN 51/60DF

IMO Tier II requirements:

For detailed information see section Cooling water system, Page 335.

IMO: International Maritime Organization

MARPOL 73/78; Revised Annex VI-2008, Regulation 13

Tier II: NO_x technical code on control of emission of nitrogen oxides from diesel engines.

2.16.2 Urea consumption for emission standard IMO Tier III

The table below shows indicative the urea consumption values for the reduction from IMO Tier II to IMO Tier III level according to MARPOL Annex VI – Regulation 13 for different engine types. The indicative values are based on aqueous solution having an urea content of 40 %, engine type related standard performance map, load point 100 % MCR and DMA grade fuel. See also section Specification of urea solution, Page 295.

Speed (rpm)	500/514	600	720/750	800	900	1,000
Urea consumption (g/kWh)	13.5	13.0	13.0	12.5	12.0	11.5

Table 40: Urea consumption

Note:
Urea consumption could be different for engines with specific load point optimisation. For more detailed information on expected level of urea consumption, contact MAN Energy Solutions with your project-specific request.

2.16.3 Lube oil consumption

1,050 kW/cyl., 500 rpm or 514 rpm or 1,150 kW/cyl., 500 rpm or 514 rpm

Specific lube oil consumption:

$$0.4\ \text{g/kWh}^{1)} \times \frac{100\%}{\text{load\%}} \times \frac{1{,}000\ \text{kW/cyl.}}{\text{nominal output per cyl.}}$$

load%	Actual engine load	[%]
nominal output per cyl.	Insert the nominal output per cyl.	[kW/cyl.]

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[1] The value stated above is without any losses due to cleaning of filter and centrifuge or lube oil charge replacement. Tolerance for warranty +20 %.

Example:

For nominal output 1,050 kW/cyl. and 100 % actual engine load: 0.38 g/kWh

For nominal output 1,150 kW/cyl. and 100 % actual engine load: 0.35 g/kWh

2.16.4 Compressed air consumption – SCR system

Soot blowing and urea injection requires compressed air. Depending on the SCR reactor size the following volumes are permanent and in SCR operation required.

Mixing device	Engine power approximately	Mixing pipe	Permanent air consumption by sootblower	Additional air consumption in SCR operation by urea injection[1) 2)]
No.	kW	DN	Nm^3/h[3)]	Nm^3/h[3)]
1	0 – 1,000	500	3	9
2	1,001 – 2,000	600	3	12
3	2,001 – 3,000	800	3	23
4	3,001 – 4,200	1,000	3	37
5	4,201 – 5,400	1,100	3	46
6	5,401 – 6,800	1,200	3	50
7	6,801 – 8,500	1,400	3	64
8	8,501 – 10,500	1,500	5	90
9	10,501 – 13,000	1,600	5	112
10	13,001 – 20,000	2,100	5	140
11	20,001 – 21,600	2,300	5	210

[1)] Starting and shutdown of the SCR system can result in short-term higher compressed air consumption caused by cooling and scavenging phases.

[2)] Tolerance: Quantity +10 %.

[3)] Nm^3 corresponds to one cubic metre of gas at 20 °C and 100.0 kPa abs.

Table 41: Compressed air consumption – SCR system

2.16.5 Starting air and control air consumption

No. of cylinders, config.		6L	7L	8L	9L	12V	14V	16V
Control air consumption	Nm^3/h[1)]	1.5						
Air consumption per start[2)]								
Liquid fuel mode:	Nm^3 [1)]	3.2	4.5	4.0	4.0	4.8	9.0	6.0
Gas mode:	Nm^3 [1)]	3.9	5.4	4.8	4.8	5.8	10.8	7.2
Air consumption per slow turn manoeuvre[2) 3)]	Nm^3 [1)]	5.6	6.4	7.0	7.6	9.6	11.0	12.0

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No. of cylinders, config.		6L	7L	8L	9L	12V	14V	16V
Reference moment of inertia for stated air consumption figures[2]	kgm²	5,735	6,518	6,576	6,881	7,705	8,291	8,878
Air consumption per jet assist activation[4], 1,050 kW/cyl.	Nm³ [1]	3.9	3.9	3.9	5.4	5.4	7.8	7.8
Air consumption per jet assist activation[4], 1,150 kW/cyl.	Nm³ [1]	3.9	3.9	5.4	5.4	7.8	7.8	11.3
Air consumption jet assist in case of emergency loading	Nm³ [5]	To be considered: 20 jet assist activations during loading from 0 % to 100 % load						

[1] Nm³ corresponds to one cubic metre of gas at 20 °C and 100.0 kPa abs.

[2] The stated air consumption values are based on the "Reference moment of inertia" in this table. The air consumption per starting manoeuvre/slow turn of the unit (e.g. engine plus alternator) increases in relation to its total moment of inertia. Please consider also the "Required minimum total moments of inertia" as stated within section Moments of inertia – Crankshaft, damper, flywheel, Page 174.

[3] Required for plants with power management system demanding automatic engine start. The air consumption per slow turn activation depends on the inertia moment of the unit. This value does not include air consumption required for the automatically activated engine start after the end of the slow turn manoeuvre.

[4] The mentioned above air consumption per jet assist activation is valid for a jet duration of 5 seconds. The jet duration may vary between 3 sec and 10 sec, depending on the loading (average jet duration 5 sec).

[5] See accordingly section Load application – Continuous loading, Page 56.

Table 42: Starting air and control air consumption

Note:
See also section External compressed air system – Dimensioning starting air receivers, compressors, Page 418.

2.16.6 Recalculation of total gas consumption and NOx emission dependent on ambient conditions

In accordance to ISO standard ISO 3046-1 *"Reciprocating internal combustion engines – Performance, Part 1: Declarations of power, fuel and lubricating oil consumptions, and test methods – Additional requirements for engines for general use"* MAN Energy Solutions has specified the method for recalculation of total gas consumption dependent on ambient conditions.

Details will be clarified during project handling.

2.16.7 Recalculation of liquid fuel consumption dependent on ambient conditions

In accordance to ISO standard ISO 3046-1 "*Reciprocating internal combustion engines – Performance, Part 1: Declarations of power, fuel and lube oil consumptions, and test methods – Additional requirements for engines for general use*" MAN Energy Solutions has specified the method for recalculation of fuel consumption for liquid fuel dependent on ambient conditions for single-stage turbocharged engines as follows:

$$\beta = 1 + 0.0006 \times (t_x - t_r) + 0.0004 \times (t_{bax} - t_{bar}) + 0.07 \times (p_r - p_x)$$

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The formula is valid within the following limits:	
Ambient air temperature	5 °C – 55 °C
Charge air temperature before cylinder	25 °C – 75 °C
Ambient air pressure	0.885 bar – 1.030 bar

Table 43: Limit values for recalculation of liquid fuel consumption

$$b_x = b_r \times \beta \qquad b_r = \frac{b_x}{\beta}$$

β	Fuel consumption factor
t_{bar}	Engine type specific reference charge air temperature before cylinder see table Reference conditions for fuel consumption, Page 94

	Unit	Reference	At test run or at site
Specific fuel consumption	[g/kWh]	b_r	b_x
Ambient air temperature	[°C]	t_r	t_x
Charge air temperature before cylinder	[°C]	t_{bar}	t_{bax}
Ambient air pressure	[bar]	p_r	p_x

Table 44: Recalculation of liquid fuel consumption – Units and references

Example

Reference values:

b_r = 200 g/kWh, t_r = 25 °C, t_{bar} = 40 °C, p_r = 1.0 bar

At site:

t_x = 45 °C, t_{bax} = 50 °C, p_x = 0.9 bar

ß = 1+ 0.0006 (45 – 25) + 0.0004 (50 – 40) + 0.07 (1.0 – 0.9) = 1.023

b_x = ß x b_r = 1.023 x 200 = 204.6 g/kWh

2.16.8 Influence of engine aging on fuel consumption

The fuel oil consumption will increase over the running time of the engine. Timely service can reduce or eliminate this increase. For dependencies see following figures.

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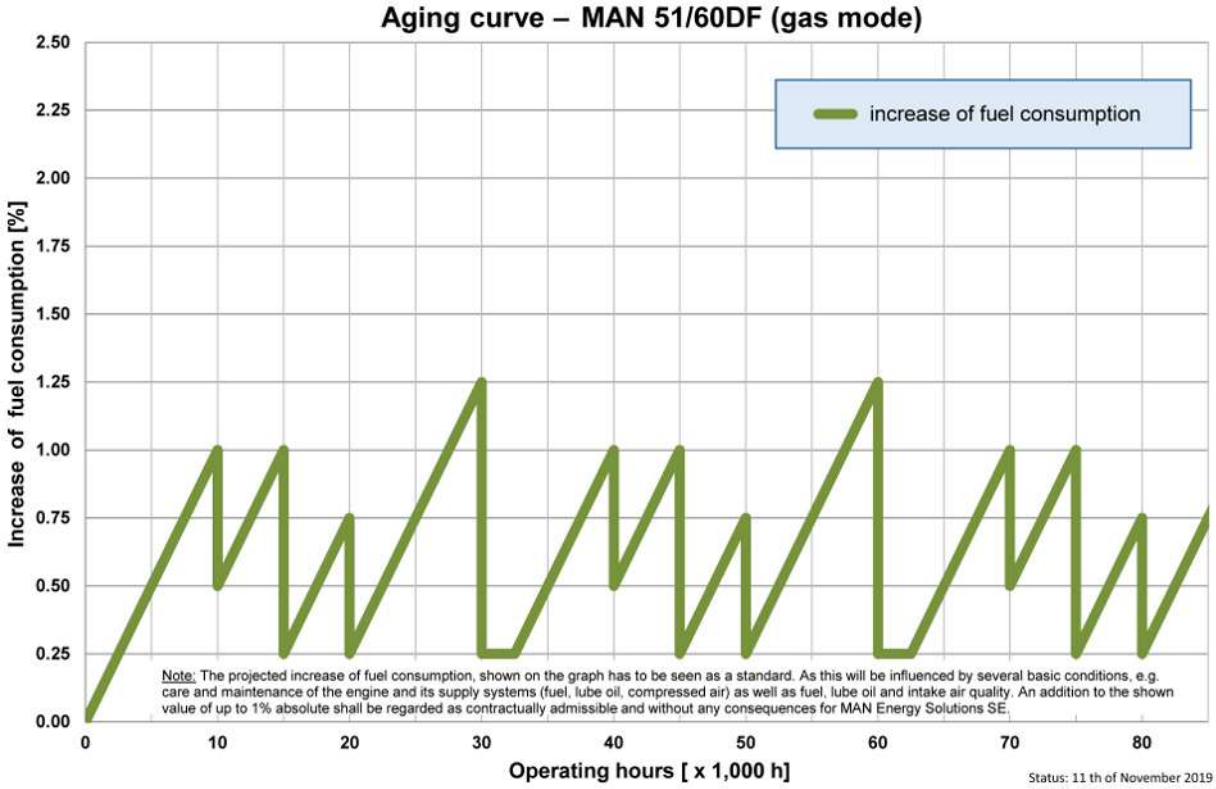

Figure 42: Influence of total engine running time and service intervals on fuel consumption in gas mode

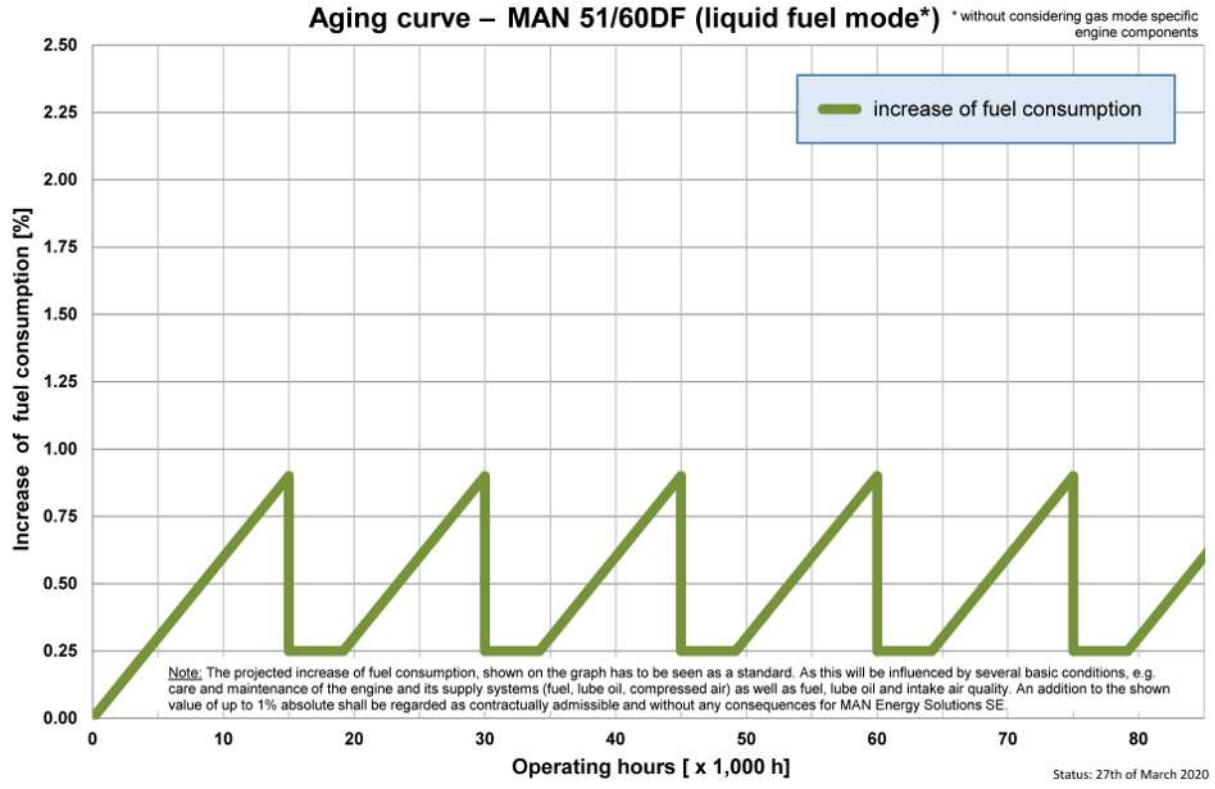

Figure 43: Influence of total engine running time and service intervals on fuel oil consumption in liquid fuel mode

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2.17 Planning data for emission standard: IMO Tier II, 1,050 kW/cyl. – Electric propulsion

2.17.1 Nominal values for cooler specification – MAN L51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode/gas mode – Electric propulsion

Note:
If an advanced HT cooling water system for increased freshwater generation is to be applied, contact MAN Energy Solutions for corresponding planning data.

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 45: Reference conditions: Tropics

No. of cylinders, config.		**6L**		**7L**		**8L**		**9L**	
Engine output	kW	6,300		7,350		8,400		9,450	
Speed	rpm	500/514							
Heat to be dissipated[1]		liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode
Charge air: Charge air cooler (HT stage) Charge air cooler (LT stage)	kW	2,297 873	1,777 815	2,617 1,012	2,434 937	2,913 1,270	2,622 1,144	3,203 1,417	3,050 1,322
Lube oil cooler[2]		686	489	802	568	915	649	1,032	730
Jacket cooling		670	557	785	649	895	741	1,013	834
Nozzle cooling water		14	14	17	17	19	19	21	21
Turbocharger compressor wheel cooling		26.4	26.4	26.4	26.4	26.4	26.4	37.8	37.8
Heat radiation engine (based on 55 °C engine room temperature)		157	157	183	183	209	209	235	235
Flow rates[3]									
HT circuit (jacket cooling + charge air cooler HT stage)	m^3/h	77		88		99		110	
LT circuit (lube oil cooler + charge air cooler LT stage)		85		100		110		125	
Lube oil		140		158		176		194	
Cooling water fuel nozzles		1.7		2.0		2.2		2.5	

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No. of cylinders, config.		6L	7L	8L	9L
LT cooling water turbocharger compressor wheel		2.3	2.3	2.3	3.3
Pumps					
a) Attached					
HT CW service pump	m^3/h	77	88	99	110
LT CW service pump		85	100	110	125
Lube oil service pump		182	182	218	252
b) Free-standing[4]					
HT CW stand-by pump	m^3/h	77	88	99	110
LT CW stand-by pump		Depending on plant design			
Lube oil stand-by pump		147+z	166+z	185+z	204+z
Nozzle CW pump		1.7	2.0	2.2	2.5
Prelubrication pump		28.0 – 33.0 +0.5z	31.5 – 37.0 +0.5z	35.0 – 41.0 +0.5z	38.5 – 45.0 +0.5z
MGO/MDO supply pump		4.2	4.9	5.6	6.3
HFO supply pump		2.1	2.5	2.8	3.2
HFO circulation pump		4.2	4.9	5.6	6.3
Pilot fuel supply[5]	l/h	66.1	73.1	80.1	87.0

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] Addition required for separator heat (30 kJ/kWh).

[3] Basic values for layout design of the coolers.

[4] Tolerances of the pumps delivery capacities must be considered by the manufacturer.

[5] In case of an activation of the pressure limiting valve of the common rail system by the safety system, the required pilot fuel supply increases to following amount: L engine: 179 l/h, V engine: 296 l/h.
This needs to be considered for each external pilot fuel supply systems at least for 1 of the connected engines.

z = Flushing oil of automatic filter.

Table 46: Nominal values for cooler specification – MAN L51/60DF, 1,050 kW/cyl., liquid fuel mode/gas mode – Electric propulsion

Note:
You will find further planning data for the listed subjects in the corresponding sections.

- Minimal heating power required for preheating HT cooling water see paragraph HT cooling water preheating module (MOD-004), Page 343.
- Minimal heating power required for preheating lube oil see paragraph H-002/Lube oil preheater, Page 313.
- Capacities of preheating pumps see paragraph HT cooling water preheating module (MOD-004), Page 343.

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2.17.2 Nominal values for cooler specification – MAN V51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode/gas mode – Electric propulsion

Note:
If an advanced HT cooling water system for increased freshwater generation is to be applied, contact MAN Energy Solutions for corresponding planning data.

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 47: Reference conditions: Tropics

No. of cylinders, config.		**12V**		**14V**		**16V**	
Engine output	kW	12,600		14,700		16,800	
Speed	rpm	500/514					
Heat to be dissipated[1]		liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode
Charge air: Charge air cooler (HT stage) Charge air cooler (LT stage)	kW	 4,593 1,744	 3,921 1,563	 5,234 2,023	 4,867 1,875	 5,827 2,541	 5,243 2,289
Lube oil cooler[2]		1,371	975	1,604	1,136	1,831	1,299
Jacket cooling		1,339	1,112	1,571	1,297	1,792	1,483
Nozzle cooling water		28	28	33	33	38	38
Turbocharger compressor wheel cooling		37.8	37.8	52.7	52.7	52.7	52.7
Heat radiation engine (based on 55 °C engine room temperature)		314	314	366	366	418	418
Flow rates[3]							
HT circuit (jacket cooling + charge air cooler HT stage)	m^3/h	154		176		198	
LT circuit (lube oil cooler + charge air cooler LT stage)		170		200		220	
Lube oil		340		370		400	
Cooling water fuel nozzles		3.5		4.1		4.8	
LT cooling water turbocharger compressor wheel		3.3		4.6		4.6	
Pumps							
a) Attached							
HT CW service pump	m^3/h	154		176		198	
LT CW service pump		170		200		220	
Lube oil service pump		364		408		436	

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No. of cylinders, config.		12V	14V	16V
b) Free-standing[4]				
HT CW stand-by pump	m³/h	154	176	198
LT CW stand-by pump		Depending on plant design		
Lube oil stand-by pump		357+z	389+z	420+z
Nozzle CW pump		3.5	4.1	4.8
Prelubrication pump		58 – 68 +0.5z	63 – 74 +0.5z	68 – 80 +0.5z
MGO/MDO supply pump		8.4	9.8	11.2
HFO supply pump		4.2	4.9	5.6
HFO circulation pump		8.4	9.8	11.2
Pilot fuel supply[5]	l/h	108.0	122.0	136.0

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] Addition required for separator heat (30 kJ/kWh).

[3] Basic values for layout design of the coolers.

[4] Tolerances of the pumps delivery capacities must be considered by the manufacturer.

[5] In case of an activation of the pressure limiting valve of the common rail system by the safety system, the required pilot fuel supply increases to following amount: L engine: 179 l/h, V engine: 296 l/h.
This needs to be considered for each external pilot fuel supply systems at least for 1 of the connected engines.

z = Flushing oil of automatic filter.

Table 48: Nominal values for cooler specification – MAN V51/60DF, 1,050 kW/cyl., liquid fuel mode/gas mode – Electric propulsion

Note:
You will find further planning data for the listed subjects in the corresponding sections.

- Minimal heating power required for preheating HT cooling water see paragraph HT cooling water preheating module (MOD-004), Page 343.
- Minimal heating power required for preheating lube oil see paragraph H-002/Lube oil preheater, Page 313.
- Capacities of preheating pumps see paragraph HT cooling water preheating module (MOD-004), Page 343.

2.17.3 Temperature basis, nominal air and exhaust gas data – MAN L51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode – Electric propulsion

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 49: Reference conditions: Tropics

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No. of cylinders, config.		6L	7L	8L	9L
Engine output	kW	6,300	7,350	8,400	9,450
Speed	rpm	500/514			
Temperature basis					
HT cooling water engine outlet	°C	90			
LT cooling water charge air cooler inlet		38 [1]			
Lube oil engine inlet		55			
Cooling water fuel nozzles inlet		60			
Air data					
Temperature of charge air at charge air cooler outlet	°C	58.0	59.3	58.8	60.0
Air flow rate[2]	m^3/h	40,316	47,020	53,745	60,440
	t/h	44.1	51.5	58.8	66.1
Charge air pressure (absolute)	bar abs	4.93	4.94	4.93	4.94
Air required to dissipate heat radiation (engine) ($t_2 - t_1$ = 10 °C)	m^3/h	50,365	58,760	67,155	75,550
Heat radiation engine (based on 55 °C engine room temperature)	kW	157	183	209	235
Exhaust gas data[3]					
Volume flow (temperature turbine outlet)[4]	m^3/h	78,571	91,748	104,814	118,024
Mass flow	t/h	45.4	52.9	60.5	68.0
Temperature at turbine outlet	°C	330	331	331	331
Heat content (190 °C)	kW	1,894	2,220	2,532	2,865
Permissible exhaust gas back pressure	mbar	≤ 50			

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 50: Temperature basis, nominal air and exhaust gas data – MAN L51/60DF, 1,050 kW/cyl., liquid fuel mode – Electric propulsion

2.17.4 Temperature basis, nominal air and exhaust gas data – MAN L51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Electric propulsion

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 51: Reference conditions: Tropics

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No. of cylinders, config.		6L	7L	8L	9L
Engine output	kW	6,300	7,350	8,400	9,450
Speed	rpm	500/514			
Temperature basis					
HT cooling water engine outlet	°C	90			
LT cooling water charge air cooler inlet		38 [1]			
Lube oil engine inlet		55			
Cooling water fuel nozzles inlet		60			
Air data					
Temperature of charge air at charge air cooler outlet	°C	52.0	57.5	56.4	58.4
Air flow rate[2]	m^3/h	31,860	40,520	45,390	52,620
	t/h	34.9	44.4	49.7	57.6
Charge air pressure (absolute)	bar abs	4.44	4.96	4.86	5.02
Air required to dissipate heat radiation (engine) ($t_2 - t_1$ = 10 °C)	m^3/h	50,365	58,760	67,155	75,550
Heat radiation engine (based on 55 °C engine room temperature)	kW	157	183	209	235
Exhaust gas data[3]					
Volume flow (temperature turbine outlet)[4]	m^3/h	58,320	70,810	80,090	91,750
Mass flow	t/h	35.9	45.6	51.1	59.3
Temperature at turbine outlet	°C	293	268	273	266
Heat content (180 °C)	kW	1,239	1,217	1,445	1,534
Permissible exhaust gas back pressure	mbar	≤ 50			

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 52: Temperature basis, nominal air and exhaust gas data – MAN L51/60DF, 1,050 kW/cyl., gas mode – Electric propulsion

2.17.5 Temperature basis, nominal air and exhaust gas data – MAN V51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode – Electric propulsion

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 53: Reference conditions: Tropics

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No. of cylinders, config.		12V	14V	16V
Engine output	kW	12,600	14,700	16,800
Speed	rpm	500/514		
Temperature basis				
HT cooling water engine outlet	°C	90		
LT cooling water charge air cooler inlet		38 [1]		
Lube oil engine inlet		55		
Cooling water fuel nozzles inlet		60		
Air data				
Temperature of charge air at charge air cooler outlet	°C	58.0	59.3	58.8
Air flow rate[2]	m^3/h	80,637	94,042	107,494
	t/h	88.3	102.9	117.6
Charge air pressure (absolute)	bar abs	4.94		
Air required to dissipate heat radiation (engine) ($t_2 - t_1$ = 10 °C)	m^3/h	100,730	117,520	134,305
Heat radiation engine (based on 55 °C engine room temperature)	kW	314	366	418
Exhaust gas data[3]				
Volume flow (temperature turbine outlet)[4]	m^3/h	157,140	183,500	209,630
Mass flow	t/h	90.7	105.8	120.9
Temperature at turbine outlet	°C	331		
Heat content (190 °C)	kW	3,789	4,444	5,067
Permissible exhaust gas back pressure	mbar	≤ 50		

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 54: Temperature basis, nominal air and exhaust gas data – MAN V51/60DF, 1,050 kW/cyl., liquid fuel mode – Electric propulsion

2.17.6 Temperature basis, nominal air and exhaust gas data – MAN V51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Electric propulsion

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 55: Reference conditions: Tropics

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No. of cylinders, config.		12V	14V	16V
Engine output	kW	12,600	14,700	16,800
Speed	rpm	500/514		
Temperature basis				
HT cooling water engine outlet	°C	90		
LT cooling water charge air cooler inlet		38 [1]		
Lube oil engine inlet		55		
Cooling water fuel nozzles inlet		60		
Air data				
Temperature of charge air at charge air cooler outlet	°C	55.0	57.5	56.4
Air flow rate[2]	m^3/h	66,480	81,045	90,780
	t/h	72.8	88.8	99.5
Charge air pressure (absolute)	bar abs	4.73	4.96	4.86
Air required to dissipate heat radiation (engine) ($t_2 - t_1$ = 10 °C)	m^3/h	100,730	117,520	134,305
Heat radiation engine (based on 55 °C engine room temperature)	kW	314	366	418
Exhaust gas data[3]				
Volume flow (temperature turbine outlet)[4]	m^3/h	118,680	141,780	160,330
Mass flow	t/h	74.9	91.3	102.3
Temperature at turbine outlet	°C	279	268	273
Heat content (180 °C)	kW	2,256	2,434	2,890
Permissible exhaust gas back pressure	mbar	≤ 50		

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 56: Temperature basis, nominal air and exhaust gas data – MAN V51/60DF, 1,050 kW/cyl., gas mode – Electric propulsion

2.17.7 Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode – Electric propulsion

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: ISO		
Air temperature	°C	25
Cooling water temp. before charge air cooler (LT stage)		25
Total atmospheric pressure	mbar	1,000
Relative humidity	%	30

Table 57: Reference conditions: ISO

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,115 352	 997 353	 1,043 379	 629 375
Lube oil cooler[3]		363	391	421	567
Jacket cooling		331	332	364	446
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 35 °C engine room temperature)		115	116	124	156
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 234 43.0	 212 43.0	 207 43.0	 155 43.0
Air flow rate	kg/kWh	7.48	7.80	8.47	8.75
Charge air pressure (absolute)	bar abs	5.04	4.44	4.25	2.94
Exhaust gas data[4]					
Mass flow	kg/kWh	7.67	7.99	8.66	8.95
Temperature at turbine outlet	°C	292	280	276	296
Heat content (190 °C)	kJ/kWh	831	761	789	1,010
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 58: Load specific values at ISO conditions – MAN L/V51/60DF, 1,050 kW/cyl., liquid fuel mode – Electric propulsion

2.17.8 Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Electric propulsion

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: ISO		
Air temperature	°C	25
Cooling water temp. before charge air cooler (LT stage)		25
Total atmospheric pressure	mbar	1,000
Relative humidity	%	30

Table 59: Reference conditions: ISO

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 728 249	 584 229	 478 225	 223 225
Lube oil cooler[3]		275	350	344	488
Jacket cooling		317	350	366	484
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 35 °C engine room temperature)		115	116	125	157
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 207 43.0	 182 45.0	 164 46.0	 121 48.0
Air flow rate	kg/kWh	5.81	5.78	5.83	6.05
Charge air pressure (absolute)	bar abs	4.02	3.41	3.03	2.13
Exhaust gas data[4]					
Mass flow	kg/kWh	5.97	5.94	5.99	6.23
Temperature at turbine outlet	°C	336	361	378	432
Heat content (180 °C)	kJ/kWh	1,021	1,181	1,307	1,742
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 60: Load specific values at ISO conditions – MAN L/V51/60DF, 1,050 kW/cyl., gas mode – Electric propulsion

2.17.9 Load specific values at tropic conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode – Electric propulsion

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 61: Reference conditions: Tropics

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,312 498	 1,200 438	 1,263 445	 847 225
Lube oil cooler[3]		392	423	455	613
Jacket cooling		383	384	421	515
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 55 °C engine room temperature)		90	90	96	122
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 266 58.0	 242 58.0	 237 58.0	 182 58.0
Air flow rate	kg/kWh	7.00	7.30	7.93	8.19
Charge air pressure (absolute)	bar abs	4.93	4.34	4.16	2.87
Exhaust gas data[4]					
Mass flow	kg/kWh	7.20	7.49	8.13	8.40
Temperature at turbine outlet	°C	330	317	313	335
Heat content (190 °C)	kJ/kWh	1,083	1,022	1,070	1,302
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 62: Load specific values at tropic conditions – MAN L/V51/60DF, 1,050 kW/cyl., liquid fuel mode – Electric propulsion

2.17.10 Load specific values at tropic conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Electric propulsion

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 63: Reference conditions: Tropics

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,120 447	 790 411	 722 416	 402 292
Lube oil cooler[3]		279	346	350	488
Jacket cooling		318	352	382	484
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 55 °C engine room temperature)		90	90	97	122
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 269 55.0	 219 51.0	 201 49.3	 148 48.0
Air flow rate	kg/kWh	5.78	5.55	5.81	6.01
Charge air pressure (absolute)	bar abs	4.73	3.59	3.11	2.13
Exhaust gas data[4]					
Mass flow	kg/kWh	5.95	5.71	5.97	6.19
Temperature at turbine outlet	°C	279	332	363	428
Heat content (180 °C)	kJ/kWh	645	953	1,206	1,705
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 64: Load specific values at tropic conditions – MAN L/V51/60DF, 1,050 kW/cyl., gas mode – Electric propulsion

2.18 Planning data for emission standard: IMO Tier II, 1,150 kW/cyl. – Electric propulsion

2.18.1 Nominal values for cooler specification – MAN L51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode/gas mode – Electric propulsion

Note:
If an advanced HT cooling water system for increased freshwater generation is to be applied, contact MAN Energy Solutions for corresponding planning data.

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

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Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 65: Reference conditions: Tropics

No. of cylinders, config.		6L		7L		8L		9L	
Engine output	kW	6,900		8,050		9,200		10,350	
Speed	rpm	500/514							
Heat to be dissipated[1]		liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode
Charge air: Charge air cooler (HT stage) Charge air cooler (LT stage)	kW	2,731 912	2,240 892	3,096 1,119	2,662 1,042	3,434 1,418	2,916 1,300	3,760 1,588	3,294 1,471
Lube oil cooler[2]		777	521	907	588	1,036	680	1,168	744
Jacket cooling		767	612	897	719	1,024	820	1,158	927
Nozzle cooling water		16	16	18	18	21	21	23	23
Turbocharger compressor wheel cooling		26.4	26.4	26.4	26.4	37.8	37.8	37.8	37.8
Heat radiation engine (based on 55 °C engine room temperature)		172	172	200	200	229	229	257	257
Flow rates[3]									
HT circuit (jacket cooling + charge air cooler HT stage)	m^3/h	77		88		99		110	
LT circuit (lube oil cooler + charge air cooler LT stage)		85		100		110		125	
Lube oil		140		158		176		194	
Cooling water fuel nozzles		1.7		2.0		2.2		2.5	
LT cooling water turbocharger compressor wheel		2.3		2.3		3.3		3.3	
Pumps									
a) Attached									
HT CW service pump	m^3/h	77		88		99		110	
LT CW service pump		85		100		110		125	
Lube oil service pump		182		182		218		252	
b) Free-standing[4]									
HT CW stand-by pump	m^3/h	77		88		99		110	
LT CW stand-by pump		Depending on plant design							

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No. of cylinders, config.		6L	7L	8L	9L
Lube oil stand-by pump		147+z	166+z	185+z	204+z
Nozzle CW pump		1.7	2.0	2.2	2.5
Prelubrication pump		28.0 – 33.0 +0.5z	31.5 – 37.0 +0.5z	35.0 – 41.0 +0.5z	38.5 – 45.0 +0.5z
MGO/MDO supply pump		4.6	5.4	6.1	6.9
HFO supply pump		2.3	2.7	3.1	3.5
HFO circulation pump		4.6	5.4	6.1	6.9
Pilot fuel supply[5]	l/h	66.1	73.1	80.1	87.0

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] Addition required for separator heat (30 kJ/kWh).

[3] Basic values for layout design of the coolers.

[4] Tolerances of the pumps delivery capacities must be considered by the manufacturer.

[5] In case of an activation of the pressure limiting valve of the common rail system by the safety system, the required pilot fuel supply increases to following amount: L engine: 179 l/h, V engine: 296 l/h.
This needs to be considered for each external pilot fuel supply systems at least for 1 of the connected engines.

z = Flushing oil of automatic filter.

Table 66: Nominal values for cooler specification – MAN L51/60DF, 1,150 kW/cyl., liquid fuel mode/gas mode – Electric propulsion

Note:
You will find further planning data for the listed subjects in the corresponding sections.

- Minimal heating power required for preheating HT cooling water see paragraph HT cooling water preheating module (MOD-004), Page 343.
- Minimal heating power required for preheating lube oil see paragraph H-002/Lube oil preheater, Page 313.
- Capacities of preheating pumps see paragraph HT cooling water preheating module (MOD-004), Page 343.

2.18.2 Nominal values for cooler specification – MAN V51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode/gas mode – Electric propulsion

Note:
If an advanced HT cooling water system for increased freshwater generation is to be applied, contact MAN Energy Solutions for corresponding planning data.

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 67: Reference conditions: Tropics

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No. of cylinders, config.		**12V**		**14V**	
Engine output	kW	13,800		16,100	
Speed	rpm	500/514			
Heat to be dissipated[1]		liquid fuel mode	gas mode	liquid fuel mode	gas mode
Charge air: Charge air cooler (HT stage) Charge air cooler (LT stage)	kW	 5,462 1,825	 4,480 1,784	 6,192 2,237	 5,323 2,084
Lube oil cooler[2]		1,553	1,041	1,814	1,176
Jacket cooling		1,534	1,224	1,795	1,438
Nozzle cooling water		31	31	36	36
Turbocharger compressor wheel cooling		52.7	52.7	52.7	52.7
Heat radiation engine (based on 55 °C engine room temperature)		343	343	400	400
Flow rates[3]					
HT circuit (jacket cooling + charge air cooler HT stage)	m^3/h	154		176	
LT circuit (lube oil cooler + charge air cooler LT stage)		170		200	
Lube oil		340		370	
Cooling water fuel nozzles		3.5		4.1	
LT cooling water turbocharger compressor wheel		4.6		4.6	
Pumps					
a) Attached					
HT CW service pump	m^3/h	154		176	
LT CW service pump		170		200	
Lube oil service pump		364		408	
b) Free-standing[4]					
HT CW stand-by pump	m^3/h	154		176	
LT CW stand-by pump		Depending on plant design			
Lube oil stand-by pump		357+z		389+z	
Nozzle CW pump		3.5		4.1	
Prelubrication pump		58.0 – 68.0 +0.5z		63.0 – 74.0 +0.5z	
MGO/MDO supply pump		9.2		10.7	
HFO supply pump		4.6		5.4	
HFO circulation pump		9.2		10.7	
Pilot fuel supply[5]	l/h	108.0		122.0	

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No. of cylinders, config.		12V	14V

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] Addition required for separator heat (30 kJ/kWh).

[3] Basic values for layout design of the coolers.

[4] Tolerances of the pumps delivery capacities must be considered by the manufacturer.

[5] In case of an activation of the pressure limiting valve of the common rail system by the safety system, the required pilot fuel supply increases to following amount: L engine: 179 l/h, V engine: 296 l/h.
This needs to be considered for each external pilot fuel supply systems at least for 1 of the connected engines.

z = Flushing oil of automatic filter.

Table 68: Nominal values for cooler specification – MAN V51/60DF, 1,150 kW/cyl., liquid fuel mode/gas mode – Electric propulsion

Note:
You will find further planning data for the listed subjects in the corresponding sections.

- Minimal heating power required for preheating HT cooling water see paragraph HT cooling water preheating module (MOD-004), Page 343.
- Minimal heating power required for preheating lube oil see paragraph H-002/Lube oil preheater, Page 313.
- Capacities of preheating pumps see paragraph HT cooling water preheating module (MOD-004), Page 343.

2.18.3 Temperature basis, nominal air and exhaust gas data – MAN L51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode – Electric propulsion

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 69: Reference conditions: Tropics

No. of cylinders, config.		6L	7L	8L	9L
Engine output	kW	6,900	8,050	9,200	10,350
Speed	rpm	500/514			
Temperature basis					
HT cooling water engine outlet	°C	90			
LT cooling water charge air cooler inlet		38 [1]			
Lube oil engine inlet		55			
Cooling water fuel nozzles inlet		60			
Air data					
Temperature of charge air at charge air cooler outlet	°C	61.0	61.6	61.2	62.5
Air flow rate[2]	m^3/h	46,924	54,734	62,562	70,355

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No. of cylinders, config.		6L	7L	8L	9L
	t/h	51.4	59.9	68.5	77.0
Charge air pressure (absolute)	bar abs	5.05	5.06	5.06	5.07
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	55,107	64,291	73,476	82,660
Heat radiation engine (based on 55 °C engine room temperature)	kW	172	200	229	257
Exhaust gas data[3]					
Volume flow (temperature turbine outlet)[4]	m^3/h	92,620	108,110	123,510	139,075
Mass flow	t/h	52.7	61.5	70.3	79.1
Temperature at turbine outlet	°C	338	339	338	339
Heat content (190 °C)	kW	2,332	2,728	3,112	3,518
Permissible exhaust gas back pressure	mbar	≤ 50			

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 70: Temperature basis, nominal air and exhaust gas data – MAN L51/60DF, 1,150 kW/cyl., liquid fuel mode – Electric propulsion

2.18.4 Temperature basis, nominal air and exhaust gas data – MAN L51/60DF IMO Tier II, 1,150 kW/cyl., gas mode – Electric propulsion

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 71: Reference conditions: Tropics

No. of cylinders, config.		6L	7L	8L	9L
Engine output	kW	6,900	8,050	9,200	10,350
Speed	rpm	500/514			
Temperature basis					
HT cooling water engine outlet	°C	90			
LT cooling water charge air cooler inlet		38 [1]			
Lube oil engine inlet		55			
Cooling water fuel nozzles inlet		60			
Air data					

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No. of cylinders, config.		6L	7L	8L	9L
Temperature of charge air at charge air cooler outlet	°C	59.2	59.9	59.2	60.8
Air flow rate[2]	m^3/h	41,380	49,085	55,760	63,015
	t/h	45.3	53.8	61.1	69.0
Charge air pressure (absolute)	bar abs	4.80	4.93	4.89	4.99
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	55,107	64,291	73,476	82,660
Heat radiation engine (based on 55 °C engine room temperature)	kW	172	200	229	257
Exhaust gas data[3]					
Volume flow (temperature turbine outlet)[4]	m^3/h	74,880	86,670	99,010	109,335
Mass flow	t/h	46.5	55.1	62.6	70.8
Temperature at turbine outlet	°C	288	275	278	271
Heat content (180 °C)	kW	1,522	1,583	1,855	1,953
Permissible exhaust gas back pressure	mbar	≤ 50			

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 72: Temperature basis, nominal air and exhaust gas data – MAN L51/60DF, 1,150 kW/cyl., gas mode – Electric propulsion

2.18.5 Temperature basis, nominal air and exhaust gas data – MAN V51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode – Electric propulsion

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 73: Reference conditions: Tropics

No. of cylinders, config.		12V	14V
Engine output	kW	13,800	16,100
Speed	rpm	500/514	
Temperature basis			
HT cooling water engine outlet	°C	90	
LT cooling water charge air cooler inlet		38 [1]	
Lube oil engine inlet		55	

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No. of cylinders, config.		12V	14V
Cooling water fuel nozzles inlet		60	
Air data			
Temperature of charge air at charge air cooler outlet	°C	61.0	61.6
Air flow rate[2]	m^3/h	93,849	109,468
	t/h	102.7	119.8
Charge air pressure (absolute)	bar abs	5.05	5.06
Air required to dissipate heat radiation (engine) ($t_2 - t_1$ = 10 °C)	m^3/h	110,213	128,580
Heat radiation engine (based on 55 °C engine room temperature)	kW	343	400
Exhaust gas data[3]			
Volume flow (temperature turbine outlet)[4]	m^3/h	185,240	216,210
Mass flow	t/h	105.5	123.1
Temperature at turbine outlet	°C	338	339
Heat content (190 °C)	kW	4,665	5,456
Permissible exhaust gas back pressure	mbar	≤ 50	

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 74: Temperature basis, nominal air and exhaust gas data – MAN V51/60DF, 1,150 kW/cyl., liquid fuel mode – Electric propulsion

2.18.6 Temperature basis, nominal air and exhaust gas data – MAN V51/60DF IMO Tier II, 1,150 kW/cyl., gas mode – Electric propulsion

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 75: Reference conditions: Tropics

No. of cylinders, config.		12V	14V
Engine output	kW	13,800	16,100
Speed	rpm	500/514	
Temperature basis			
HT cooling water engine outlet	°C	90	
LT cooling water charge air cooler inlet		38 [1]	

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No. of cylinders, config.		12V	14V
Lube oil engine inlet		55	
Cooling water fuel nozzles inlet		60	
Air data			
Temperature of charge air at charge air cooler outlet	°C	57.9	59.9
Air flow rate[2]	m^3/h	82,760	98,170
	t/h	90.7	107.5
Charge air pressure (absolute)	bar abs	4.80	4.93
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	110,213	128,580
Heat radiation engine (based on 55 °C engine room temperature)	kW	343	400
Exhaust gas data[3]			
Volume flow (temperature turbine outlet)[4]	m^3/h	146,930	171,920
Mass flow	t/h	92.9	110.3
Temperature at turbine outlet	°C	278	270
Heat content (180 °C)	kW	2,761	2,998
Permissible exhaust gas back pressure	mbar	≤ 50	

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 76: Temperature basis, nominal air and exhaust gas data – MAN V51/60DF, 1,150 kW/cyl., gas mode – Electric propulsion

2.18.7 Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode – Electric propulsion

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: ISO		
Air temperature	°C	25
Cooling water temp. before charge air cooler (LT stage)		25
Total atmospheric pressure	mbar	1,000
Relative humidity	%	30

Table 77: Reference conditions: ISO

Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,210 399	 1,105 398	 1,149 430	 753 415

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Engine output	%	100	85	75	50
Lube oil cooler[3]		373	415	423	575
Jacket cooling		342	355	381	468
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 35 °C engine room temperature)		115	115	124	156
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 240 43	 217 43	 211 43	 161 43
Air flow rate	kg/kWh	7.96	8.41	9.19	9.64
Charge air pressure (absolute)	bar abs	5.14	4.55	4.39	3.11
Exhaust gas data[4]					
Mass flow	kg/kWh	8.15	8.60	9.39	9.84
Temperature at turbine outlet	°C	297	275	269	277
Heat content (190 °C)	kJ/kWh	934	781	791	904
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 78: Load specific values at ISO conditions – MAN L/V51/60DF, 1,150 kW/cyl., liquid fuel mode – Electric propulsion

2.18.8 Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,150 kW/cyl., gas mode – Electric propulsion

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: ISO		
Air temperature	°C	25
Cooling water temp. before charge air cooler (LT stage)		25
Total atmospheric pressure	mbar	1,000
Relative humidity	%	30

Table 79: Reference conditions: ISO

Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 807 255	 670 237	 517 223	 241 212

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Engine output	%	100	85	75	50
Lube oil cooler[3]		307	323	354	501
Jacket cooling		310	347	382	495
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 35 °C engine room temperature)		115	115	125	191
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 206 50	 185 50	 163 50	 119 52.9
Air flow rate	kg/kWh	6.64	6.57	6.41	6.68
Charge air pressure (absolute)	bar abs	4.36	3.80	3.19	2.24
Exhaust gas data[4]					
Mass flow	kg/kWh	6.80	6.73	6.57	6.85
Temperature at turbine outlet	°C	316	328	354	397
Heat content (180 °C)	kJ/kWh	1,004	1,085	1,255	1,640
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 80: Load specific values at ISO conditions – MAN L/V51/60DF, 1,150 kW/cyl., gas mode – Electric propulsion

2.18.9 Load specific values at tropic conditions – MAN L/V51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode – Electric propulsion

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 81: Reference conditions: Tropics

Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,425 476	 1,329 402	 1,393 405	 996 239

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Engine output	%	100	85	75	50
Lube oil cooler[3]		405	451	459	623
Jacket cooling		400	415	446	544
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 55 °C engine room temperature)		90	90	96	121
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 273 61	 249 61	 242 61	 189 59.1
Air flow rate	kg/kWh	7.44	7.87	8.60	9.02
Charge air pressure (absolute)	bar	5.05	4.48	4.32	3.05
Exhaust gas data[4]					
Mass flow	kg/kWh	7.64	8.06	8.80	9.23
Temperature at turbine outlet	°C	338	315	308	315
Heat content (190 °C)	kJ/kWh	1,217	1,075	1,110	1,229
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 82: Load specific values at tropic conditions – MAN L/V51/60DF, 1,150 kW/cyl., liquid fuel mode – Electric propulsion

2.18.10 Load specific values at tropic conditions – MAN L/V51/60DF IMO Tier II, 1,150 kW/cyl., gas mode – Electric propulsion

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Electric propulsion

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 83: Reference conditions: Tropics

Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,169 465	 888 440	 745 429	 436 215

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Engine output	%	100	85	75	50
Lube oil cooler[3]		272	338	355	501
Jacket cooling		319	350	382	495
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 55 °C engine room temperature)		90	90	97	149
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 258 57.9	 218 53.6	 195 51.2	 146 53.0
Air flow rate	kg/kWh	6.57	6.35	6.41	6.64
Charge air pressure (absolute)	bar abs	4.80	3.82	3.23	2.24
Exhaust gas data[4]					
Mass flow	kg/kWh	6.73	6.52	6.57	6.81
Temperature at turbine outlet	°C	278	314	347	394
Heat content (180 °C)	kJ/kWh	720	952	1,201	1,603
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50		-	

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 84: Load specific values at tropic conditions – MAN L/V51/60DF, 1,150 kW/cyl., gas mode – Electric propulsion

2.19 Planning data for emission standard: IMO Tier II, 1,050 kW/cyl. – Mechanical propulsion with CPP

2.19.1 Nominal values for cooler specification – MAN L51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode/gas mode – Mechanical propulsion with CPP

Note:
If an advanced HT cooling water system for increased freshwater generation is to be applied, contact MAN Energy Solutions for corresponding planning data.

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000

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Reference conditions: Tropics		
Relative humidity	%	60

Table 85: Reference conditions: Tropics

No. of cylinders, config.		6L		7L		8L		9L	
Engine output	kW	6,300		7,350		8,400		9,450	
Speed	rpm	500/514							
Heat to be dissipated[1]		liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode
Charge air: Charge air cooler (HT stage) Charge air cooler (LT stage)	kW	 2,297 873	 1,777 815	 2,617 1,012	 2,434 937	 2,913 1,270	 2,622 1,144	 3,203 1,417	 3,050 1,322
Lube oil cooler[2]		686	489	802	568	915	649	1,032	730
Jacket cooling		670	557	785	649	895	741	1,013	834
Nozzle cooling water		14	14	17	17	19	19	21	21
Turbocharger compressor wheel cooling		26.4	26.4	26.4	26.4	26.4	26.4	37.8	37.8
Heat radiation engine (based on 55 °C engine room temperature)		157	157	183	183	209	209	235	235
Flow rates[3]									
HT circuit (jacket cooling + charge air cooler HT stage)	m^3/h	77		88		99		110	
LT circuit (lube oil cooler + charge air cooler LT stage)		85		100		110		125	
Lube oil		140		158		176		194	
Cooling water fuel nozzles		1.7		2.0		2.2		2.5	
LT cooling water turbocharger compressor wheel		2.3		2.3		2.3		3.3	
Pumps									
a) Attached									
HT CW service pump	m^3/h	77		88		99		110	
LT CW service pump		85		100		110		125	
Lube oil service pump		182		182		218		252	
b) Free-standing[4]									
HT CW stand-by pump	m^3/h	77		88		99		110	
LT CW stand-by pump		Depending on plant design							
Lube oil stand-by pump		147+z		166+z		185+z		204+z	
Nozzle CW pump		1.7		2.0		2.2		2.5	

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No. of cylinders, config.		6L	7L	8L	9L
Prelubrication pump		28.0 – 33.0 +0.5z	31.5 – 37.0 +0.5z	35.0 – 41.0 +0.5z	38.5 – 45.0 +0.5z
MGO/MDO supply pump		4.2	4.9	5.6	6.3
HFO supply pump		2.1	2.5	2.8	3.2
HFO circulation pump		4.2	4.9	5.6	6.3
Pilot fuel supply[5]	l/h	66.1	73.1	80.1	87.0

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] Addition required for separator heat (30 kJ/kWh).

[3] Basic values for layout design of the coolers.

[4] Tolerances of the pumps delivery capacities must be considered by the manufacturer.

[5] In case of an activation of the pressure limiting valve of the common rail system by the safety system, the required pilot fuel supply increases to following amount: L engine: 179 l/h, V engine: 296 l/h.
This needs to be considered for each external pilot fuel supply systems at least for 1 of the connected engines.

z = Flushing oil of automatic filter.

Table 86: Nominal values for cooler specification – MAN L51/60DF, 1,050 kW/cyl., liquid fuel mode/gas mode – Mechanical propulsion with CPP

Note:
You will find further planning data for the listed subjects in the corresponding sections.

- Minimal heating power required for preheating HT cooling water see paragraph HT cooling water preheating module (MOD-004), Page 343.
- Minimal heating power required for preheating lube oil see paragraph H-002/Lube oil preheater, Page 313.
- Capacities of preheating pumps see paragraph HT cooling water preheating module (MOD-004), Page 343.

2.19.2 Nominal values for cooler specification – MAN V51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode/gas mode – Mechanical propulsion with CPP

Note:
If an advanced HT cooling water system for increased freshwater generation is to be applied, contact MAN Energy Solutions for corresponding planning data.

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 87: Reference conditions: Tropics

No. of cylinders, config.		12V	14V	16V
Engine output	kW	12,600	14,700	16,800

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No. of cylinders, config.		12V		14V		16V	
Speed	rpm	500/514					
Heat to be dissipated[1]		liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode
Charge air: Charge air cooler (HT stage) Charge air cooler (LT stage)	kW	 4,593 1,744	 3,921 1,563	 5,234 2,023	 4,867 1,875	 5,827 2,541	 5,243 2,289
Lube oil cooler[2]		1,371	975	1,604	1,136	1,831	1,299
Jacket cooling		1,339	1,112	1,571	1,297	1,792	1,483
Nozzle cooling water		28	28	33	33	38	38
Turbocharger compressor wheel cooling		37.8	37.8	52.7	52.7	52.7	52.7
Heat radiation engine (based on 55 °C engine room temperature)		314	314	366	366	418	418
Flow rates[3]							
HT circuit (jacket cooling + charge air cooler HT stage)	m^3/h	154		176		198	
LT circuit (lube oil cooler + charge air cooler LT stage)		170		200		220	
Lube oil		340		370		400	
Cooling water fuel nozzles		3.5		4.1		4.8	
LT cooling water turbocharger compressor wheel		3.3		4.6		4.6	
Pumps							
a) Attached							
HT CW service pump	m^3/h	154		176		198	
LT CW service pump		170		200		220	
Lube oil service pump		364		408		436	
b) Free-standing[4]							
HT CW stand-by pump	m^3/h	154		176		198	
LT CW stand-by pump		Depending on plant design					
Lube oil stand-by pump		357+z		389+z		420+z	
Nozzle CW pump		3.5		4.1		4.8	
Prelubrication pump		58.0 – 68.0 +0.5z		63.0 – 74.0 +0.5z		68.0 – 80.0 +0.5z	
MGO/MDO supply pump		8.4		9.8		11.2	
HFO supply pump		4.2		4.9		5.6	
HFO circulation pump		8.4		9.8		11.2	
Pilot fuel supply[5]	l/h	108.0		122.0		136.0	

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No. of cylinders, config.		12V	14V	16V

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] Addition required for separator heat (30 kJ/kWh).

[3] Basic values for layout design of the coolers.

[4] Tolerances of the pumps delivery capacities must be considered by the manufacturer.

[5] In case of an activation of the pressure limiting valve of the common rail system by the safety system, the required pilot fuel supply increases to following amount: L engine: 179 l/h, V engine: 296 l/h.
This needs to be considered for each external pilot fuel supply systems at least for 1 of the connected engines.

z = Flushing oil of automatic filter.

Table 88: Nominal values for cooler specification – MAN V51/60DF, 1,050 kW/cyl., liquid fuel mode/gas mode – Mechanical propulsion with CPP

Note:
You will find further planning data for the listed subjects in the corresponding sections.

- Minimal heating power required for preheating HT cooling water see paragraph HT cooling water preheating module (MOD-004), Page 343.
- Minimal heating power required for preheating lube oil see paragraph H-002/Lube oil preheater, Page 313.
- Capacities of preheating pumps see paragraph HT cooling water preheating module (MOD-004), Page 343.

2.19.3 Temperature basis, nominal air and exhaust gas data – MAN L51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 89: Reference conditions: Tropics

No. of cylinders, config.		6L	7L	8L	9L
Engine output	kW	6,300	7,350	8,400	9,450
Speed	rpm	500/514			
Temperature basis					
HT cooling water engine outlet	°C	90			
LT cooling water charge air cooler inlet		38 [1]			
Lube oil engine inlet		55			
Cooling water fuel nozzles inlet		60			
Air data					
Temperature of charge air at charge air cooler outlet	°C	58.0	59.3	58.8	60.0

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No. of cylinders, config.		6L	7L	8L	9L
Air flow rate[2]	m^3/h	40,316	47,020	53,745	60,440
	t/h	44.1	51.5	58.8	66.1
Charge air pressure (absolute)	bar abs	4.93	4.94	4.93	4.94
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	50,365	58,760	67,155	75,550
Heat radiation engine (based on 55 °C engine room temperature)	kW	157	183	209	235
Exhaust gas data[3]					
Volume flow (temperature turbine outlet)[4]	m^3/h	78,571	91,748	104,814	118,024
Mass flow	t/h	45.4	52.9	60.5	68.0
Temperature at turbine outlet	°C	330	331	331	331
Heat content (190 °C)	kW	1,894	2,220	2,532	2,865
Permissible exhaust gas back pressure	mbar	≤ 50			

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 90: Temperature basis, nominal air and exhaust gas data – MAN L51/60DF, 1,050 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

2.19.4 Temperature basis, nominal air and exhaust gas data – MAN L51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 91: Reference conditions: Tropics

No. of cylinders, config.		6L	7L	8L	9L
Engine output	kW	6,300	7,350	8,400	9,450
Speed	rpm	500/514			
Temperature basis					
HT cooling water engine outlet	°C	90			
LT cooling water charge air cooler inlet		38 [1]			
Lube oil engine inlet		55			
Cooling water fuel nozzles inlet		60			

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No. of cylinders, config.		6L	7L	8L	9L
Air data					
Temperature of charge air at charge air cooler outlet	°C	52.0	57.5	56.4	58.4
Air flow rate[2]	m^3/h	31,860	40,520	45,390	52,620
	t/h	34.9	44.4	49.7	57.6
Charge air pressure (absolute)	bar abs	4.44	4.96	4.86	5.02
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	50,365	58,760	67,155	75,550
Heat radiation engine (based on 55 °C engine room temperature)	kW	157	183	209	235
Exhaust gas data[3]					
Volume flow (temperature turbine outlet)[4]	m^3/h	58,320	70,810	80,090	91,750
Mass flow	t/h	35.9	45.6	51.1	59.3
Temperature at turbine outlet	°C	293	268	273	266
Heat content (180 °C)	kW	1,239	1,251	1,445	1,534
Permissible exhaust gas back pressure	mbar	≤ 50			

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 92: Temperature basis, nominal air and exhaust gas data – MAN L51/60DF, 1,050 kW/cyl., gas mode – Mechanical propulsion with CPP

2.19.5 Temperature basis, nominal air and exhaust gas data – MAN V51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 93: Reference conditions: Tropics

No. of cylinders, config.		12V	14V	16V
Engine output	kW	12,600	14,700	16,800
Speed	rpm	500/514		
Temperature basis				
HT cooling water engine outlet	°C	90		
LT cooling water charge air cooler inlet		38 [1]		

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No. of cylinders, config.		12V	14V	16V
Lube oil engine inlet		55		
Cooling water fuel nozzles inlet		60		
Air data				
Temperature of charge air at charge air cooler outlet	°C	58.0	59.3	58.8
Air flow rate[2]	m^3/h	80,637	94,042	107,494
	t/h	88.3	102.9	117.6
Charge air pressure (absolute)	bar abs	4.94		
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	100,730	117,520	134,305
Heat radiation engine (based on 55 °C engine room temperature)	kW	314	366	418
Exhaust gas data[3]				
Volume flow (temperature turbine outlet)[4]	m^3/h	157,140	183,500	209,630
Mass flow	t/h	90.7	105.8	120.9
Temperature at turbine outlet	°C	331		
Heat content (190 °C)	kW	3,789	4,444	5,067
Permissible exhaust gas back pressure	mbar	≤ 50		

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 94: Temperature basis, nominal air and exhaust gas data – MAN V51/60DF, 1,050 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

2.19.6 Temperature basis, nominal air and exhaust gas data – MAN V51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 95: Reference conditions: Tropics

No. of cylinders, config.		12V	14V	16V
Engine output	kW	12,600	14,700	16,800
Speed	rpm	500/514		
Temperature basis				

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No. of cylinders, config.		12V	14V	16V
HT cooling water engine outlet	°C	90		
LT cooling water charge air cooler inlet		38 [1]		
Lube oil engine inlet		55		
Cooling water fuel nozzles inlet		60		
Air data				
Temperature of charge air at charge air cooler outlet	°C	55.0	57.5	56.4
Air flow rate[2]	m^3/h	66,480	81,045	90,780
	t/h	72.8	88.8	99.5
Charge air pressure (absolute)	bar abs	4.73	4.96	4.86
Air required to dissipate heat radiation (engine) ($t_2 - t_1$ = 10 °C)	m^3/h	100,730	117,520	134,305
Heat radiation engine (based on 55 °C engine room temperature)	kW	314	366	418
Exhaust gas data[3]				
Volume flow (temperature turbine outlet)[4]	m^3/h	118,680	141,780	160,330
Mass flow	t/h	74.9	91.3	102.3
Temperature at turbine outlet	°C	279	268	273
Heat content (180 °C)	kW	2,256	2,434	2,890
Permissible exhaust gas back pressure	mbar	≤ 50		

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 96: Temperature basis, nominal air and exhaust gas data – MAN V51/60DF, 1,050 kW/cyl., gas mode – Mechanical propulsion with CPP

2.19.7 Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: ISO		
Air temperature	°C	25
Cooling water temp. before charge air cooler (LT stage)		25
Total atmospheric pressure	mbar	1,000
Relative humidity	%	30

Table 97: Reference conditions: ISO

Engine output	%	100	85	75	50
	rpm	500/514			

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Engine output	%	100	85	75	50
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,115 352	 997 353	 1,043 379	 629 375
Lube oil cooler[3]		363	391	421	567
Jacket cooling		331	332	364	446
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 35 °C engine room temperature)		115	116	124	156
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 234 43.0	 212 43.0	 207 43.0	 155 43.0
Air flow rate	kg/kWh	7.48	7.80	8.47	8.75
Charge air pressure (absolute)	bar abs	5.04	4.44	4.25	2.94
Exhaust gas data[4]					
Mass flow	kg/kWh	7.67	7.99	8.66	8.95
Temperature at turbine outlet	°C	292	280	276	296
Heat content (190 °C)	kJ/kWh	831	761	789	1,010
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 98: Load specific values at ISO conditions – MAN L/V51/60DF, 1,050 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

2.19.8 Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: ISO		
Air temperature	°C	25
Cooling water temp. before charge air cooler (LT stage)		25
Total atmospheric pressure	mbar	1,000
Relative humidity	%	30

Table 99: Reference conditions: ISO

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 728 249	 584 229	 478 225	 223 225
Lube oil cooler[3]		275	350	344	488
Jacket cooling		317	350	366	484
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 35 °C engine room temperature)		115	116	125	157
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 207 43.0	 182 45.0	 164 46.0	 121 48.0
Air flow rate	kg/kWh	5.81	5.78	5.83	6.05
Charge air pressure (absolute)	bar abs	4.02	3.41	3.03	2.13
Exhaust gas data[4]					
Mass flow	kg/kWh	5.97	5.94	5.99	6.23
Temperature at turbine outlet	°C	336	361	378	432
Heat content (180 °C)	kJ/kWh	1,021	1,181	1,307	1,742
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50		-	

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 100: Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Mechanical propulsion with CPP

2.19.9 Load specific values at tropic conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 101: Reference conditions: Tropics

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,312 498	 1,200 438	 1,263 445	 847 225
Lube oil cooler[3]		392	423	455	613
Jacket cooling		383	384	421	515
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 55 °C engine room temperature)		90	90	96	122
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 266 58.0	 242 58.0	 237 58.0	 182 58.0
Air flow rate	kg/kWh	7.00	7.30	7.93	8.19
Charge air pressure (absolute)	bar abs	4.93	4.34	4.16	2.87
Exhaust gas data[4]					
Mass flow	kg/kWh	7.20	7.49	8.13	8.40
Temperature at turbine outlet	°C	330	317	313	335
Heat content (190 °C)	kJ/kWh	1,083	1,022	1,070	1,302
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 102: Load specific values at tropic conditions – MAN L/V51/60DF, 1,050 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

2.19.10 Load specific values at tropic conditions – MAN L/V51/60DF IMO Tier II, 1,050 kW/cyl., gas mode – Mechanical propulsion with CPP

1,050 kW/cyl., 500 rpm or 1,050 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 103: Reference conditions: Tropics

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,120 447	 790 411	 722 416	 402 292
Lube oil cooler[3]		279	346	350	488
Jacket cooling		318	352	382	484
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 55 °C engine room temperature)		90	90	97	122
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 269 55.0	 219 51.0	 201 49.3	 148 48.0
Air flow rate	kg/kWh	5.78	5.55	5.81	6.01
Charge air pressure (absolute)	bar abs	4.73	3.59	3.11	2.13
Exhaust gas data[4]					
Mass flow	kg/kWh	5.95	5.71	5.97	6.19
Temperature at turbine outlet	°C	279	332	363	428
Heat content (190 °C)	kJ/kWh	645	953	1,206	1,705
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 104: Load specific values at tropic conditions – MAN L/V51/60DF, 1,050 kW/cyl., gas mode – Mechanical propulsion with CPP

2.20 Planning data for emission standard: IMO Tier II, 1,150 kW/cyl. – Mechanical propulsion with CPP

2.20.1 Nominal values for cooler specification – MAN L51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode/gas mode – Mechanical propulsion with CPP

Note:
If an advanced HT cooling water system for increased freshwater generation is to be applied, contact MAN Energy Solutions for corresponding planning data.

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

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Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 105: Reference conditions: Tropics

No. of cylinders, config.		6L		7L		8L		9L	
Engine output	kW	6,900		8,050		9,200		10,350	
Speed	rpm	500/514							
Heat to be dissipated[1]		liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode	liquid fuel mode	gas mode
Charge air: Charge air cooler (HT stage) Charge air cooler (LT stage)	kW	2,731 912	2,240 892	3,096 1,119	2,662 1,042	3,434 1,418	2,916 1,300	3,760 1,588	3,294 1,471
Lube oil cooler[2]		777	521	907	588	1,036	680	1,168	744
Jacket cooling		767	612	897	719	1,024	820	1,158	927
Nozzle cooling water		16	16	18	18	21	21	23	23
Turbocharger compressor wheel cooling		26.4	26.4	26.4	26.4	37.8	37.8	37.8	37.8
Heat radiation engine (based on 55 °C engine room temperature)		172	172	200	200	229	229	257	257
Flow rates[3]									
HT circuit (jacket cooling + charge air cooler HT stage)	m^3/h	77		88		99		110	
LT circuit (lube oil cooler + charge air cooler LT stage)		85		100		110		125	
Lube oil		140		158		176		194	
Cooling water fuel nozzles		1.7		2.0		2.2		2.5	
LT cooling water turbocharger compressor wheel		2.3		2.3		3.3		3.3	
Pumps									
a) Attached									
HT CW service pump	m^3/h	77		88		99		110	
LT CW service pump		85		100		110		125	
Lube oil service pump		182		182		218		252	
b) Free-standing[4]									
HT CW stand-by pump	m^3/h	77		88		99		110	
LT CW stand-by pump		Depending on plant design							

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No. of cylinders, config.		6L	7L	8L	9L
Lube oil stand-by pump		147+z	166+z	185+z	204+z
Nozzle CW pump		1.7	2.0	2.2	2.5
Prelubrication pump		28.0 – 33.0 +0.5z	31.5 – 37.0 +0.5z	35.0 – 41.0 +0.5z	38.5 – 45.0 +0.5z
MGO/MDO supply pump		4.6	5.4	6.1	6.9
HFO supply pump		2.3	2.7	3.1	3.5
HFO circulation pump		4.6	5.4	6.1	6.9
Pilot fuel supply[5]	l/h	66.1	73.1	80.1	87.8

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] Addition required for separator heat (30 kJ/kWh).

[3] Basic values for layout design of the coolers.

[4] Tolerances of the pumps delivery capacities must be considered by the manufacturer.

[5] In case of an activation of the pressure limiting valve of the common rail system by the safety system, the required pilot fuel supply increases to following amount: L engine: 179 l/h, V engine: 296 l/h.
This needs to be considered for each external pilot fuel supply systems at least for 1 of the connected engines.

z = Flushing oil of automatic filter.

Table 106: Nominal values for cooler specification – MAN L51/60DF, 1,150 kW/cyl., liquid fuel mode/gas mode – Mechanical propulsion with CPP

Note:
You will find further planning data for the listed subjects in the corresponding sections.

- Minimal heating power required for preheating HT cooling water see paragraph HT cooling water preheating module (MOD-004), Page 343.
- Minimal heating power required for preheating lube oil see paragraph H-002/Lube oil preheater, Page 313.
- Capacities of preheating pumps see paragraph HT cooling water preheating module (MOD-004), Page 343.

2.20.2 Nominal values for cooler specification – MAN V51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode/gas mode – Mechanical propulsion with CPP

Note:
If an advanced HT cooling water system for increased freshwater generation is to be applied, contact MAN Energy Solutions for corresponding planning data.

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 107: Reference conditions: Tropics

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No. of cylinders, config.		**12V**		**14V**	
Engine output	kW	13,800		16,100	
Speed	rpm	500/514			
Heat to be dissipated[1]		liquid fuel mode	gas mode	liquid fuel mode	gas mode
Charge air: Charge air cooler (HT stage) Charge air cooler (LT stage)	kW	 5,462 1,825	 4,480 1,784	 6,192 2,237	 5,323 2,084
Lube oil cooler[2]		1,553	1,041	1,814	1,176
Jacket cooling		1,534	1,224	1,795	1,438
Nozzle cooling water		31	31	36	36
Turbocharger compressor wheel cooling		52.7	52.7	52.7	52.7
Heat radiation engine (based on 55 °C engine room temperature)		343	343	400	400
Flow rates[3]					
HT circuit (jacket cooling + charge air cooler HT stage)	m^3/h	154		176	
LT circuit (lube oil cooler + charge air cooler LT stage)		170		200	
Lube oil		340		370	
Cooling water fuel nozzles		3.5		4.1	
LT cooling water turbocharger compressor wheel		4.6		4.6	
Pumps					
a) Attached					
HT CW service pump	m^3/h	154		176	
LT CW service pump		170		200	
Lube oil service pump		364		408	
b) Free-standing[4]					
HT CW stand-by pump	m^3/h	154		176	
LT CW stand-by pump		Depending on plant design			
Lube oil stand-by pump		357+z		389+z	
Nozzle CW pump		3.5		4.1	
Prelubrication pump		58.0 – 68.0 +0.5z		63.0 – 74.0 +0.5z	
MGO/MDO supply pump		9.2		10.7	
HFO supply pump		4.6		5.4	
HFO circulation pump		9.2		10.7	
Pilot fuel supply[5]	l/h	108.0		122.0	

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No. of cylinders, config.		12V	14V
[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.			
[2] Addition required for separator heat (30 kJ/kWh).			
[3] Basic values for layout design of the coolers.			
[4] Tolerances of the pumps delivery capacities must be considered by the manufacturer.			
[5] In case of an activation of the pressure limiting valve of the common rail system by the safety system, the required pilot fuel supply increases to following amount: L engine: 179 l/h, V engine: 296 l/h. This needs to be considered for each external pilot fuel supply systems at least for 1 of the connected engines.			
z = Flushing oil of automatic filter.			

Table 108: Nominal values for cooler specification – MAN V51/60DF, 1,150 kW/cyl., liquid fuel mode/gas mode – Mechanical propulsion with CPP

Note:
You will find further planning data for the listed subjects in the corresponding sections.

- Minimal heating power required for preheating HT cooling water see paragraph HT cooling water preheating module (MOD-004), Page 343.
- Minimal heating power required for preheating lube oil see paragraph H-002/Lube oil preheater, Page 313.
- Capacities of preheating pumps see paragraph HT cooling water preheating module (MOD-004), Page 343.

2.20.3 Temperature basis, nominal air and exhaust gas data – MAN L51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 109: Reference conditions: Tropics

No. of cylinders, config.		6L	7L	8L	9L
Engine output	kW	6,900	8,050	9,200	10,350
Speed	rpm	500/514			
Temperature basis					
HT cooling water engine outlet	°C	90			
LT cooling water charge air cooler inlet		38 [1]			
Lube oil engine inlet		55			
Cooling water fuel nozzles inlet		60			
Air data					
Temperature of charge air at charge air cooler outlet	°C	61.0	61.6	61.2	62.5

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No. of cylinders, config.		6L	7L	8L	9L
Air flow rate[2]	m^3/h	46,924	54,734	62,562	70,355
	t/h	51.4	59.9	68.5	77.0
Charge air pressure (absolute)	bar abs	5.05	5.06	5.06	5.07
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	55,107	64,291	73,476	82,660
Heat radiation engine (based on 55 °C engine room temperature)	kW	172	200	229	257
Exhaust gas data[3]					
Volume flow (temperature turbine outlet)[4]	m^3/h	92,620	108,110	123,510	139,075
Mass flow	t/h	52.7	61.5	70.3	79.1
Temperature at turbine outlet	°C	338	339	338	339
Heat content (190 °C)	kW	2,332	2,728	3,112	3,518
Permissible exhaust gas back pressure	mbar	≤ 50			

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 110: Temperature basis, nominal air and exhaust gas data – MAN L51/60DF, 1,150 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

2.20.4 Temperature basis, nominal air and exhaust gas data – MAN L51/60DF IMO Tier II, 1,150 kW/cyl., gas mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 111: Reference conditions: Tropics

No. of cylinders, config.		6L	7L	8L	9L
Engine output	kW	6,900	8,050	9,200	10,350
Speed	rpm	500/514			
Temperature basis					
HT cooling water engine outlet	°C	90			
LT cooling water charge air cooler inlet		38 [1]			
Lube oil engine inlet		55			
Cooling water fuel nozzles inlet		60			

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No. of cylinders, config.		6L	7L	8L	9L
Air data					
Temperature of charge air at charge air cooler outlet	°C	59.2	59.9	59.2	60.8
Air flow rate[2]	m^3/h	41,380	49,085	55,760	63,015
	t/h	45.3	53.8	61.1	69.0
Charge air pressure (absolute)	bar abs	4.80	4.93	4.89	4.99
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	55,107	64,291	73,476	82,660
Heat radiation engine (based on 55 °C engine room temperature)	kW	172	200	229	257
Exhaust gas data[3]					
Volume flow (temperature turbine outlet)[4]	m^3/h	74,880	86,670	99,010	109,335
Mass flow	t/h	46.5	55.1	62.6	70.8
Temperature at turbine outlet	°C	288	275	278	271
Heat content (180 °C)	kW	1,522	1,583	1,855	1,953
Permissible exhaust gas back pressure	mbar	≤ 50			

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 112: Temperature basis, nominal air and exhaust gas data – MAN L51/60DF, 1,150 kW/cyl., gas mode – Mechanical propulsion with CPP

2.20.5 Temperature basis, nominal air and exhaust gas data – MAN V51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 113: Reference conditions: Tropics

No. of cylinders, config.		12V	14V
Engine output	kW	13,800	16,100
Speed	rpm	500/514	
Temperature basis			
HT cooling water engine outlet	°C	90	
LT cooling water charge air cooler inlet		38 [1]	

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No. of cylinders, config.		12V	14V
Lube oil engine inlet		55	
Cooling water fuel nozzles inlet		60	
Air data			
Temperature of charge air at charge air cooler outlet	°C	61.0	61.6
Air flow rate[2]	m^3/h	93,849	109,468
	t/h	102.7	119.8
Charge air pressure (absolute)	bar abs	5.05	5.06
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	110,213	128,580
Heat radiation engine (based on 55 °C engine room temperature)	kW	343	400
Exhaust gas data[3]			
Volume flow (temperature turbine outlet)[4]	m^3/h	185,240	216,210
Mass flow	t/h	105.5	123.1
Temperature at turbine outlet	°C	338	339
Heat content (190 °C)	kW	4,665	5,456
Permissible exhaust gas back pressure	mbar	≤ 50	

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 114: Temperature basis, nominal air and exhaust gas data – MAN V51/60DF, 1,150 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

2.20.6 Temperature basis, nominal air and exhaust gas data – MAN V51/60DF IMO Tier II, 1,150 kW/cyl., gas mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 115: Reference conditions: Tropics

No. of cylinders, config.		12V	14V
Engine output	kW	13,800	16,100
Speed	rpm	500/514	
Temperature basis			

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No. of cylinders, config.		12V	14V
HT cooling water engine outlet	°C	90	
LT cooling water charge air cooler inlet		38 [1]	
Lube oil engine inlet		55	
Cooling water fuel nozzles inlet		60	
Air data			
Temperature of charge air at charge air cooler outlet	°C	57.9	59.9
Air flow rate[2]	m^3/h	82,760	98,170
	t/h	90.7	107.5
Charge air pressure (absolute)	bar abs	4.80	4.93
Air required to dissipate heat radiation (engine) (t_2 – t_1 = 10 °C)	m^3/h	110,213	128,580
Heat radiation engine (based on 55 °C engine room temperature)	kW	343	400
Exhaust gas data[3]			
Volume flow (temperature turbine outlet)[4]	m^3/h	146,930	171,920
Mass flow	t/h	92.9	110.3
Temperature at turbine outlet	°C	278	270
Heat content (180 °C)	kW	2,761	2,998
Permissible exhaust gas back pressure	mbar	≤ 50	

[1] For design see figures Cooling water system diagrams, Page 347.

[2] Under mentioned above reference conditions.

[3] Tolerance: Quantity ±5 %, temperature ±20 °C.

[4] Calculated based on stated temperature at turbine outlet and total atmospheric pressure according mentioned above reference conditions.

Table 116: Temperature basis, nominal air and exhaust gas data – MAN V51/60DF, 1,150 kW/cyl., gas mode – Mechanical propulsion with CPP

2.20.7 Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: ISO		
Air temperature	°C	25
Cooling water temp. before charge air cooler (LT stage)		25
Total atmospheric pressure	mbar	1,000
Relative humidity	%	30

Table 117: Reference conditions: ISO

Engine output	%	100	85	75	50
	rpm	500/514			

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Engine output	%	100	85	75	50
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,210 399	 1,105 398	 1,149 430	 753 415
Lube oil cooler[3]		373	415	423	575
Jacket cooling		342	355	381	468
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 35 °C engine room temperature)		115	115	124	156
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 240 43.0	 217 43.0	 211 43.0	 161 43.0
Air flow rate	kg/kWh	7.96	8.41	9.19	9.64
Charge air pressure (absolute)	bar abs	5.14	4.55	4.39	3.11
Exhaust gas data[4]					
Mass flow	kg/kWh	8.15	8.60	9.39	9.84
Temperature at turbine outlet	°C	297	275	269	277
Heat content (190 °C)	kJ/kWh	934	781	791	904
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 118: Load specific values at ISO conditions – MAN L/V51/60DF, 1,150 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

2.20.8 Load specific values at ISO conditions – MAN L/V51/60DF IMO Tier II, 1,150 kW/cyl., gas mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: ISO		
Air temperature	°C	25
Cooling water temp. before charge air cooler (LT stage)		25
Total atmospheric pressure	mbar	1,000
Relative humidity	%	30

Table 119: Reference conditions: ISO

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 807 255	 607 237	 517 223	 241 212
Lube oil cooler[3]		307	323	354	501
Jacket cooling		310	347	382	495
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 35 °C engine room temperature)		115	115	125	191
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 206 50.0	 185 50.0	 163 50.0	 119 52.9
Air flow rate	kg/kWh	6.64	6.57	6.41	6.68
Charge air pressure (absolute)	bar abs	4.36	3.80	3.19	2.24
Exhaust gas data[4]					
Mass flow	kg/kWh	6.80	6.73	6.57	6.85
Temperature at turbine outlet	°C	316	328	354	397
Heat content (180 °C)	kJ/kWh	1,004	1,085	1,255	1,640
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 120: Load specific values at ISO conditions – MAN L/V51/60DF, 1,150 kW/cyl., gas mode – Mechanical propulsion with CPP

2.20.9 Load specific values at tropic conditions – MAN L/V51/60DF IMO Tier II, 1,150 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 121: Reference conditions: Tropics

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,425 476	 1,329 402	 1,393 405	 996 239
Lube oil cooler[3]		405	451	459	623
Jacket cooling		400	415	446	544
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 55 °C engine room temperature)		90	90	96	121
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 273 61.0	 249 61.0	 242 61.0	 189 59.1
Air flow rate	kg/kWh	7.44	7.87	8.60	9.02
Charge air pressure (absolute)	bar abs	5.05	4.48	4.32	3.05
Exhaust gas data[4]					
Mass flow	kg/kWh	7.64	8.06	8.80	9.23
Temperature at turbine outlet	°C	338	315	308	315
Heat content (190 °C)	kJ/kWh	1,217	1,075	1,110	1,229
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, –15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 122: Load specific values at tropic conditions – MAN L/V51/60DF, 1,150 kW/cyl., liquid fuel mode – Mechanical propulsion with CPP

2.20.10 Load specific values at tropic conditions – MAN L/V51/60DF IMO Tier II, 1,150 kW/cyl., gas mode – Mechanical propulsion with CPP

1,150 kW/cyl., 500 rpm or 1,150 kW/cyl., 514 rpm – Mechanical propulsion with CPP

Reference conditions: Tropics		
Air temperature	°C	45
Cooling water temp. before charge air cooler (LT stage)		38
Total atmospheric pressure	mbar	1,000
Relative humidity	%	60

Table 123: Reference conditions: Tropics

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Engine output	%	100	85	75	50
	rpm	500/514			
Heat to be dissipated[1]					
Charge air: Charge air cooler (HT stage)[2] Charge air cooler (LT stage)[2]	kJ/kWh	 1,169 465	 888 440	 745 429	 436 215
Lube oil cooler[3]		272	338	355	501
Jacket cooling		319	350	382	495
Water for fuel valves		8	8	8	8
Heat radiation engine (based on 55 °C engine room temperature)		90	90	97	149
Air data					
Temperature of charge air: After compressor At charge air cooler outlet	°C	 258 57.9	 218 53.6	 195 51.2	 146 53.0
Air flow rate	kg/kWh	6.57	6.35	6.41	6.64
Charge air pressure (absolute)	bar abs	4.80	3.82	3.23	2.24
Exhaust gas data[4]					
Mass flow	kg/kWh	6.73	6.52	6.57	6.81
Temperature at turbine outlet	°C	278	314	347	394
Heat content (180 °C)	kJ/kWh	720	952	1,201	1,603
Permissible exhaust gas back pressure after turbocharger (maximum)	mbar	50	-		

[1] Tolerance: +10 % for rating coolers, −15 % for heat recovery.

[2] The values of the particular cylinder numbers can differ depending on the charge air cooler specification.

[3] Addition required for separator heat (30 kJ/kWh).

[4] Tolerance: Quantity ±5 %, temperature ±20 °C.

Table 124: Load specific values at tropic conditions – MAN L/V51/60DF, 1,150 kW/cyl., gas mode – Mechanical propulsion with CPP

2.21 Operating/service temperatures and pressures

Intake air (conditions before compressor of turbocharger)

	Min.	Max.
Intake air temperature compressor inlet	5 °C[1]	45 °C[2]
Intake air pressure compressor inlet	−20 mbar	-

[1] Conditions below this temperature are defined as "arctic conditions" – see section Engine operation under arctic conditions, Page 70.

[2] In accordance with power definition. A reduction in power is required at higher temperatures/lower pressures.

Table 125: Intake air (conditions before compressor of turbocharger)

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Charge air (conditions within charge air pipe before cylinder)

	Min.	Max.
Charge air temperature cylinder inlet[1)]	34 °C	55 °C
[1)] Aim for a higher value in conditions of high air humidity (to reduce condensate amount).		

Table 126: Charge air (conditions within charge air pipe before cylinder)

HT cooling water – Engine

	Min.	Max.
HT cooling water temperature at jacket cooling outlet[1)]	90 °C nominal[2)]	95 °C[3)]
HT cooling water temperature engine inlet – Preheated before start	60 °C	90 °C
HT cooling water pressure engine inlet; nominal value 4 bar[4)]	3 bar	6 bar
Pressure loss engine (total, for nominal flow rate)	-	1.3 bar
Only for information: + Pressure loss engine (without charge air cooler) + Pressure loss HT piping engine + Pressure loss charge air cooler (HT stage)	 0.3 bar 0.2 bar 0.2 bar	 0.5 bar 0.4 bar 0.4 bar
Pressure rise attached HT cooling water pump (optional)	3.2 bar	3.8 bar
[1)] SaCoSone measuring point is jacket cooling outlet of the engine. [2)] Regulated temperature. [3)] Operation at alarm level. [4)] SaCoSone measuring point is jacket cooling inlet of the engine.		

Table 127: HT cooling water – Engine

HT cooling water – Plant

	Min.	Max.
Permitted pressure loss of external HT system (plant)	-	1.9 bar
Minimum required pressure rise of free-standing HT cooling water stand-by pump (plant)	3.2 bar	-
Cooling water expansion tank + Pre-pressure due to expansion tank at suction side of cooling water pump + Pressure loss from expansion tank to suction side of cooling water pump	 0.6 bar -	 0.9 bar 0.1 bar

Table 128: HT cooling water – Plant

LT cooling water – Engine

	Min.	Max.
LT cooling water temperature charge air cooler inlet (LT stage)[1)]	32 °C	38 °C[2)]
LT cooling water pressure charge air cooler inlet (LT stage); nominal value 4 bar	2 bar	6 bar
Pressure loss charge air cooler (LT stage, for nominal flow rate)	-	0.8 bar
Only for information: + Pressure loss LT piping engine + Pressure loss charge air cooler (LT stage)	 - -	 0.3 bar 0.5 bar

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	Min.	Max.
Pressure rise attached LT cooling water pump (optional)	3.2 bar	3.8 bar

[1] Regulated temperature.

[2] In accordance with power definition. A reduction in power is required at higher temperatures/lower pressures.

Table 129: LT cooling water – Engine

LT cooling water – Plant

	Connection number internal media schemata "Cooling water system"	Min.	Max.
Permitted pressure loss of external LT system (plant)	-	-	2.4 bar
Minimum required pressure rise of free-standing LT cooling water stand-by pump (plant)	-	3.2 bar	-
Cooling water expansion tank: Pre-pressure due to expansion tank at suction side of cooling water pump Pressure loss from expansion tank to suction side of cooling water pump	- -	0.6 bar -	0.9 bar 0.1 bar

Table 130: LT cooling water – Plant

Nozzle cooling water

	Min.	Max.
Nozzle cooling water temperature engine inlet	55 °C	70 °C[1]
Nozzle cooling water pressure engine inlet + Open system + Closed system	 2 bar 3 bar	 3 bar 5 bar
Pressure loss engine (fuel nozzles, for nominal flow rate)	-	1.5 bar

[1] Operation at alarm level.

Table 131: Nozzle cooling water

Lube oil

	Min.	Max.
Lube oil temperature engine inlet	50 °C[1]	60 °C[2]
Lube oil temperature engine inlet – Preheated before start	40 °C	50 °C[3]
Lube oil pressure (during engine operation) – L engine inlet – V engine inlet – Turbocharger inlet	 4 bar 5 bar 1.3 bar	 5 bar 5.5 bar 2.2 bar
Prelubrication/postlubrication (duration ≤ 10 min) lube oil pressure – L engine inlet – V engine inlet – Turbocharger inlet	 0.3 bar[4] 0.3 bar[4] 0.2 bar	 5 bar 5.5 bar 2.2 bar

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	Min.	Max.
Prelubrication/postlubrication (duration > 10 min) lube oil pressure – Engine inlet – Turbocharger inlet	 0.3 bar[4] 0.2 bar	 0.6 bar 0.6 bar
Lube oil pump (attached, free-standing) – Design pressure – Opening pressure safety valve	 7 bar -	 - 8 bar

[1] Regulated temperature.

[2] Operation at alarm level.

[3] If higher temperatures of lube oil in system will be reached, e.g. due to separator operation, at engine start this temperature needs to be reduced as quickly as possible below alarm level to avoid a start failure.

[4] Note: Oil pressure > 0.3 bar must be ensured also for lube oil temperatures up to 70 °C.

Table 132: Lube oil

Fuel – Main fuel

	Min.	Max.
Fuel temperature engine inlet – MGO (DMA, DFA) and MDO (DMB, DFB) according ISO 8217 – HFO according ISO 8217	 –10 °C[1] -	 45 °C[2] 150 °C[2]
Fuel viscosity engine inlet – MGO (DMA, DFA) and MDO (DMB, DFB) according ISO 8217 – HFO according ISO 8217, recommended viscosity	 1.9 cSt 12.0 cSt	 14.0 cSt 14.0 cSt
Fuel pressure engine inlet	5.0 bar	8.0 bar
Fuel pressure engine inlet in case of black out (only engine start idling)	0.6 bar	-
Differential pressure (engine inlet/engine outlet)	1.0 bar	-
Fuel return, fuel pressure engine outlet	2.0 bar	-
Maximum pressure variation at engine inlet	-	±0.5 bar
HFO supply system + Minimum required pressure rise of free-standing HFO supply pump (plant) + Minimum required pressure rise of free-standing HFO circulating pump (booster pumps, plant) + Minimum required absolute design pressure free-standing HFO circulating pump (booster pumps, plant)	 7.0 bar 7.0 bar 10.0 bar	 - - -
MDO/MGO supply system + Minimum required pressure rise of free-standing MDO/MGO supply pump (plant)	7.0 bar	-
Fuel temperature within HFO day tank (preheating)	75 °C	90 °C[3]

[1] Maximum viscosity not to be exceeded. “Pour point” and “Cold filter plugging point” have to be observed.

[2] Not permissible to fall below minimum viscosity.

[3] If flash point is below 100 °C, than the limit is: 10 degree distance to the flash point.

Table 133: Fuel – Main fuel

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Fuel – Pilot fuel

	Min.	Max.
Fuel temperature engine inlet – MGO (DMA, DFA) and MDO (DMB, DFB) according ISO 8217	–10 °C[1]	45 °C[2]
Fuel viscosity engine inlet – MGO (DMA, DFA) and MDO (DMB, DFB) according ISO 8217	1.9 cSt	11.0 cSt
Pilot fuel pressure engine inlet	5.0 bar	9.0 bar
Pilot fuel return, fuel pressure engine outlet	0.0 bar	0.2 bar

[1] Maximum viscosity not to be exceeded. “Pour point” and “Cold filter plugging point” have to be observed.

[2] Not permissible to fall below minimum viscosity.

Table 134: Fuel – Pilot fuel

Gas

See section Specifications and requirements for the gas supply of the engine, Page 154.

Nitrogen for purging of fuel gas line

	Min.	Max.
Nitrogen temperature	5 °C	50 °C
Nitrogen pressure	5.5 bar	6.5 bar

Note:
The duration of the purging process will be adjusted during commissioning according to the actual pipe length of the fuel gas line.

Table 135: Nitrogen for purging of fuel gas line

Compressed air in the starting air system

	Min.	Max.
Starting air pressure	15.0 bar[1]	30.0 bar[2] [3]
Jet assist pressure engine inlet	18.0 bar[4]	30.0 bar

[1] Operation at alarm level.

[2] Nominal value.

[3] For layout of the starting air system, note:

- Starting air consumption and start duration stated within section Starting air and control air consumption, Page 96.
- Maximum velocity (m/s) see section External pipe dimensioning, Page 298 - consider multiple start of engines, if needed.

[4] Below this value jet assist will be deactivated.

Table 136: Compressed air in the starting air system

Compressed air in the control air system

	Min.	Max.
Control air pressure engine inlet	5.5 bar[1]	8.0 bar

	Min.	Max.
[1] Operation at alarm level.		

Table 137: Compressed air in the control air system

Crankcase pressure (engine)

	Min.	Max.
Pressure within crankcase	–2.5 mbar	3.0 mbar

Table 138: Crankcase pressure (engine)

	Setting
Safety valve attached to the crankcase (opening pressure)	50 – 70 mbar

Table 139: Safety valve

Exhaust gas

	Min.	Max.
Exhaust gas temperature turbine outlet (normal operation under tropic conditions)	-	440 °C
Exhaust gas temperature turbine outlet (with SCR within regeneration mode)	360 °C	400 °C
Exhaust gas temperature turbine outlet (emergency operation – According classification rules – One failure of TC)	-	485 °C
Minimum exhaust gas temperature after recooling due to exhaust gas heat utilisation	190 °C[1]	-
Recommended design exhaust gas temperature turbine outlet for layout of exhaust gas line (plant)	450 °C[2]	-
Exhaust gas back pressure after turbocharger (static)	-	50 mbar[3]

[1] To avoid sulfur corrosion in exhaust gas line (plant)

[2] Project specific evaluation required, figure given as minimum value for guidance only.

[3] If this value is exceeded by the total exhaust gas back pressure of the designed exhaust gas line, sections Derating, definition of P Operating, Page 44 and Increased exhaust gas pressure due to exhaust gas after treatment installations, Page 48 need to be considered.

Table 140: Exhaust gas

Note:
Operating pressures without further specification are below/above atmospheric pressure.

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2.22 Leakage rate

	Max. leak rate (clean fuel)		Burst leak rate in case of pipe break (for max. 1 min)
	l/cyl. x h		l/min
	HFO	DO	HFO/DO
Main fuel (conventional)	0.11	0.6	4.9

Table 141: Leakage rate – MAN 51/60DF main injection system with SP injection pumps

	Max. leak rate injector (clean fuel)	Max. leak rate system (clean fuel)	Max. leak rate total (clean fuel)	Burst leak rate in case of pipe break (for max. 1 min)
	l/h/cyl.	l/h/engine	l/h	l/min
Pilot fuel (CR injection) DO	1.3	53.9	1.3 x no. cyl. + 53.9	L engine: 1.6 V engine: 3.8

Table 142: Leakage rate – MAN 51/60DF pilot fuel system

Note:

- A high flow of dirty leakage oil will occur in case of a pipe break, for short time only (< 1 min).
  Engine will run down immediately after a pipe break alarm.
  This leakage can be reused, if the entire fuel treatment of separation and filtration is done.
- The operating leakage (clean) of the pilot fuel system (CR injection) includes the leakage amount of the high-pressure pumps, injection valves and valve groups, which occur during normal operation due to their function. This leakage can be reused, if the entire fuel treatment of separation and filtration is done.
- The operating leakage (clean) of the main fuel system with sealed plunger pumps (conventional injection) includes the leakage amount of the injection pumps and the injection valves, which occur during normal operation due to their function. This leakage can be reused, if the entire fuel treatment of separation and filtration is done.
- All other leakage amounts (dirt fuel oil from filters or from engine drains) have to be discharged into the sludge tank.

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2.23 Filling volumes

Cooling water and oil volume of engine[1]								
No. of cylinders		**6**	**7**	**8**	**9**	**12**	**14**	**16**
Cooling water approximately	litres	470	540	615	685	1,250	1,400	1,550
Lube oil		170	190	220	240	325	380	435

[1] Be aware: This is just the amount inside the engine. By this amount the level in the service or expansion tank will be lowered when media systems are put in operation.

Table 143: Cooling water and oil volume of engine

Service tanks	**Installation height[1]**	**Minimum effective capacity**						
	m	**m³**						
No. of cylinders		**6**	**7**	**8**	**9**	**12**	**14**	**16**
Cooling water cylinder	6 – 9	1.0				1.5		
Required diameter for expansion pipeline	-	≥ DN50 [2]						
Lube oil in lube oil service tank (1,050 kW/cyl./ 1,150 kW/cyl. engine variant)	-	6.3/ 6.9	7.4/ 8.1	8.4/ 9.2	9.5/ 10.4	12.6/ 13.8	14.7/ 16.1	16.8/ -

[1] Installation height refers to tank bottom and crankshaft centre line.

[2] Pipe diameter to be increased for multi engine plants.

Table 144: Service tanks capacities

2.24 Specifications and requirements for the gas supply of the engine

General items regarding the GVU, see also section Fuel gas system, Page 404.

For perfect dynamic engine performance, the following has to be ensured:

Natural gas		
Permitted temperature range	°C	+5 °C[1] up to 50 °C before GVU and +0 °C[1] up to 50 °C before engine
Calorific value (LHV)	kJ/Nm³	≥ 28,000
Methane number[2] (for nominal engine output)	-	≥ 80
Permissible rate of change of the methane number	-	Max. Δ of 5 MN/min
Gas supply at inlet engine		
Minimum gas pressure at inlet engine	bar	See figure(s) below.
Maximum allowable fluctuation at inlet engine	bar/s	≤ ±0.2
Maximum gas pressure at inlet engine **(SAFETY ISSUE!)**	bar	**6.5**
Gas supply at inlet GVU		

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Maximum admissible supply gas pressure at inlet GVU	bar	9.0
Minimum supply gas pressure at inlet GVU (recommended)	bar	6.4 [3]
Minimum supply gas pressure at inlet GVU with pre-filter at engine (recommended)	bar	6.9 [3] [4]
Maximum length of the piping between GVU and engine inlet	m	20

[1] The temperature- and pressure-dependent dew point of natural gas must always be exceeded to prevent condensation.

[2] Methane number (MN) calculated according to EN 16726 using GasCalc at least version 2.5.2.

[3] Considering: LHV 28.0 MJ/Nm³, pressure losses and reserve for governing purposes.

[4] Pre-filter before engine is required if gas line between GVU and engine is not made of stainless steel (contrary to the requirements in section Specification of materials for piping, Page 299).

Table 145: Specifications and requirements for the gas supply of the engine

Note:
Operating pressures without further specification are below/above atmospheric pressure.
Nm³ corresponds to one cubic metre of gas at 0 °C and 101.32 kPa abs.

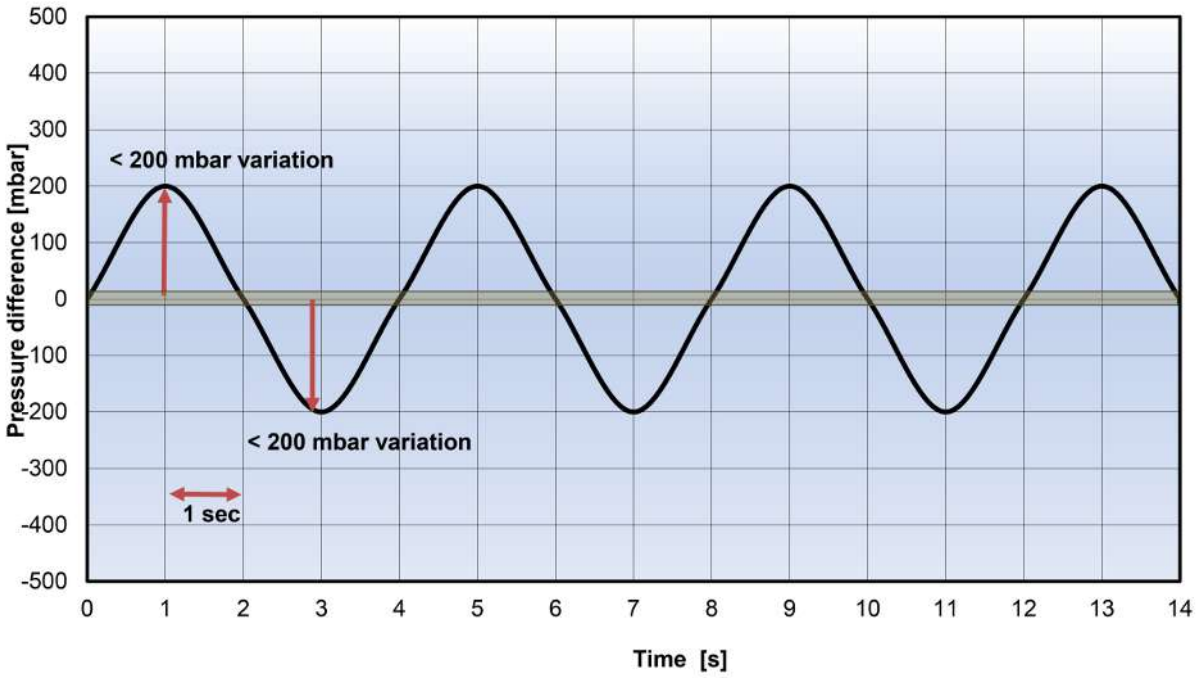

Figure 44: Maximum allowable oscilation of inlet pressure at GVU

As the required supply gas pressure is not only dependent on engine related conditions like the charge air pressure and accordingly required gas pressure at the gas valves, but is also influenced by the difference pressure of the gas valve unit, the piping of the plant and the caloric value of the fuel gas, a project-specific layout is required. Therefore details must be clarified with MAN Energy Solutions in an early project stage.

Additional note:
To clarify the relevance of the dependencies, the following figure illustrates that the lower the caloric value of the fuel gas is, the higher the gas pressure must be in order to achieve the same engine performance.

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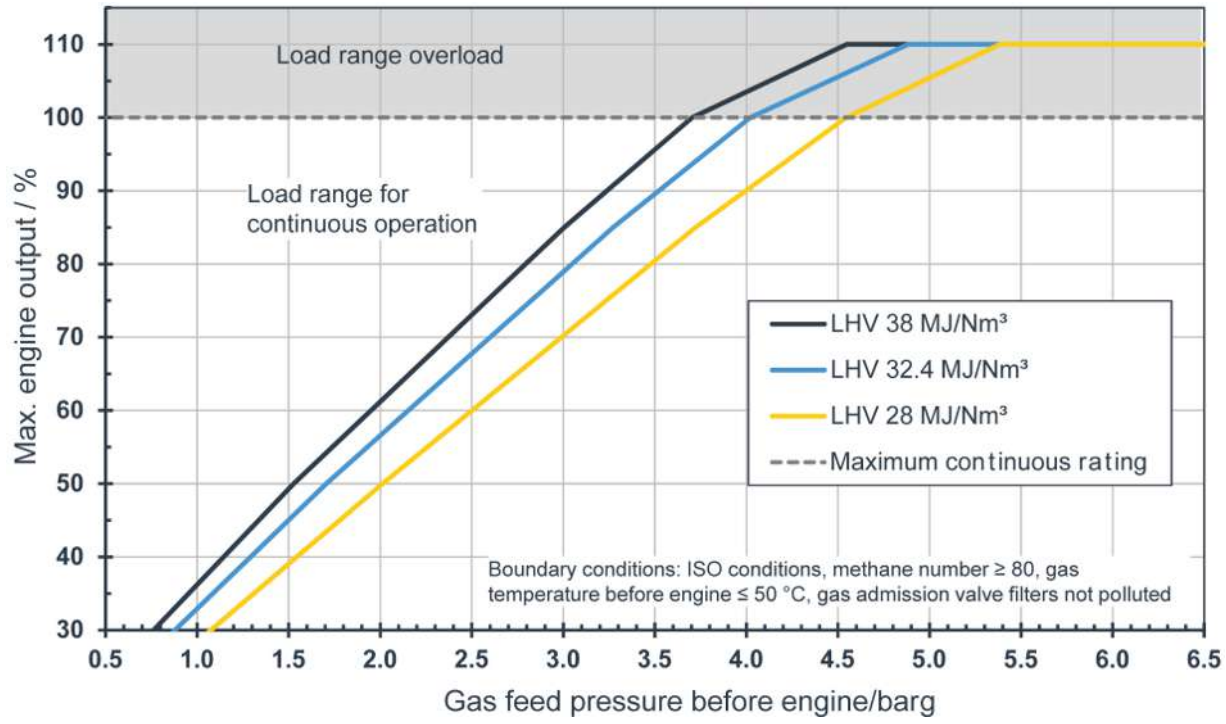

Figure 45: Gas feed pressure before engine inlet dependent on LHV – MAN 51/60DF, 1,050 kW/cyl.

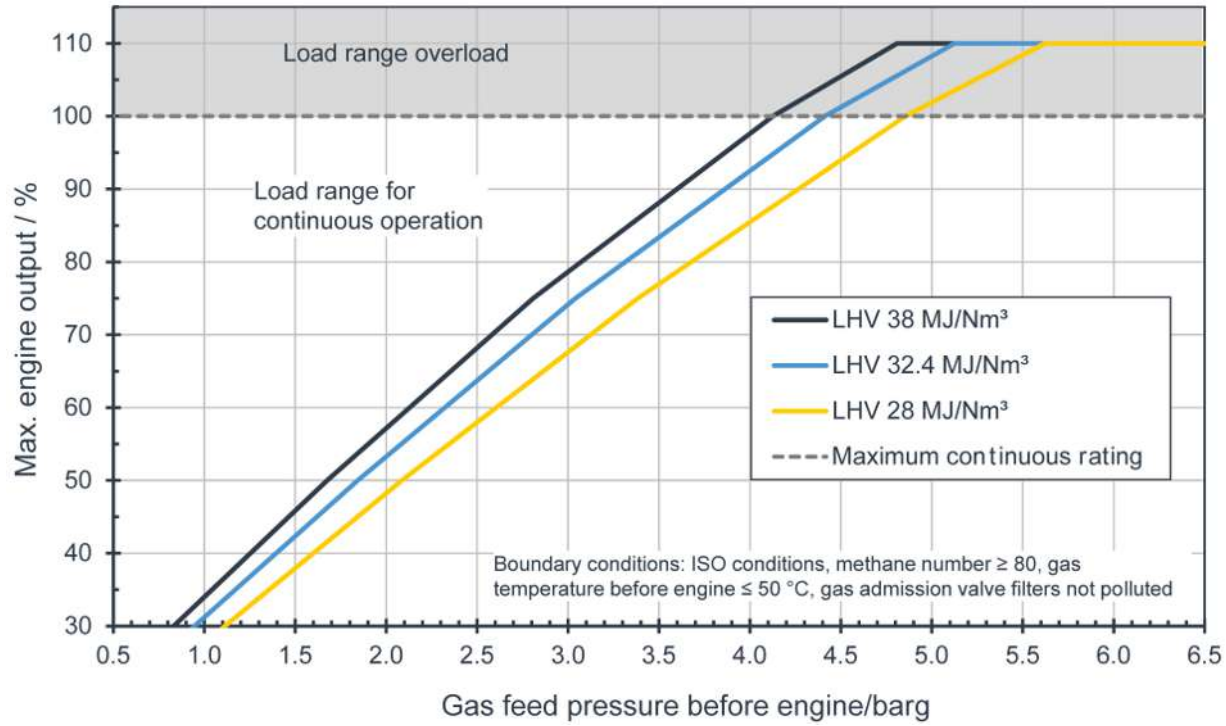

Figure 46: Gas feed pressure before engine inlet dependent on LHV – MAN 51/60DF, 1,150 kW/cyl.

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Load range overload

According to DIN ISO 8528-1 load > 100 % of the rated output is permissible only for a short time to provide additional engine power for governing purposes only (e.g. transient load conditions and suddenly applied load). This additional power shall not be used for the supply of electrical consumers.

Note:
1 GVU is required per engine.

2.25 Venting amount of crankcase and turbocharger

A ventilation of the engine crankcase and the turbochargers is required, as described in section Crankcase vent and lube oil tank vent, Page 333.

For the layout of the ventilation system guidance is provided below:

Due to normal blow-by of the piston ring package small amounts of combustion chamber gases get into the crankcase and carry along oil dust.

- The amount of crankcase vent gases is approximately 0.1 % of the engine´s air flow rate.
- The temperature of the crankcase vent gases is approximately 5 K higher than the oil temperature at the engine´s oil inlet.
- The density of crankcase vent gases is 1.0 kg/m³ (assumption for calculation).

In addition, the sealing air of the turbocharger needs to be vented.

- The amount of turbocharger sealing air is approximately:
  - For single-stage turbocharged engines 0.2 % of the engine´s air flow rate.
  - For two-stage turbocharged engines 0.4 % of the engine´s air flow rate.
- The temperature of turbocharger sealing air is approximately 5 K higher than the oil temperature at the engine´s oil inlet.
- The density of turbocharger sealing air is 1.0 kg/m³ (assumption for calculation).

2.26 Exhaust gas emission

2.26.1 Maximum permissible NOx emission limit value IMO Tier II and IMO Tier III

IMO Tier III: Engine in standard version[1]

Rated speed	500 rpm	514 rpm
NO_x [1) 2)]		
IMO Tier II cycle D2/E2	10.54 g/kWh[3)]	10.47 g/kWh[3)]
IMO Tier III cycle D2/E2	2.60 g/kWh[3)]	2.58 g/kWh[3)]

Note:
The engine´s certification for compliance with the NO_x limits will be carried out during factory acceptance test as a single or a group certification.

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Rated speed	500 rpm	514 rpm
[1] Cycle values, operation on DM grade fuel (marine distillate fuel: MGO or MDO) according ISO 8217, based on a LT charge air cooling water temperature of max. 32 °C at 25 °C reference seawater temperature. [2] Calculated as NO_2. D2: Test cycle for constant speed aux. engine application. E2: Test cycle for "constant speed main propulsion application" (including electric propulsion and all controllable pitch propeller installations). [3] Maximum permissible NO_x emissions for marine diesel engines according to IMO Tier II: $130 \le n \le 2000 \rightarrow 44 * n^{-0.23}$ g/kWh (n = rated engine speed in rpm) IMO Tier III: $130 \le n \le 2000 \rightarrow 9 * n^{-0.2}$ g/kWh (n = rated engine speed in rpm).		

Table 146: Maximum permissible NO_x emission limit value

[1] Marine engines are guaranteed to meet the revised International Convention for the Prevention of Pollution from Ships, "Revised MARPOL Annex VI (Regulations for the Prevention of Air Pollution from Ships), Regulation 13.4 (Tier III)" as adopted by the International Maritime Organization (IMO).

NO_x emission in liquid fuel mode

IMO NO_x certification will be carried out while factory acceptance test with distillate fuel for compliance with IMO Tier II. On-board compliance with IMO Tier III by activation of SCR system achieved.

NO_x emission in gas mode

In gas mode the SCR system will be deactivated. Therefore IMO NO_x certification will be carried out while factory acceptance test for gas operation for compliance with IMO Tier III.

2.26.2 Smoke emission index (FSN)

Valid for normal engine operation.

Engine load	Smoke emission index (FSN)		
Fuel	MGO, MDO	HFO	Gas
1,050 kW/cyl. or 1,150 kW/cyl., equipped with VIT			
100 %	0.1 + 0.05	0.2 + 0.1	< 0.1
75 %	0.1 + 0.05	0.2 + 0.1	< 0.1
50 %	0.2 + 0.1	0.2 + 0.2	< 0.1
25 %	0.2 + 0.1	0.4 + 0.2	< 0.1
10 %	0.5 + 0.1	0.7 + 0.2	< 0.1
1,050 kW/cyl. equipped with VVT			
100 %	0.1 + 0.05	0.2 + 0.1	< 0.1
75 %	0.1 + 0.05	0.2 + 0.1	< 0.1
50 %	0.15 + 0.1	0.2 + 0.2	< 0.1
25 %	0.15 + 0.1	0.2 + 0.2	< 0.1

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Engine load	Smoke emission index (FSN)		
Fuel	MGO, MDO	HFO	Gas
10 %	0.3 + 0.1	0.3 + 0.2	< 0.1

Table 147: Smoke emission index (FSN), MAN 51/60DF

Limit of visibility is 0.4 FSN.

SCR regeneration phase

Dependent on the ambient conditions during the regeneration phase of the SCR the smoke emission index may be increased.

2.26.3 Emission related installation instruction for engines

Position of the exhaust gas sampling points

The sampling position shall be fitted:

- At least 10 times the diameter of the exhaust pipe after the outlet of the engine, turbocharger or last after-treatment device
- In a straight pipe segment:
  - With an inlet path length before sampling position of at least 5 times the diameter of the exhaust pipe after last bending/obstruction/diameter deviation
  - With a path length after sampling position of at least 2 times the diameter of the exhaust pipe
- Aat least 5 times the diameter of the exhaust pipe before end of exhaust pipe
- Sufficiently close to the engine to ensure that the exhaust gas temperature at the sampling position will be minimum 180°C

If this is not attainable, a deviation of these requirements is acceptable if:

- justified by good engineering practice
- an incorrect measurement is not expected

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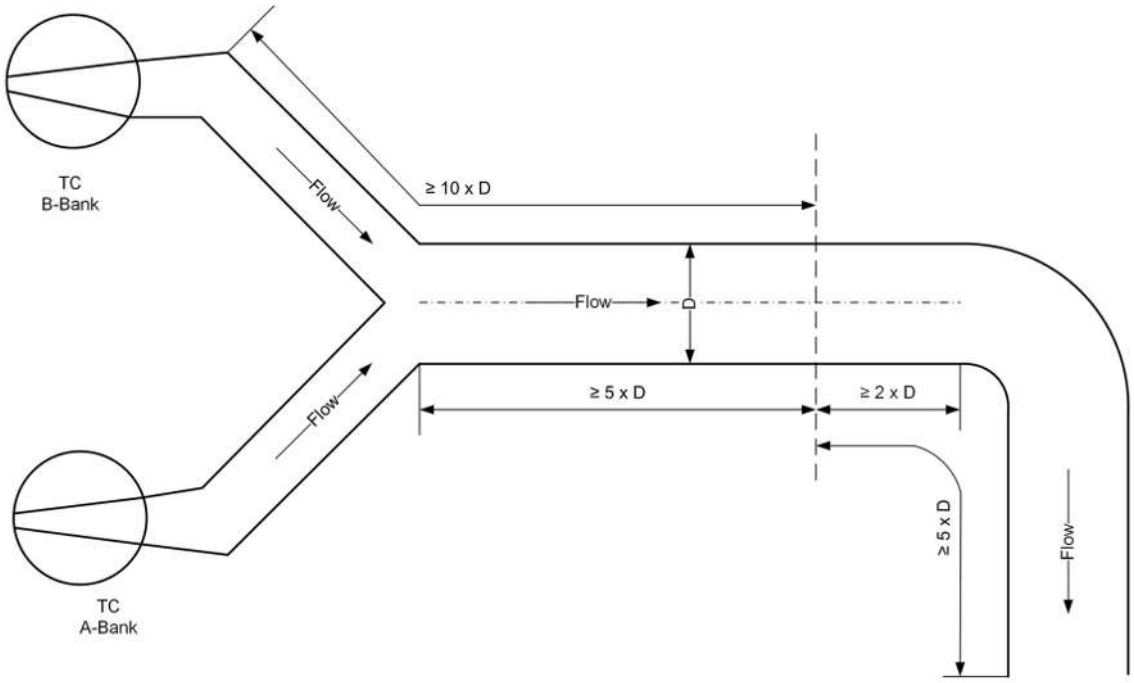

Figure 47: Position of sampling points (exemplary)

In the case an engine has distinct groups of manifolds, it is permissible to acquire a sample from each group individually and calculate an average exhaust emission. In this case each sampling point needs to be designed as mentioned above.

The exhaust gas upstream the sampling point shall be free of any dilution from surrounding air or contamination from other exhaust gas systems.

Accessibility and working area at sampling point

The sampling point shall be accessible in a save way also during engine operation. In case the sampling point is coverd by an insulation, the position shall be clearly visible marked on the insulation. The Insulation shall be designed in a way that it is easy to remove and to install again.

If necessary a step has to be established.

There shall be enough space to place and operate the measurement equipment.

Requirements regarding the working area:

- Minimum space requirements: about 2m width, 2m depth, 2m height
- Sufficient floor load capacity (at least 200 kg per m²)
- Temperature at this working area should be within +5 to + 40°C and well ventilated
- Not subjected to excessive vibrations
- If needed weather protection for personnel and equipment against sun, wind and rain

Depending on the measurement equipment it should to be considered that during the measurements following items are available at the working area:

• Adequate lighting and ventilation

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- Power supply
- Pressurized air (instrument quality, oil free)
- Lifting devices for raising and lowering the equipment, if necessary

Exhaust pipe connection

- 1 piece of pipe thread inner diameter G 3” (use for particulate measurements)
- 2 pieces of pipe thread inner diameter G ½" (use for gaseous emission measurements or for smoke number measurement)

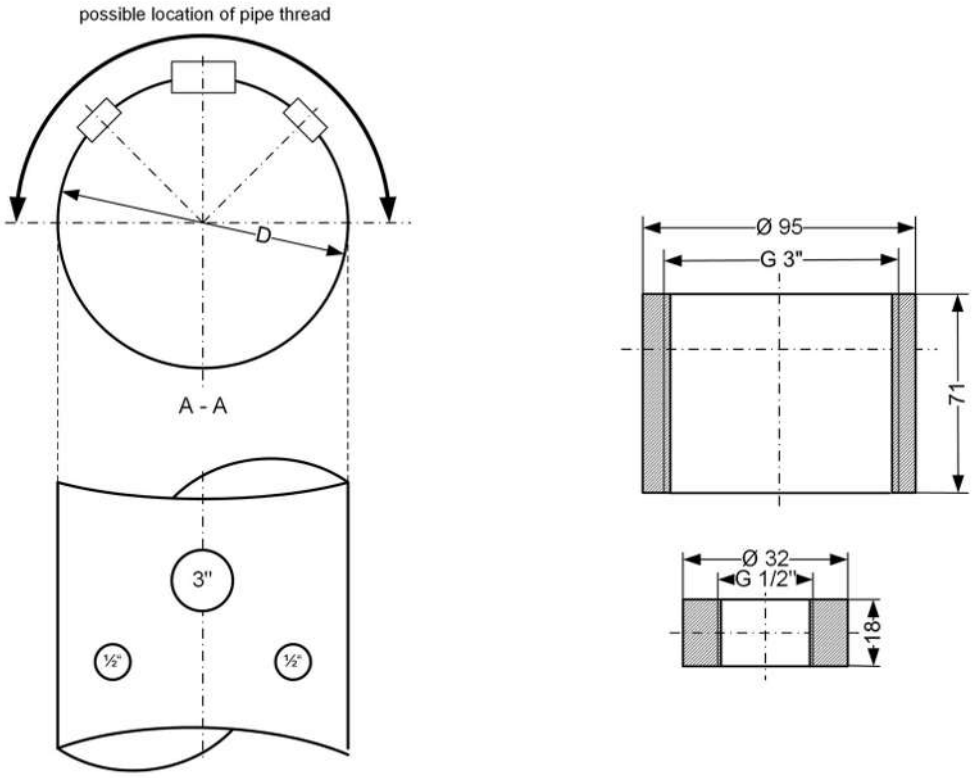


Figure 48: Sampling point, details

Installation of pipe thread G 3" and G ½”

To establish an airtight connection the pipe threads shall be welded at the exhaust gas pipe. In case of a horizontal orientated exhaust gas pipe the position shall be in the upper half of the exhaust gas pipe in radial direction to the center line (see figure 2) and easy to access.

When the sampling points are not in use, a pipe plug shall close the pipe threads.

2.27 Noise

2.27.1 Airborne noise

L engine

Sound pressure level Lp

Measurements

Approximately 20 measuring points at 1 metre distance from the engine surface are distributed evenly around the engine according to ISO 6798. The noise at the exhaust outlet is not included, but provided separately in the following sections.

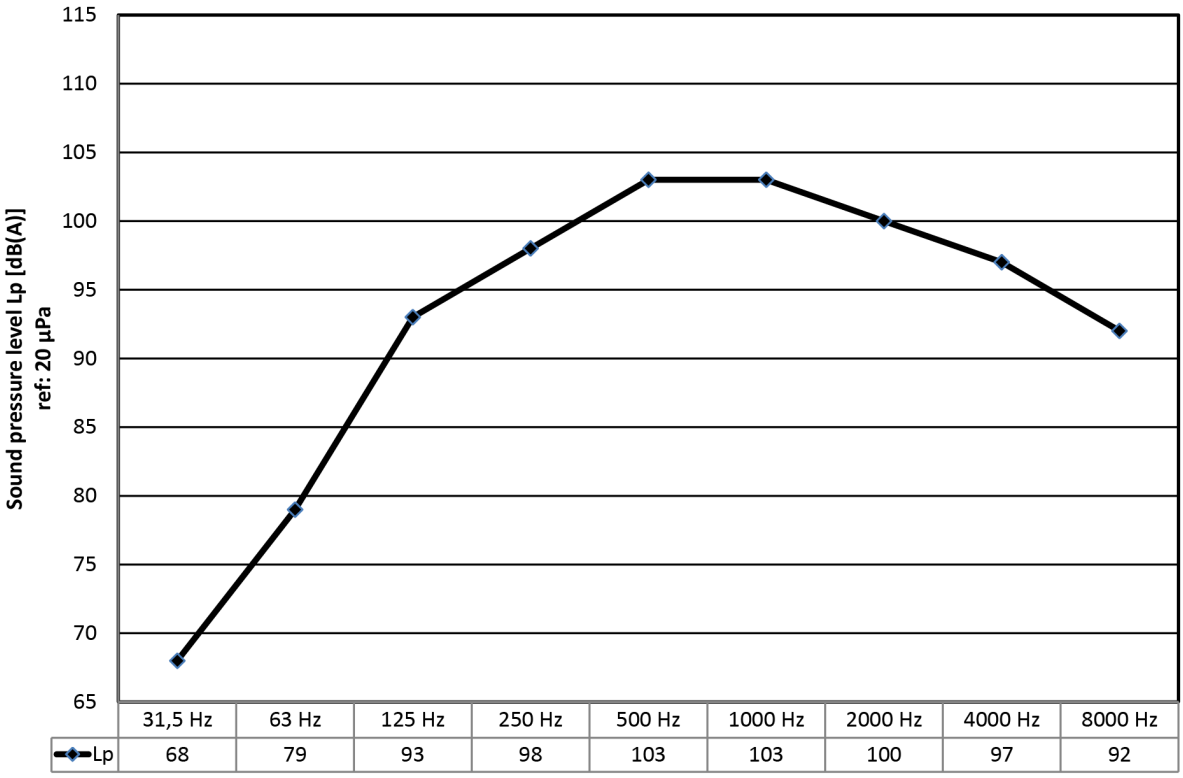
Octave level diagram

The expected sound pressure level Lp is below 107 dB(A) at 100 % MCR.

The octave level diagram below represents an envelope of averaged measured spectra for comparable engines at the testbed and is a conservative spectrum consequently. No room correction is performed. The data will change depending on the acoustical properties of the environment.

Blow-off noise

Blow-off noise is not considered in the measurements, see below.



	31,5 Hz	63 Hz	125 Hz	250 Hz	500 Hz	1000 Hz	2000 Hz	4000 Hz	8000 Hz
Lp	68	79	93	98	103	103	100	97	92

Figure 49: Airborne noise – Sound pressure level Lp – Octave level diagram

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V engine

Sound pressure level Lp

Measurements

Approximately 20 measuring points at 1 metre distance from the engine surface are distributed evenly around the engine according to ISO 6798. The noise at the exhaust outlet is not included, but provided separately in the following sections.

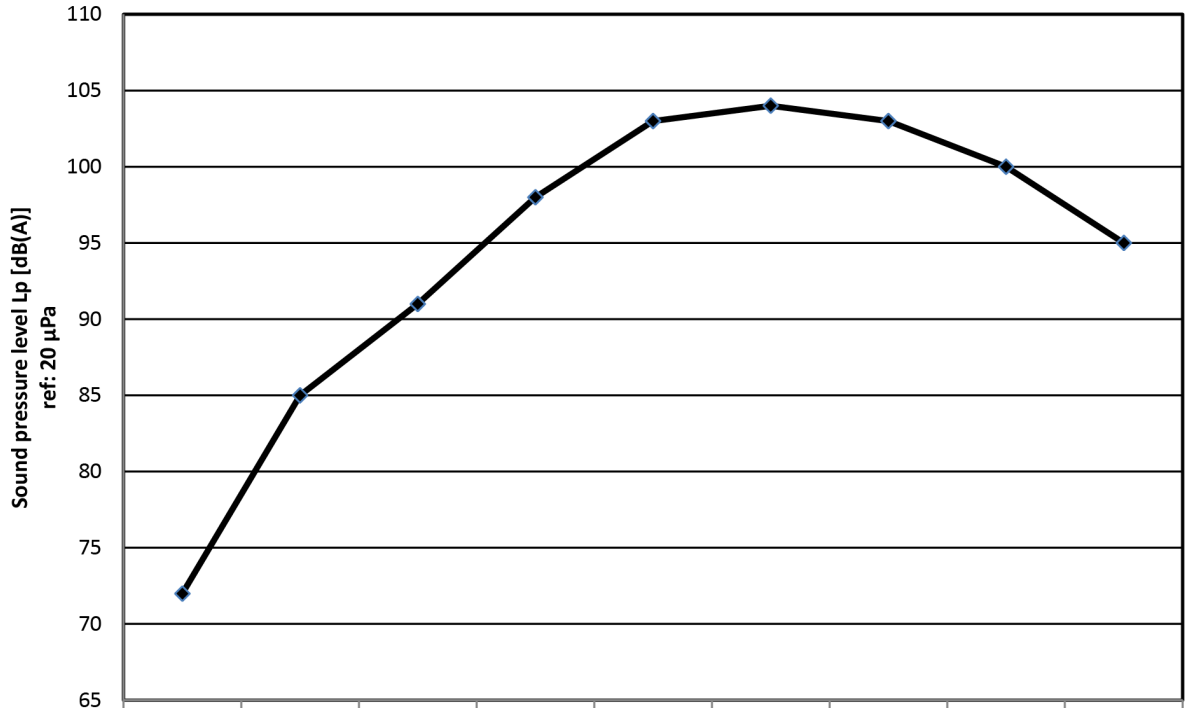
Octave level diagram

The expected sound pressure level Lp is below 110 dB(A) at 100 % MCR.

The octave level diagram below represents an envelope of averaged measured spectra for comparable engines at the testbed and is a conservative spectrum consequently. No room correction is performed. The data will change depending on the acoustical properties of the environment.

Blow-off noise

Blow-off noise is not considered in the measurements, see below.



	31,5 Hz	63 Hz	125 Hz	250 Hz	500 Hz	1000 Hz	2000 Hz	4000 Hz	8000 Hz
Lp	72	85	91	98	103	104	103	100	95

Figure 50: Airborne noise – Sound pressure level Lp – Octave level diagram

2.27.2 Intake noise

L/V engine

Sound power level Lw

Measurements

The (unsilenced) intake air noise is determined based on measurements at the turbocharger test bed and on measurements in the intake duct of typical engines at the test bed.

Octave level diagram

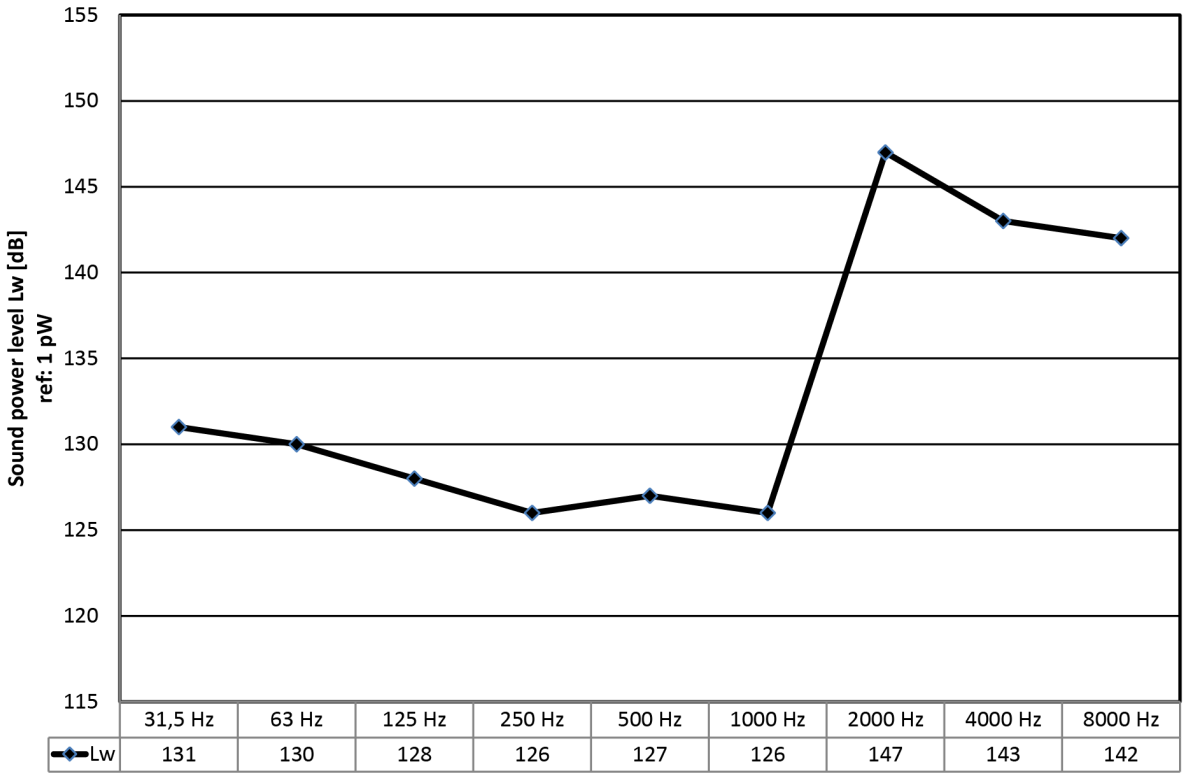
The expected sound power level Lw of the unsilenced intake noise in the intake duct is below 150 dB at 100 % MCR.

The octave level diagram below represents an envelope of averaged measured spectra for comparable engines and is a conservative spectrum consequently. The data will change depending on the acoustical properties of the environment.

Charge air blow-off noise

Charge air blow-off noise is not considered in the measurements, see below.

These data are required and valid only for ducted air intake systems. The data are not valid if the standard air filter silencer is attached to the turbocharger.



	31,5 Hz	63 Hz	125 Hz	250 Hz	500 Hz	1000 Hz	2000 Hz	4000 Hz	8000 Hz
Lw	131	130	128	126	127	126	147	143	142

Figure 51: Unsilenced intake noise MAN L51/60DF – Sound power level Lw – Octave level diagram

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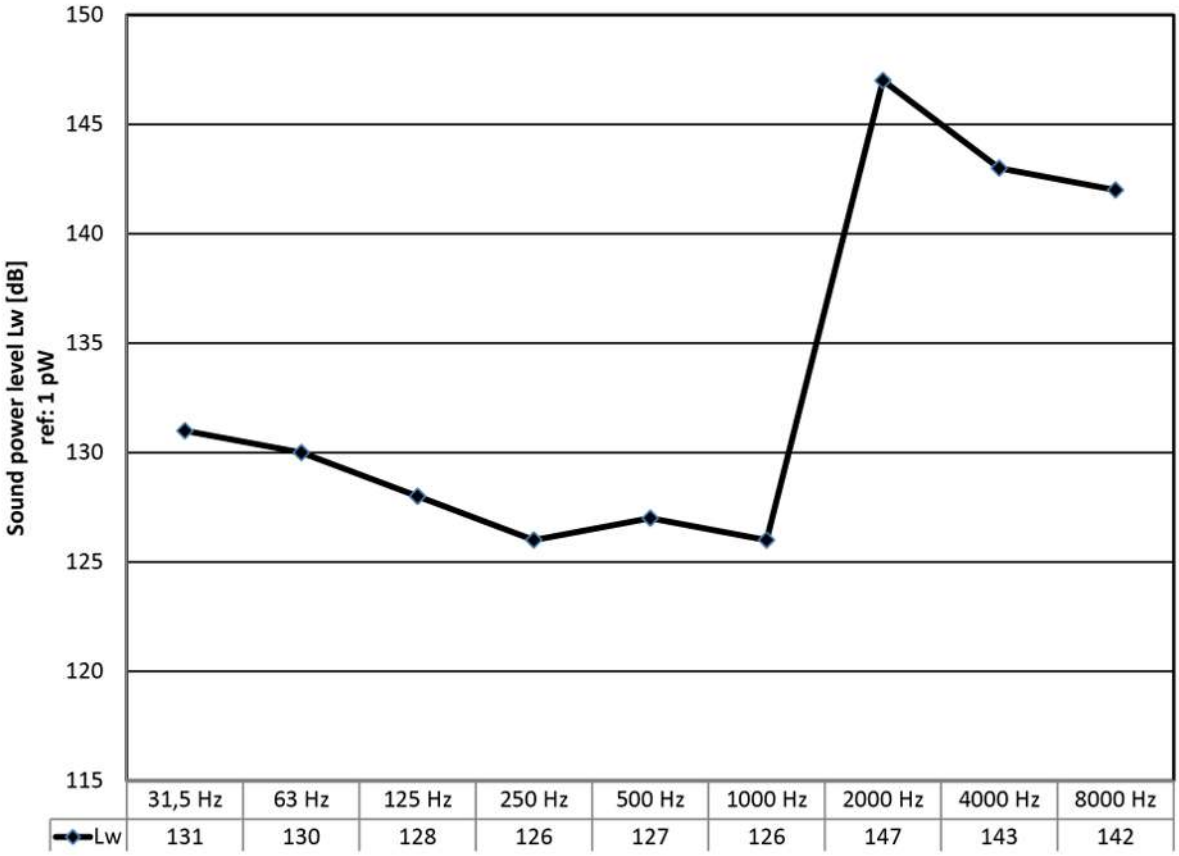

Figure 52: Unsilenced intake noise MAN V51/60DF – Sound power level Lw – Octave level diagram

2.27.3 Exhaust gas noise

L engine

Sound power level Lw at 100 % MCR

Measurements

The (unsilenced) exhaust gas noise is measured according to internal MAN Energy Solutions guidelines at several positions in the exhaust duct.

Octave level diagram

The sound power level Lw of the unsilenced exhaust gas noise in the exhaust pipe is shown at 100 % MCR.

The octave level diagram below represents an envelope of averaged measured spectra for comparable engines and is a conservative spectrum consequently. The data will change depending on the acoustical properties of the environment.

Acoustic design

To ensure an appropriate acoustic design of the exhaust gas system, the yard, MAN Energy Solutions, supplier of silencer and where necessary acoustic consultant have to cooperate.

Waste gate blow-off noise

Waste gate blow-off noise is not considered in the measurements, see below.

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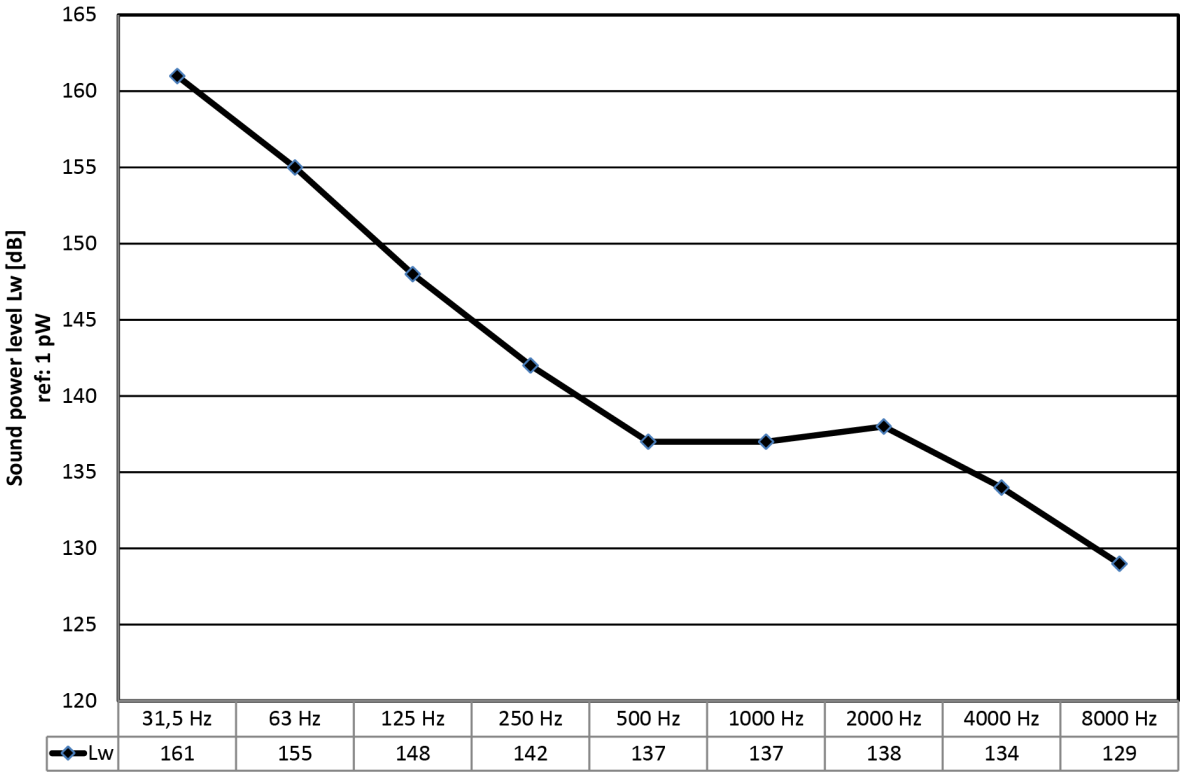

	31,5 Hz	63 Hz	125 Hz	250 Hz	500 Hz	1000 Hz	2000 Hz	4000 Hz	8000 Hz
Lw	161	155	148	142	137	137	138	134	129

Figure 53: Unsilenced exhaust gas noise – Sound power level Lw – Octave level diagram

V engine

Sound power level Lw at 100 % MCR

Measurements

The (unsilenced) exhaust gas noise is measured according to internal MAN Energy Solutions guidelines at several positions in the exhaust duct.

Octave level diagram

The sound power level Lw of the unsilenced exhaust gas noise in the exhaust pipe is shown at 100 % MCR.

The octave level diagram below represents an envelope of averaged measured spectra for comparable engines and is a conservative spectrum consequently. The data will change depending on the acoustical properties of the environment.

Acoustic design

To ensure an appropriate acoustic design of the exhaust gas system, the yard, MAN Energy Solutions, supplier of silencer and where necessary acoustic consultant have to cooperate.

Waste gate blow-off noise

Waste gate blow-off noise is not considered in the measurements, see below.

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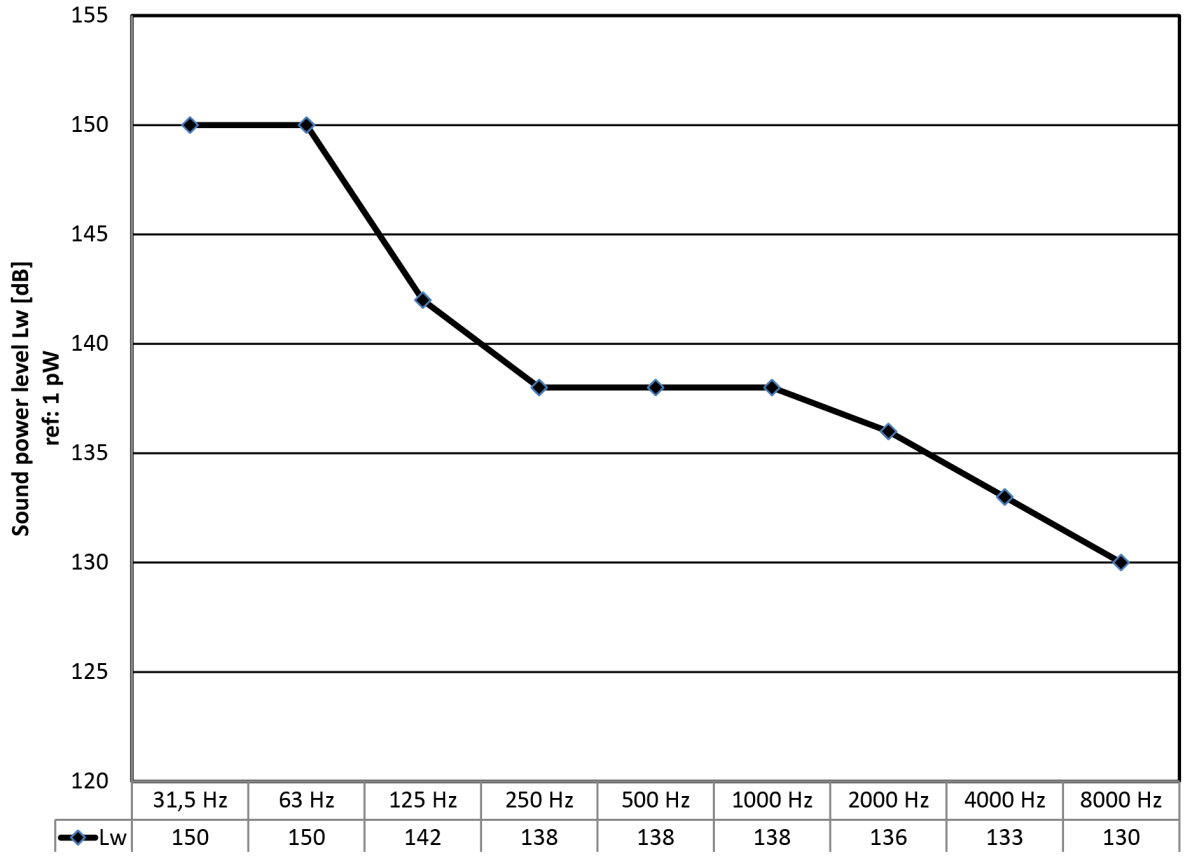

Figure 54: Unsilenced exhaust gas noise – Sound power level Lw – Octave level diagram

2.27.4 Blow-off noise example

Sound power level Lw

Measurements

The (unsilenced) charge air blow-off noise is measured according to DIN 45635, part 47 at the orifice of a duct.

Throttle body with bore size 135 mm

Expansion of charge air from 3.4 bar to ambient pressure at 42 °C

Octave level diagram

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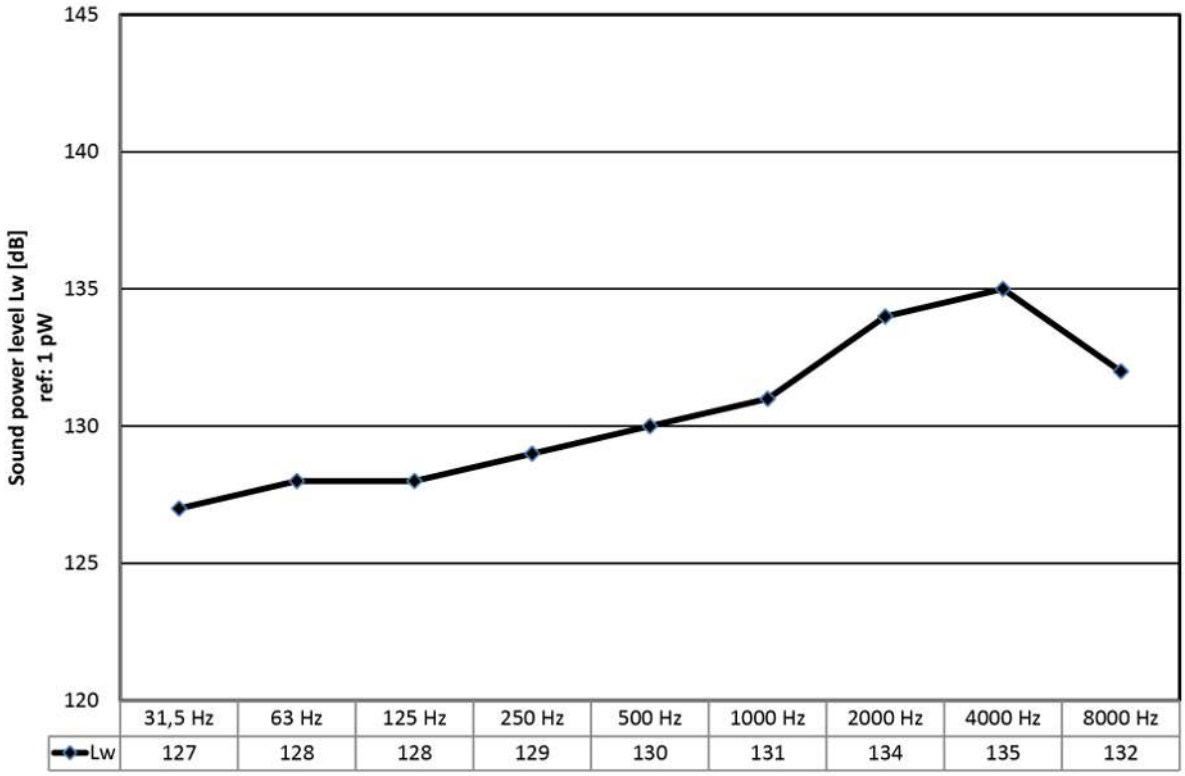

Figure 55: Unsilenced charge air blow-off noise – Sound power level Lw – Octave level diagram

2.27.5 Noise and vibration – Impact on foundation

Noise and vibration is emitted by the engine to the surrounding (see figure Noise and vibration – Impact on foundation, Page 169). The engine impact transferred through the engine mounting to the foundation is focused subsequently.

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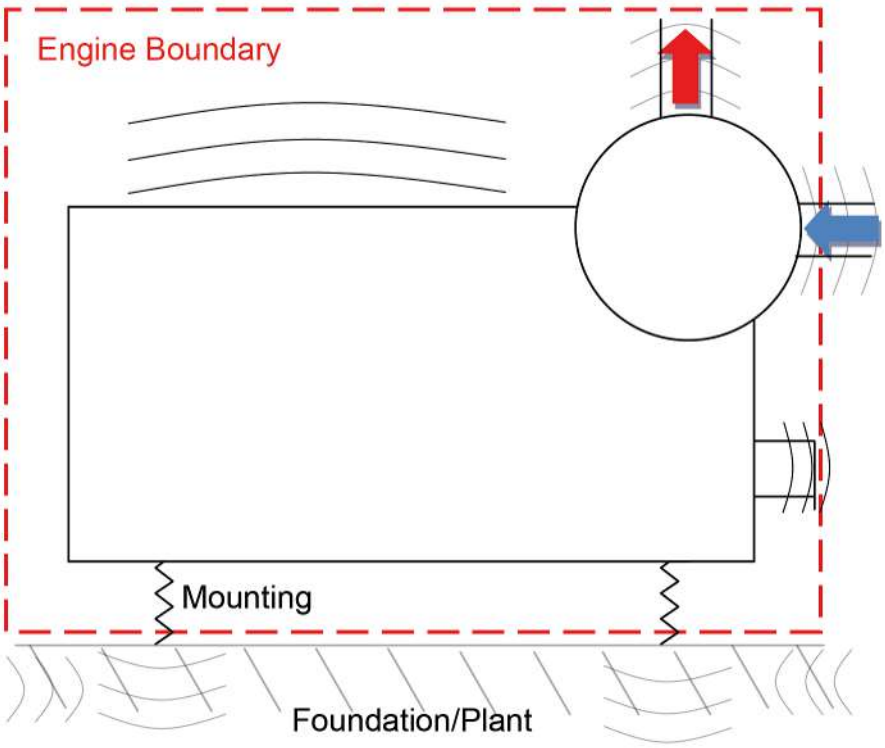

Figure 56: Noise and vibration – Impact on foundation

The foundation is excited to vibrations in a wide frequency range by the engine and by auxiliary equipment (from engine or plant). The engine is vibrating as a rigid body. Additionally, elastic engine vibrations are superimposed. Elastic vibrations are either of global (e.g. complete engine bending) or local (e.g. bending engine foot) character. If the higher frequency range is involved, the term "structure borne noise" is used instead of "vibrations".

Mechanical engine vibrations are mainly caused by mass forces of moved drive train components and by gas forces of the combustion process. For structure borne noise, further excitations are relevant as well, e.g. impacts from piston stroke and valve seating, impulsive gas force components, alternating gear train meshing forces and excitations from pumps.

For the analysis of the engine noise- and vibration-impact on the surrounding, the complete system with engine, engine mounting, foundation and plant has to be considered.

Engine related noise and vibration reduction measures cover e.g. counterbalance weights, balancing, crankshaft design with firing sequence, component design etc. The remaining, inevitable engine excitation is transmitted to the surrounding of the engine – but not completely in case of a resilient engine mounting, which is chosen according to the application-specific requirements. The resilient mounting isolates engine noise and vibration from its surrounding to a large extend. Hence, the transmitted forces are considerably reduced compared with a rigid mounting. Nevertheless, the engine itself is vibrating stronger in the low frequency range in general – especially when driving through mounting resonances.

In order to avoid resonances, it must be ensured that eigenfrequencies of foundation and coupled plant structures have a sufficient safety margin in relation to the engine excitations. Moreover, the foundation has to be designed as stiff as possible in all directions at the connections to the engine. Thus, the

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foundation mobility (measured according to ISO 7262) has to be as low as possible to ensure low structure borne noise levels. For low frequencies, the global connection of the foundation with the plant is focused for that matter. The dynamic vibration behaviour of the foundation is mostly essential for the mid frequency range. In the high frequency range, the foundation elasticity is mainly influenced by the local design at the engine mounts. E.g. for steel foundations, sufficient wall thicknesses and stiffening ribs at the connection positions shall be provided. The dimensioning of the engine foundation also has to be adjusted to other parts of the plant. For instance, it has to be avoided that engine vibrations are amplified by alternator foundation vibrations. Due to the scope of supply, the foundation design and its connection with the plant is mostly within the responsibility of the costumer. Therefore, the customer is responsible to involve MAN Energy Solutions for consultancy in case of system-related questions with interaction of engine, foundation and plant. The following information is available for MAN Energy Solutions customers, some on special request:

- Residual external forces and couples (Project Guide)

  Resulting from the summation of all mass forces from the moving drive train components. All engine components are considered rigidly in the calculation. The residual external forces and couples are only transferred completely to the foundation in case of a rigid mounting, see above.
- Static torque fluctuation (Project Guide)

  Static torque fluctuations result from the summation of gas and mass forces acting on the crank drive. All components are considered rigidly in the calculation. These couples are acting on the foundation dependent on the applied engine mounting, see above.
- Mounting forces (project-specific)

  The mounting dimensioning calculation is specific to a project and defines details of the engine mounting. Mounting forces acting on the foundation are part of the calculation results. Gas and mass forces are considered for the excitation. The engine is considered as one rigid body with elastic mounts. Thus, elastic engine vibrations are not implemented.
- Reference measurements for engine crankcase vibrations according to ISO 10816-6 (project-specific)
- Reference test bed measurements for structure borne noise (project-specific)

  Measuring points are positioned according to ISO 13332 on the engine feet above and below the mounting elements. Structure borne noise levels above elastic mounts mainly depend on the engine itself. Whereas structure borne noise levels below elastic mounts strongly depend on the foundation design. A direct transfer of the results from the test bed foundation to the plant foundation is not easily possible – even with the consideration of test bed mobilities. The results of test bed foundation mobility measurements according to ISO 7626 are available as a reference on request as well.
- Dynamic transfer stiffness properties of resilient mounts (supplier information, project-specific)

Beside the described interaction of engine, foundation and plant with transfer through the engine mounting to the foundation, additional transfer paths need to be considered. For instance with focus on the elastic coupling of the drive train, the exhaust pipe, other pipes and supports etc. Besides the engine, other sources of noise and vibration need to be considered as well (e.g. auxiliary equipment, propeller, thruster).

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2.28 Vibration

2.28.1 Torsional vibrations

Data required for torsional vibration calculation

MAN Energy Solutions calculates the torsional vibrations behaviour for each individual engine plant of their supply to determine the location and severity of resonance points. If necessary, appropriate measures will be taken to avoid excessive stresses due to torsional vibration. These investigations cover the ideal normal operation of the engine (all cylinders are firing equally) as well as the simulated emergency operation (misfiring of the cylinder exerting the greatest influence on vibrations, acting against compression). Besides the natural frequencies and the modes also the dynamic response will be calculated, normally under consideration of the 1st to 24th harmonic of the gas and mass forces of the engine.

If necessary, a torsional vibration calculation will be worked out which can be submitted for approval to a classification society or a legal authority.

To carry out the torsional vibration calculation following particulars and/or documents are required.

General

- Type of propulsion (GenSet, mechanical or electric propulsion)
- Arrangement of the whole system including all engine-driven equipment
- Definition of the operating modes
- Maximum power consumption of the individual working machines

Engine

- Rated output, rated speed
- Kind of engine load (fixed pitch propeller, controllable pitch propeller, combinator curve, operation with reduced speed at excessive load)
- Kind of mounting of the engine (can influence the determination of the flexible coupling)
- Operational speed range

Flexible coupling

- Make, size and type
- Rated torque (Nm)
- Possible application factor
- Maximum speed (rpm)
- Permissible maximum torque for passing through resonance (Nm)
- Permissible shock torque for short-term loads (Nm)
- Permanently permissible alternating torque (Nm) including influencing factors (frequency, temperature, mean torque)
- Permanently permissible power loss (W) including influencing factors (frequency, temperature)
- Dynamic torsional stiffness (Nm/rad) including influencing factors (load, frequency, temperature), if applicable

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- Relative damping (ψ) including influencing factors (load, frequency, temperature), if applicable
- Moment of inertia (kgm^2) for all parts of the coupling
- Dynamic stiffness in radial, axial and angular direction
- Permissible relative motions in radial, axial and angular direction, permanent and maximum

Gearbox

- Make and type
- Torsional multi mass system including the moments of inertia and the torsional stiffness, preferably related to the individual speed; in case of related figures, specification of the relation speed is required
- Gear ratios (number of teeth, speeds)
- Possible operating conditions (different gear ratios, clutch couplings)
- Permissible alternating torques in the gear meshes

Shaft line

- Drawing including all information about length and diameter of the shaft sections as well as the material
- Alternatively torsional stiffness (Nm/rad)

Propeller

- Kind of propeller (fixed pitch or controllable pitch propeller)
- Moment of inertia in air (kgm^2)
- Moment of inertia in water (kgm^2); for controllable pitch propellers also in dependence on pitch; for twin-engine plants separately for single- and twin-engine operation
- Relation between load and pitch
- Number of blades
- Diameter (mm)
- Possible torsional excitation in % of the rated torque for the 1st and the 2nd blade-pass frequency

Alternator for electric propulsion plants

- Drawing of the alternator shaft with all lengths and diameters
- Alternatively, torsional stiffness (Nm/rad)
- Moment of inertia of the parts mounted to the shaft (kgm^2)
- Electrical output (kVA) including power factor cos φ and efficiency
- Or mechanical output (kW)
- Complex synchronizing coefficients for idling and full load in dependence on frequency, reference torque
- Island or parallel mode
- Load profile (e.g. load steps)
- Frequency fluctuation of the net

Alternator for mechanical propulsion plants (e.g. PTO/PTH)

- Drawing of the alternator shaft with all lengths and diameters
- Torsional stiffness, if available

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- Moment of inertia of the parts mounted to the shaft (kgm²)
- Electrical output (kVA) including power factor cos φ and efficiency
- Or mechanical output (kW)
- Complex synchronizing coefficients for idling and full load in dependence on frequency, reference torque

2.29 Requirements for power drive connection (static)

Evaluation of permissible theoretical bearing loads

Limit values of masses to be coupled after the engine

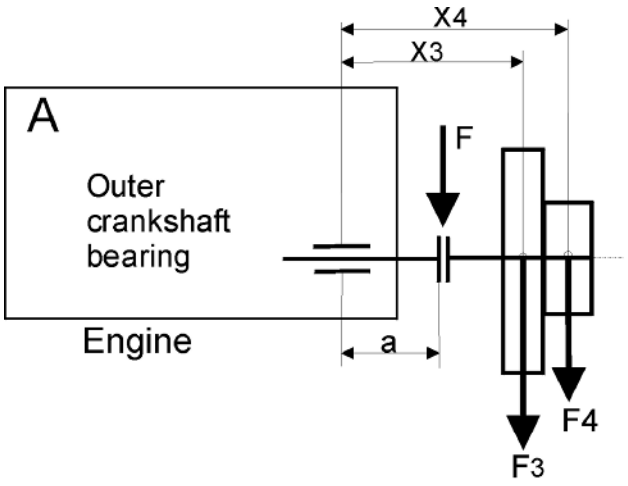

Figure 57: Case A: Overhung arrangement

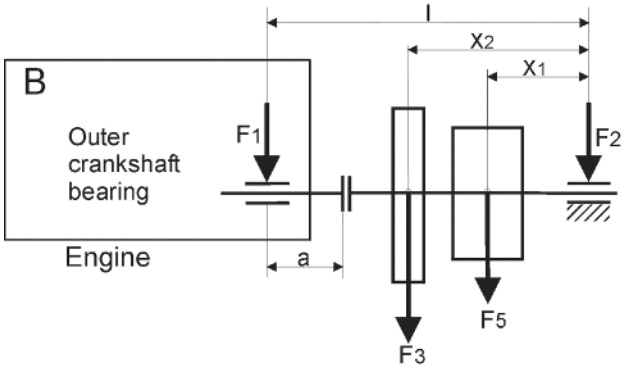

Figure 58: Case B: Rigid coupling

$M_{max} = F * a = F_3 * x_3 + F_4 * x_4$ $F_1 = (F_3 * x_2 + F_5 * x_1)/l$

F_1	Theoretical bearing force at the external engine bearing
F_2	Theoretical bearing force at the alternator bearing
F_3	Flywheel weight
F_4	Coupling weight acting on the engine, including reset forces
F_5	Rotor weight of the alternator
a	Distance between end of coupling flange and centre of outer crankshaft bearing

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l	Distance between centre of outer crankshaft bearing and alternator bearing

Engine	Distance a	Case A	Case B
		$M_{max} = F * a$	$F_{1\ max}$
	mm	kNm	kN
L engine	530	80 [1]	140
V engine	560	105	180
[1] Inclusive of couples resulting from restoring forces of the coupling.			

Table 148: Example calculation case A and B

Distance between engine seating surface and crankshaft centre line:

- L engine: 700 mm
- V engine: 830 mm

Note:
Stated values only for orientation – need to be clarified project specific and may be limited due to the torsional vibration calculation or special service conditions.

Note:
Masses which are connected downstream of the engine in the case of an overhung or rigidly coupled, arrangement result in additional crankshaft bending stress, which is mirrored in a measured web deflection during engine installation.

Provided the limit values for the masses to be coupled downstream of the engine (permissible values for M_{max} and $F_{1\ max}$) are complied with, the permitted web deflections will not be exceeded during assembly.

Observing these values ensures a sufficiently long operating time before a realignment of the crankshaft has to be carried out.

2.30 Requirements for power drive connection (dynamic)

2.30.1 Moments of inertia – Crankshaft, damper, flywheel

Engine MAN 51/60DF

1,050 kW/cyl.; 500/514 rpm

Operation with variable speed

Marine main engines						
Engine					Required minimum total moment of inertia[1]	Plant
No. of cylinders, config.	Maximum continuous rating	Moment of inertia crankshaft + damper	Moment of inertia flywheel	Mass of flywheel		Required minimum additional moment of inertia after flywheel[2]
	[kW]	[kgm²]	[kgm²]	[kg]	[kgm²]	[kgm²]
n = 500 rpm						
6L	6,000	2,633	3,102	5,324	2,736	-

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Marine main engines						
Engine					Required minimum total moment of inertia[1]	Plant
No. of cylinders, config.	Maximum continuous rating	Moment of inertia crankshaft + damper	Moment of inertia flywheel	Mass of flywheel		Required minimum additional moment of inertia after flywheel[2]
	[kW]	[kgm²]	[kgm²]	[kg]	[kgm²]	[kgm²]
7L	7,350	3,416			3,352	
8L	8,400	3,474			3,830	
9L	9,450	3,779			4,309	
12V	12,600	4,770	2,935	4,308	5,746	-
14V	14,700	5,356			6,703	
16V	16,800	5,943			7,661	
n = 514 rpm						
6L	6,000	2,633	3,102	5,324	2,589	-
7L	7,350	3,416			3,172	
8L	8,400	3,474			3,625	
9L	9,450	3,779			4,078	
12V	12,600	4,770	2,935	4,308	5,437	-
14V	14,700	5,356			6,343	
16V	16,800	5,943			7,249	

[1] Required minimum moment of inertia of engine, flywheel and arrangement after flywheel in total.

[2] Required additional moment of inertia after flywheel to achieve the required minimum total moment of inertia.

Table 149: Moments of inertia/flywheels for marine main engines – Engine MAN 51/60DF 1,050 kW/cyl.

1,150 kW/cyl.; 500/514 rpm

Operation with variable speed

Marine main engines						
Engine					Required minimum total moment of inertia[1]	Plant
No. of cylinders, config.	Maximum continuous rating	Moment of inertia crankshaft + damper	Moment of inertia flywheel	Mass of flywheel		Required minimum additional moment of inertia after flywheel[2]
	[kW]	[kgm²]	[kgm²]	[kg]	[kgm²]	[kgm²]
n = 500 rpm						
6L	6,900	2,633	3,102	5,324	3,146	-
7L	8,050	3,416			3,671	
8L	9,200	3,474			4,195	
9L	10,350	3,779			4,720	
12V	13,800	4,770	2,935	4,308	6,293	-
14V	16,100	5,356			7,342	

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Marine main engines						
Engine					Required minimum total moment of inertia[1]	Plant
No. of cylinders, config.	Maximum continuous rating	Moment of inertia crankshaft + damper	Moment of inertia flywheel	Mass of flywheel		Required minimum additional moment of inertia after flywheel[2]
	[kW]	[kgm²]	[kgm²]	[kg]	[kgm²]	[kgm²]
n = 514 rpm						
6L	6,900	2,633	3,102	5,324	2,977	-
7L	8,050	3,416			3,474	
8L	9,200	3,474			3,970	
9L	10,350	3,779			4,466	
12V	13,800	4,770	2,935	4,308	5,955	-
14V	16,100	5,356			6,947	

[1] Required minimum moment of inertia of engine, flywheel and arrangement after flywheel in total.

[2] Required additional moment of inertia after flywheel to achieve the required minimum total moment of inertia.

Table 150: Moments of inertia/flywheels for marine main engines – Engine MAN 51/60DF 1,150 kW/cyl.

1,050 kW/cyl.; 500/514 rpm

Constant speed

Marine main engines							
Engine						Required minimum total moment of inertia[1]	Plant
No. of cylinders, config.	Maximum continuous rating	Moment of inertia crankshaft + damper	Moment of inertia flywheel	Mass of flywheel	Cyclic irregularity		Required minimum additional moment of inertia after flywheel[2]
	[kW]	[kgm²]	[kgm²]	[kg]		[kgm²]	[kgm²]
n = 500 rpm							
6L	6,000	2,633	3,102	5,324	1/60	7,548	1,813
7L	7,350	3,416			1/65	9,246	2,728
8L	8,400	3,474			1/50	10,567	3,991
9L	9,450	3,779			1/64	11,887	5,006
12V	12,600	4,770	2,935	4,308	1/92	15,850	8,145
14V	14,700	5,356			1/103	18,492	10,201
16V	16,800	5,943			1/78	21,133	12,255
n = 514 rpm							
6L	6,000	2,633	3,102	5,324	1/66	7,142	1,407
7L	7,350	3,416			1/71	8,749	2,231
8L	8,400	3,474			1/48	9,999	3,423
9L	9,450	3,779			1/70	11,249	4,368

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Marine main engines							
Engine						Required minimum total moment of inertia[1]	Plant
No. of cylinders, config.	Maximum continuous rating	Moment of inertia crankshaft + damper	Moment of inertia flywheel	Mass of flywheel	Cyclic irregularity		Required minimum additional moment of inertia after flywheel[2]
	[kW]	[kgm²]	[kgm²]	[kg]		[kgm²]	[kgm²]
12V	12,600	4,770	2,935	4,308	1/97	14,998	7,293
14V	14,700	5,356			1/96	17,498	9,207
16V	16,800	5,943			1/83	19,998	11,120

[1] Required minimum moment of inertia of engine, flywheel and arrangement after flywheel in total.

[2] Required additional moment of inertia after flywheel to achieve the required minimum total moment of inertia.

Table 151: Moments of inertia/flywheels for electric propulsion plants – Engine MAN 51/60DF 1,050 kW/cyl.

1,150 kW/cyl.; 500/514 rpm

Constant speed

Marine main engines							
Engine						Required minimum total moment of inertia[1]	Plant
No. of cylinders, config.	Maximum continuous rating	Moment of inertia crankshaft + damper	Moment of inertia flywheel	Mass of flywheel	Cyclic irregularity		Required minimum additional moment of inertia after flywheel[2]
	[kW]	[kgm²]	[kgm²]	[kg]		[kgm²]	[kgm²]
n = 500 rpm							
6L	6,900	2,633	3,102	5,324	tbd.	8,680	2,945
7L	8,050	3,416				10,126	3,608
8L	9,200	3,474				11,573	4,997
9L	10,350	3,779				13,020	6,139
12V	13,800	4,770	2,935	4,308	tbd.	17,359	9,654
14V	16,100	5,356				20,253	11,962
n = 514 rpm							
6L	6,900	2,633	3,102	5,324	tbd.	8,213	2,478
7L	8,050	3,416				9,582	3,064
8L	9,200	3,474				10,951	4,375
9L	10,350	3,779				12,320	5,439
12V	13,800	4,770	2,935	4,308	tbd.	16,427	8,722
14V	16,100	5,356				19,164	10,873

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Marine main engines							
Engine						Required minimum total moment of inertia[1]	Plant
No. of cylinders, config.	Maximum continuous rating	Moment of inertia crankshaft + damper	Moment of inertia flywheel	Mass of flywheel	Cyclic irregularity		Required minimum additional moment of inertia after flywheel[2]
	[kW]	[kgm²]	[kgm²]	[kg]		[kgm²]	[kgm²]

[1] Required minimum moment of inertia of engine, flywheel and arrangement after flywheel in total.

[2] Required additional moment of inertia after flywheel to achieve the required minimum total moment of inertia.

Table 152: Moments of inertia/flywheels for electric propulsion plants – Engine MAN 51/60DF 1,150 kW/cyl.

For flywheels dimensions see section Power transmission, Page 184.

2.30.2 Balancing of masses – Firing order

Certain cylinder numbers have unbalanced forces and couples due to crank diagram. These forces and couples cause dynamic effects on the foundation. Due to a balancing of masses the forces and couples are reduced. In the following tables the remaining forces and couples are displayed.

Engine MAN L51/60DF

Rotating crank balance: 100 %

No. of cylinders, config.	Firing order	Residual external couples		
		M_{rot} (kNm) + ½ $M_{osc\ 1st\ order}$ (kNm)		$M_{osc\ 2nd\ order}$ (kNm)
Directiion		vertical	horizontal	vertical
Engine speed		500 rpm		
6L	A	0	0	0
7L	C	0	0	86.6
8L	B	0	0	0
9L	B	27.0	27.0	146.6
Engine speed		514 rpm		
6L	A	0	0	0
7L	C	0	0	91.6
8L	B	0	0	0
9L	B	28.5	28.5	155.0

For engines of type MAN 51/60DF the external mass forces are equal to zero.

M_{rot} is eliminated by means of balancing weights on resiliently mounted engines.

Table 153: Residual external couples – Engine MAN L51/60DF

Firing order: Counted from coupling side

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No. of cylinders, config.	Firing order	Clockwise rotation	Counter clockwise rotation
6L	A	1-3-5-6-4-2	1-2-4-6-5-3
7L	C[1)]	1-2-4-6-7-5-3	1-3-5-7-6-4-2
8L	B	1-4-7-6-8-5-2-3	1-3-2-5-8-6-7-4
9L	B	1-6-3-2-8-7-4-9-5	1-5-9-4-7-8-2-3-6
[1)] Irregular firing order.			

Table 154: Firing order L engine

Engine MAN V51/60DF

Rotating crank balance: 99 %

No. of cylinders, config.	Firing order	Residual external couples			
		M_{rot} (kNm)+ $M_{osc\ 1st\ order}$ (kNm)		$M_{osc\ 2nd\ order}$ (kNm)	
Direction		vertical	horizontal	vertical	horizontal
Engine speed		500 rpm			
12V	A	0	0	0	0
14V	C	0	0	123.1	68.4
16V	B	0	0	0	0
Engine speed		514 rpm			
12V	A	0	0	0	0
14V	C	0	0	130.1	72.3
16V	B	0	0	0	0
For engines of type MAN 51/60DF the external mass forces are equal to zero. M_{rot} is eliminated by means of balancing weights on resiliently mounted engines.					

Table 155: Residual external couples – Engine MAN V51/60DF

Firing order: Counted from coupling side

No. of cylinders, config.	Firing order	Clockwise rotation	Counter clockwise rotation
12V	A	A1-B1-A3-B3-A5-B5-A6-B6-A4-B4-A2-B2	A1-B2-A2-B4-A4-B6-A6-B5-A5-B3-A3-B1
14V	C[1)]	A1-B1-A2-B2-A4-B4-A6-B6-A7-B7-A5-B5-A3-B3	A1-B3-A3-B5-A5-B7-A7-B6-A6-B4-A4-B2-A2-B1
16V	B	A1-B1-A4-B4-A7-B7-A6-B6-A8-B8-A5-B5-A2-B2-A3-B3	A1-B3-A3-B2-A2-B5-A5-B8-A8-B6-A6-B7-A7-B4-A4-B1
[1)] Irregular firing order.			

Table 156: Firing order V engine

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2.30.3 Static torque fluctuation

General

The static torque fluctuation is the summation of the torques acting at all cranks around the crankshaft axis taking into account the correct phase-angles. These torques are created by the gas and mass forces acting at the crankpins, with the crank radius being used as the lever. An rigid crankshaft is assumed.

The values T_{max} and T_{min} listed in the following table(s) represent a measure for the reaction forces of the engine. The reaction forces generated by the torque fluctuation are dependent on speed and cylinder number and give a contribution to the excitations transmitted into the foundation see figure Static torque fluctuation, Page 180 and the table(s) in this section. According to different mountings these forces are reduced.

In order to avoid local vibration excitations in the vessel, it must be ensured that the natural frequencies of important part structures (e.g. panels, bulkheads, tank walls and decks, equipment and its foundation, pipe systems) have a sufficient safety margin (if possible ±30 %) in relation to all engine excitation frequencies.

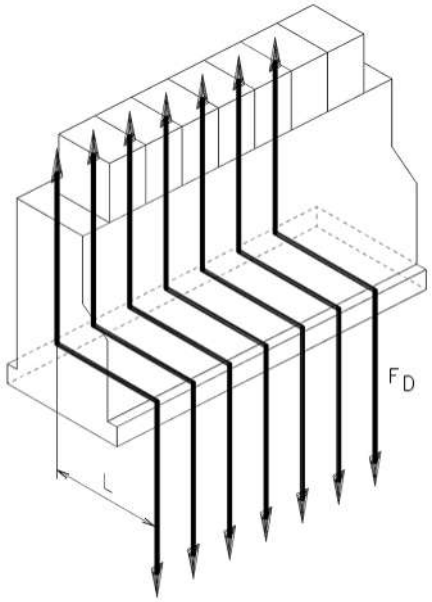

Figure 59: Static torque fluctuation

$$F_D \times L \times z = \frac{T_{max} - T_{min}}{2}$$

L Distance between foundation bolts

z Number of cylinders

Static torque fluctuation and exciting frequencies

L engine – Example to declare abbreviations

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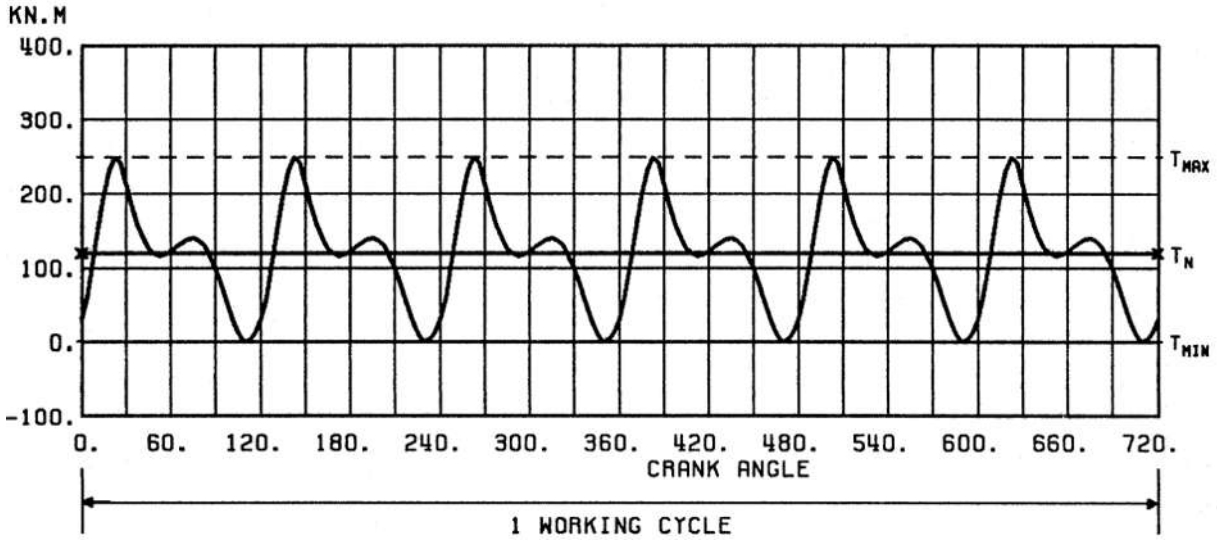

Figure 60: Example to declare abbreviations – L engine

No. of cylinders, config.	Output	Speed	T_n	T_{max}	T_{min}	Main exciting components		
						Order	Frequency[1)]	±T
	kW	rpm	kNm	kNm	kNm	rpm	Hz	kNm
6L	6,000	500	114.6	270.4	–17.1	3.0 6.0	25.0 50.0	85.4 71.3
7L	7,350		140.4	434.6	–97.7	3.5 7.0	29.2 58.3	241.6 52.5
8L	8,400		160.4	400.1	–39.3	4.0 8.0	33.3 66.7	210.2 38.9
9L	9,450		180.5	405.3	–14.0	4.5 9.0	37.5 75.0	205.9 27.8
6L	6,000	514	111.5	255.3	–12.6	3.0 6.0	25.7 51.4	70.7 70.5
7L	7,350		136.6	424.9	–96.9	3.5 7.0	30.0 60.0	235.9 52.6
8L	8,400		156.1	391.0	–38.0	4.0 8.0	34.3 68.5	204.6 39.6
9L	9,450		175.6	396.1	–15.9	4.5 9.0	38.5 77.1	202.1 28.9

[1)] Exciting frequency of the main harmonic components.

Table 157: Static torque fluctuation and exciting frequencies – L engine

No. of cylinders, config.	Output	Speed	T_n	T_{max}	T_{min}	Main exciting components		
						Order	Frequency[1)]	±T
	kW	rpm	kNm	kNm	kNm	rpm	Hz	kNm
6L	6,900	500	131.8	tbd.	tbd.	3.0 6.0	25.0 50.0	tbd.
7L	8,050		153.7			3.5 7.0	29.2 58.3	

No. of cylinders, config.	Output	Speed	T_n	T_{max}	T_{min}	Main exciting components		
						Order	Frequency[1)]	±T
	kW	rpm	kNm	kNm	kNm	rpm	Hz	kNm
8L	9,200		175.7			4.0 8.0	33.3 66.7	
9L	10,350		197.7			4.5 9.0	37.5 75.0	
6L	6,900	514	128.2	tbd.	tbd.	3.0 6.0	25.7 51.4	tbd.
7L	8,050		149.6			3.5 7.0	30.0 60.0	
8L	9,200		170.9			4.0 8.0	34.3 68.5	
9L	10,350		192.3			4.5 9.0	38.5 77.1	

[1)] Exciting frequency of the main harmonic components.

Table 158: Static torque fluctuation and exciting frequencies – L engine

V engine – Example to declare abbreviations

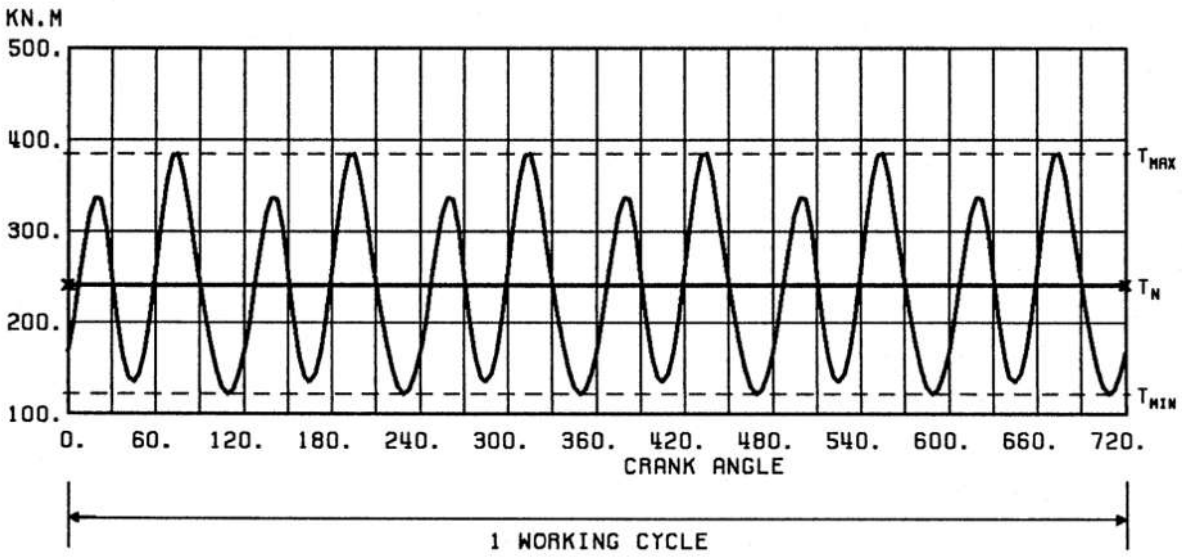

Figure 61: Example to declare abbreviation – V engine

No. of cylinders, config.	Output	Speed	T_n	T_{max}	T_{min}	Main exciting components		
						Order	Frequency[1)]	±T
	kW	rpm	kNm	kNm	kNm	rpm	Hz	kNm
12V	12,600	500	240.6	409.4	101.7	3.0 6.0	25.0 50.0	50.0 125.3
14V	14,700		280.7	405.8	140.6	3.5 7.0	29.2 58.3	21.1 104.5

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No. of cylinders, config.	Output	Speed	T_n	T_{max}	T_{min}	Main exciting components		
						Order	Frequency[1)]	±T
	kW	rpm	kNm	kNm	kNm	rpm	Hz	kNm
16V	16,800		320.9	442.5	191.8	4.0 8.0	33.3 66.7	73.0 73.0
12V	12,600	514	234.1	396.2	102.8	3.0 6.0	25.7 51.4	42.3 124.4
14V	14,700		273.1	397.0	134.7	3.5 7.0	30.0 60.0	20.6 104.8
16V	16,800		312.1	433.0	183.8	4.0 8.0	34.3 68.5	71.0 74.5

[1)] Exciting frequency of the main harmonic components.

Table 159: Static torque fluctuation and exciting frequencies – V engine

No. of cylinders, config.	Output	Speed	T_n	T_{max}	T_{min}	Main exciting components		
						Order	Frequency[1)]	±T
	kW	rpm	kNm	kNm	kNm	rpm	Hz	kNm
12V	13,800	500	263.6	tbd.	tbd.	3.0 6.0	25.0 50.0	tbd.
14V	16,100		307.5			3.5 7.0	29.2 58.3	
12V	13,800	514	256.4	tbd.	tbd.	3.0 6.0	25.7 51.4	tbd.
14V	16,100		299.1			3.5 7.0	30.0 60.0	

[1)] Exciting frequency of the main harmonic components.

Table 160: Static torque fluctuation and exciting frequencies – V engine

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2.31 Power transmission

2.31.1 Flywheel arrangement

Flywheel with flexible coupling

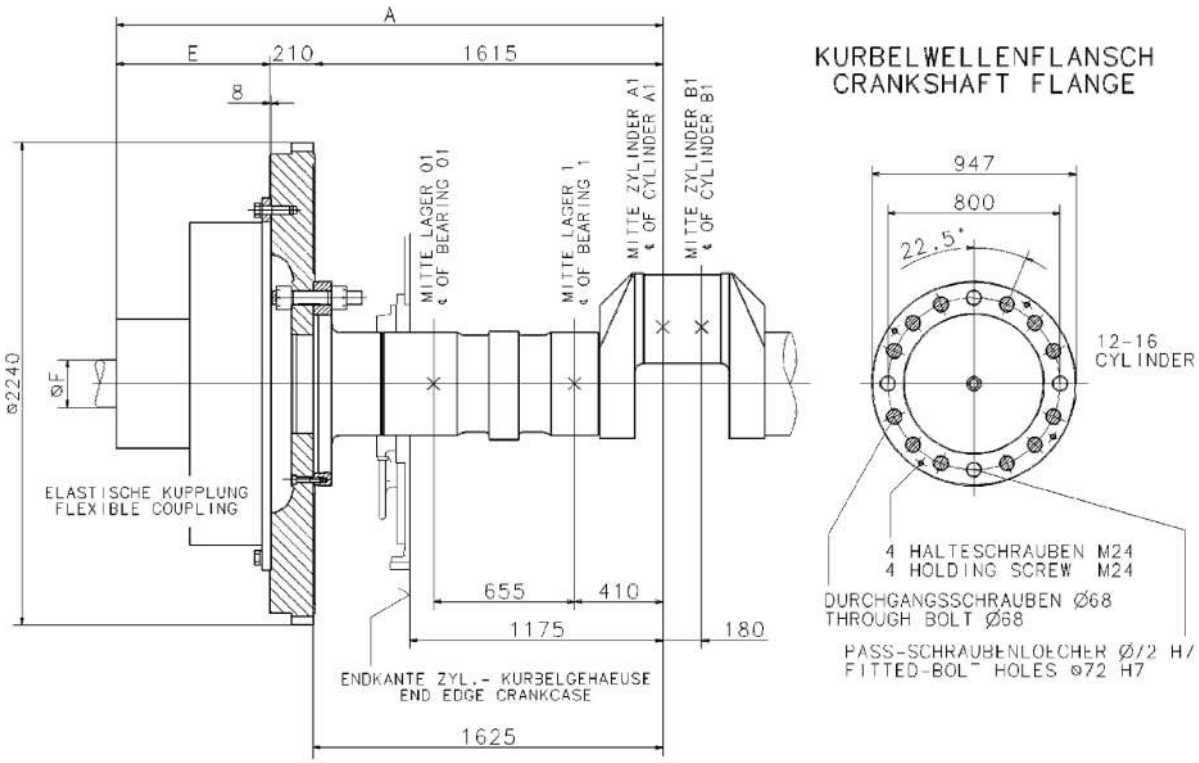

Figure 62: Flywheel with flexible coupling

No. of cylinders, config.	A[1]	A[2]	E[1]	E[2]	F_{min}	F_{max}	No. of through bolts	No. of fitted bolts
	mm							
12V	Dimensions will result from clarification of technical details of propulsion drive						12	2
14V								
16V								

[1] Without torsional limit device.

[2] With torsional limit device.

For mass of flywheel Moments of inertia – Crankshaft, damper, flywheel, Page 174.

Note:
The flexible coupling will be part of MAN Energy Solutions supply and thus we will produce a contract specific flywheel/coupling/driven machine arrangement drawing giving all necessary installation dimensions. Final dimensions of flywheel and flexible coupling will result from clarification of technical details of drive and from the result of the torsional vibration calculation. Flywheel diameter must not be changed.

Arrangement of flywheel, coupling and alternator

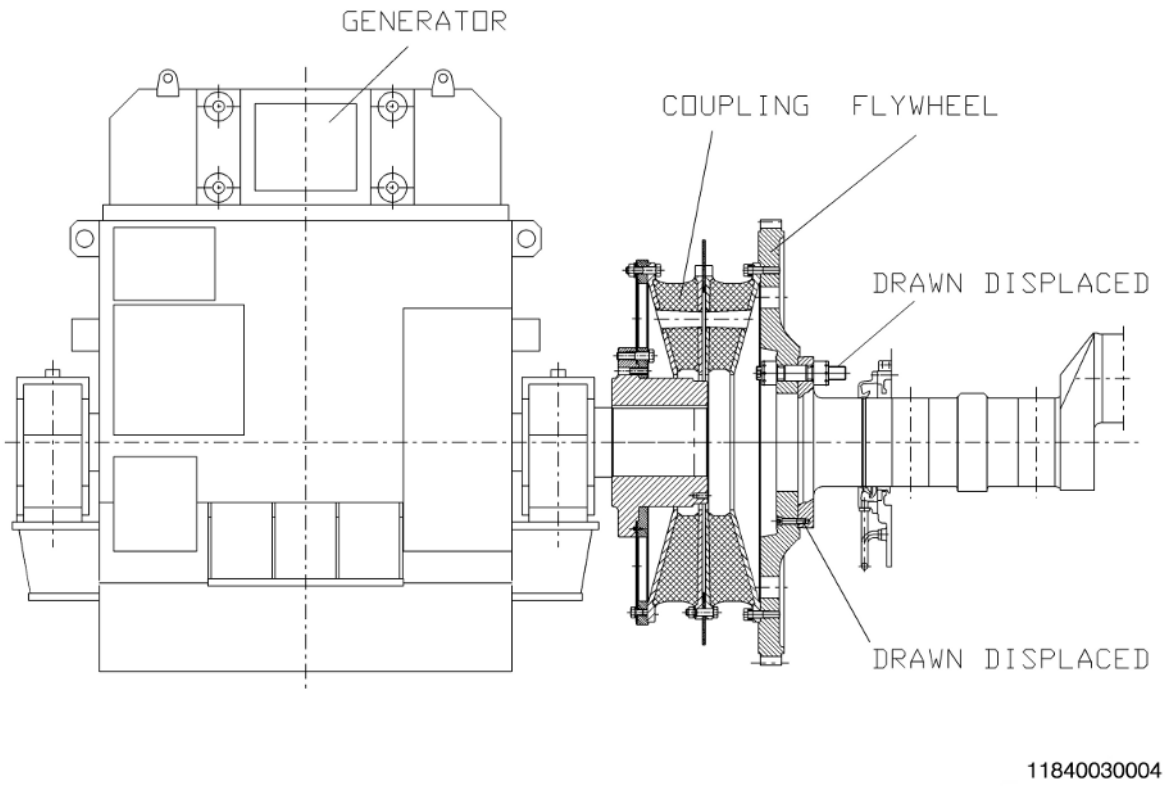

Figure 63: Example for an arrangement of flywheel, coupling and alternator

2.32 Arrangement of attached pumps

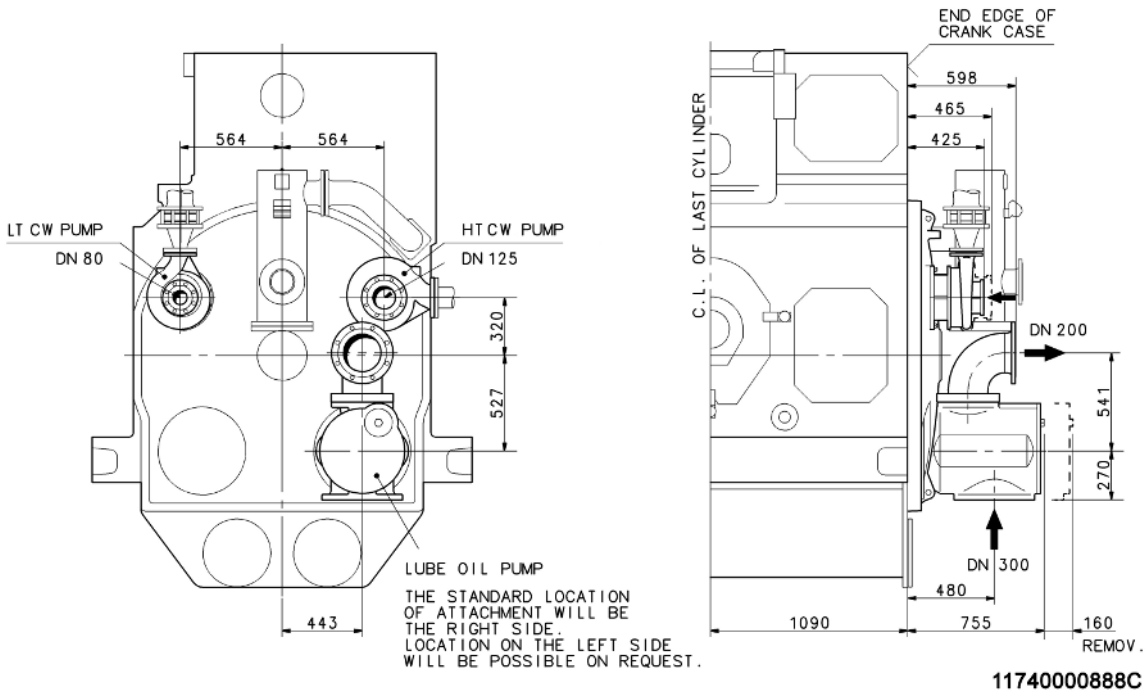

Figure 64: Attached pumps L engine

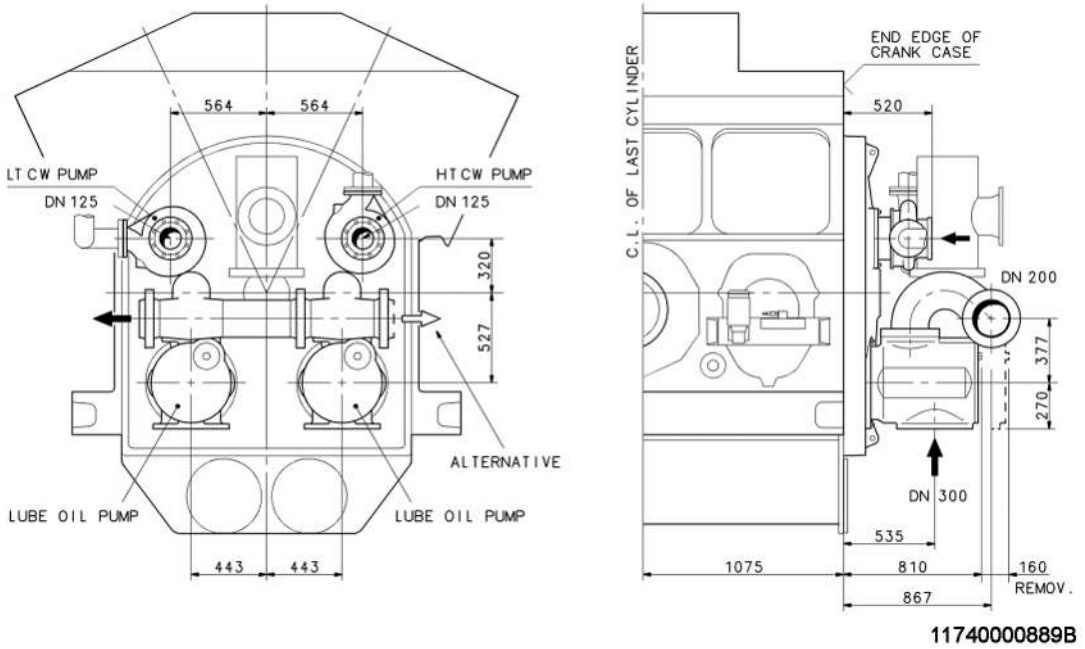

Figure 65: Attached pumps V engine

Note:
The final arrangement of the lube oil and cooling water pumps will be made at inquiry or order.

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2.33 Foundation

2.33.1 General requirements for engine foundation

Plate thicknesses

The stated material dimensions are recommendations, calculated for steel plates. Thicknesses smaller than these are not permissible. When using other materials (e.g. aluminium), a sufficient margin has to be added.

Top plates

Before or after having been welded in place, the bearing surfaces should be machined and freed from rolling scale. Surface finish corresponding to Ra 3.2 peak-to-valley roughness in the area of the chocks shall be accomplished.

The thickness given is the finished size after machining.

Downward inclination outwards, not exceeding 0.7 %.

Prior to fitting the chocks, clean the bearing surfaces from dirt and rust that may have formed. After the drilling of the foundation bolt holes, spotface the lower contact face normal to the bolt hole.

Foundation girders

The distance of the inner girders must be observed. We recommend that the distance of the outer girders (only required for larger types) is observed as well.

The girders must be aligned exactly above and underneath the tank top.

Floor plates

No manholes are permitted in the floor plates in the area of the box-shaped foundation. Welding is to be carried out through the manholes in the outer girders.

Top plate supporting

Provide support in the area of the frames from the nearest girder below.

Dynamic foundation requirements

The eigenfrequencies of the foundation and the supporting structures, including GenSet weight (without engine) shall be higher than 20 Hz. Occasionally, even higher foundation eigenfrequencies are required. For further information refer to section Noise and vibration – Impact on foundation, Page 168.

Note:
The requirements of the respective classification society have to be considered in addition.

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2.33.2 Rigid seating

L engine

Recommended configuration of foundation

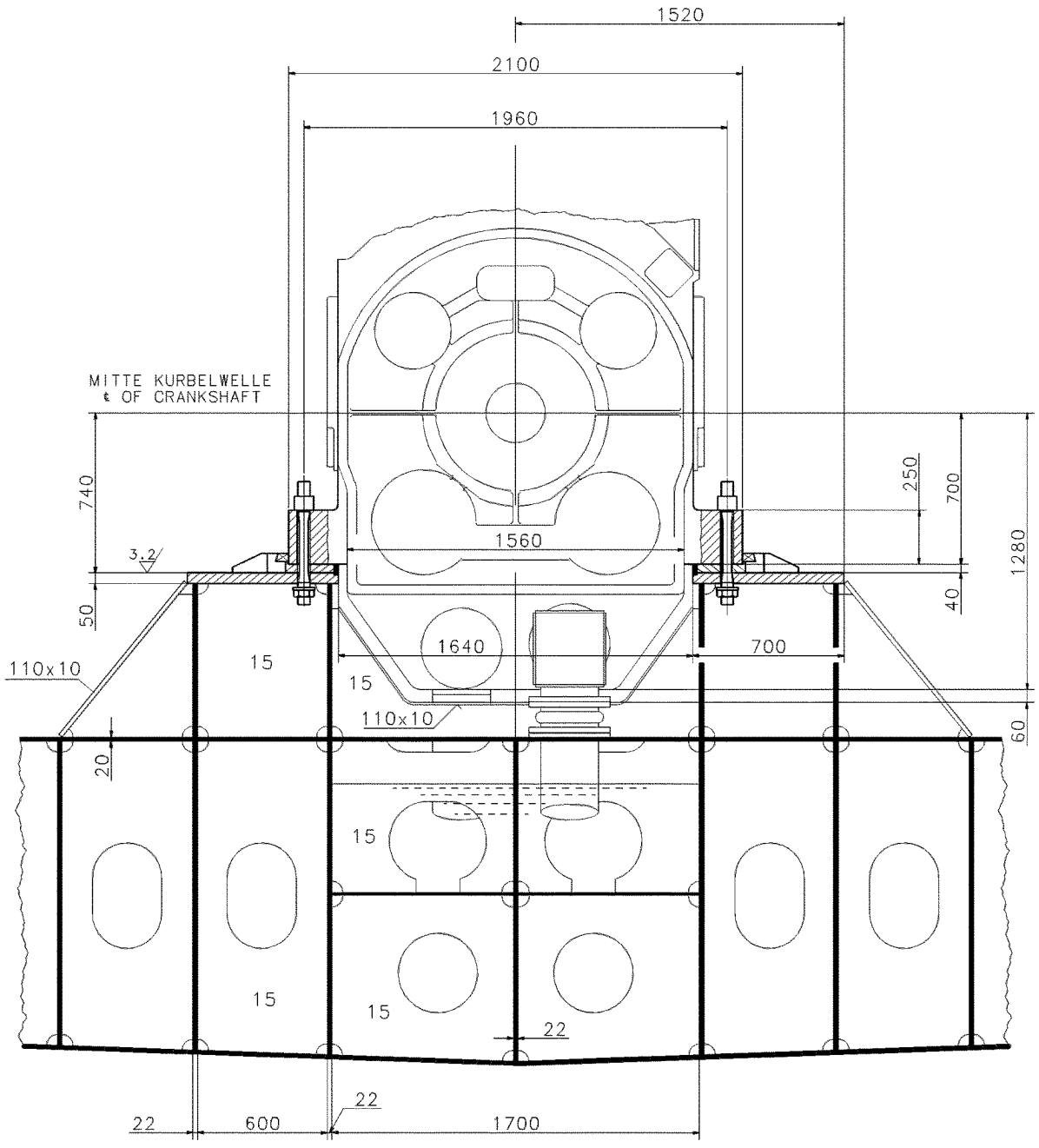

Figure 66: Recommended configuration of foundation L engine

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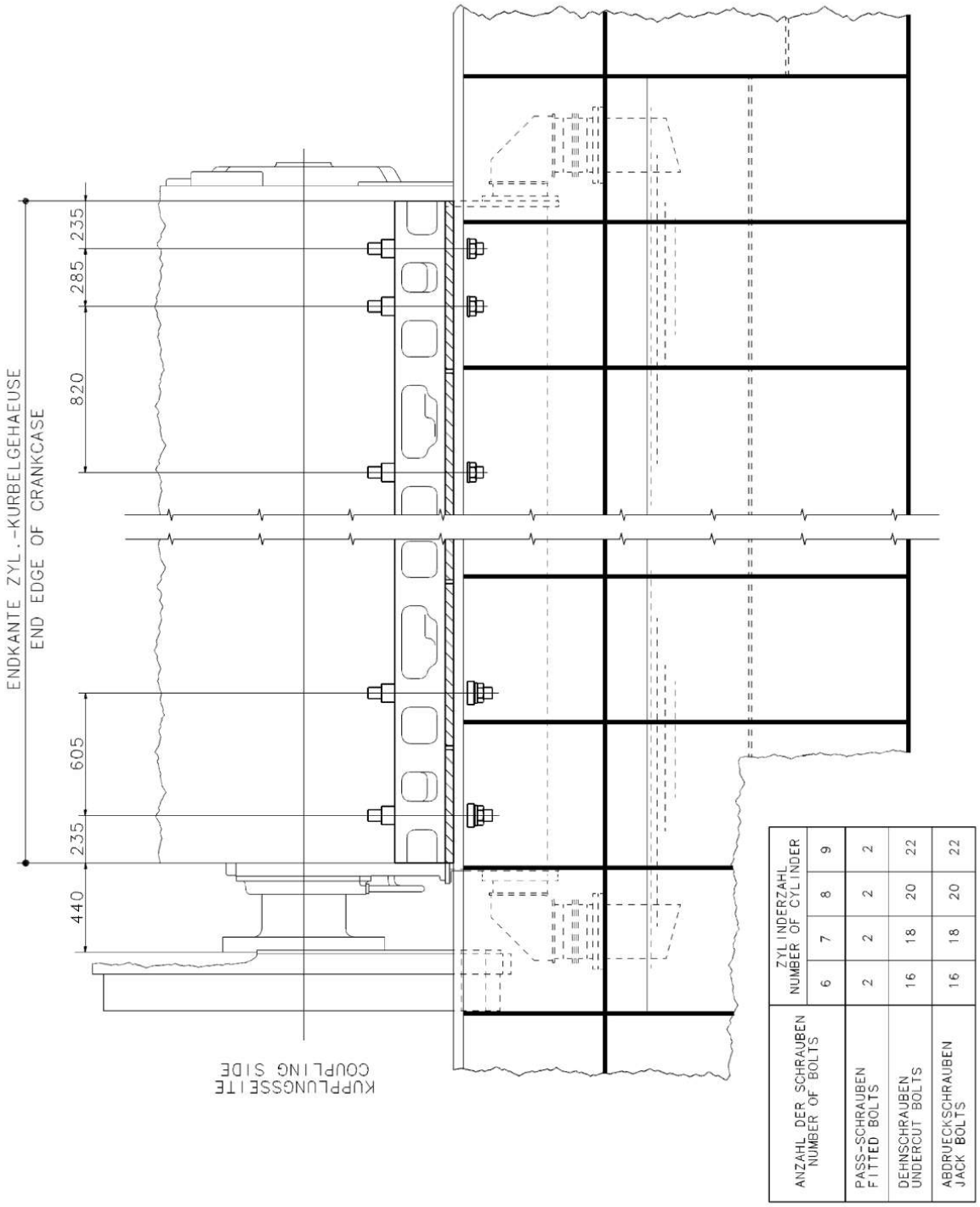
Recommended configuration of foundation – Number of bolts



ANZAHL DER SCHRAUBEN NUMBER OF BOLTS	ZYLINDERZAHL NUMBER OF CYLINDER			
	6	7	8	9
PASS-SCHRAUBEN FITTED BOLTS	2	2	2	2
DEHNSCHRAUBEN UNDERCUT BOLTS	16	18	20	22
ABDRUECKSCHRAUBEN JACK BOLTS	16	18	20	22

Figure 67: Recommended configuration of foundation L engine – Number of bolts

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Arrangement of foundation bolt holes

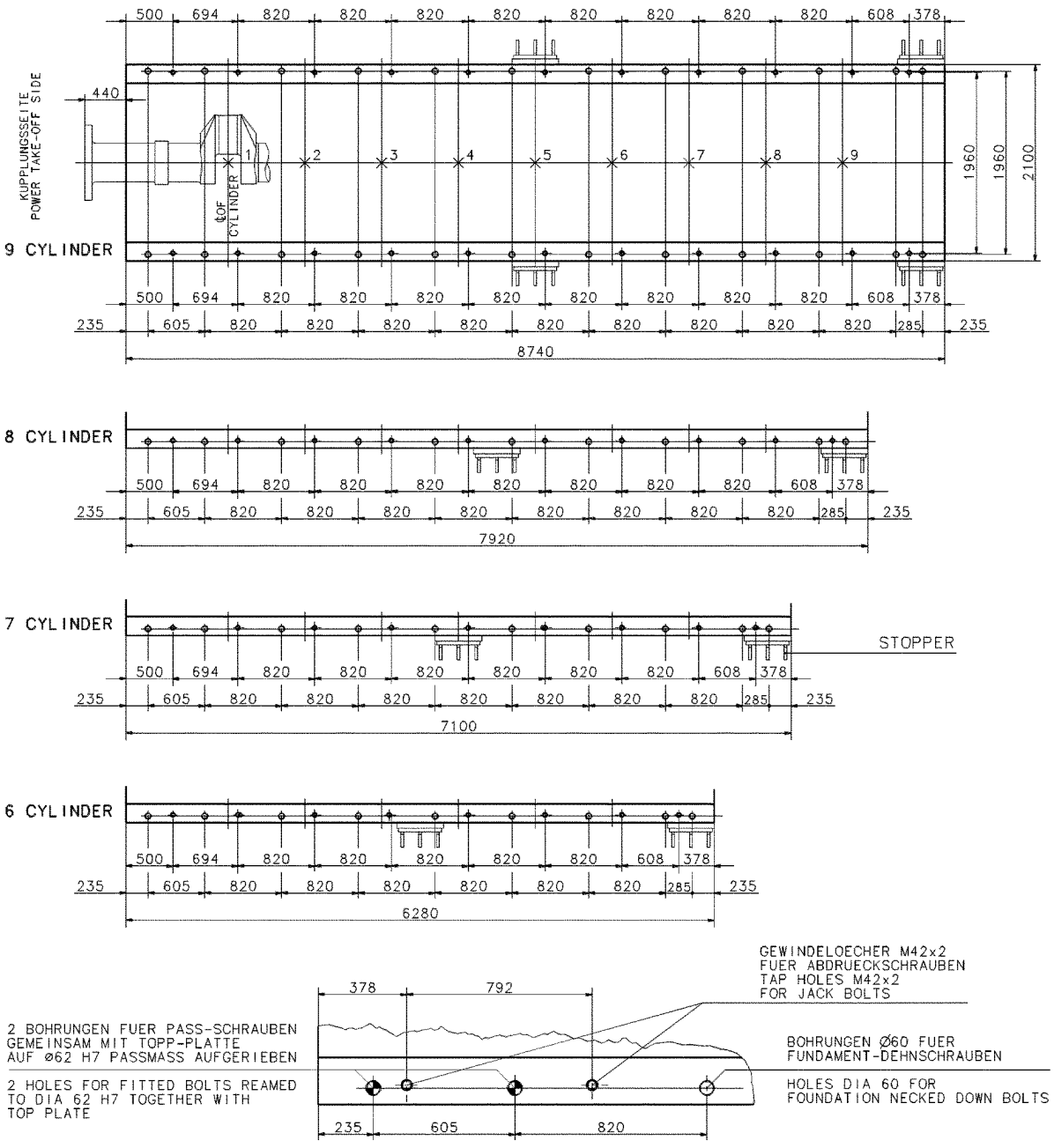

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Figure 68: Arrangement of foundation bolt holes L engine

Two **fitted bolts** have to be provided either on starboard side or portside.

In any case they have to be positioned on the coupling side.

Number and position of the **stoppers** have to be provided according to the figure above.

12V, 14V, 16V engine

Recommended configuration of foundation

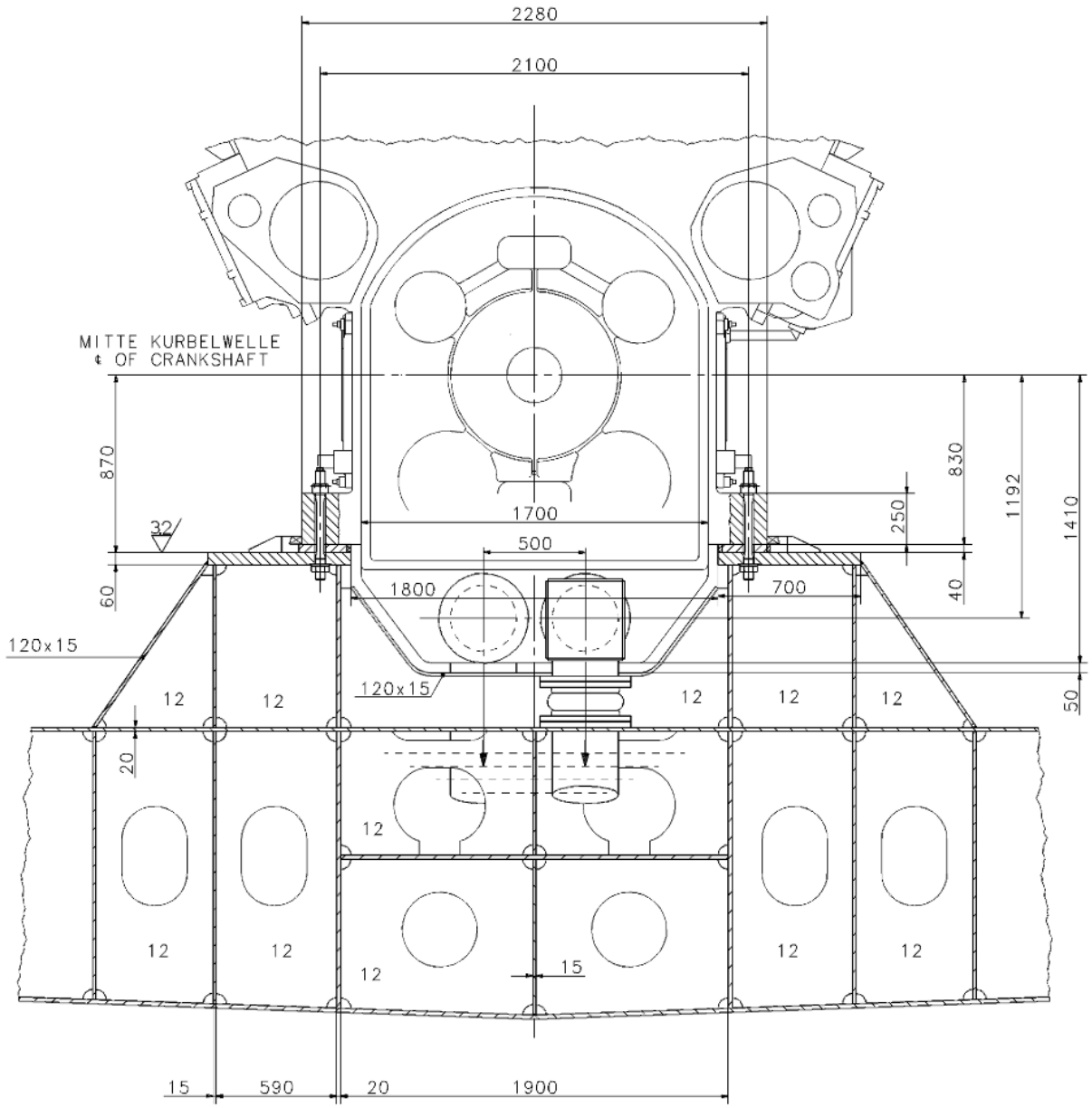

Figure 69: Recommended configuration of foundation 12V, 14V, 16V engine

V engine

Recommended configuration of foundation – Number of bolts

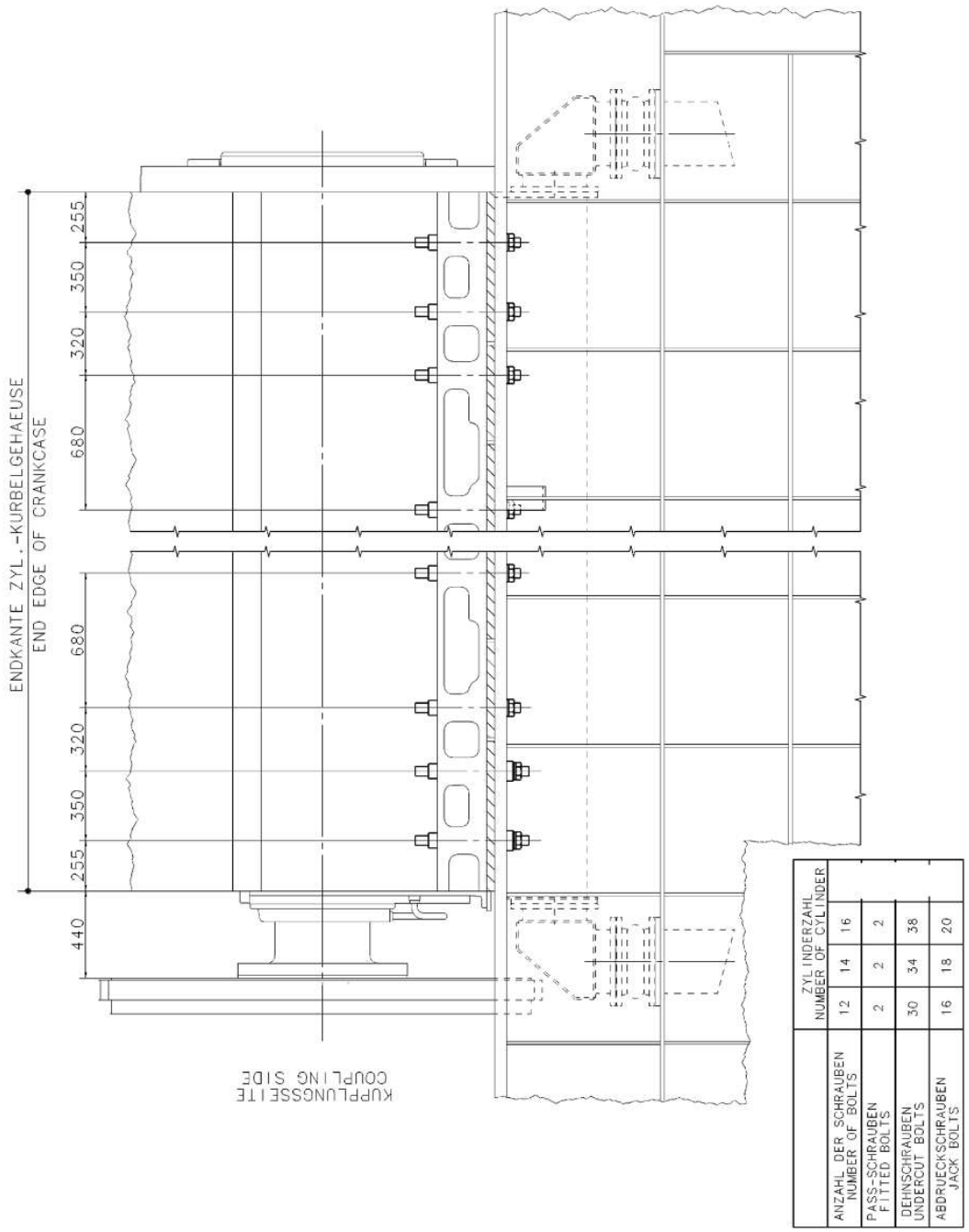

Figure 70: Recommended configuration of foundation V engine – Number of bolts

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Arrangement of foundation bolt holes

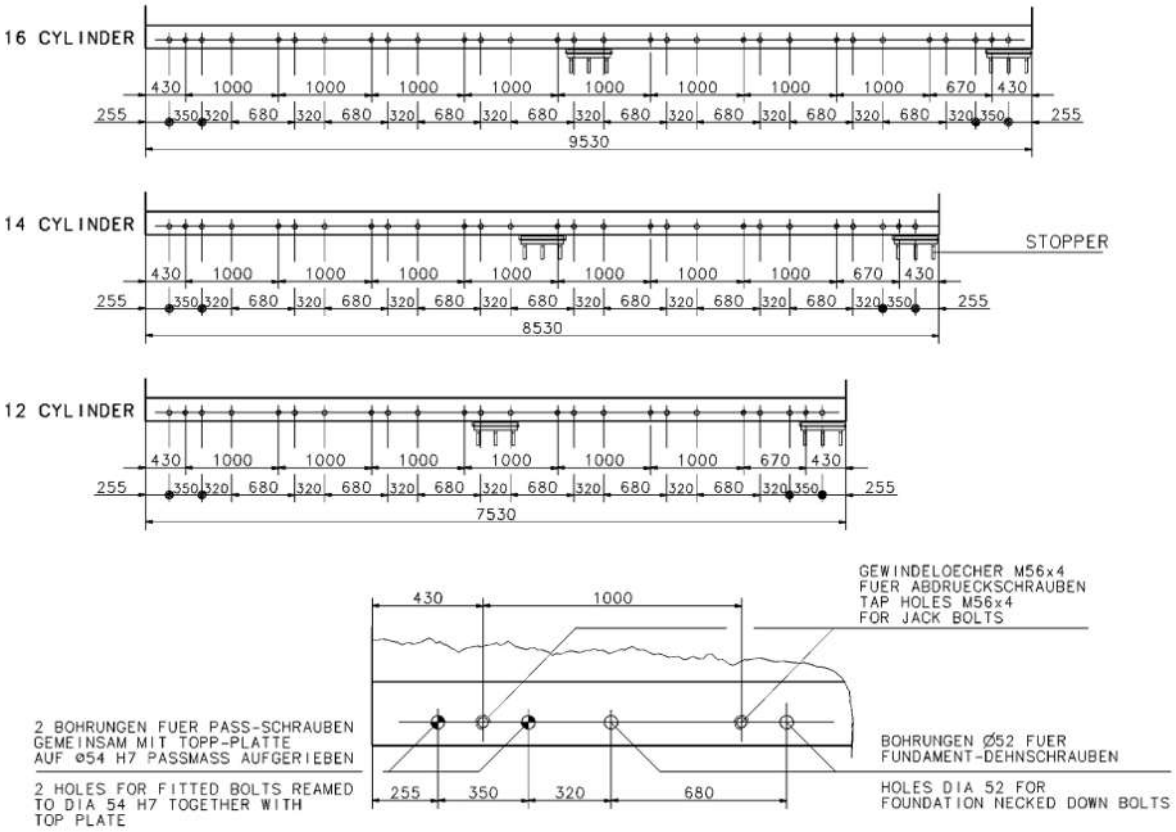

Figure 71: Arrangement of foundation bolt holes V engine

Two **fitted bolts** have to be provided either on starboard side or portside.

In any case they have to be positioned on the coupling side.

Number and position of the **stoppers** have to be provided according to the figure above.

2.33.3 Chocking with synthetic resin

Most classification societies permit the use of the following synthetic resins for chocking diesel engines:

- Chockfast Orange
  (Philadelphia Resins Corp. U.S.A)
- Epocast 36
  (H.A. Springer, Kiel)

MAN Energy Solutions accepts engines being chocked with synthetic resin provided:

- If processing is done by authorised agents of the above companies.
- If the classification society responsible has approved the synthetic resin to be used for a unit pressure (engine weight + foundation bolt preloading) of 450 N/cm^2 and a chock temperature of at least 80 °C.

The loaded area of the chocks must be dimensioned in a way, that the pressure effected by the engines dead weight does not exceed 70 N/cm^2 (requirement of some classification societies).

The pretensioning force of the foundation bolts was chosen so that the permissible total surface area load of 450 N/cm^2 is not exceeded. This will ensure that the horizontal thrust resulting from the mass forces is safely transmitted by the chocks.

The shipyard is responsible for the execution and must also grant the warranty.

Tightening of the foundation bolts only permissible with hydraulic tensioning device. Nuts definitely must not be tightened with hook spanner and hammer, even for later inspections.

Tightening of foundation bolts

The tensioning tools with tensioning nut and pressure sleeve are included in the standard scope of supply of tools for the engine.

Dedicated installation values (e.g. pre-tensioning forces) will be given in the customer documentation specific to each project.

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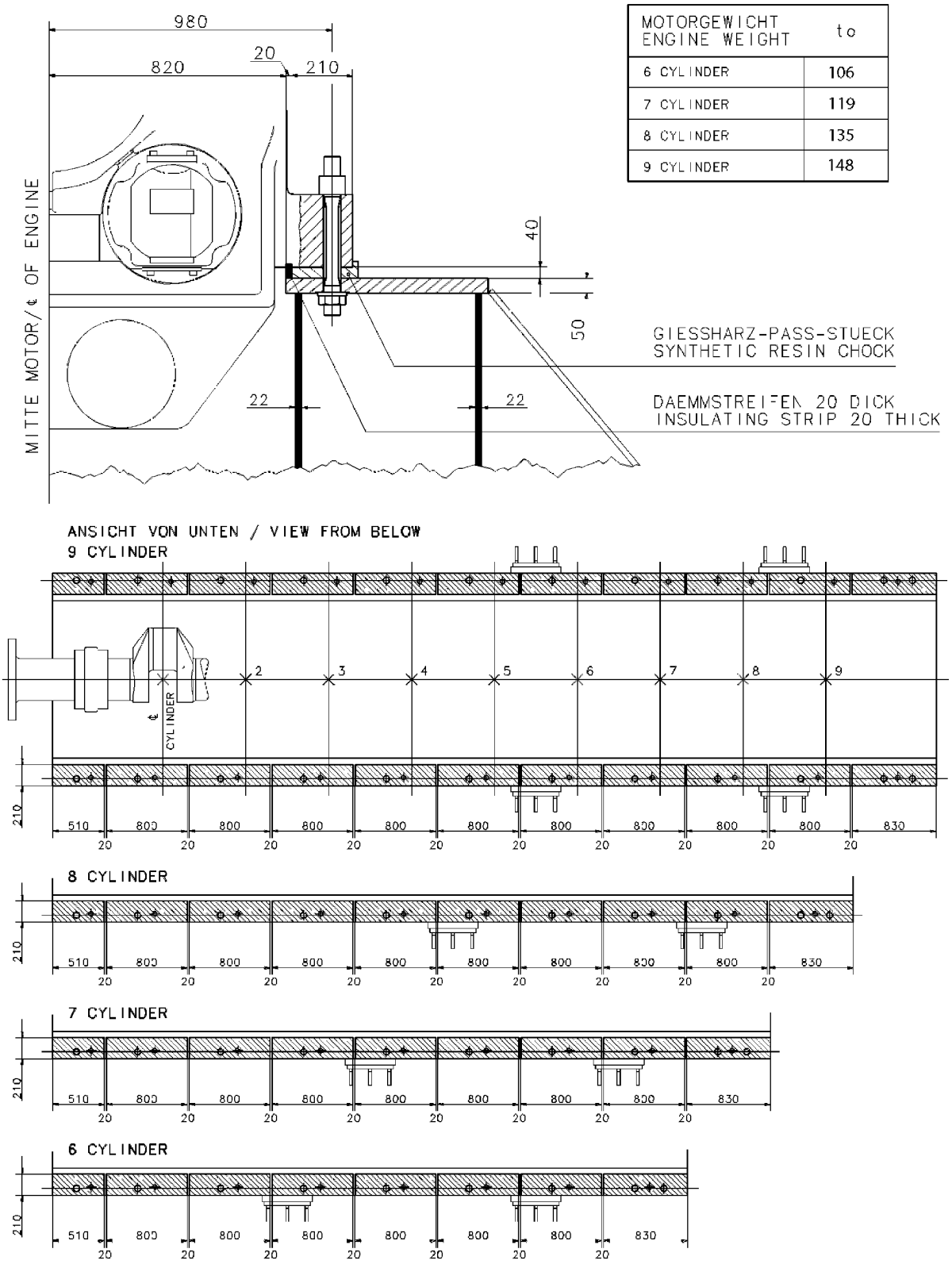

Figure 72: Chocking with synthetic resin MAN L51/60DF

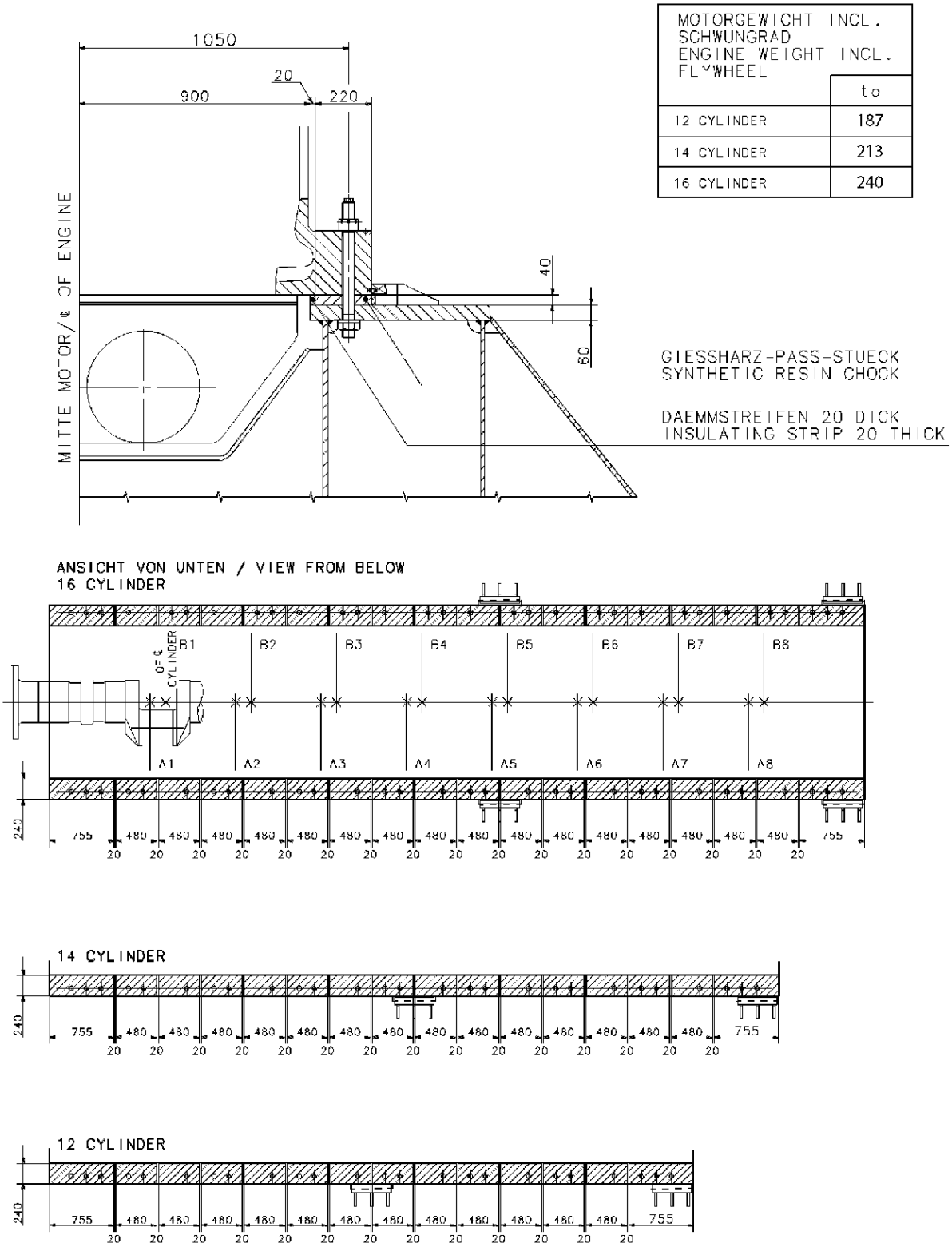

Figure 73: Chocking with synthetic resin MAN 12V, 14V, 16V51/60DF

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2.33.4 Resilient seating

General

The vibration of the engine causes dynamic effects on the foundation. These effects are attributed to the pulsating reaction forces due to the fluctuating torque. Additionally, in engines with certain cylinder numbers these effects are increased by unbalanced forces and couples brought about by rotating or reciprocating masses which – considering their vector sum – do not equate to zero.

The direct resilient support makes it possible to reduce the dynamic forces acting on the foundation, which are generated by every reciprocating engine and may – under adverse conditions – have harmful effects on the environment of the engine.

With respect to large engines (bore > 400 mm) MAN Energy Solutions offers two different versions of the resilient mounting (one using conical – the other inclined sandwich elements).

The inclined resilient mounting was developed especially for ships with high comfort demands, e.g. passenger ferries and cruise vessels. This mounting system is characterised by natural frequencies of the resiliently supported engine being lower than approximately 7 Hz. The resonances are located away from the excitation frequencies related to operation at nominal speed.

For average demands of comfort, e.g. for merchant ships, and for smaller engines (bore < 400 mm) mountings using conical mounts can be judged as being fully sufficient. Because of the stiffer design of the elements the natural frequencies of the system are significantly higher than in case of the inclined resilient mounting. The natural frequencies of engines mounted with this kind of mounts are lower than approximately 18 Hz. The vibration isolation is thus of lower quality. It is however, still considerably better than a rigid or semi resilient engine support.

The appropriate design of the resilient support will be selected in accordance with the demands of the customer, i.e. it will be adjusted to the special requirements of each plant.

In both versions the supporting elements will be connected directly to the engine feet by special brackets.

The number, rubber hardness and distribution of the supporting elements depend on:

- The weight of the engine
- The centre of gravity of the engine
- The desired natural frequencies

Where resilient mounting is applied, the following has to be taken into consideration when designing a propulsion plant:

- Resilient mountings always feature several resonances resulting from the natural mounting frequencies. In spite of the endeavour to keep resonances as far as possible from nominal speed the lower bound of the speed range free from resonances will rarely be lower than 70 % of nominal speed for mountings using inclined mounts and rarely lower than 85 % for mountings using conical mounts. It must be pointed out that these percentages are only guide values. The speed interval being free from resonances may be larger or smaller. These restrictions in speed will mostly require the deployment of a controllable pitch propeller.

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- Between the resiliently mounted engine and the rigidly mounted gearbox or alternator, a flexible coupling with minimum axial and radial elastic forces and large axial and radial displacement capacities should be provided.
- The media connections (compensators) to and from the engine must be highly flexible whereas the fixations of the compensators on the one hand with the engine and on the other hand with the environment must be realised as stiff as possible.
- For the inclined resilient support, provision for stopper elements has to be made because of the sea-state-related movement of the vessel. In the case of conical mounting, these stoppers are integrated in the element.
- In order to achieve a good vibration isolation, the lower brackets used to connect the supporting elements with the ship's foundation are to be fitted at sufficiently rigid points of the foundation. Influences of the foundation's stiffness on the natural frequencies of the resilient support of the engine will not be considered in the mounting design calculation.
- The yard must specify with which inclination related to the plane keel the engine will be installed in the ship. The inclination must be defined and communicated before entering the dimensioning process.

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2.33.5 Recommended configuration of foundation

Engine mounting using inclined sandwich elements

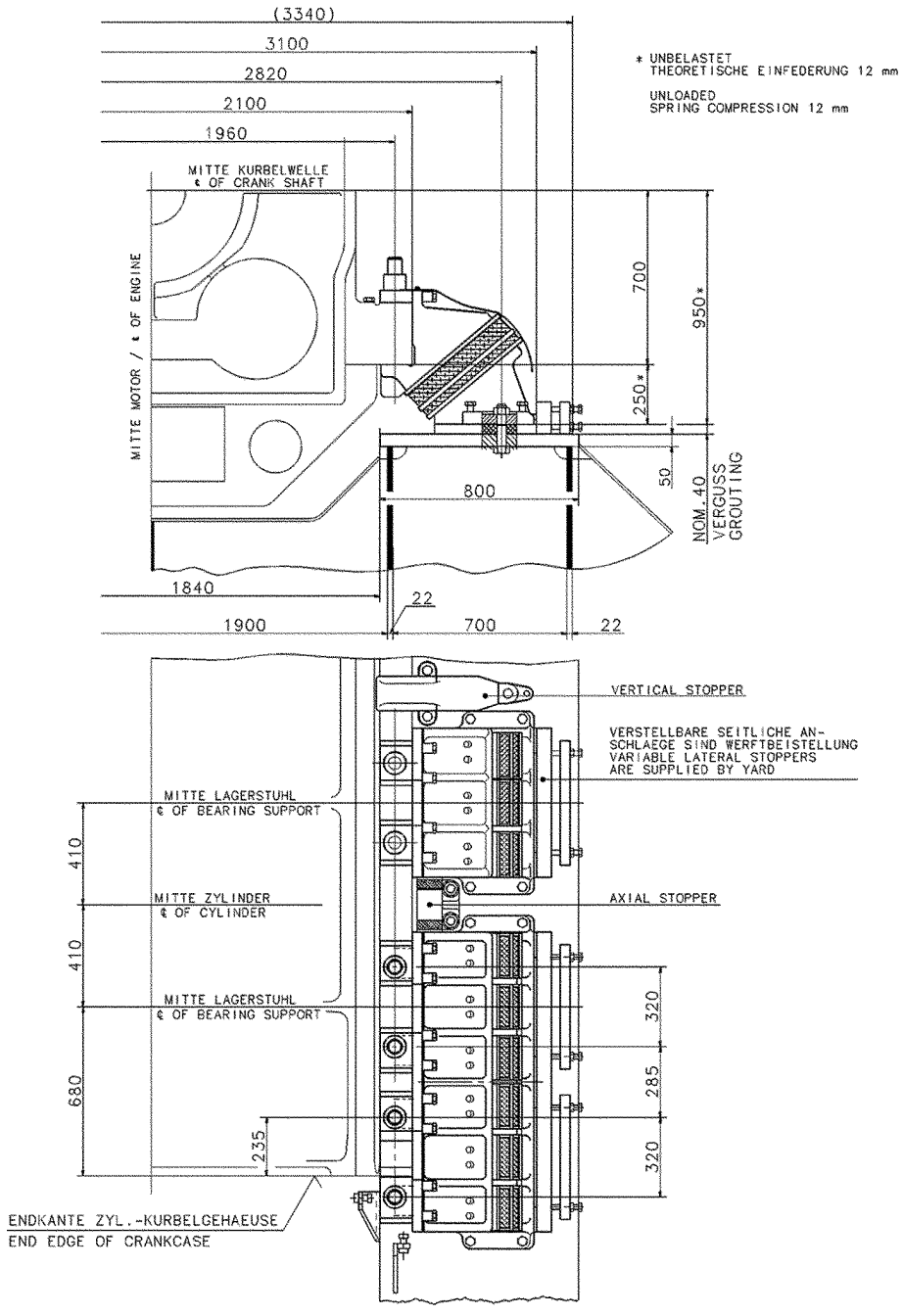

Figure 74: Recommended configuration of foundation – L engine, resilient seating 1

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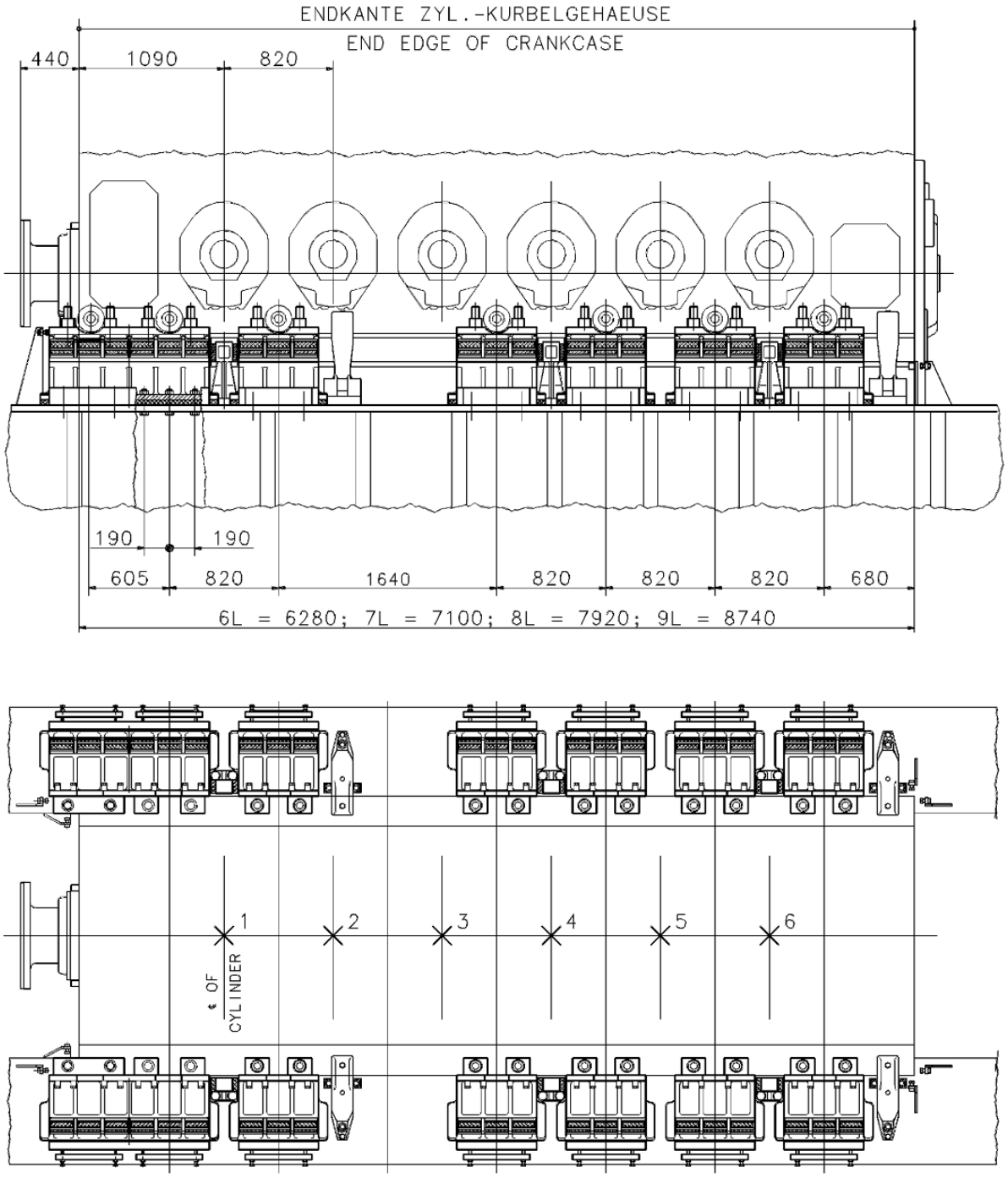

Figure 75: Recommended configuration of foundation – L engine, resilient seating 2

12V, 14V and 16V engine

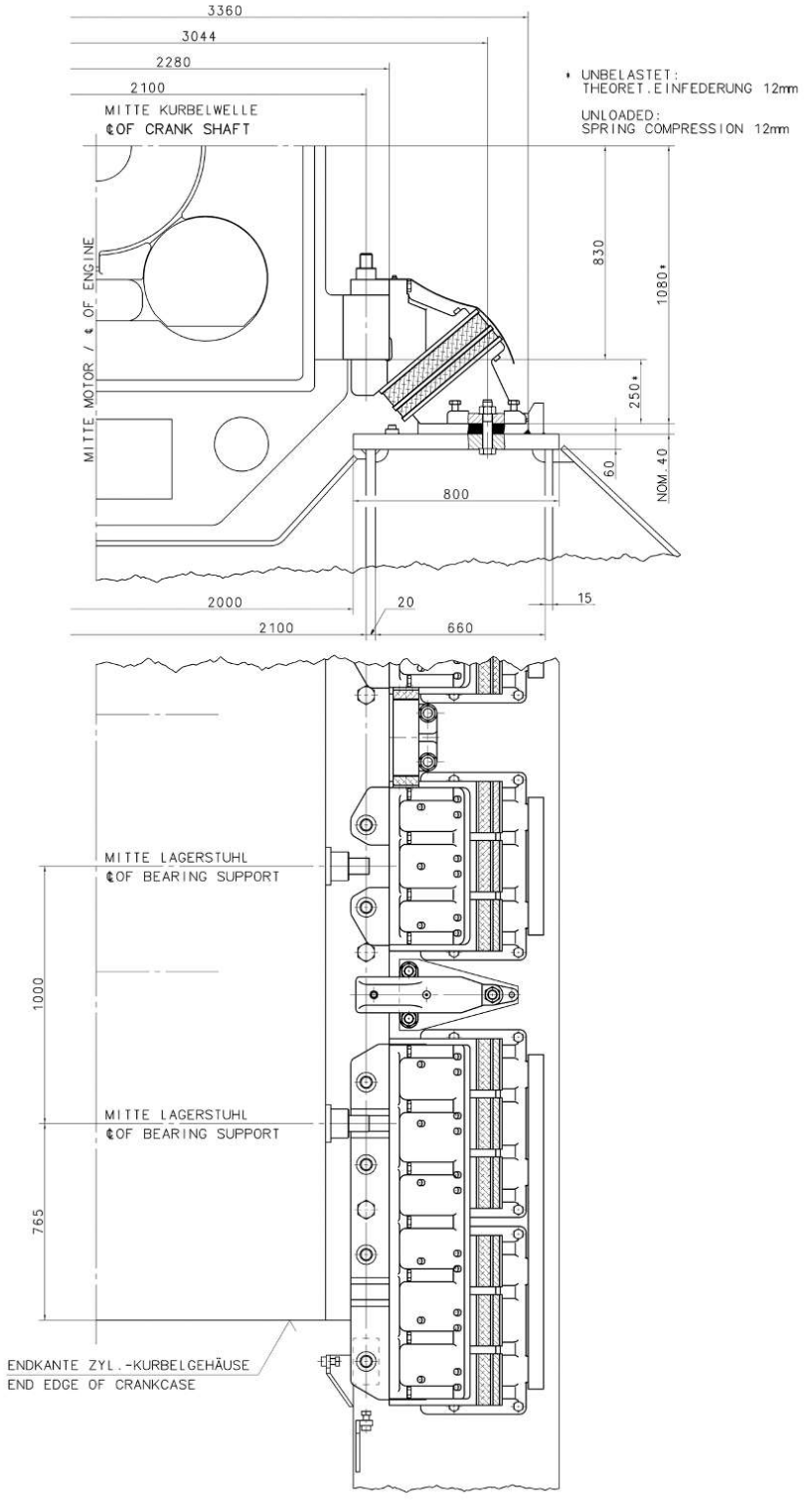

Figure 76: Recommended configuration of foundation – 12V, 14V and 16V engine, resilient seating

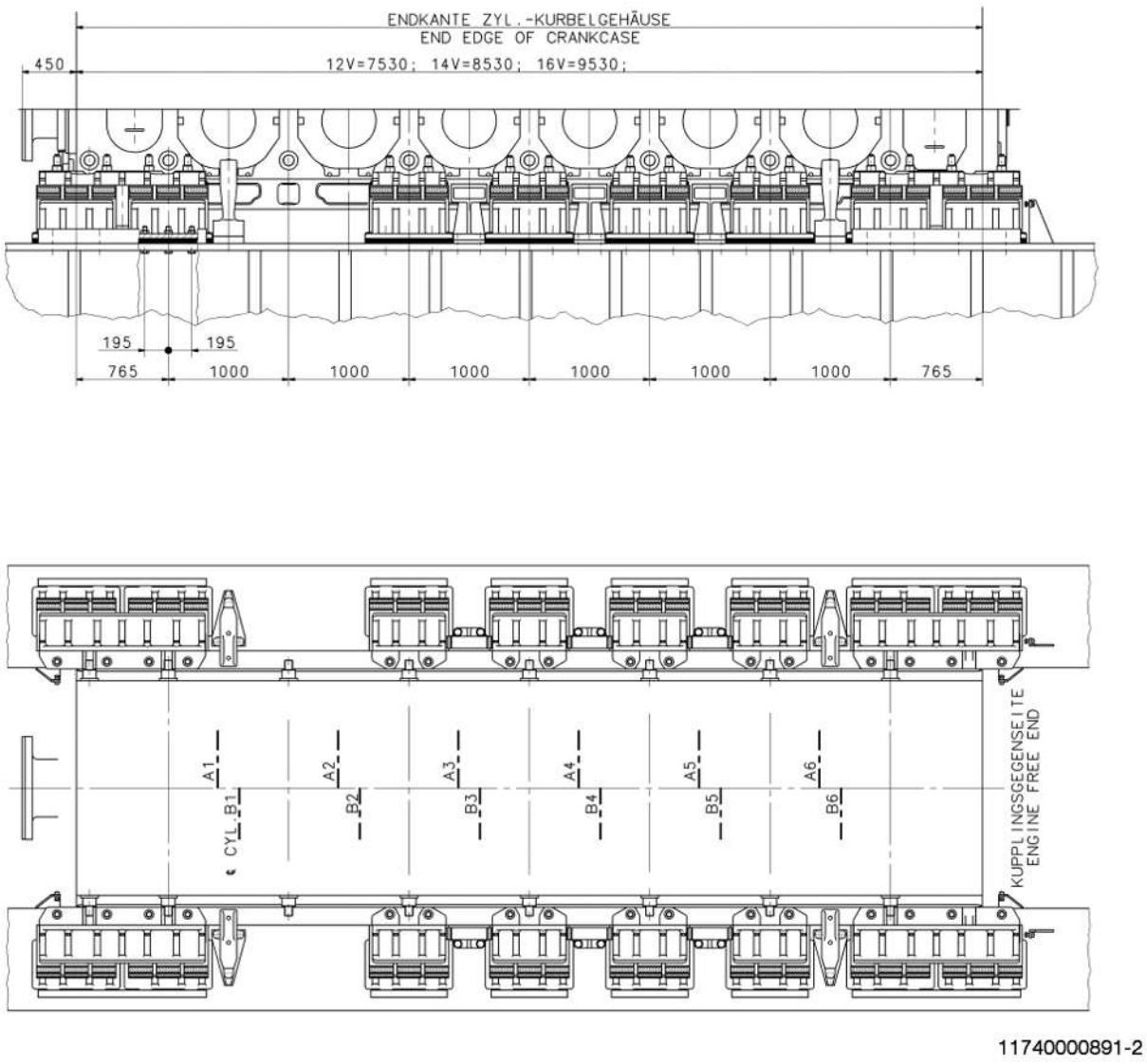

Figure 77: Recommended configuration of foundation – V engine, resilient seating

Engine mounting using conical mounts

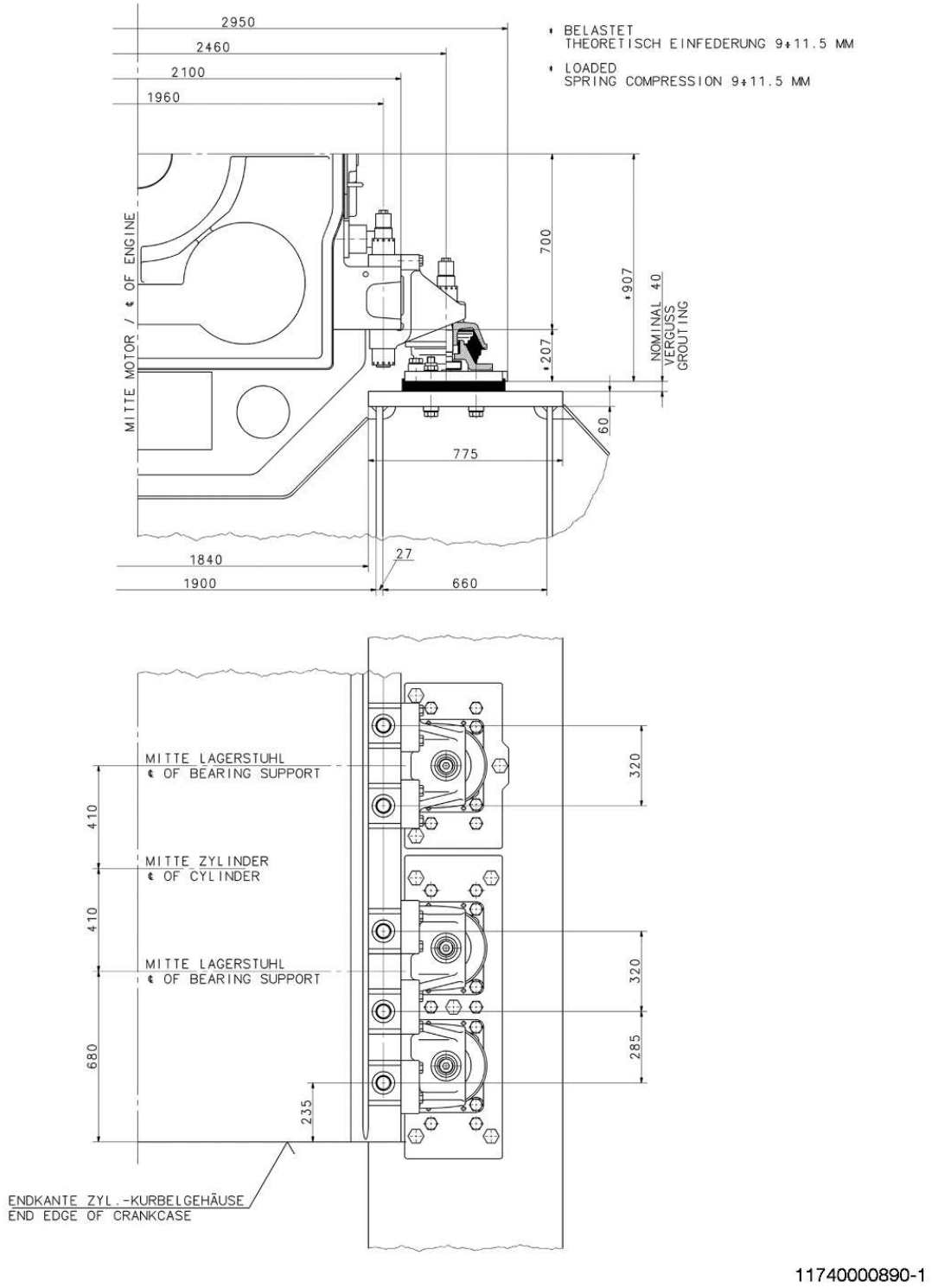

Figure 78: Recommended configuration of foundation – L engine, resilient seating 1

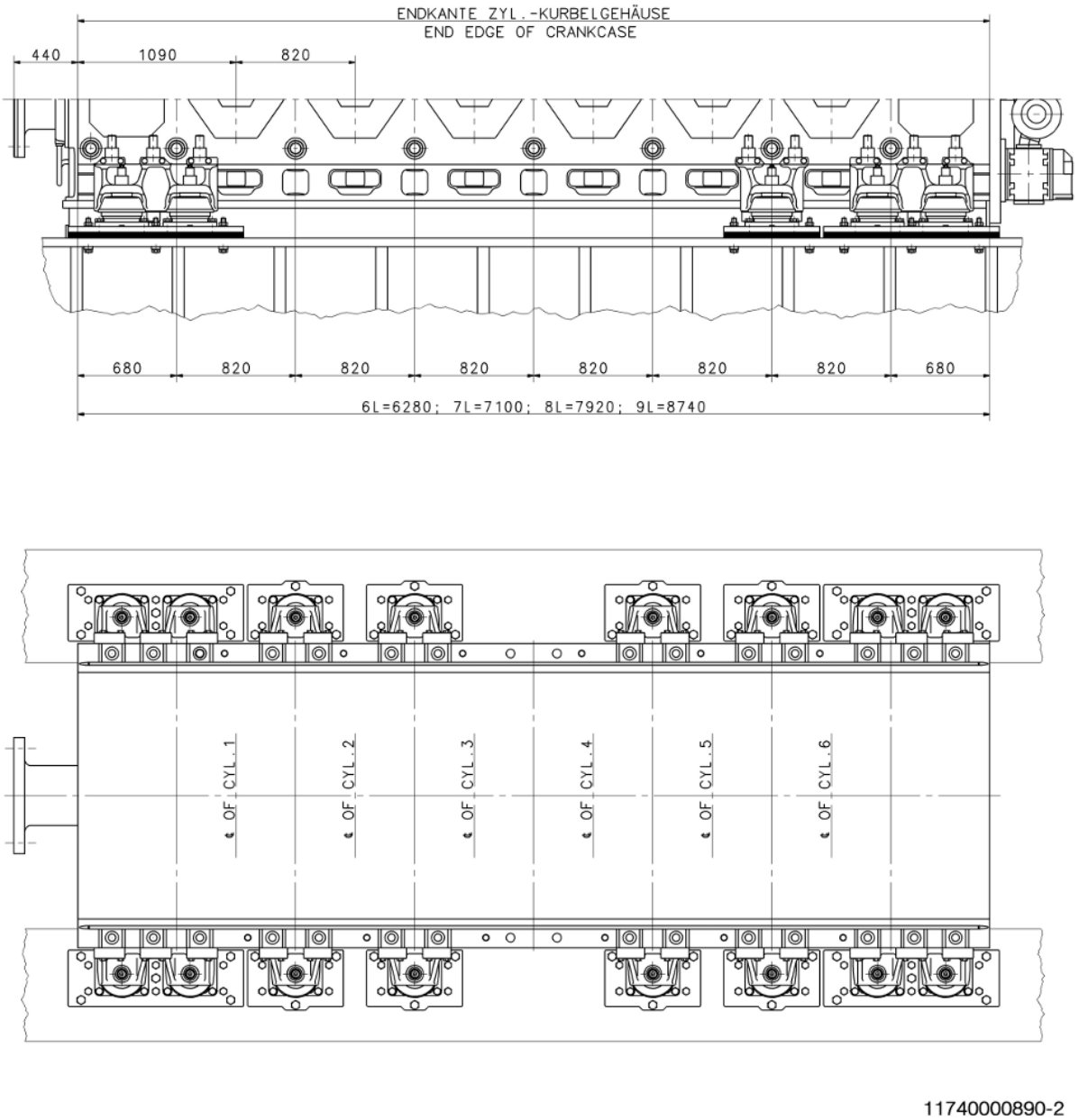

Figure 79: Recommended configuration of foundation – L engine, resilient seating 2

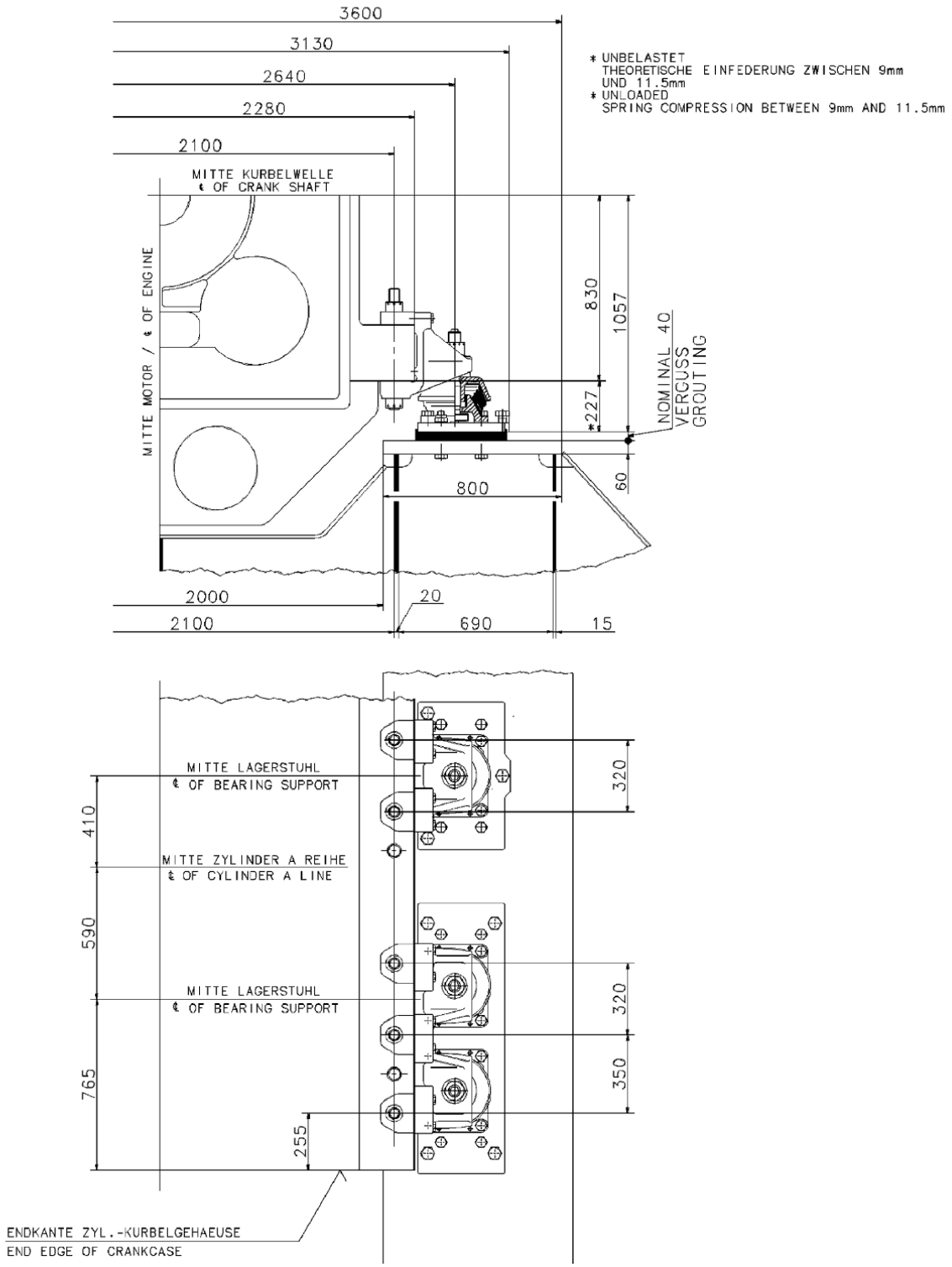

Figure 80: Recommended configuration of foundation V engine – Resilient seating 1

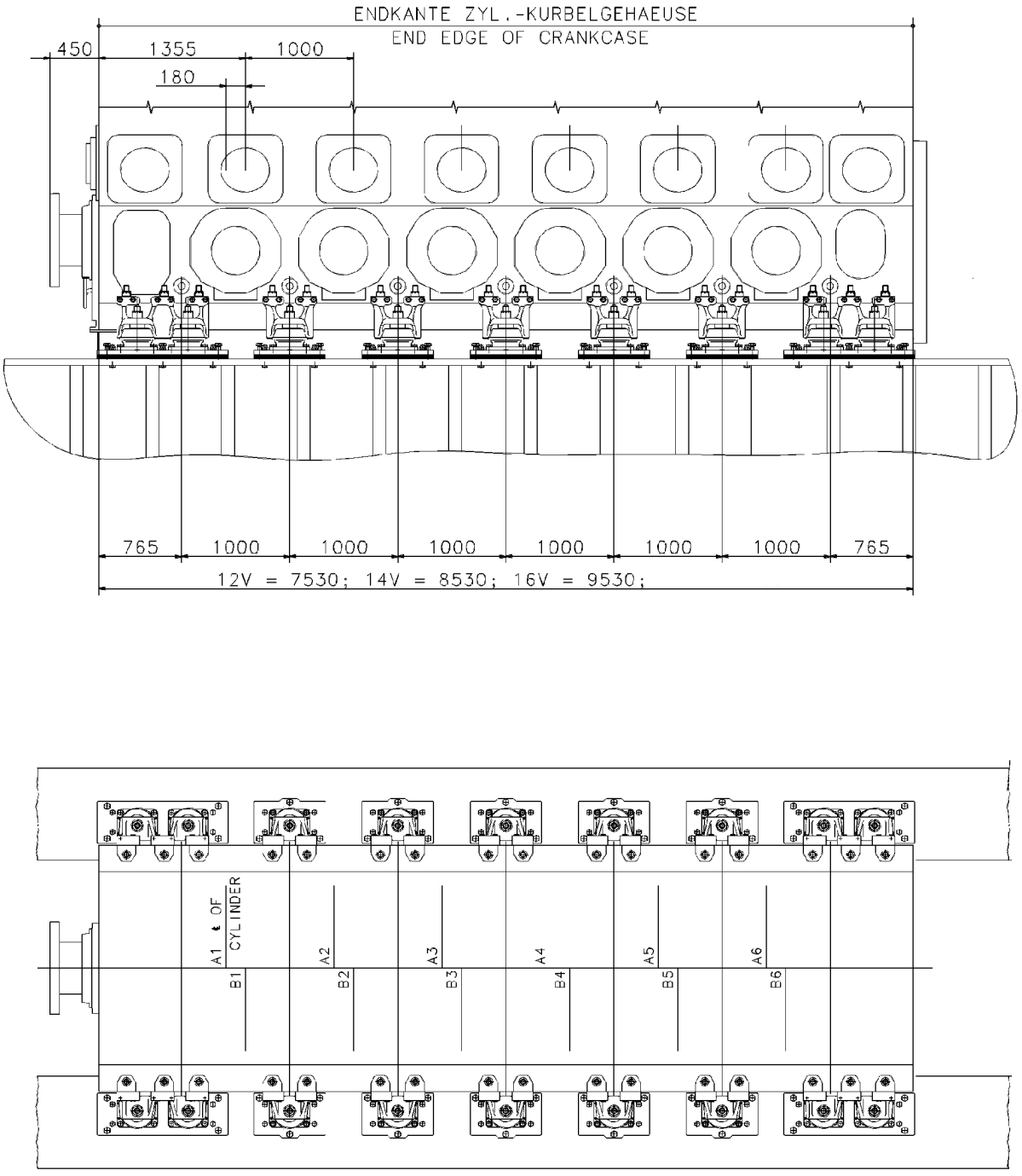

Figure 81: Recommended configuration of foundation V engine – Resilient seating 2

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2.33.6 Engine alignment

The alignment of the engine to the attached power train is crucial for trouble-free operation.

Dependent on the plant installation influencing factors on the alignment might be:

- Thermal expansion of the foundations
- Thermal expansion of the engine, alternator or the gearbox
- Thermal expansion of the rubber elements in the case of resilient mounting
- The settling behaviour of the resilient mounting
- Shaft misalignment under pressure
- Necessary axial pre-tensioning of the flex-coupling

Therefore take care that a special alignment calculation, resulting in alignment tolerance limits will be carried out.

Follow the relevant working instructions of this specific engine type. Alignment tolerance limits must not be exceeded.

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3 Engine automation

3.1 SaCoSone system overview

The monitoring and safety system SaCoSone is responsible for complete engine operation, control, alarming and safety. All sensors and operating devices are wired to the engine-attached units. The interface to the plant is done by means of an Interface Cabinet.

During engine installation, only the bus connections, the power supply and safety-related signal cables between the units/modules on engine and the cabinets are to be laid, as well as connections to external modules, electrical motors on the engine and parts on site.

The SaCoSone design is based on highly reliable and approved components as well as modules specially designed for installation on medium-speed engines. The used components are harmonised to an homogenous system. The system has already been tested and parameterised in the factory.

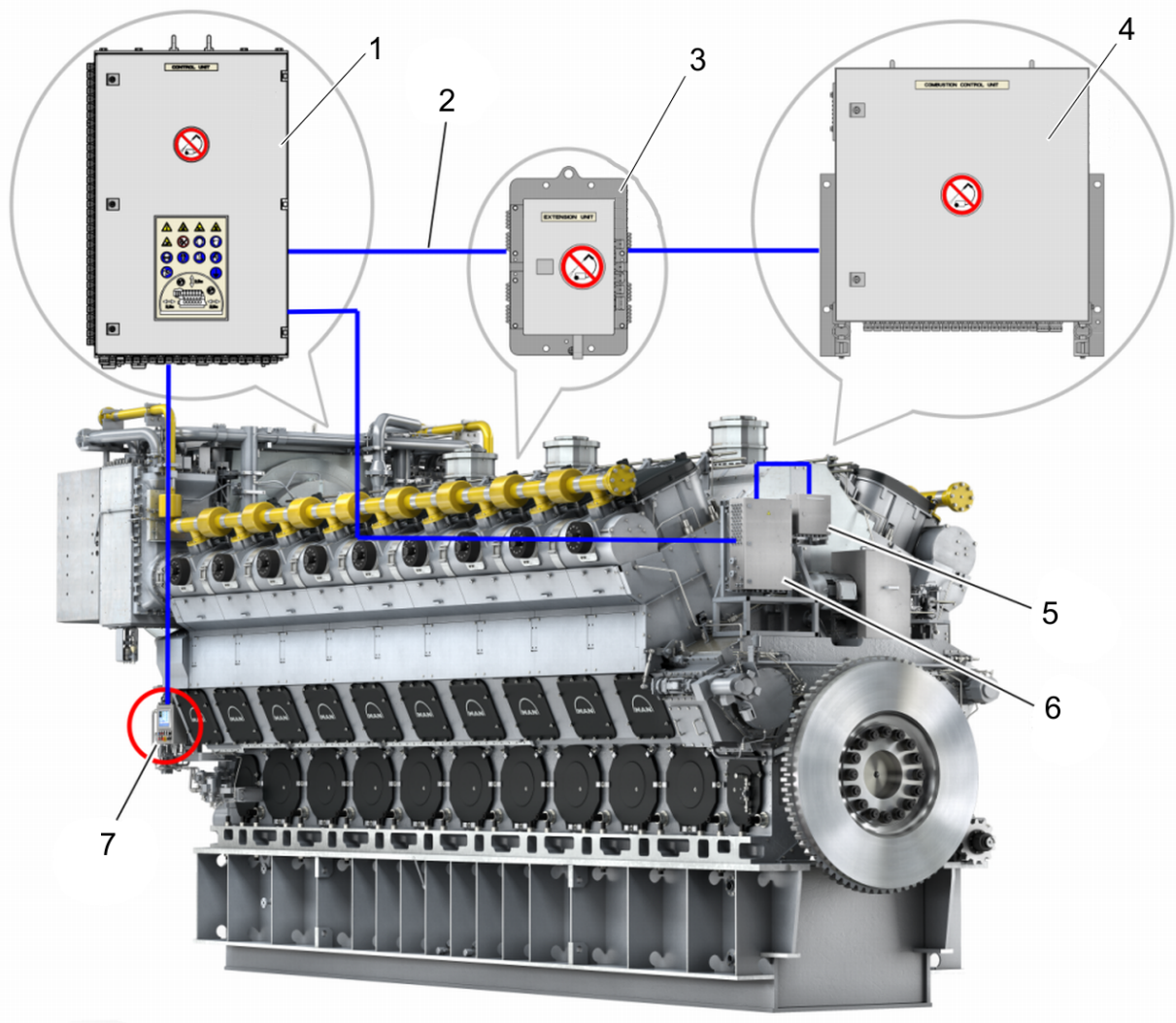

Figure 82: SaCoSone system overview

1	Control Unit	5	Knock Control Module

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2	System bus	6	Injection Unit
3	Extension Unit	7	Local operating Panel
4	Combustion Control Unit		

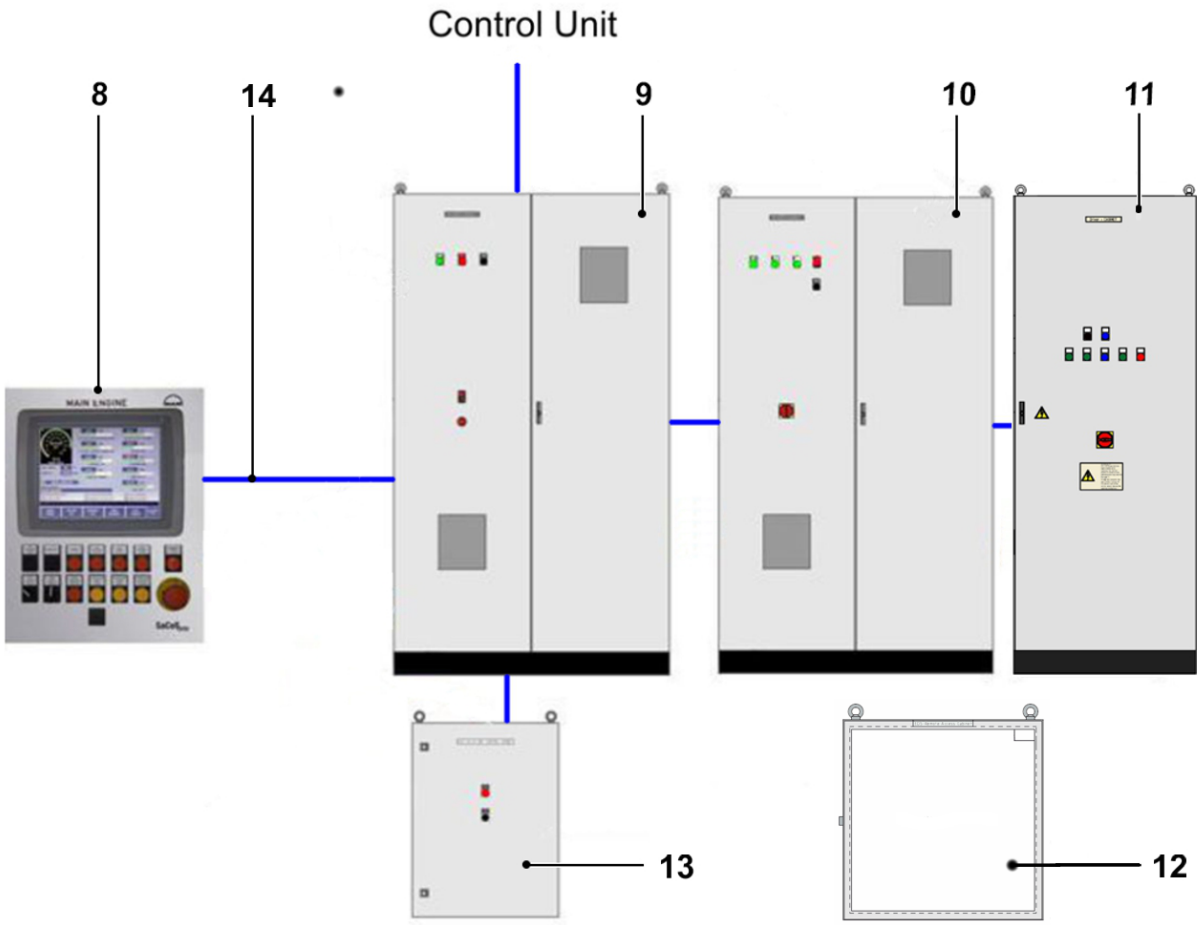

Figure 83: SaCoSone system overview

8	Remote Operating Panel (optional)	12	Data Logging Cabinet (optional)
9	Interface Cabinet (for stationary applications AC and IC are connected to each other as one unit)	13	Gas Valve Unit Cabinet
10	Auxiliary Cabinet (for stationary applications AC and IC are connected to each other as one unit)	14	System bus
11	Driver Cabinet (optional)	-	Synchro Device Cabinet (optional)

Control Unit

The Control Unit is mounted on the engine cushioned against any vibration. It includes two identical, highly integrated Control Modules: One for safety functions and the other one for engine control and alarming.

The modules work independently of each other and collect engine measuring data by means of separate sensors.

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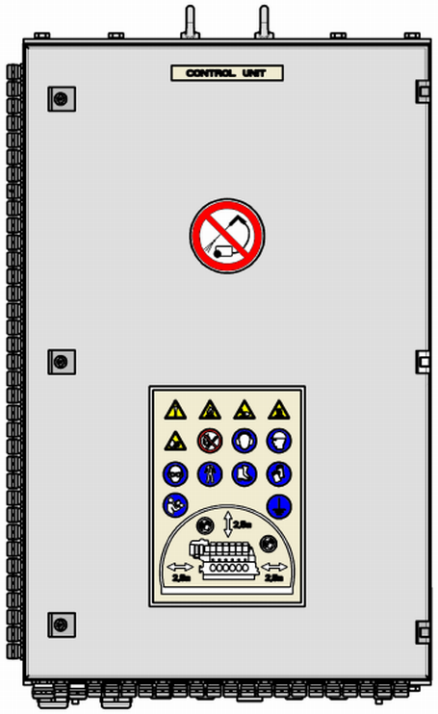

Figure 84: Control Unit

Injection Unit

The Injection Unit is mounted on the engine cushioned against any vibration. Depending on the usage of the engine, it includes one or two identical, highly integrated Injection Modules.

The Injection Module is used for speed control and for the actuation of the injection valves.

Injection Module I is used for L engines. At V engines it is used for bank A.

Injection Module II is used for bank B (only used for V engines).

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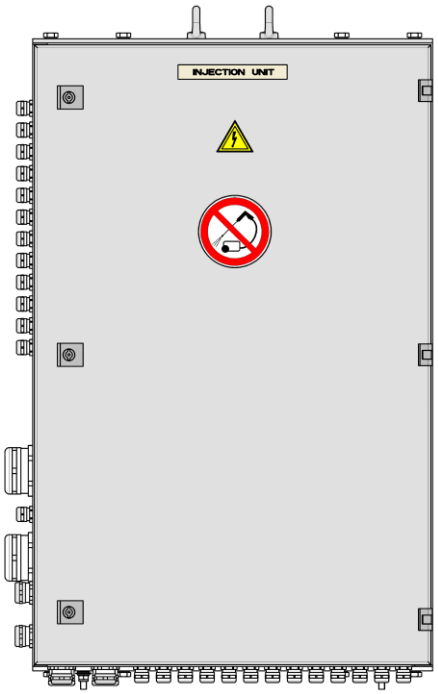

Figure 85: Injection Unit

Extension Unit

The Extension Unit provides additional I/O for the leakage monitoring sensors and the sensors of the Variable Valve Timing. The Extension Unit is mounted on the engine.

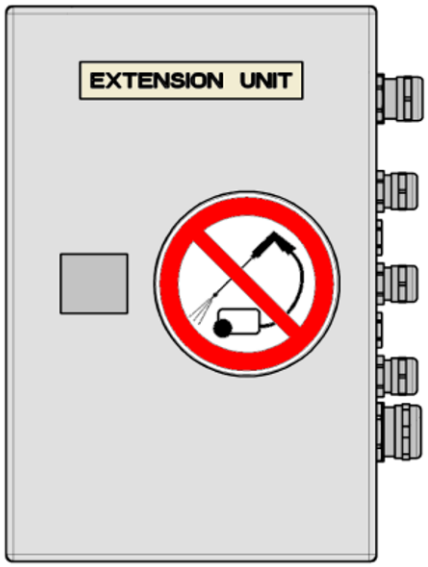

Figure 86: Extension Unit

Knock Control Module

For the purpose of knock recognition, a Knock Control Module is mounted on the engine and connected to the engine control via the CAN bus. The Knock Control Module is mainly designed for the connection of knocking sensors for engines with gas combustion. The module ensures decentralised knocking data analysing of each cylinder.

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Figure 87: Knock Control Module

Combustion Control Unit

The Combustion Control Unit (CCU) is mounted on the engine and it contains a Cylinder Pressure Module and a Gateway Module/Extension. The Cylinder Pressure Module logs the peak pressure of the cylinders via the pressure transmitter. Within the Gateway Module, an offset for the opening time of the injection valves is calculated for each respective cylinder in accordance with the logged cylinder pressures in order to keep the ignition pressure of all cylinders on an uniform level.

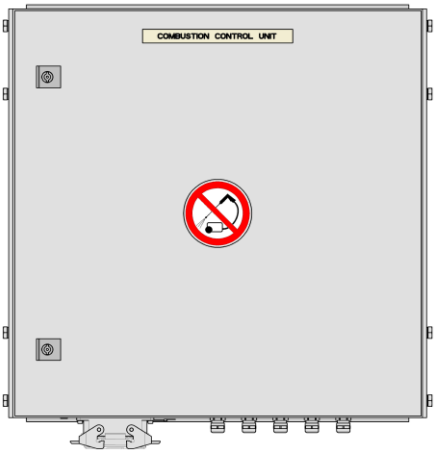

Figure 88: Combustion Control Unit

Interface Cabinet

The Interface Cabinet is a floor-standing cabinet, that optionally will be equipped with an air condition. The Interface Cabinet serves as a communication interface between SaCoSone, the overall plant control system and the supply system for the plant. The Interface Cabinet has two Gateway Modules, each of which has input and output channels as well as various interfaces for connecting automated plant/ship systems, ROP and PrimeServ Online Service.

The Interface Cabinet serves as a central connection point for the following power supplies:

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- 230 V AC power supply for the control cabinet lighting, air conditioning system, temperature control valves, condensation heater, etc.

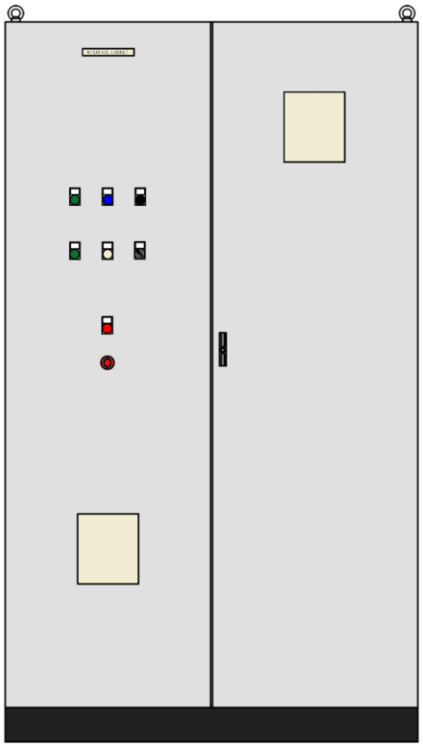

Figure 89: Interface Cabinet

Auxiliary Cabinet

The Auxiliary Cabinet is a floor-standing cabinet, that optionally will be equipped with an air condition. The Auxiliary Cabinet is the central connection point with the power grid of the plant or ship for the 24 V DC, 230 V AC and 400 V AC power supply of the engine.

It contains the starter for the engine oil pumps, temperature control valves, the electrical high-pressure pump for pilot oil injection as well as the Driver Units of the fuel, VVT or VTA actuator.

The Auxiliary Cabinet serves as a central connection point for the following power supplies:

- 24 V DC power supply and distribution for SaCoSone
- 230 V AC power supply for the control cabinet lighting, air conditioning system, temperature control valves, condensation heater, etc.
- 400 V AC power supply for pumps and actuators on the engine

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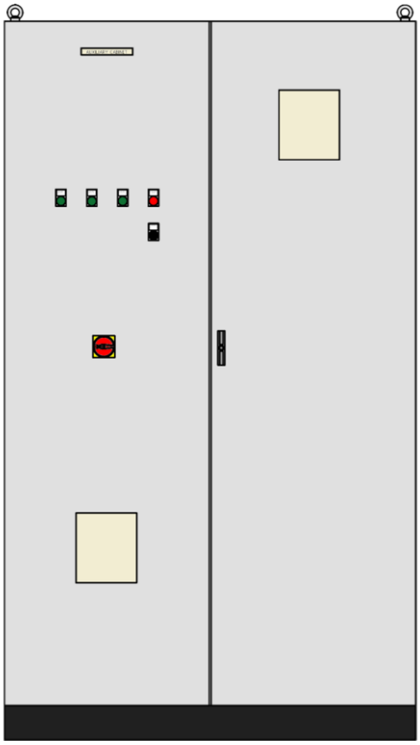

Figure 90: Auxiliary Cabinet

Driver Cabinet (optional)

The Driver Cabinet contains the frequency inverter for activation of the speed actuator as well as the 400 V AC power supply for the consumers on the engine and the 230 V AC power supply for the control cabinet lighting, grid socket, condensation heater and the air conditioning system.

The 24 V DC power supply is carried out via the Auxiliary Cabinet.

If a VVT and/or a VTA is mounted, the Driver Cabinet contains additional the control system for the VVT and/or the VTA.

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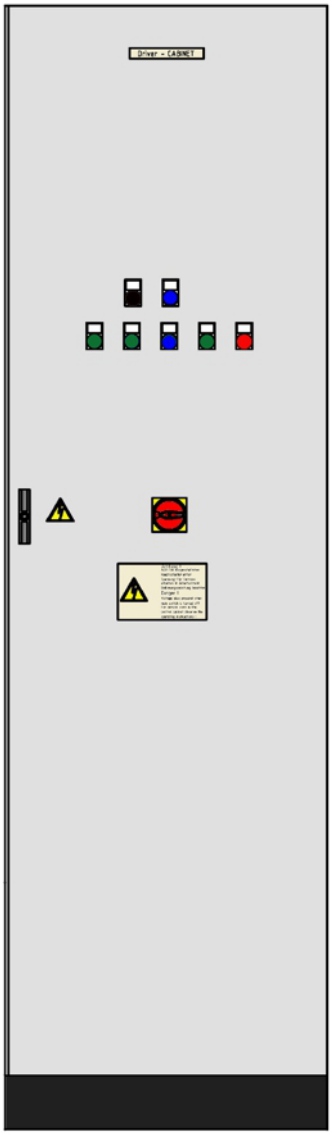

Figure 91: Driver Cabinet

Gas Valve Unit Cabinet

The Gas Valve Unit Cabinet (GVUC) is an extension which is designed specially for control of the gas valve unit within the gas supply system of the engine. SaCoSone specifies the required gas pressure and monitors and regulates it with the GVUC.

The GVUC must be installed in a suitable position outside the installation location of the gas valve unit.

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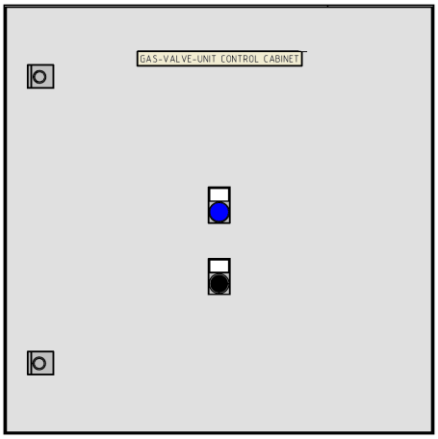

Figure 92: Gas Valve Unit Cabinet

Data Logging Cabinet (optional)

The data logging cabinet is used for the connection to Online Assist. The data logging module is installed in the data logging cabinet.

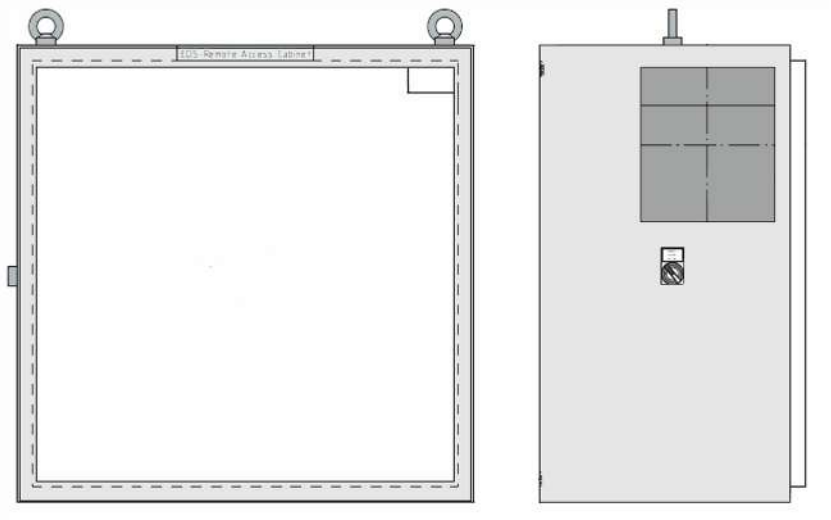

Figure 93: Data Logging Cabinet

Synchro Device Cabinet (optional)

The Synchro Device Cabinet (SDC) is installed between the Injection Unit and the GPC and synchronizes the engine with the available network as quickly as possible.

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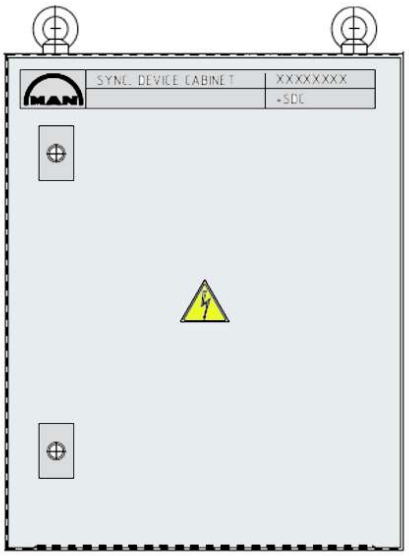

Figure 94: Synchro Device Cabinet

Local Operating Panel

The Local Operating Panel is mounted on the engine cushioned against vibration. This panel is equipped with a TFT display for visualisation of all engine operating and measuring data. At the Local Operating Panel the engine can be fully operated. Additional hardwired switches are available for relevant functions.

Stationary engines are not equipped with a backup display as shown on top of the Local Operating Panel.

Figure 95: Local Operating Panel

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Remote Operating Panel (optional)

The Remote Operating Panel serves for engine operation from a control room. The Remote Operating Panel has the same functions as the Local Operating Panel.

From this operating device it is possible to transfer the engine operation functions to a super-ordinated automation system. In plants with integrated automation systems, this panel can be replaced by IAS.

The panel can be delivered as loose supply for installation in the control room desk.

Figure 96: Remote Operating Panel (optional)

SaCoSone system bus

The SaCoSone system bus connects all system modules. This redundant field bus system provides the basis of data exchange between the modules and allows the takeover of redundant measuring values from other modules in case of a sensor failure.

SaCoSone is connected to the plant by the Gateway Module. This module is equipped with decentral input and output channels as well as with different interfaces for connection to a super-ordinated automation system, the Remote Operating Panel and the online service.

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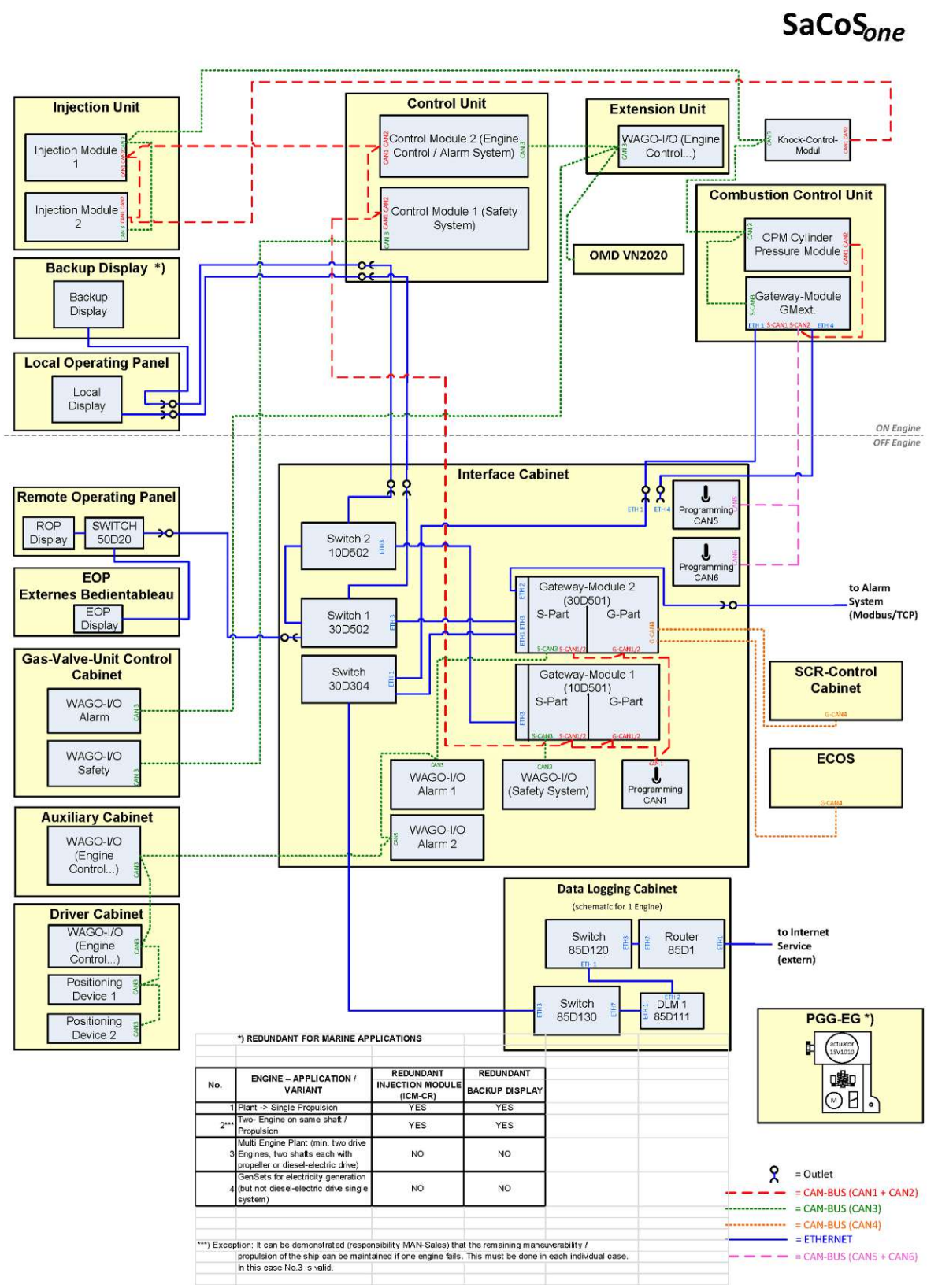

*) REDUNDANT FOR MARINE APPLICATIONS

No.	ENGINE – APPLICATION / VARIANT	REDUNDANT INJECTION MODULE (ICM-CR)	REDUNDANT BACKUP DISPLAY
1	Plant -> Single Propulsion	YES	YES
2***	Two- Engine on same shaft / Propulsion	YES	YES
3	Multi Engine Plant (min. two drive Engines, two shafts each with propeller or diesel-electric drive)	NO	NO
4	GenSets for electricity generation (but not diesel-electric drive single system)	NO	NO

***) Exception: It can be demonstrated (responsibility MAN-Sales) that the remaining maneuverability / propulsion of the ship can be maintained if one engine fails. This must be done in each individual case. In this case No.3 is valid.

Figure 97: SaCoSone system bus

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Monitoring network

The monitoring network interconnects monitoring interface of all available engine controls. This network is the basis of data exchange between monitoring applications, e.g. CoCoS EDS PC or MAN Online Services. Within each engine control, a component is installed which is responsible for data exchange of TCP/IP level. A firewall is implemented to protect the system which also regulates communication between monitoring network, internet access network and MAN Online Services.

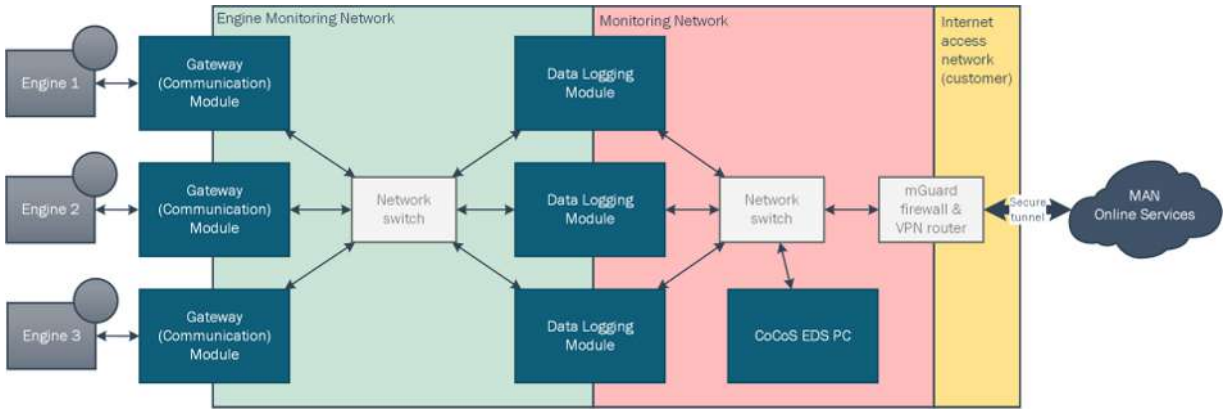

Figure 98: Engine monitoring network, monitoring network, internet access network

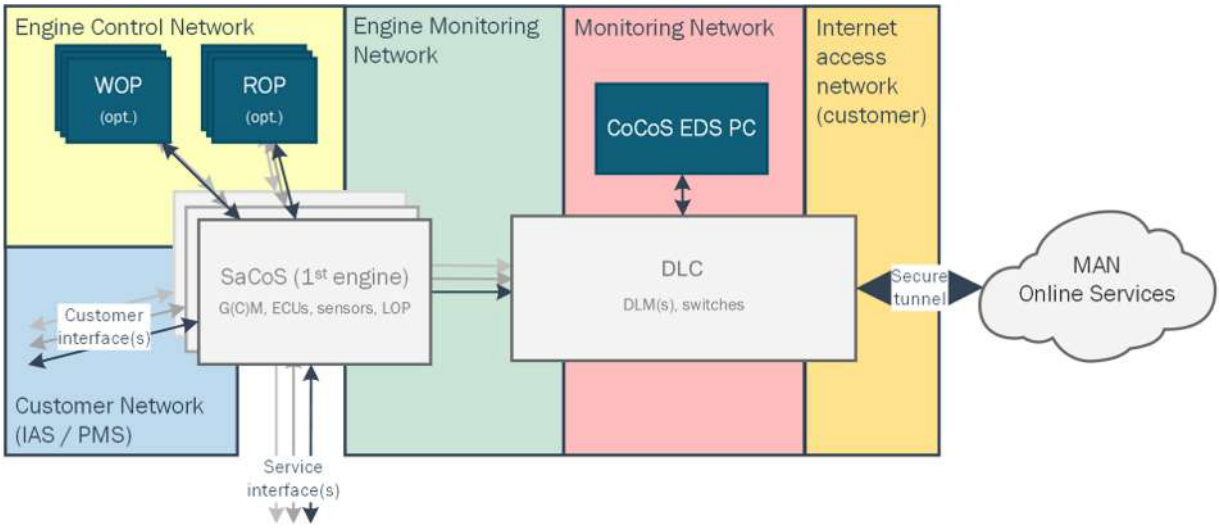

Figure 99: Engine control network, engine monitoring network, monitoring network, internet access network

Cyber security guidelines

The following cyber security guidelines represent a recommendation of protective measures to prevent unauthorized access, among other things. Some recommendations may not be applicable or only partially applicable or achievable, depending on the system.

Assure physical SaCoS security

Due to its design, certain parts and connections of SaCoSone have to be physically protected against access by unauthorized personnel. Only authorized staff must be allowed access, and only after their authentication. Such access should be logged, and the logs kept for future audit purposes. Physical security measures (walls, locked doors, cable routing that makes network and serial cables inaccessible to unauthorized personnel) must be established

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to prevent attackers from accessing and tampering with SaCoSone hardware, software, cables, and data, disconnecting SaCoSone components from their intended networks, or connecting unauthorized devices to the system.

The level of physical security measures must be assessed by the customer and may be based on but is not limited to system context, intended system use, system location, assumptions about current threats to the customer business, ease of access, and applicable regulations.

Such measures are particularly necessary for the following SaCoSone components and connections associated with safety-critical engine operation and control; depending on the SaCoS variant, those may include but are not limited to the following:

- SaCoSone modules connected via the CAN bus ("Core control network"), including but not limited to Gateway Module, Control Module, Injection Module, WAGO-I/O, VTT frequency converters.
- CAN bus.
- SaCoSone sensors and actuators.
- Network cables connecting operating panels (esp. those placed outside the engine room such as ROP and EOP) to SaCoSone.
- All operating panels incl. LOP, ROP, EOP, and WOP.
  Software update of operating panels can also be performed over USB. It is therefore imperative for the customer to limit and control access to the room(s) where such panels are installed. Only trained MAN service staff is allowed to interact with the USB port of the operating panel, unless the customer is explicitly permitted to do so by MAN Energy Solutions in case of emergency. The customer must not expose the operating panel's USB port on the outside of the panel's enclosure. The enclosure must remain closed except during authorized servicing tasks.
- (Serial or Ethernet) cable(s) connecting the system to external networks (e.g., Customer Network).
- All cables used to connect the SaCoSone Toolbox with the SaCoSone modules, as well as the Toolbox itself.
- Network switches.

Failing to follow those guidelines may result in, but is not limited to the following:

- Catastrophic impact on the safety of the engine, plant, as well as their environment.
- Loss of engine control, potentially to a threat actor.
- Loss of access to (authentic) information on engine and system state (incl. operational and alarm data), potentially to a threat actor.

Devices within Security Zones 1 and higher have limited communication access to devices within Security Zone 0, and thus pose a lower risk of compromising the safety of the engine. However, those devices and their connections may have to be protected against unauthorized access. This may include, but is not limited to the following security measures:

- Communication between the GM and the DLM has to be physically secured against unauthorized access.
- Any network switches installed between the GM and the DLM may not be physically accessed, or connected to.

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Failure to follow the outlined guidelines to prevent attack scenarios may result in, but is not limited to the following risk scenarios (Depending on the resources available to the attacker, a targeted attack on SaCoSone may have grave consequences to the engine and its environment. The risk scenarios listed in this document are not exhaustive):

- Loss of engine control, potentially to a threat actor.
- Loss of access to (authentic) information on engine and system state (incl. operational and alarm data), potentially to a threat actor.

Guarantee and verify the security of connection to the Customer Network and operating panels

SaCoSone can be operated remotely using a remote operating panel such as ROP or WOP, as well as by issuing commands from the Customer Network using a serial or Ethernet connection. For that purpose, the SaCoSone Gateway Module (GM) provides an interface that is used by the operating panels and devices within the customer network.

The customer must therefore ensure the following:

- Physical access to network and serial cables connecting SaCoSone with devices on the Customer Network and the operating panels must be restricted.
- Communication with operating panel must not be routed over an already existing network that has other participants or to which other participants can be added in the future without authorization.
- Communication between SaCoSone and devices on the Customer Network must not be exposed to potential aggressors (e.g., potentially compromised or infected devices on the same network).

Failure to follow the outlined guidelines to prevent attack scenarios may result in, but is not limited to the following risk scenarios:

- Loss of engine control, potentially to a threat actor.
- Loss of access to (authentic) information on engine and system state (incl. operational and alarm data), potentially to a threat actor.

Ensure secure access and storage of SaCoS related data

Data that is created by or for SaCoS, including SDIs, parameter sets, or operational data is not protected by the system. It is therefore imperative to store such data securely and protect it against unauthorized access.

Data access and secure storage may be implemented by following the following guidelines:

- Only use separate, dedicated hardware (e.g., SaCoSone Toolbox, dedicated “MAN Energy Solutions Service Stick” USB mass storage devices) for separate, dedicated tasks (e.g., parametrizing SaCoSone, storing SDIs).
- Consider encrypting and/or authenticating SaCoS related data stored outside of SaCoS.
- Never connecting such hardware to devices or networks other than those dedicated to their purpose (e.g., do not connect the MAN Energy Solutions Service Stick used to store SDIs to the DLM).
- Physically secure such hardware by storing it in a secure, access controlled location.

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- Do not disclose any authenticators used to access any SaCoS device or system to unauthorized parties. Such authenticators may include the eToken password, LOP passwords, or credentials for devices managed by the customer that connect to SaCoS.
- Data related to servicing the operating panels (e.g., software updates, extracted alarm history) must be stored in a secure, tamper-proof location.

Failing to follow these guidelines may result in, but is not limited to the following scenarios:

- Attacker learns details about system and engine usage, properties, and state.
- Attacker is able to manipulate engine state.

Secure the Customer Network

SaCoS is exposing an unauthenticated network service that the customer can connect to in order to monitor and control the engine. This logical network is referred to as the Customer Network, and must be secured by the customer in order to prevent the loss of engine monitoring and control, potentially to an attacker. Assuring the security of this network is the customer's responsibility.

While the risk of an attacker within the Customer Network being able to threaten the safety of the engine is low, they may gain the ability to control the engine, which may present a safety risk to the overall plant. Therefore, the customer must implement security measures to secure that network that may include, but are not limited to the following:

- Devices on the Customer Network may only communicate with SaCoS via the Eth2 network interface of the GM.
- Limit and protect physical and communication access to the devices on the network that the GM connects to via its Eth2 network interface, as well as all network connections between the GM and any of the devices on the Customer Network.
- Use of wireless technologies is strongly discouraged.
- In addition to the SaCoSone GM, the Customer Network may only contain IAS/PMS devices.
- The "Customer Network" must be segregated from any other customer's networks.
- Do not connect any untrustworthy or vulnerable devices to the Customer Network.
- Do not bridge the Customer Network with public or otherwise untrustworthy networks, or any other plant networks that may have untrusted participants.
- Secure every device on the customer network according to the customer's own risk assessment and CSMS. The measures may include a hardening of the devices, providing them with regular security updates, disabling USB ports, performing antivirus scans, and/or the maintenance of an "air gap" between the Customer Network devices and any other network.
- Avoid running any Modbus services on the Customer Network to avoid incidentally connecting to those services instead of the Modbus service provided by the SaCoSone GM.

Failing to follow the outlined guideline may result in, but is not limited to the following risk scenarios:

- Threat actor may manipulate what is displayed on the operating panels.

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- Threat actor may learn operational data reported to the operating panels and to the customer network.
- Threat actor may be able to control engine (within safety constraints).
- Threat actor may be able to execute engine safety test(s) that are usually executing using the operating panel(s).
- Threat actor may be able to deny engine operation and control via the operating panels or via Modbus signals from within the Customer Network.
- Threat actor may be able to disable some of the safety functions implemented by the GM.

In addition, a compromised SaCoS may pose a threat to the Customer Network, and compromise or infect devices connected to it. While the risk is estimated by MAN Energy Solutions to be very low in the context of a risk assessment performed in the context of Security Level 1 as per IEC 62443, the customer may still want to implement additional security measures in order to protect against this threat. Such measures may include, but are not limited to the following:

- Install a dedicated network security device (e.g., firewall, IPS) that protects the conduit between SaCoS and the Customer Network. Such a device may, e.g., only allow Modbus communication initiated from within the Customer Network and the Modbus responses by the GM, and/or monitor, log, and notify the customer about suspicious or unusual communication activity between SaCoS and the Customer Network.
- Harden devices on the Customer Network under the assumption that SaCoS may act as an aggressor.
- Do not expose any critical assets (e.g., confidential data, other plant control or monitoring functionality) on the Customer Network.

Secure the Operating Panels

The SaCoSone operating panels and their connection to the rest of the system are not hardened or kept up to date, which may allow a potential attacker to compromise them. The customer is therefore advised to ensure the following:

- Physical access to the operating panel must be controlled and restricted to authorized personnel.
- Physical access to the communication ports of the operating panels (including USB and Ethernet ports) must be limited and restricted to authorized MAN Energy Solutions service staff. The customer may not change the connections set up by MAN Energy Solutions, or connect their own devices to the panels except in case of an emergency, and only by following exactly the directions provided by MAN Energy Solutions.
- Software update of an operating panel may only be performed by the customer in case of an emergency, and only if directed by MAN Energy Solutions. USB drive used for transferring the update to the LOP must be wiped prior to its use. Software update files must be obtained directly from MAN Energy Solutions, and only over a secure channel as directed by MAN Energy Solutions, using a secure and trusted device to assure that they have not been tampered with. Customer must make sure that the update they are applying is the version that MAN Energy Solutions directed them to use, and afterwards verify that the installed version is correct.

Failing to follow those guidelines may result in, but is not limited to the following (see also “Secure the Customer Network” section):

- Engine control is lost (to a potential attacker).

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- Inauthentic data is shown on the operating panels or reported to devices on the Customer Network.
- Operating panels become inoperable.

Maintain the network segmentation at all times

The network segmentation implemented for SaCoSone is a major part in keeping the system secure against outside threats. Therefore, the network structure that as established during system setup must not be changed. Specifically, the following has to be assured:

- Individual network segments may not be bridged, e.g., by connecting a service device to several network segments at the same time, interconnecting network switches from different network segments, or connecting devices from one segment to devices on other segments.
- No additional devices may be connected to any of the network zones.

Failing to uphold the network segmentation prescribed by MAN Energy Solutions may result in, but is not limited to the full and persistent compromise of SaCoSone and the safety of the engine.

Prepare for security incidents

Following the guidelines outlined in this document helps to minimize the risk of possible system compromise by an attacker, but residual risks will remain. The customer must be prepared for potential compromise scenarios, and be able to respond to a security incident. The customer must prepare an incident response plan that may include, but is not limited to the following steps:

- Purchase spare parts to replace SaCoSone devices that are suspected to have been compromised.
- If making changes to the system (e.g., parameter changes through Toolbox), also remember to take backups of those changes.
- In case a Toolbox is not available, contract specific agreements apply (availability of MAN Energy Solutions service, spare parts, etc.).
- Report the incident to MAN Energy Solutions and follow the instructions of MAN Energy Solutions support service.
- Secure the forensic evidence (including, but not limited to audit and access logs created by SaCoSone and the customer's devices and processes) that may point to potential cause(s) of the incident.
- Analyze the evidence and enact steps to mitigate the vulnerability that led to the incident.

The customer shall also monitor at the best of their abilities the system for indications of a possible security incident. Any of the following indicators may point to the fact that the SaCoS protection measures have been breached:

- Inconsistencies between data displayed on any of the operating panels, and/or reported to devices within the Customer Network are detected.
- Unrealistic or suspicious data is displayed on any or all of the operating panels or the EDS PC, and/or reported to devices within the Customer Network.
- Unusual system behavior is detected that can be attributed to unintended system operation commands.
- Remote engine operation is no longer possible.

To address such a breach, the customer is advised to consider the following mitigations:

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- Put the system into a safe state depending on the current situation (e.g., island mode, engine shutdown etc.), inform MAN Energy Solutions service of the suspicious behavior, and continue operation only after the issue has been resolved.
- For the duration of continued operation, switch the system into island mode (i.e. local engine control via LOP) by disconnecting all remote connections to the GM (i.e., network connections to Eth1, Eth2 interfaces of the GM). If this does not resolve the issue, and the current situation allows for it without compromising system safety, also disconnect the network connection to the Eth3 interface of the GM, and operate the system via hardwire.

Failing to follow those guidelines may result in much longer times until an incident has been detected and corrected, leading to potential financial damages, and/or posing a risk to the safety of the system.

Secure disposal / decommissioning

For the purposes of secure device disposal, consider that SaCoS components (including, but not limited to ECUs, Data Logging Modules, EDS PC, eTokens, and the Toolbox) store sensitive information including engine parameters, engine operational data, and cryptographic keys.

Account management

Most SaCoS components do not support interactive access by the customer. Notable exceptions are:

- mGuard configuration account,
- Toolbox account,
- EDS PC account, and
- GM service access using the EXPERT eToken.

mGuard account and EXPERT eToken management is handled by MAN Energy Solutions, whereas SaCoS Toolbox and EDS PC are Windows-based machines to be managed and secured by the customer according to this guideline. For this purpose, the onboard operating system facilities can be utilized.

3.2 Power supply and distribution

The plant has to provide electric power for the automation and monitoring system. In general an uninterrupted 24 V DC power supply is required for SaCoSone.

For the supply of the electronic backup fuel actuator an uninterrupted 230 V AC distribution must be provided. For pumps and other consumers a 400 V AC power is required.

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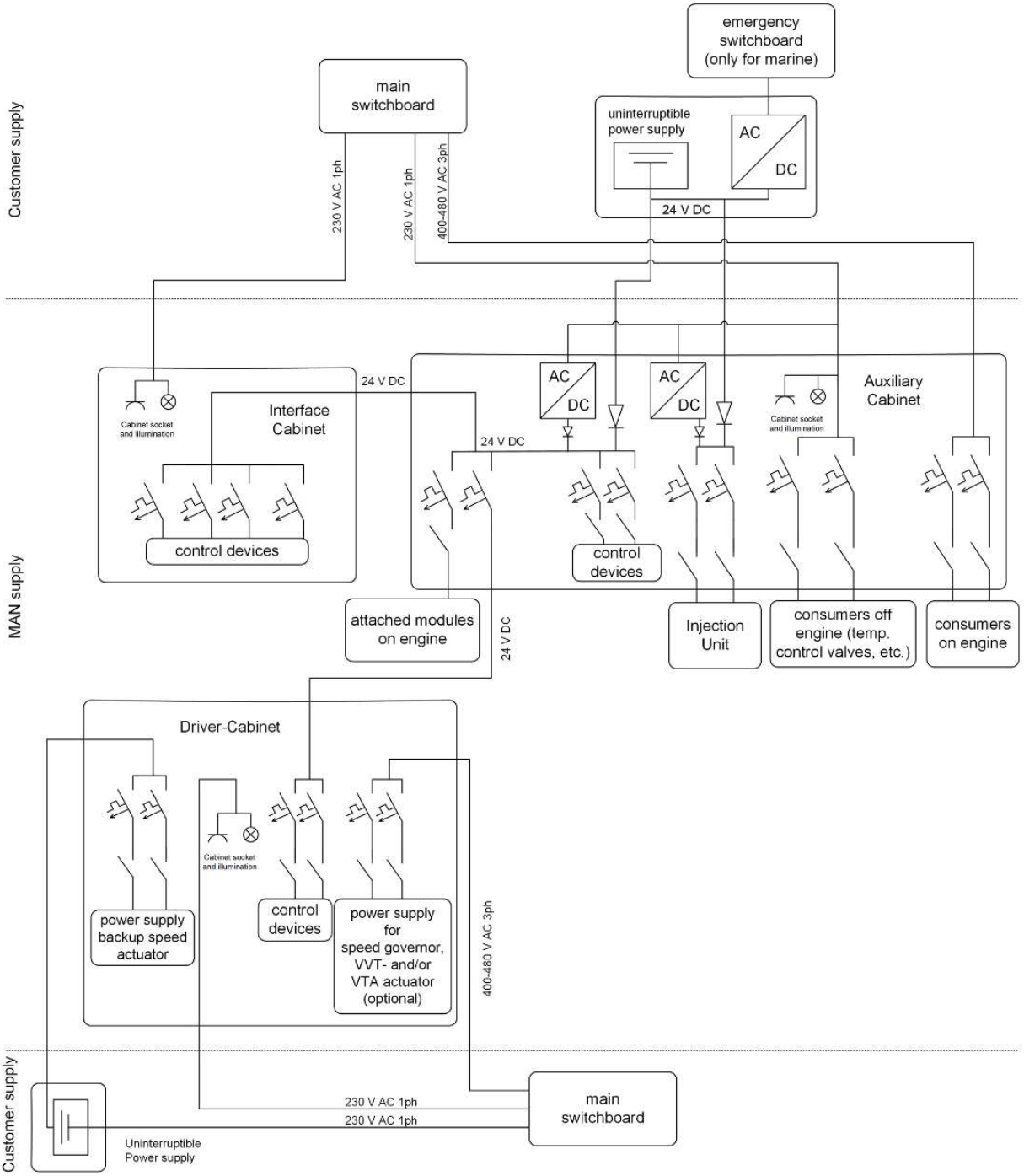

Figure 100: Supply diagram

Required power supplies

Voltage	Consumer	Notes
230 V 50/60 Hz	SaCoSone Interface Cabinet	Cabinet illumination, socket, anticondensation heater
24 V DC	SaCoSone Auxiliary Cabinet	All SaCoSone components in the Interface Cabinet and on the engine

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Voltage	Consumer	Notes
230 V 50/60 Hz	SaCoSone Auxiliary Cabinet	Cabinet illumination, socket, temperature control valves, anticondensation heater
400 – 480 V 50/60 Hz	SaCoSone Auxiliary Cabinet	Power supply for consumers on engine (e.g. cylinder lubricator)
230 V 50/60 Hz	Driver Cabinet	Cabinet illumination, socket, temperature control valves, anticondensation heater
230 V 50/60 Hz	Driver Cabinet	UPS-buffered power supply for speed actuator backup
400 – 480 V 50/60 Hz	Driver Cabinet	Power supply for speed actuator, drive of the VVT (optional) and/or VTA (optional)

Table 161: Required power supplies

Galvanic isolation

It is important that at least one of the two 24 V DC power supplies per engine is foreseen as **isolated unit with earth fault monitoring** to improve the localisation of possible earth faults. This isolated unit can either be the UPS-buffered 24 V DC power supply or the 24 V DC power supply without UPS.

Example:
The following overviews shows the exemplary layout for a plant consisting of four engines. In this example the 24 V DC power supply without UPS is the isolated unit. The UPS-buffered 24 V DC power supply is used for several engines. In this case there must be the possibility to disconnect the UPS from each engine (e.g. via double-pole circuit breaker) for earth fault detection.

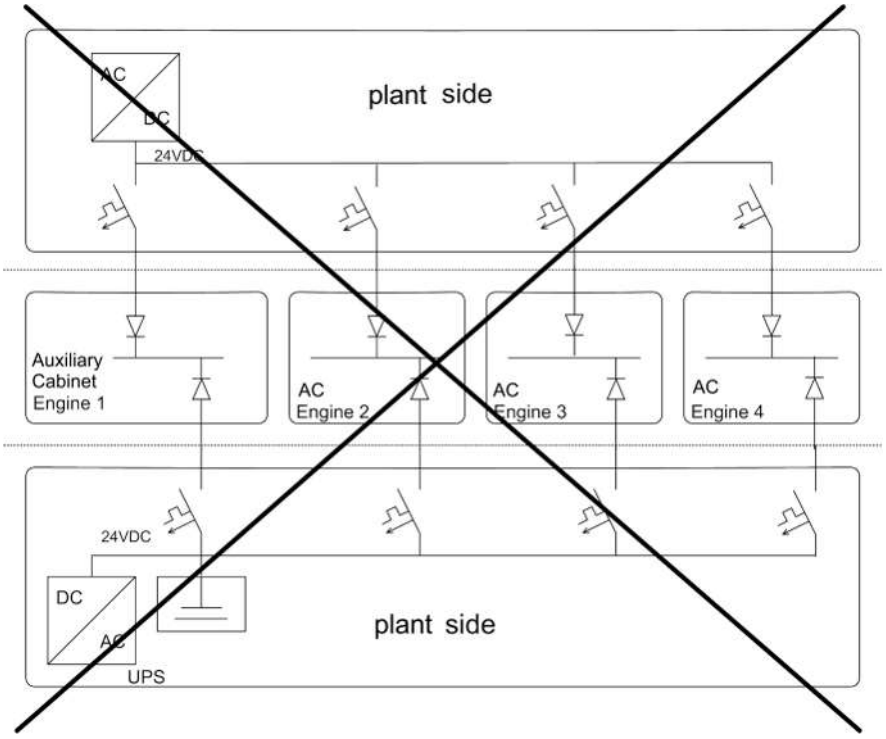

Figure 101: Wrong installation of the 24 V DC power supplies

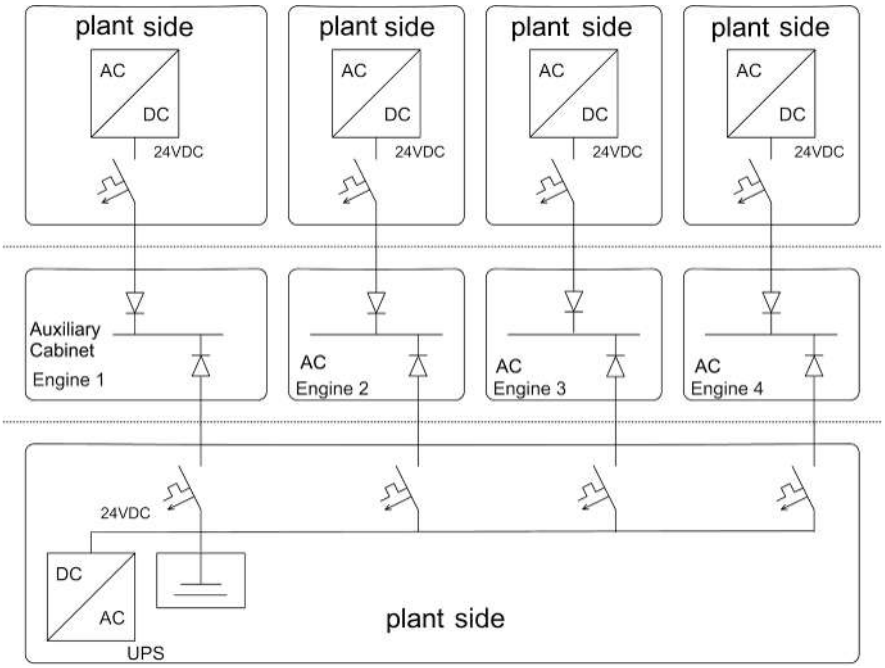

Figure 102: Correct installation of the 24 V DC power supplies

3.3 Safety architecture

Connection of external digital outputs to safety-relevant dual-channel digital inputs of SaCoSone

SaCoS owns a double-channel safety architecture. External emergency stops or automatic shutdowns have to support this architecture to avoid false alarms of the plausibility check. Supporting could be done by having a double-channel architecture too (recommended by MAN Energy Solutions) or by having a single-channel architecture effecting both channels. To support single-channel architectures of marine applications an additional box is necessary, which is effecting both channels and provide a wire break detection (see right figure below).

Note:
A single-channel architecture has a higher probability of failure than a double-channel architecture.

MAN Energy Solutions will not be responsible for any increase of risk, which might be caused by the use of such a single-channel architecture instead of double-channel architecture.

The relais have to be energized on demand of the safety function. The wire-break-monitor will be de-energized at wire-break.

Attention, wire-break-monitoring is not supported for all variants of SaCoS at redundant inputs (because a plausibility check of the two channels, when the safety function is requested, is sufficient to detect errors).

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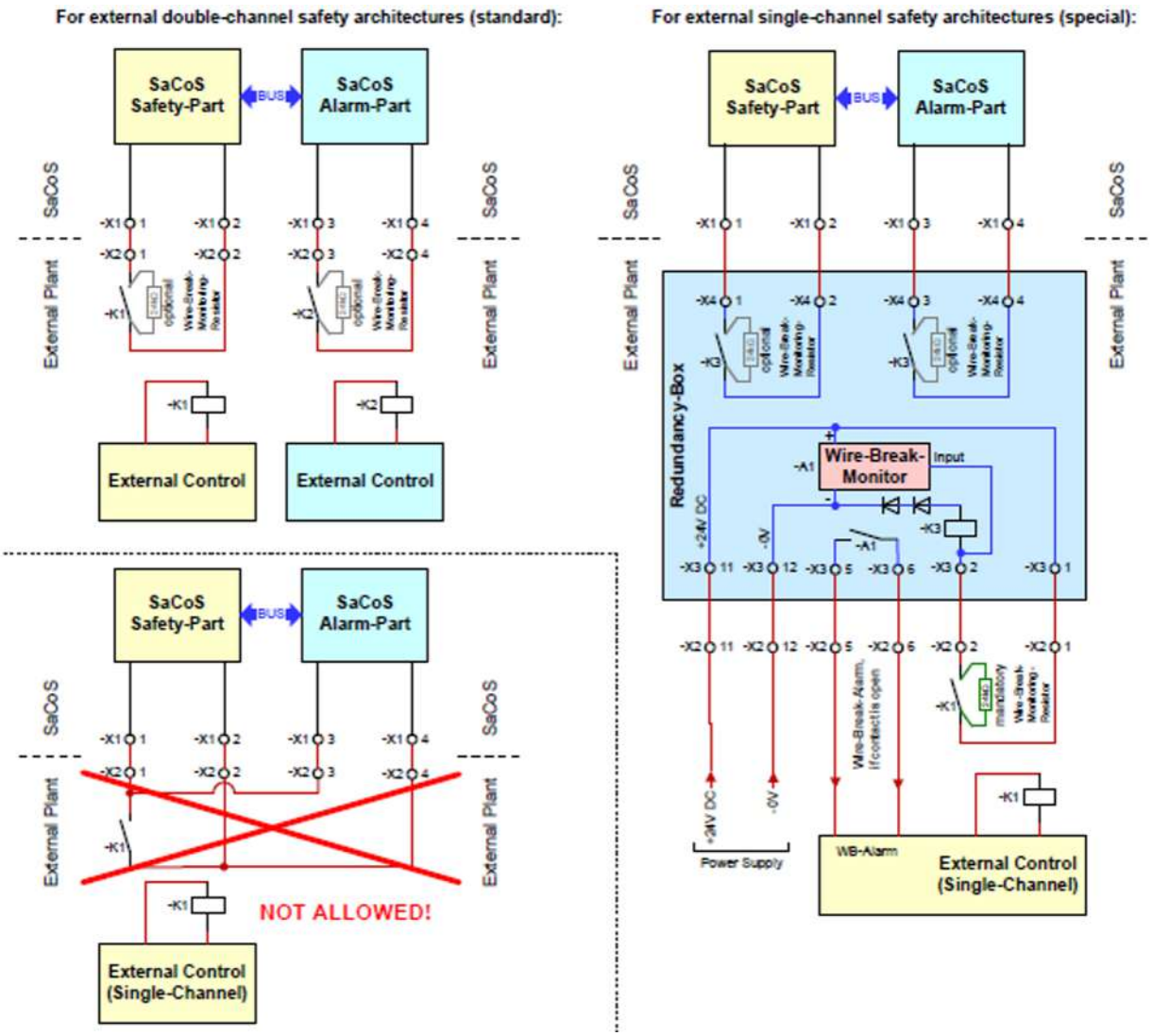

⚠ WARNING It is not allowed to connect both redundant channels of SaCoS direct which each other without redundancy box and connect it then to the external single channel.

Connection of single-channel digital inputs to safety-relevant dual-channel digital outputs

Where double-channel outputs are provided by MAN Energy Solutions for safety related functions, they have to be connected to a double-channel inputs at site. Where this is not possible, due to the lack of a double-channel architecture in the external controlling system, then a single-channel input can be converted to a double-channel architecture with the circuit proposed in this document.

MAN Energy Solutions will not be responsible for any increase of risk, which might be caused by the use of such a single-channel control.

The relais are deenergized in normal case and energized at failure. The main contacts are open in normal case and closed at failure (the feedback contacts must be complementary). All relays must have forcibly guided contacts.

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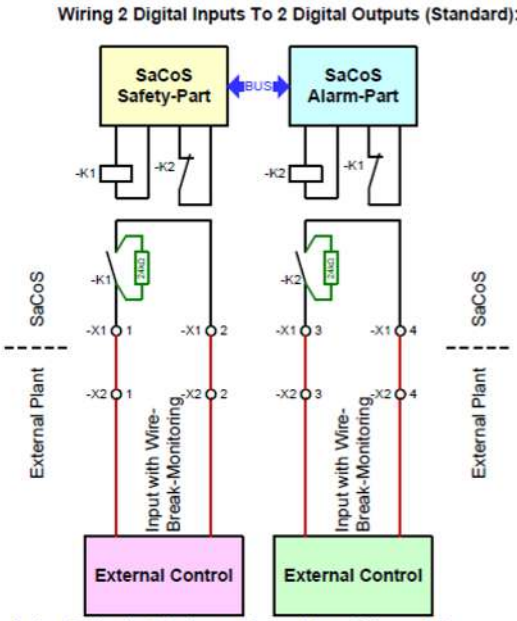
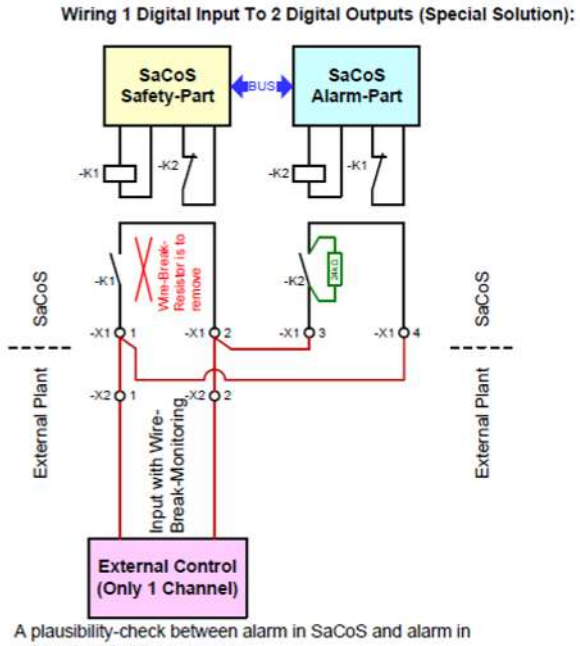


A plausibility-check between Channel 1 and Channel 2 is necessary at external plant.

A wire-break monitoring is optional at double-channel architectures, if plausibility-check is implemented.



A plausibility-check between alarm in SaCoS and alarm in external control is necessary.

A wire-break monitoring is mandatory at single-channel architectures.

3.4 Operation

Control station changeover

The operation and control can be done from both operating panels. Selection and activation of the control stations is possible at the Local Operating Panel. On the displays, all the measuring points acquired by means of SaCoSone can be shown in clearly arranged drawings and figures. It is not necessary to install additional speed indicators separately.

The operating rights can be handed over from the Remote Operating Panel to another Remote Operating Panel or to an external automatic system. Therefore a handshake is necessary.

For applications with Integrated Automation Systems (IAS) also the functionality of the Remote Operating Panel can be taken over by the IAS.

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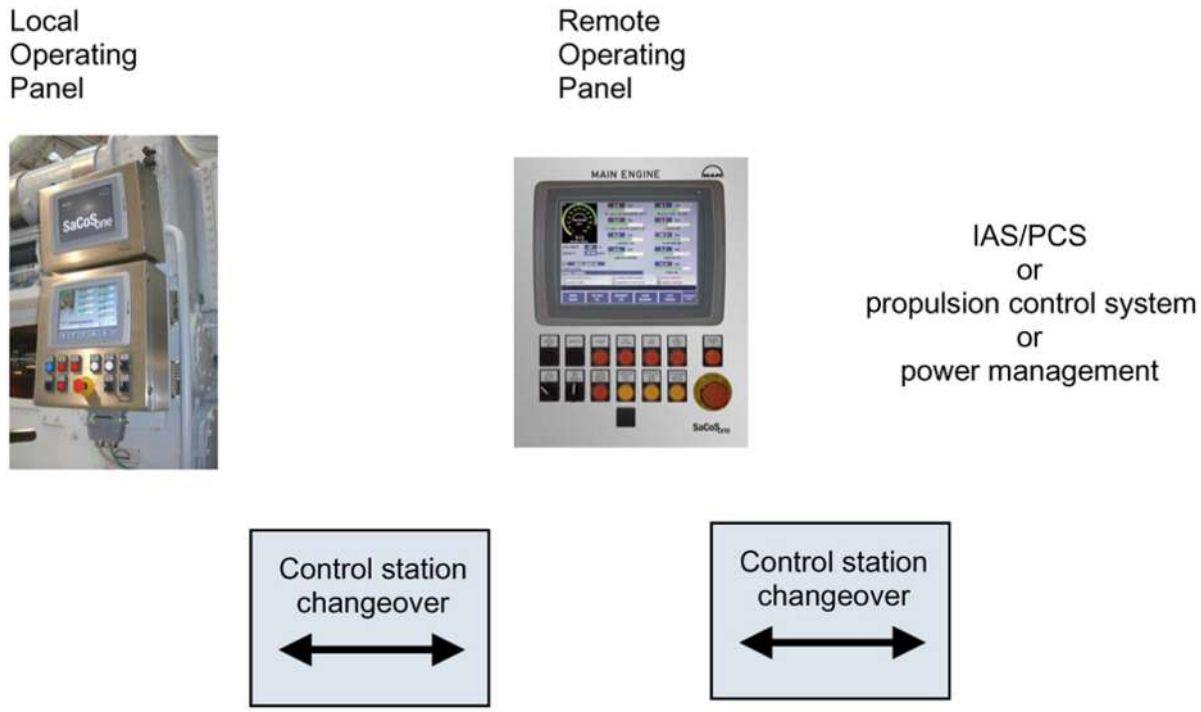

Figure 103: Control station changeover

Speed setting

In case of operating with one of the SaCoSone panels, the engine speed setting is carried out manually by a decrease/increase switch button.
If the operation is controlled by an external system, the speed setting can be done either by means of binary contacts (e.g. for synchronisation) or by an active 4 – 20 mA analogue signal alternatively. The signal type for this is to be defined in the project planning period.

Operating modes

For alternator applications:

- Droop (5-percent speed increase between nominal load and no load)

For propulsion engines:

- Isochronous
- Master/Slave Operation for operation of two engines on one gear box

The operating mode is pre-selected via the SaCoS interface and has to be defined during the application period.

Details regarding special operating modes on request.

3.5 Functionality

Safety functions

The safety system monitors all operating data of the engine and initiates the required actions, i.e. load reduction or engine shutdown, in case any limit values are exceeded. The safety system is separated into Control Module and

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Gateway Module. The Control Module supervises the engine, while the Gateway Module examines all functions relevant for the security of the connected plant components.

The system is designed to ensure that all functions are achieved in accordance with the classification societies' requirements for marine main engines.

The safety system directly effects the emergency shutdown, the speed control, the Gas Valve Unit Control Cabinet and the Prechamber Gas Valve Unit Cabinet (optional for gas engines).

It is possible to establish additional shutdowns and blockings of external systems in SaCoSone.

Load reduction

The exceeding of certain parameters requires a load reduction to 60 %. The safety system supervises these parameters and requests a load reduction, if necessary. The load reduction has to be carried out by an external system (IAS, PMS, PCS). For safety reasons, SaCoSone will not reduce the power by itself.

Auto shutdown

Auto shutdown is an engine shutdown initiated by any automatic supervision of either engine internal parameters or mentioned above external control systems. If an engine shutdown is triggered by the safety system, the emergency stop signal has an immediate effect on the emergency shutdown device, and the speed control.

Some auto shutdowns may also be initiated redundantly by the alarm system.

Emergency stop

Emergency stop is an engine shutdown initiated by an operator's manual action by pressing an emergency stop button.

Note:
A manual emergency stop stops the engine, but does not affect the interface signals requesting the auxiliary units. During the integration, the respective integrator must determine how the system reacts. Therefore manual emergency stops are provided to the plant explicit.

Cylinder cut off gas

In certain fault cases that only affect individual cylinders, the engine can continue to operate with only the affected cylinder switched off (max. duration 1h with 80% engine load). In other words, CCOG (cylinder cut off gas) replaces the previous shutdown in these cases.

Since when one cylinder is switched off, the load on the other cylinders is increased and the torsional vibrations also increase at the same time, the output is reduced with the transition to CCOG operation. A CCOG load reduction request is triggered for this. The CCOG load reduction request differs from the general load reduction request in terms of the power that is still available.

Override

Only during operation in diesel mode safety actions can be suppressed by the override function. In gas mode, if override is selected, an automatic changeover to diesel mode will be performed. The override has to be selected before a safety action is actuated. The scope of parameters prepared for override is different and depends on the chosen classification society. The availability of the override function depends on the application.

Alarming

The alarm function of SaCoSone supervises all necessary parameters and generates alarms to indicate discrepancies when required.

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Self-monitoring

SaCoSone carries out independent self-monitoring functions. Thus, for example the connected sensors are checked constantly for function and wire break. In case of a fault SaCoSone reports the occurred malfunctions in single system components via system alarms.

Speed control

The engine speed control is realised by software functions of the Control Module/Alarm and the Injection Modules. Engine speed and crankshaft turn angle indication is carried out by means of redundant pick ups at the gear drive.

Load distribution in multi-engine plants

With electronic speed control, the load distribution is carried out by speed droop, isochronously by load sharing lines or master/slave operation.

Load limit curves

- Start fuel limiter
- Charge air pressure dependent fuel limiter
- Torque limiter
- Jump-rate limiter
- External limiter
- Lambda limiter

Note:
In the case of controllable pitch propeller (CPP) units with combinator mode, the combinator curves must be sent to MAN Energy Solutions for assessment in the design stage. If load control systems of the CPP-supplier are used, the load control curve is to be sent to MAN Energy Solutions in order to check whether it is below the load limit curve of the engine.

Shutdown

The engine shutdown, initiated by safety functions and manual emergency stops, is carried out by redundant fast closing valves of the gas valve unit and independent interrupt of power supply of gas admission valves.

Note:
The engine shutdown may have impact on the function of the plant. These effects can be very diverse depending on the overall design of the plant and must already be considered in early phase of the project planning.

Overspeed protection

The engine speed is monitored in both Control Modules independently. In case of overspeed each Control Module actuates the shutdown device by a separate hardware channel. Overspeed is monitored in Ignition Module too to interrupt igntion.

Control

SaCoSone controls all engine-internal functions as well as external components, for example:

Start/stop sequences

- Requests of lube oil and cooling water pumps
- Monitoring of the prelubrication and post-cooling period
- Monitoring of the acceleration period
- Request of start-up air blower

Fuel changeover

- Control of the switch-over from one type of fuel to another

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- Fuel injection flow is controlled by the electric fuel injection
- Release of the gas operating mode

Control station switch-over

Switch-over from local operation in the engine room to remote control from the engine control room.

Knock control

For the purpose of knock recognition, a special evaluation unit is fitted to the engine and connected to the engine control via the CAN bus.

Air-fuel ratio control

For air-fuel ratio control, part of the charge air is rerouted via a by-pass flap. The exhaust gas temperature upstream of the turbine, as well as characteristic fields stored in the engine control, are used for control purposes. The air-fuel ratio control is only active in gas operating mode. In Diesel operating mode, the flap remains closed.

Control of the gas valve unit

The gas pressure at the engine inlet is specified by the engine control and regulated by the gas valve unit. The main gas valves are activated by the engine control system. Prior to every engine start and switch-over to the gas operating mode respectively, the block-and-bleed valves are checked for tightness (see also section Fuel gas system, Page 404).

External functions

- Electrical lube oil pump
- Electrical driven HT cooling water pump
- Electrical driven LT cooling water pump
- Nozzle cooling water module
- HT preheating unit
- Clutches

The scope of control functions depends on plant configuration and must be coordinated during the project engineering phase.

Media temperature control

Various media flows must be controlled to ensure trouble-free engine operation.

The temperature controllers are available as software functions inside the Gateway Module of SaCoSone. The temperature controllers are operated by the displays at the operating panels as far as it is necessary. From the Interface Cabinet the relays actuate the control valves.

- The cylinder cooling water (HT) temperature control is equipped with performance-related feed forward control, in order to guarantee the best control accuracy possible (refer also to section Cooling water system, Page 335).
- The low temperature (LT) cooling water temperature control works similarly to the HT cooling water temperature control and can be used if the LT cooling water system is designed as one individual cooling water system per engine.

  In case several engines are operated with a combined LT cooling water system, it is necessary to use an external temperature controller.

  This external controller must be mounted on the engine control room desk and is to be wired to the temperature control valve (refer also to section Cooling water system, Page 335).
- The charge air temperature control is designed similar to the HT cooling water temperature control.

  The cooling water quantity in the LT part of the charge air cooler is regulated by the charge air temperature control valve (refer also to section Cooling water system, Page 335).

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- The design of the lube oil temperature control depends on the engine type. It is designed either as a thermostatic valve (waxcartridge type) or as an electric driven control valve with electronic control similar to the HT temperature controller. Refer also to section External lube oil system, Page 313.

Starters

For engine attached pumps and motors the starters are installed in the Auxiliary Cabinet. Starters for external pumps and consumers are not included in the SaCoSone scope of supply in general.

3.6 Interfaces

Data bus interface (machinery alarm system)

This interface serves for data exchange to ship alarm systems or Integrated Automation Systems (IAS).

The interface is actuated with MODBUS protocol and is available as:

- Ethernet interface (MODBUS over TCP) or as
- Serial interface (MODBUS RTU) RS422/RS485, standard 5 wire with electrical isolation (cable length ≤ 100 m)

Only if the ethernet interface is used, the transfer of data can be handled with timestamps from SaCoSone.

The status messages, alarms and safety actions, which are generated in the system, can be transferred. All measuring values acquired by SaCoSone are available for transfer.

Alternator control

Hardwired interface, used for example for synchronisation, load indication, etc.

Alternator electric power (active power) signal

To keep, despite natural long-term deterioration effects, engine operation within its optimum range MAN Energy Solutions' engine safety and control system SaCoSone must be provided with an alternator electric power (active power) signal. Interface and signal shall comply with the following requirements:

1. The electric power of the generator (active power) shall be measured with the following components:
   - Current transformer with accuracy class: cl. 0.2 s
   - Voltage transformer with accuracy class: cl. 0.2
   - Measuring transducer with accuracy class: cl. 1
2. Measuring transducer shall provide the current active power as 4 – 20 mA signal and shall provide 0 – 90 % of measured value with response time ≤ 300 ms (EN 60688). The maximum response time of the 4 – 20 mA signal shall not exceed 350 ms.
3. The 4 – 20 mA generator power signal shall be hard-wired with shielded cable. The analogue value of 4 mA shall be approximately equivalent to 0 % generator power, the value of 20 mA shall be approximately equivalent to nominal generator power, plus 10 %. A proper scaling shall be determ-

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ined during project execution. Furthermore the signal for “Generator CB is closed” from power management system to SaCoSone Interface Cabinet shall be provide.

Power management

Hardwired interface, for remote start/stop, load setting, fuel mode selection, etc.

Propulsion control system

Standardised hardwired interface including all signals for control and safety actions between SaCoSone and the propulsion control system.

Others

In addition, interfaces to auxiliary systems are available, such as:

- Nozzle cooling water module (for DF engines)
- HT preheating unit
- Electric driven pumps for lube oil, HT and LT cooling water
- Clutches
- Gearbox
- Propulsion control system

On request additional hard wired interfaces can be provided for special applications.

Cables – Scope of supply

The bus cables between engine and interface are scope of the MAN Energy Solutions supply.

The control cables and power cables are not included in the scope of the MAN Energy Solutions supply. This cabling has to be carried out by the customer.

3.7 Technical data

Design

Cabinet

- Floor-standing cabinets with plinth and fan/air condition
- Cable entries: From below, through cabinet base
- Accessible by front door(s), doors with locks
- Opening angle: 90°
- Standard colour: Light grey (RAL7035)
- Ingress protection: IP54

Cabinet	Dimensions (mm) including base			Approx. weight (kg)
	Width	Height	Depth	
Interface Cabinet, equipped with fan	1,200	2,100	400	300
Interface Cabinet, equipped with air condition	1,550	2,100	400	360

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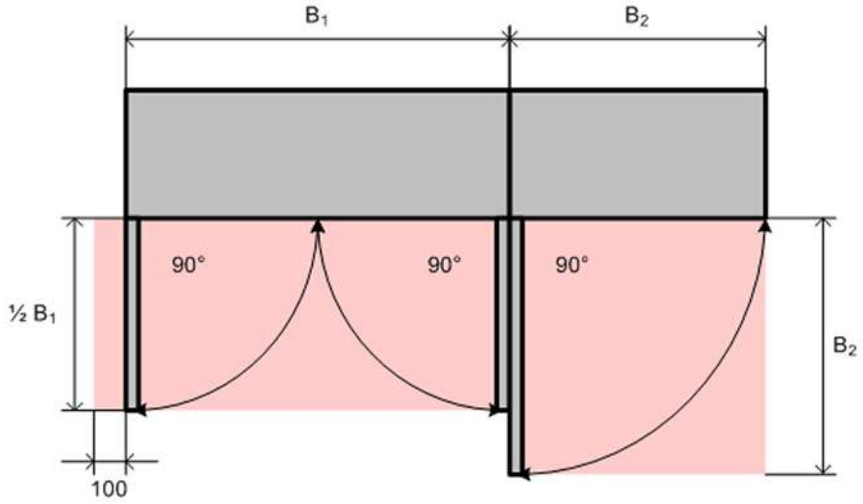
Cabinet	Dimensions (mm) including base			Approx. weight (kg)
	Width	Height	Depth	
Auxiliary Cabinet, equipped with fan	1,200	2,100	400	300
Auxiliary Cabinet, equipped with air condition	1,550	2,100	400	360
Driver Cabinet, equipped with fan	600	2,100	400	180
Driver Cabinet, equipped with air condition	950	2,100	400	240
Gas Valve Unit Cabinet	500	500	300	40

Table 162: Dimensions and weights of cabinets

Door opening area of cabinets



B_1 Width of cabinet 1 B_2 Width of cabinet 2

Figure 104: Exemplary arrangement of control cabinets with door opening areas (top view)

Remote Operating Panel (optional)

Design

- Panel for control desk installation with 3 m cable to terminal bar for installation inside control desk
- Front colour: White aluminium (RAL9006)
- Weight: 15 kg
- Ingress of protection: IP23
- Dimensions: 370 x 480 x 150 mm[1)]

[1)] width x height x depth (including base)

Environmental conditions

Components on engine and wall-mounted cabinets

- Ambient air temperature: +5 °C to +55 °C (Exception: an ambient air temperature of +5 °C to +45 °C applies to the Data Logging Cabinet)
- Relative humidity: < 96 %
- Vibrations: < 0.7 g

Floor-standing cabinets

- Ambient air temperature:
  - 0 °C to +45 °C: Floor-standing cabinets will be equipped with a fan

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  - Over +45 °C: Floor-standing cabinets will be mandatory equipped with an air condition
- Relative humidity: < 96 %
- Vibrations: < 0.7 g

3.8 Installation requirements

Location

The cabinets are designed for installation in non-hazardous areas.

Wall-mounted cabinets and floor-standing cabinets

The cabinets must be installed at a location suitable for service inspection.

Do not install the cabinets close to heat-generating devices.

Floor-standing cabinets

In case of installation at walls, the distance between the floor-standing cabinets and the wall has to be at least 100 mm in order to allow air convection.

The foundation at the installation site must be solid enough to withstand the weight of the floor-standing cabinets.

All floor-standing cabinets have to be fixed to the floor and additionally be secured against tipping over (e.g. by attaching the roof to the wall) with suitable mounting support, proven to withstand at minimum the maximum allowed inclination, see accordingly section Engine inclination, Page 37.

Note:
If the restrictions for ambient temperature can not be kept, the floor-standing cabinet must be ordered with an optional air condition system.

Ambient air conditions

For restrictions of ambient conditions, refer to the section Technical data, Page 238.

Cabling

The interconnection cables between the engine and the cabinets have to be installed according to the rules of electromagnetic compatibility. Control cables and power cables have to be routed in separate cable ducts.

The cables for the connection of sensors and actuators which are not mounted on the engine are not included in the scope of MAN Energy Solutions supply. Shielded cables have to be used for the cabling of sensors. For electrical noise protection, an electric ground connection must be made from the cabinets to the ship's hull.

All cabling between the cabinets and the controlled device is scope of customer supply.

The cabinets are equipped with spring loaded terminal clamps. All wiring to external systems should be carried out without conductor sleeves.

The redundant CAN cables are MAN Energy Solutions scope of supply. If the customer provides these cables, the cable must have a characteristic impedance of 120 Ω.

Maximum cable length

Connection	Max. cable length
Cables between engine and cabinets	≤ 45 m

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Connection	Max. cable length
MODBUS cable between Interface Cabinet and superordinated automation system (only for ethernet)	≤ 100 m
Cable between Interface Cabinet and Remote Operating Panel	≤ 100 m

Table 163: Maximum cable length

Installation works

During the installation period the customer has to protect the cabinets against water, dust and fire. It is not permissible to do any welding near the cabinets. The cabinets have to be fixed to the floor by screws.

If it is inevitable to do welding near the cabinets, the cabinets and panels have to be protected against heat, electric current and electromagnetic influences. To guarantee protection against current, all of the cabling must be disconnected from the affected components.

The installation of additional components inside the cabinets is only permissible after approval by the responsible project manager of MAN Energy Solutions.

Installation of sensor 1TE6000 „Ambient air temp”

The sensor 1TE6000 “Ambient air temp” (double Pt1000) measures the temperature of the (outdoor) ambient air. The temperature of the ambient air will typically differ from that in the engine room.

The sensor may be installed in the ventilation duct of the fan blowing the (outdoor) ambient air into the engine room. Ensure to keep the sensor away from the influence of heat sources or radiation. The image below shows two options of installing the sensors correctly:

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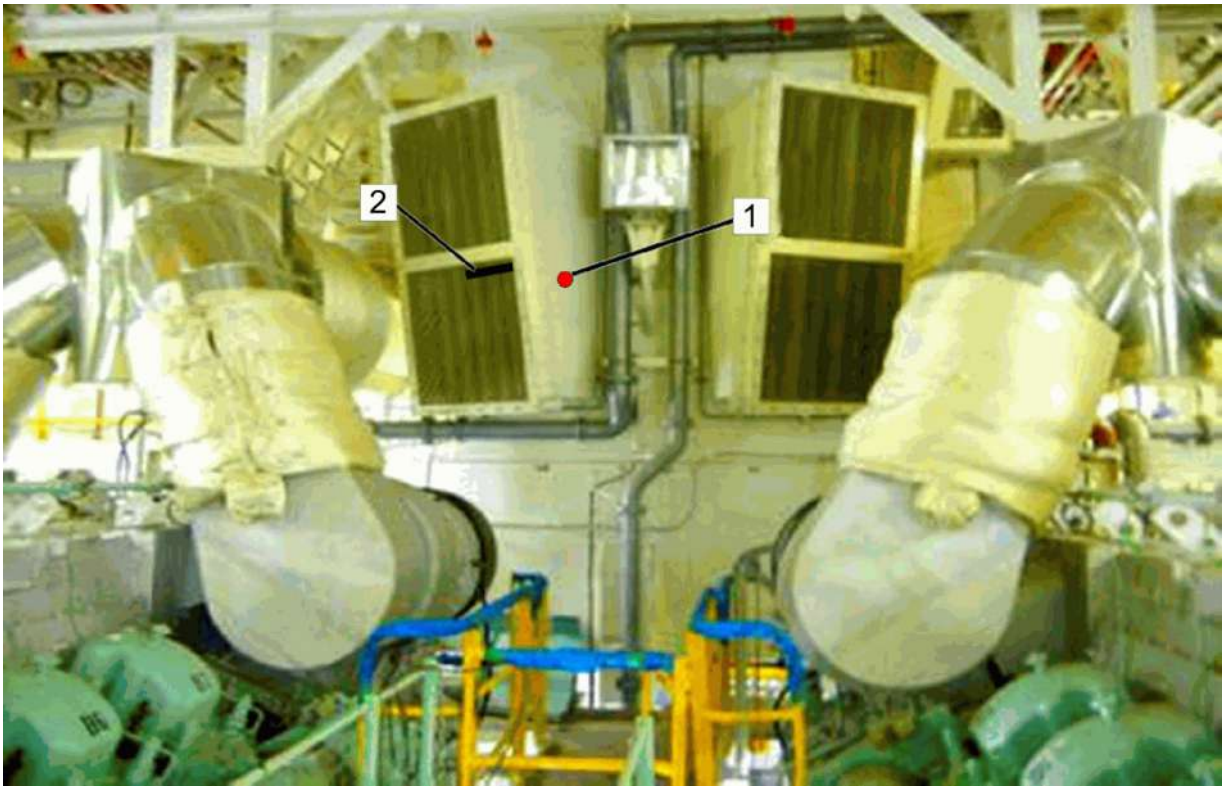

1 Hole drilled into the duct of the engine room ventilation. Sensor measuring the temperature of the airstream.

2 Self-designed holder in front of the duct.

Figure 105: Possible locations for installing the sensor 1TE6000

Note:
The sensor 1TE6100 “intake air temp” is not suitable for this purpose.

3.9 Engine-located measuring and control devices

Exemplary list for project planning

No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
Speed pickups							
1	1SE1004	speed pickup turbocharger speed	indication, supervision	-	turbocharger	Control Module/ Safety	-
2	1SE1005	speed pickup engine speed	camshaft speed and position detection	0–600 rpm/ 0–1,200 Hz	camshaft drive wheel	Control Module/ Alarm	-
3	2SE1005	speed pickup engine speed	camshaft speed and position detection	0–600 rpm/ 0–1,200 Hz	camshaft drive wheel	Control Module/ Safety	-

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
4	1SV1010	actuator engine fuel admission	speed and load governing in diesel mode	-	engine	Auxiliary Cabinet	-
5	1SCS1010	electric motor speed setpoint adjustment	integrated in 1SV1010, for remote speed setting in mech. mode	-	engine	Interface/Auxiliary Cabinet	Governor = PGG-EG 200
6	1GOS1010	limit switch mech. speed setpoint min	integrated in 1SV1010	-	engine	Control Module/ Alarm	Governor = PGG-EG 200
7	2GOS1010	limit switch mech. speed setpoint max	integrated in 1SV1010	-	engine	Control Module/ Alarm	Governor = PGG-EG 200
8	1SZ1010	solenoid in governor for engine stop	integrated in 1SV1010, for manual stop and auto shutdown	-	engine	Control Module/ Alarm	Governor = PGG-EG 200
Start and stop of engine							
9	1PS1011	pressure switch start air pressure after start valve	feedback start valve activated	0–10 bar	engine	Control Module/ Alarm	-
10	1SSV1011	solenoid valve engine start	actuated during engine start and slowturn	-	engine	Control Module/ Alarm	-
11	1HZ1012	push button local emergency stop	emergency stop from local control station	-	Local Operating Panel	Gateway Module	-
12	1SZV1012	solenoid valve engine shutdown	manual and auto-emergency shutdown	-	engine	Control Module/ Safety	-

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
13	1PS1012	pressure switch emergency stop air	feedback emergency stop, start-blocking active	0–10 bar	emergency stop air pipe on engine	Control Module/ Safety	-
14	1ESV1016	Solenoid valve speed governor mechanical operation switch-over	integrated in 1SV1010	-	engine	Control Module/ Alarm	Governor = PGG-EG 200
15	1SSV1017	solenoid valve starting interlock	3/2-way valve M371/1, blocking of manual start on engine	-	engine	Control Module/ Alarm	-
Variable Valve Timing (VVT)							
16	1EM1024	electric motor VVT setting row A/B	Variable Valve Timing	-	engine	Interface Cabinet	VVT
17	1GOS1024A/ B[1)]	limit switch VVT part load position row A/B, CS	feedback VVT part load position reached	-	engine cs	Extension Unit	VVT
18	2GOS1024A/ B[1)]	limit switch VVT full load position row A/B, CS	feedback VVT full load position reached	-	engine ccs	Extension Unit	VVT
19	3GOS1024A/ B[1)]	limit switch VVT part load position row A/B, CCS	feedback VVT part load position reached	-	engine ccs	Extension Unit	VVT
20	4GOS1024A/ B[1)]	limit switch VVT full load position row A/B, CCS	feedback VVT full load position reached	-	engine ccs	Extension Unit	VVT
21	1PT1024A/ B[1)]	pressure transmitter VVT hydraulic system "part load", row A/B	monitoring, alarm	-	engine	Extension Unit	VVT
22	2PT1024A/ B[1)]	pressure transmitter VVT hydraulic system "part load", row A/B	monitoring, alarm	-	engine	Extension Unit	VVT
Variable Injection Timing (VIT)							

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
23	1EM1028	electric motor VIT-setting	injection time setting	-	engine	Auxiliary Cabinet	variable injection timing
24	1UV1028	solenoid valve VIT adjustment	energise valve means remove hydraulic brake for VIT-adjustment	-	engine	Control Module/ Alarm	variable injection timing
25	2UV1028	solenoid valve VIT adjustment	energise valve means remove hydraulic brake for VIT-adjustment	-	engine	Control Module/ Alarm	variable injection timing
26	1PS1028	pressure switch hydraulic oil VIT-brake 1	release VIT-motor at sufficient pressure	0–6 bar	engine	Control Module/ Alarm	variable injection timing
27	2PS1028	pressure switch hydraulic oil VIT-brake 2	release VIT-motor at sufficient pressure	0–6 bar	engine	Control Module/ Alarm	variable injection timing
28	1GOS1028	limit switch VIT early position	VIT position feedback	-	engine	Control Module/ Alarm	variable injection timing
29	2GOS1028	limit switch VIT late position	VIT position feedback	-	engine	Control Module/ Alarm	variable injection timing
Variable Turbine Area (VTA)							
30	1EM1040	Electric motor VTA positioning	nozzle ring vanes pitch adjustment	-	TC on engine	Auxiliary Cabinet	variable turbine area
31	2EM1040	Electric motor VTA positioning	nozzle ring vanes pitch adjustment	-	TC on engine	Auxiliary Cabinet	variable turbine area
Main bearings							
32	xTE1064	double temp sensors, main bearings	indication, alarm, engine protection	0–120 °C	engine	Control Modules	main bearing temp monitoring
Turning gear							

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
33	1GOS1070	limit switch turning gear engaged	start blocking while turning gear engaged	-	engine	Control Module/ Alarm	-
34	1SSV1070	pneumatic valve turning gear engaged	3/2-way valve M306, start blocking while turning gear engaged	-	engine	-	-
Slow turn							
35	1SSV1075	solenoid valve slow turn	3/2-way valve M329/3, slow turn	-	engine	Control Module/ Alarm	-
36	2SSV1075	solenoid valve slow turn	3/2-way valve M371/2, start air blocking during slow turn	-	engine	Control Module/ Alarm	-
Jet assist							
37	1SSV1080	solenoid valve for jet assist	turbocharger acceleration by jet assist	-	engine	Control Module/ Alarm	Jet assist
Knock control							
38	xXE1200A/B[1)]	knock sensor cylinder x	knock event detection	0–100	engine	Knock Control Module	-
Combustion Control							
39	xPT1300A/B[1)]	Pressure transmitter combustion pressure cylinder x	-	0–250 bar	engine	Cylinder Pressure Module	-
Lube oil system							
40	1PT2170	pressure transmitter, lube oil pressure engine inlet	alarm at low lube oil pressure	0–10 bar	engine	Control Module/ Alarm	-

No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
41	2PT2170	pressure transmitter, lube oil pressure engine inlet	auto shut-down at low pressure	0–10 bar	Local Operating Panel	Control Module/ Safety	-
42	1TE2170	double temp sensor, lube oil temp engine inlet	alarm at high temp	0–120 °C	engine	Control Modules	-
43	1EM2470A/B[1)]	electric pump cylinder lubrication row A/B	cylinder lubrication line A/B	-	engine	Auxiliary Cabinet	-
44	1FE2470A/B[1)]	proximity switch cylinder lubrication row A/B	proximity switch cylinder lubrication row A	-	engine	Auxiliary Cabinet	-
45	1PT2570	pressure transmitter, lube oil pressure turbocharger inlet	alarm at low lube oil pressure	0–6 bar	engine	Control Module/ Alarm	-
46	2PT2570	pressure transmitter, lube oil pressure turbocharger inlet	auto shut-down at low lube oil pressure	0–6 bar	engine	Control Module/ Safety	-
47	1TE2580	double temp sensor, lube oil temp turbocharger drain	alarm at high temp	0–120 °C	engine	Control Modules	-
Crankcase ventilation							
48	1PT2800	pressure transmitter crankcase pressure	input for alarm system	–20 – +20 mbar	engine	Control Module/ Alarm	-
49	2PT2800	pressure transmitter crankcase pressure	input for safety system	–20 – +20 mbar	engine	Control Module/ Safety	-
Oil mist detection							
50	xQE2870	opacity sensor crankcase compartment x	oil-mist detection	-	engine	OMD	OMD= Triton
51	1QTIA2870	oil mist detector, oil mist concentration in crankcase	oil mist monitoring	-	engine	-	oil mist detection
52	1QS2870	opacity switch oil mist in crankcase	integrated in 1QTIA2870	-	engine	Control Module/ Alarm	oil mist detection
53	2QS2870	opacity switch oil mist in crankcase	integrated in 1QTIA2870	-	engine	Control Module/ Safety	oil mist detection

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
54	1ES2870	binary contact oil mist detector system ready	integrated in 1QTIA2870	-	engine	Control Module/ Safety	oil mist detection
Splash oil							
55	xTE2880	double temp sensors, splash oil temp rod bearings	splash oil supervision	0–120 °C	engine	Control Modules	-
Cooling water systems							
56	1TE3168	double temp sensor HT water temp charge air cooler inlet	for EDS visualisation and control of preheater valve	0–120 °C	turbocharger	Control Module/ Alarm	-
57	1PT3170	pressure transmitter, HT cooling water pressure engine inlet	alarm at low pressure	0–6 bar	engine	Control Module/ Alarm	-
58	2PT3170	pressure transmitter, HT cooling water pressure engine inlet	detection of low cooling water pressure	0–6 bar	engine	Control Module/ Alarm	-
59	1TE3170	double temp sensor, HTCW temp engine inlet	alarm, indication	0–120 °C	engine	Control Modules	-
60	1TE3180	temp sensor, HT water temp engine outlet	-	0–120 °C	engine	Control Modules	-
61	1PT3470	pressure transmitter, nozzle cooling water pressure engine inlet	alarm at low cooling water pressure	0–10 bar	engine	Control Module/ Alarm	-
62	2PT3470	pressure transmitter, nozzle cooling water pressure engine inlet	alarm at low cooling water pressure	0–10 bar	engine	Control Module/ Safety	-
63	1TE3470	double temp sensor, nozzle cooling water temp engine inlet	alarm at high cooling water temp	0–120 °C	engine	Control Modules	-
64	1PT4170	pressure transmitter, LT water pressure charge air cooler inlet	alarm at low cooling water pressure	0–6 bar	engine	Control Module/ Alarm	-

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
65	2PT4170	pressure transmitter, LT water pressure charge air cooler inlet	alarm at low cooling water pressure	0–6 bar	engine	Control Unit	-
66	1TE4170	double temp sensor, LT water temp charge air cooler inlet	alarm, indication	0–120 °C	LT pipe charge air cooler inlet	Control Modules	-
Fuel system							
67	1PT5070	pressure transmitter, fuel pressure engine inlet	remote indication and alarm	0–16 bar	engine	Control Module/ Alarm	-
68	2PT5070	pressure transmitter, fuel pressure engine inlet	remote indication and alarm	0–16 bar	engine	Control Module/ Safety	-
69	1TE5070	double temp sensor, fuel temp engine inlet	alarm at high temp in MDO-mode and for EDS use	0–200 °C	engine	Control Modules	-
70	1LS5076A/B[1)]	level switch fuel pipe break leakage	high pressure fuel system leakage detection	0–2,000 bar	engine	Control Module/ Alarm	-
71	1LS5080A/B[1)]	level switch pump- and nozzle leakage row A/B	alarm at high level	-	leakage fuel oil monitoring tank FSH-001	Control Module/ Alarm	-
72	2LS5080A/B[1)]	level switch dirty oil leakage pump bank CS row A/B	alarm at high level	-	pump bank leakage monitoring CS	Control Module/ Alarm	-
73	3LS5080A/B[1)]	level switch dirty oil leakage pump bank CCS row A/B	alarm at high level	-	pump bank leakage monitoring CCS	Control Module/ Alarm	-
Pilot fuel system							
74	1FCV5275	suction throttle valve pilot fuel high-pressure pump	pilot fuel quantity control	-	engine	Injection Module 1	-
75	1PT5275	pressure transmitter pilot fuel supply pressure	pilot fuel low pressure system	0–16 bar	engine	Control Module/ Alarm	-

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
76	2PT5275	pressure transmitter pilot fuel supply pressure	pilot fuel low pressure system	0–16 bar	engine	Control Module/ Safety	-
77	1PDS5275	differential pressure switch pilot fuel fine filter	fine filter contamination monitoring	-	engine	Control Module/ Alarm	-
78	1TE5275	temp sensor pilot fuel temp engine inlet	-	-	engine	Control Module/ Alarm	-
79	1PT5276	pressure transmitter pilot fuel rail	-	0–2,000 bar	engine	Injection Module 1	-
80	2PT5276	pressure transmitter pilot fuel rail	-	0–2,000 bar	engine	Injection Module 1	-
81	1LS5276	level switch pilot fuel leakage high-pressure pump	-	-	engine	Control Module/ Alarm	-
82	1EM5276	electric motor pilot fuel high-pressure pump	-	-	engine	Auxiliary Cabinet	-
83	xFSV5278A/ B[1)]	solenoid valve pilot fuel injector x	-	-	engine	Injection Module 1/2	-
84	1FSV5280	flushing valve pilot fuel rail	unloading of pilot fuel high pressure fuel system	-	engine	-	-
85	1PZV5281	pressure limiting valve pilot fuel rail	mechanical pressure relief pilot fuel rail	-	engine	-	-
86	1TE5282	temp sensor temp after pilot fuel flushing- and pressure limiting valve	-	-	engine	-	-
Gas system							
87	1PT5870	pressure transmitter double-walled gas pipe	double-walled gas pipe ventilation monitoring	–10 – 0 mbar	engine	GVUCC	-

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
88	2PT5870	pressure transmitter double-walled gas pipe	double-walled gas pipe ventilation monitoring	–10 – 0 mbar	engine	GVUCC	-
89	1PT5884	pressure transmitter main gas pressure engine inlet	-	0–10 bar	engine	Control Module/ Alarm; Injection Module 1	-
90	2PT5884	pressure transmitter main gas pressure engine inlet	-	0–10 bar	engine	Control Module/ Safety	-
91	xFSV5885A/B[1)]	solenoid valve main gas injector x	-	-	engine	Injection Module 1/2	-
92	1PT5887A/B[1)]	pressure transmitter gas pressure inert gas purge valve A/B outlet	-	-	engine	CM/Alarm Module 1	-
93	1FSV5888A/B[1)]	purge valve inert gas	purging of gas system with inert gas	0–10 bar	engine	Control Module/ Alarm	-
94	1PT5889	pressure transmitter gas pressure inert gas purge valve inlet	for inert gas availability monitoring	0–10 bar	engine	Control Module/ Alarm	-
Charge air system							
95	1PT6100	pressure transmitter, intake air pressure	for EDS visualisation	–20 – +20 mbar	intake air duct after filter	Control Module/ Alarm	-
96	1TE6100	double temp sensor, intake air temp	temp input for charge air blow-off and EDS visualisation	0–120 °C	intake air duct after filter	Control Module/ Alarm	-
97	1TE6170 A/B[1)]	double temp sensor, charge air temp charge air cooler inlet	-	0–300 °C	engine	Control Modules	-
98	1PT6180A/B[1)]	pressure transmitter, charge air pressure before cylinders	input for alarm system	0–6 bar	engine	Control Modules	-
99	2PT6180A/B[1)]	pressure transmitter, charge air pressure before	input for safety system	0–6 bar	engine	Control Modules	-

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
100	3PT6180A/B[1)]	pressure transmitter, charge air pressure before cylinders	input for injection module	0–6 bar	engine	Injection Module 1	-
101	1TE6180A/B[1)]	double temp sensor, charge air temp after charge air cooler	alarm at high temp	0–120 °C	engine	Control Modules	-
102	1TCV6180	temp control valve CA temp	control of LTCW temp for CA cooler stage 2	-	engine	Auxiliary Cabinet	-
103	1ES6180	binary contact decrease charge air temp	desired value output to 1TCV6180	-	engine	Auxiliary Cabinet	-
104	2ES6180	binary contact increase charge air temp	desired value output to 1TCV6180	-	engine	Auxiliary Cabinet	-
105	1GT6180	analog input signal charge air temp control valve position feedback	actual value input from 1TCV6180	-	engine	Auxiliary Cabinet	-
106	7GOS6180	limit switch charge air temp control valve closed	position feedback 1TCV6180 for preheating release	-	engine	Auxiliary Cabinet	-
107	1PT6182	pressure transmitter cooling air pressure TC inlet	monitoring of cooling air flow for turbine disc cooling	-	turbocharger	Control Module/ Alarm	Turbine disc cooling
108	1PCV6185A/B[1)]	variable flap compressor bypass A/B	lambda control, CA pressure relief	-	engine	-	-
109	1GT6185A/B[1)]	position feedback signal from compressor bypass A/B	actual value input from bypass flap	-	engine	Control Module/ Alarm	-
110	1ET6185A/B[1)]	position setpoint for compressor bypass A/B	desired value output to bypass flap	-	engine	Control Module/ Alarm	-

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No.	Measuring point	Description	Function	Measuring Range	Location	Connected to	Depending on option
Exhaust gas system							
111	1XCV6570	variable flap waste gate	exhaust gas blow off and lambda-control	-	engine	Extension Unit	-
112	1ET6570	position setpoint for waste gate	-	-	engine	Extension Unit	-
113	1GT6570	position feedback signal from waste gate	-	-	engine	Extension Unit	-
114	xTE6570A/B[1)]	double thermo-couples, exhaust gas temp cylinders A/B	indication, alarm, engine protection	0–800 °C	engine	Control Modules	-
115	1TE6575	double thermo-couples, exhaust gas temp before turbocharger	indication, alarm, engine protection	0–800 °C	engine	Control Modules	-
116	1TE6580	double thermo-couples, exhaust gas temp after turbocharger	indication	0–800 °C	engine	Control Modules	-
Control air, start air, stop air							
117	1PT7170	pressure transmitter, starting air pressure	engine control, remote indication	0–40 bar	engine	Control Module/ Alarm	-
118	2PT7170	pressure transmitter, starting air pressure	engine control, remote indication	0–40 bar	engine	Control Module/ Safety	-
119	1PT7180	pressure transmitter, emergency stop air pressure	alarm at low air pressure	0–40 bar	engine	Control Module/ Alarm	-
120	2PT7180	pressure transmitter, emergency stop air pressure	alarm at low air pressure	0–40 bar	engine	Control Module/ Safety	-
121	1PT7400	pressure transmitter, control air pressure	remote indication	0–10 bar	engine	Control Module/ Alarm	-
122	2PT7400	pressure transmitter, control air pressure	remote indication	0–10 bar	engine	Control Module/ Safety	-

[1)] A-sensors: All engines; B-sensors: V engines only.

Table 164: List of engine-located measuring and control devices

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4 Specification for engine supplies

4.1 Explanatory notes for operating supplies – Dual fuel engines

4.1.1 Lube oil

The selection is mainly affected by the used fuel grade.

Main fuel	Lube oil type	Viscosity class	Base No. (BN)	
MGO (ISO-F-DMA, DFA)	Doped (HD) + additives	SAE 40	10 – 16 mg KOH/g	Depending on sulphur content
MDO (ISO-F-DMB, DFB)			10 – 20 mg KOH/g	
HFO (ISO-F-RM grades)	Medium-alkaline + additives		20 – 55 mg KOH/g	

Table 165: Main fuel/lube oil type

Selection of the lube oil must be in accordance with section Specification of lubricating oil (SAE 40) for dual fuel engines, Page 258, where it distinguishes between following operation modes:

- Pure gas operation
- Pure diesel operation or alternating gas/diesel operation
- Pure heavy fuel oil operation (> 2,000 h)
- Alternating gas/heavy oil operation

A base number (BN) that is too low is critical due to the risk of corrosion. A base number that is too high is, could lead to deposits/sedimentation and takes the risk of self ignition/knocking in gas mode.

In general DF engines would be assigned to the operating mode "Alternating gas/heavy oil operation". The aim of the lube oil concept for flexible fuel operation is to keep the BN of the lube oil between 20 and 30 mg KOH/g. The BN should not be less than 20 mg KOH/g with HFO operation and the BN should not be more then 30 mg KOH/g with gas operation.

Therefore it is recommended to use two lube oil storage tanks with BN20 (for gas mode) and BN40 (for HFO operation). First filling on lube oil servcie tank to be done with BN30 (mixture of both lube oils). During gas operation the specific lube oil consumption is replenished with BN20. During HFO operation the specific lube oil consumption is replenished with BN40.

The oils used (BN20 and BN40) must be of the same brand without fail (same supplier). This ensures that the oils are fully compatible with each other.

Be aware that a change from HFO to MDO/MGO as main fuel for an extended period will demand a change of the lube oil accordingly.

4.1.2 Operation with gaseous fuel

In gas mode, natural gas is to be used according to the qualities mentioned in the relevant section. If the engine is operated with liquid fuel, the gas valves and gas supply pipes are to be purged and vented.

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4.1.3 Operation with liquid fuel

The engine is designed for operation with heavy fuel oil, marine gas oil (DMA, DFA) and marine diesel oil (DMB, DFB) as per ISO 8217 fulfilling the stated requirements.

Additional requirements for HFO before engine:

- Water content before engine: Max. 0.20 %
- Al + Si content before engine: Max. 15 mg/kg

Engine operation with DM-grade fuel according to ISO 8217, viscosity ≥ 2 cSt at 40 °C

A) Short-term operation, max. 72 hours

Engines that are normally operated with heavy fuel, can also be operated with DM-grade fuel for short periods.

Boundary conditions:

- DM-grade fuel in accordance with stated specifications and a viscosity of ≥ 2 cSt at 40 °C.
- MGO-operation maximum 72 hours within a two-week period (cumulative with distribution as required).
- Fuel oil cooler switched on and fuel oil temperature before engine ≤ 45 °C. In general, the minimum viscosity before engine of 1.9 cSt must not be undershoot!

B) Long-term (> 72 h) or continuous operation

For long-term (> 72 h) or continuous operation with DM-grade fuel special engine- and plant-related planning prerequisites must be set and special actions are necessary during operation.

Following features are required on engine side:

- None.

Following features are required on plant side:

- Layout of fuel system to be adapted for low-viscosity fuel (capacity and design of fuel supply and booster pump).
- Cooler layout in fuel system for a fuel oil temperature before engine of ≤ 45 °C (min. permissible viscosity before engine 1.9 cSt).
- Nozzle cooling system with possibility to be turned off and on during engine operation.

Boundary conditions for operation:

- Fuel in accordance with MGO (DMA, DFA) and a viscosity of ≥ 2 cSt at 40 °C.
- Fuel oil cooler activated and fuel oil temperature before engine ≤ 45 °C. In general the minimum viscosity before engine of 1.9 cSt must not be undershoot!
- Nozzle cooling system switched off.

Continuous operation with MGO (DMA, DFA):

- Lube oil for diesel operation (BN10-BN16) has to be used.

Operation with heavy fuel oil of a sulphur content of < 1.5 %

Previous experience with stationary engines using heavy fuel of a low sulphur content does not show any restriction in the utilisation of these fuels, provided that the combustion properties are not affected negatively.

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This may well change if in the future new methods are developed to produce low sulphur-containing heavy fuels.

If it is intended to run continuously with low sulphur-containing heavy fuel, lube oil with a low BN (BN30) has to be used. This is required, in spite of experiences that engines have been proven to be very robust with regard to the continuous usage of the standard lube oil (BN40) for this purpose.

Instruction for minimum admissible fuel temperature

- In general the minimum viscosity before engine of 1.9 cSt must not be undershoot.
- The fuel specific characteristic values “pour point” and “cold filter plugging point” have to be observed to ensure pumpability respectively filterability of the fuel oil.
- Fuel temperatures of approximately minus 10 °C and less have to be avoided, due to temporarily embrittlement of seals used in the engines fuel oil system and as a result their possibly loss of function.

4.1.4 Pilot fuel

- For ignition in gas mode, a small amount of pilot fuel is required. Table "Engine type and suitability of diesel fuel as pilot fuel" in section Specification of diesel fuel used as pilot oil for DF engines, Page 270, lists for each engine type the fuel types that are suitable for the respective pilot fuel system. Pilot fuel is to be used during operation with liquid fuel too, for cooling the injector needles.
- A filtering of the pilot fuel has to be provided to achieve cleanliness level 12/9/7 according to ISO 4406.

4.1.5 Engine cooling water

The quality of the engine cooling water required in relevant section has to be ensured.

Nozzle cooling system activation	
Kind of fuel	**Activated**
MGO (DMA, DFA)	No, see section Operation with liquid fuel, Page 256
MDO (DMB, DFB)	No
HFO	Yes
Gas	Yes

Table 166: Nozzle cooling system activation

4.1.6 Intake air

The quality of the intake air as stated in the relevant sections has to be ensured.

4.1.7 Urea

The quality of the urea as stated in section Specification of urea solution, Page 295.

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4.1.8 Compressed air – SCR catalyst

The compressed air must be free of oil and other contaminations. The quality of the compressed air as stated in section Specification of compressed air, Page 293.

4.2 Specification of lubricating oil (SAE 40) for dual-fuel engines 35/44DF, 51/60DF and 58/64RDF

General

The specific output achieved by modern diesel engines combined with the use of fuels that satisfy the quality requirements more and more frequently increase the demands on the performance of the lubricating oil which must therefore be carefully selected.

Doped lubricating oils (HD oils) have a proven track record as lubricants for the drive, cylinder, turbocharger and also for cooling the piston. Doped lubricating oils contain additives that, amongst other things, ensure dirt absorption capability, cleaning of the engine and the neutralisation of acidic combustion products.

Only lubricating oils that have been approved by MAN Energy Solutions may be used. These are listed in the tables below.

Specifications

Base oil

The base oil (doped lubricating oil = base oil + additives) must have a narrow distillation range and be refined using modern methods. If it contains paraffins, they must not impair the thermal stability or oxidation stability.

The base oil must comply with the limit values in the table Target values for base oils, Page 258, particularly in terms of its resistance to ageing.

Evaporation tendency

The evaporation tendency must be as low as possible as otherwise the oil consumption will be adversely affected.

Additives

The additives must be dissolved in the oil and their composition must ensure that as little ash as possible remains following combustion.

The ash must be soft. If this prerequisite is not met, it is likely the rate of deposition in the combustion chamber will be higher, particularly at the outlet valves and at the turbocharger inlet housing. Hard additive ash promotes pitting of the valve seats, and causes valve burn-out, it also increases mechanical wear of the cylinder liners.

Additives must not increase the rate, at which the filter elements in the active or used condition are blocked.

Lubricating oil additives

The use of other additives with the lubricating oil, or the mixing of different brands (oils by different manufacturers and different brands of the same manufacturer), is not permitted as this may impair the performance of the existing additives which have been carefully harmonised with each another, and also specially tailored to the base oil.

Properties/characteristics	Unit	Test method	Limit value
Structure	–	–	Preferably on paraffin basis
Cold behaviour, still fluid	°C	ASTM D2500	-15

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Properties/characteristics	Unit	Test method	Limit value
Flash point (Cleveland)	°C	ASTM D92	> 200
Ash content (oxidised ash)	weight %	ASTM D482	< 0.02
Coke residue (according to Conradson)	weight %	ASTM D189	< 0.50
Insoluble n-heptane	weight %	ASTM D4055 or DIN 51592	< 0.2
Evaporation loss	weight %	-	< 2

Table 167: Target value for base oils

Oil for mechanical-hydraulic speed governor

Multigrade oil 5W40 should ideally be used in mechanical-hydraulic controllers with a separate oil sump, unless the technical documentation for the speed governor specifies otherwise. If this oil is not available when filling, 15W40 oil may be used instead in exceptional cases. In this case, it makes no difference whether synthetic or mineral-based oils are used.

The military specification for these oils is O-236.

The oil quality prescribed by the manufacturer must be used for the remaining engine system components.

Selection of lubricating oils/ warranty

Most of the oil manufacturers are in close regular contact with engine manufacturers, and can therefore provide information on which oil in their specific product range has been approved by the engine manufacturer for the particular application. Irrespective of the above, the lubricating oil manufacturers are in any case responsible for the quality and characteristics of their products. If you have any questions, we will be happy to provide you with further information.

Oil during operation

There are no prescribed oil change intervals for MAN Energy Solutions medium-speed engines. The oil properties must be regularly analysed. The oil can be used for as long as the oil properties remain within the defined limit values (see tables entitled Limit values). An oil sample must be analysed every 1 – 3 months (see maintenance schedule).

Safety/environmental protection

If operating fluids are not handled correctly, this can pose a risk to health, safety and the environment. The corresponding manufacturer's instructions must be followed.

Analyses

A monthly analysis of lube oil samples is mandatory for safe engine operation. We can analyse samples for customers in the MAN Energy Solutions PrimeServLab.

Operating modes

Operating modes

The engine has an extremely high flexibility, as it can run on gas, diesel and heavy fuel oil (HFO).

Every fuel places different demands on the lubricating oil. To ensure that the right lubricating oil is found for the application concerned, four different operating modes have been identified:

1. Pure gas operation (> 2000 h)
2. Pure diesel operation or alternating gas/diesel operation
3. Pure heavy fuel oil operation (> 2000 h)
4. Alternating gas/heavy oil operation

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Lubrication oil for 100% gas operation

A special lubricating oil for gas engines with low ash content must be used in engines that are only operated with gas. The proportion of sulphated ash must not exceed 1%.

Only lubrication oils approved by MAN Energy Solutions may be used.

The list of the currently approved lubricating oils is available at https://corporate.man-es.com/lubrication.

Property	Unit	Limit value	Method
Viscosity at 40° C	mm²/s	min. 100	ISO 3104; ASTM D445, ASTM D7042
		max. 190	
TBN	% fresh oil BN	min. 50	ISO 3771
Water content	% vol.	max. 0.2%	DIN 51777, ASTM D6304, ISO 3733
TAN	mg KOH/g	max. 4.0 for fresh oil and TBN > TAN	ASTM D664
Oxidation	Abs/cm	max. 25	DIN 51453
Metal content	(Reference value only)	–	–
Iron	ppm	max. 50	ASTM D5185
Chromium		max. 10	
Copper		max. 15	
Lead		max. 20	
Tin		max. 10	
Aluminium		max. 20	

Table 168: Limit values for lubrication oils during operation (100% gas operation)

Lubricating oil for diesel operation or alternating gas/diesel operation

A lubricating oil with a higher BN (10 –16 mg KOH/g) is recommended due to the sulphur content of the fuel in dual-fuel engines that are exclusively operated with diesel oil, are operated more than 40 % of the time with diesel oil or are operated for more than 500 hours a year using diesel with an extremely high sulphur content (S > 0.5 %).

Neutralisation capability

The neutralisation capability (ASTM D2896) must be high enough to neutralise the acidic products produced during combustion. The reaction time of the additive must be harmonised with the process in the combustion chamber.

The list of the currently approved lubricating oils is available at https://corporate.man-es.com/lubrication.

	Limit value	Procedure
Viscosity at 40 °C	110 – 220 mm²/s	ASTM D7042, ASTM D445, DIN EN 16896 or ISO 3104
Base number (BN)	at least 50 % of the fresh oil	ISO 3771
Flash point (PM)	min. 185 °C	ISO 2719
Water content	max. 0.20 % (max. 0.5 % for brief periods)	DIN 51777 or ASTM D6304

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	Limit value	Procedure
n-heptane insoluble	max. 1.5 %	DIN 51592 or IP 316
Metal content	dependent on engine type and operating conditions	–
Guide value only Fe Cr Cu Pb Sn Al	. max. 50 ppm max. 10 ppm max. 15 ppm max. 20 ppm max. 10 ppm max. 20 ppm	ASTM D5185 or DIN 51399-1
For operation with bio-fuels: proportion of bio-fuel	max. 12 %	FT-IR

Table 169: Limit values for used lubricating oil

Lubrication oil for 100% heavy fuel oil operation (HFO) (51/60DF and 35/44DF only)

Lubricating oils of medium alkalinity must be used for engines that run on HFO. HFO engines must not be operated with lubricating oil for gas engines. Oils of medium alkalinity contain additives that, among other things, increase the neutralisation capacity of the oil and facilitate high solubility of fuel constituents.

Cleaning efficiency

The cleaning efficiency must be high enough to prevent formation of combustion-related carbon deposits and tarry residues. The lubricating oil must prevent fuel-related deposits.

Dispersion capability

The selected dispersibility must be such that commercially-available lubricating oil cleaning systems can remove harmful contaminants from the oil used, i.e. the oil must possess good filtering properties and separability.

Neutralisation capability

The neutralisation capability (ASTM D2896) must be high enough to neutralise the acidic products produced during combustion. The reaction time of the additive must be harmonised with the process in the combustion chamber.

Information on selecting a suitable BN is provided in the table below.

Approximate BN (mg KOH/g Öl)	Engines/Operating conditions
20	Marine diesel oil (MDO) with a poor quality or heavy fuel oil with a sulphur content of less than 0.5 %.
30	For pure HFO operation only with a sulphur content < 1.5 %.
40	For pure HFO operation in general, providing the sulphur content is > 1.5 %.
50	If BN 40 is not sufficient in terms of the oil service life or maintaining engine cleanliness (high sulphur content in fuel, extremely low lubricating oil consumption).

Table 170: Selecting the base number (BN)

The list of the currently approved lubricating oils is available at https://corporate.man-es.com/lubrication.

	Limit value	Procedure
Viscosity at 40° C	110 – 220 mm²/s	ASTM D7042, ASTM D445, DIN EN 16896 or ISO 3104

	Limit value	Procedure
Base number (BN)	at least 50% of the fresh oil	ISO 3771
Flash point (PM)	min. 185°C	ISO 2719
Water content	max. 0.2% (max. 0.5% for brief periods)	DIN 51777 or ASTM D6304
n-heptane insoluble	max. 1.5%	DIN 51592 or IP 316
Metal content	dependent on engine type and operating conditions	–
Guide value only Fe Cr Cu Pb Sn Al	 max. 50 ppm max. 10 ppm max. 15 ppm max. 20 ppm max. 10 ppm max. 20 ppm	ASTM D5185 or DIN 51399-1

Table 171: Limit values for used lubricating oil (100% heavy fuel oil operation)

Alternating gas/heavy fuel oil operation (51/60DF only)

As already explained above, when operating with heavy fuel oil (HFO) a lubricating oil with a high base number (BN) is required so as to ensure the neutralisation of acidic combustion products and also a strong cleaning action to counter the effects of the fuel components (prevention of deposits). This high neutralisation capacity (BN) is accompanied by a high ash content of the lubricating oil.

Ash from the lubricating oil can accumulate in the combustion chamber and exhaust system. Ash from unburned BN additives in particular can accumulate in the combustion chamber. In gas engines, these kinds of deposits can act as "hot spots" at which the gas-air mixture ignites at the wrong time thus causing knocking.

The engine has been proven to have an exceptionally low sensitivity to lubricating oils with high ash content. Long-term gas operation using lubricating oil with BN 30 has given no cause for concern.

The aim of the lubricating oil concept for flexible fuel operation is to keep the BN of the lubricating oil between 20 and 30 mg KOH/g. The BN should not be less than 20 with HFO operation and the BN should not be more than 30 with gas operation. This can be achieved by using two oils when refilling. Oil with BN 40 is refilled during HFO operation, and oil with BN 20 is refilled during gas operation. Initial filling is carried out using oil with BN 30, which can be produced by blending oils with BN 20 and BN 40 in the engine. The oils used (BN 20 and BN 40) must be of the same brand without fail (same supplier). This ensures that the oils are fully compatible with one another.

If only low-sulphur fuel (< 1,5 %) is used as HFO, BN 30 lubrication oil can be used for operation with HFO as well as gas.

The list of the currently approved lubricating oils is available at https://corporate.man-es.com/lubrication.

	Limit value	Procedure
Viscosity at 40° C	110 – 220 mm²/s	ASTM D7042, ASTM D445, DIN EN 16896 or ISO 3104
Base number (BN)	20 – 30 mgKOH/g	ISO 3771

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	Limit value	Procedure
Flash point (PM)	min. 185°C	ISO 2719
Water content	max. 0.2% (max. 0.5% for brief periods)	DIN 51777 or ASTM D6304
n-heptane insoluble	max. 1.5%	DIN 51592 or IP 316
Metal content	dependent on engine type and operating conditions	–
Guide value only Fe Cr Cu Pb Sn Al	 max. 50 ppm max. 10 ppm max. 15 ppm max. 20 ppm max. 10 ppm max. 20 ppm	ASTM D5185 or DIN 51399-1

Table 172: Limit values for used lubrication oil (alternating gas/heavy fuel oil operation)

4.3 Specification of natural gas

Gas types and gas quality

Natural gas is obtained from a wide range of sources. They can be differentiated not only in terms of their composition and processing, but also their energy content and calorific value.

Combustion in engines places special demands on the quality of the gas composition.

The following section explains the most important gas properties.

Requirements for natural gas

The gas should:

- comply with the general applicable specifications for natural gas, as well as with specific requirements indicated in the table Requirements for natural gas, Page 265.
- be free of dirt, dry and cooled (free of water, hydrocarbon condensate and oil) when fed to the engine. If the dirt concentration is higher than 50 mg/Nm^3, a gas filter must be installed upstream of the supply system.

You can check the gas quality using a gas analyser.

Measures

In the gas distribution systems of different cities that are supplied by a central natural gas pipeline, if not enough natural gas is available at peak times, a mixture of propane, butane and air is added to the natural gas in order to keep the calorific value of Wobbe index constant. Although this does not actually change the combustion characteristics for gas burners in relation to natural gas, the methane number is decisive in the case of turbocharged gas engines. It falls drastically when these kind of additions are made.

To protect the engine against damage in such cases, the MAN Energy Solutions gas engines are provided with antiknock control.

Methane number

The most important prerequisite that must be met by the gas used for combustion in the gas engine is knock resistance. The reference for this evaluation is pure methane which is extremely knock-resistant and is therefore the name used for the evaluation basis:

- Methane number (MN). Calculation according to EN 16726 using GasCalc 2.5.2

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Pure methane contains the methane number 100; hydrogen was chosen as the zero reference point for the methane number series as it is extremely prone to knocking. See the table titled Anti-knocking characteristic and methane number, Page 264.

However, pure gases are very rarely used as fuel in engines. These are normally natural gases that also contain components that are made up of high-quality hydrocarbons in addition to knock-resistant methane and often significantly affect the methane number. It is clearly evident that the propane and butane components of natural gas reduce the anti-knock characteristic. In contrast, inert components, such as N_2 and CO_2, increase the anti-knock characteristic. This means that methane numbers higher than 100 are also possible.

Anti-knock characteristic of different gases expressed as methane number (MN)

Gas	Methane number (MN)
Hydrogen	0.0
N-butane 99 %	2.0
Butane	10.5
Butadiene	11.5
Ethylene	15.5
β-butylene	20.0
Propylene	20.0
Isobutylene	26.0
Propane	35.0
Ethane	43.5
Carbon monoxide	73.0
Natural gas	70.0 – 96.0
Natural gas + 8% N_2	92.0
Natural gas + 8% CO_2	95.0
Pure methane	100.0
Natural gas + 15% CO_2	104.4
Natural gas + 40% N_2	105.5

Table 173: Anti-knock characteristic and methane number

Determining the methane number

MAN Energy Solutions can determine the gas methane number with high precision by analyzing the gas chemistry.

The gas analysis should contain the following components in vol. % or mol %:

Carbon dioxide	CO_2
Nitrogen	N_2
Oxygen	O_2
Hydrogen	H_2

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Carbon monoxide	CO
Water	H_2O
Hydrogen sulphide	H_2S
Methane	CH_4
Ethane	C_2H_6
Propane	C_3H_8
I-butane	I-C_4H_{10}
N-butane	n-C_4H_{10}
Higher hydrocarbons	
Ethylene	C_2H_4
Propylene	C_3H_6

The sum of the individual components must be 100 %.

Gas	mol %
CH_4	94.80
C_2H_6	1.03
C_3H_8	3.15
C_4H_{10}	0.16
C_5H_{12}	0.02
CO_2	0.06
N_2	0.78

Table 174: Exemplary composition natural gas MN 80

Gas specification for gaseous fuels

The fuel at the inlet of the gas engine's gas valve unit must match the following specification.

Also note the requirements stated within section Specifications and requirements forthe gas supply of the engine, Page 154.

The following components must be within our limits (0° C and 101.325 kPa)[2)]

Characteristics	Unit	Limit	Value	Test method(s)
Methane number	-	min.[3)]	60	Calculated according EN 16726; correction with nitrogen and on factors C5+
Wobbe index	MJ/m^3	min.[4)]	34.6	-
Heating Value (LHV)	MJ/m^3	min.	28	-
Methane	mol-%	min.	60	DIN EN ISO 6975; DIN ISO 6974-4; DIN EN 15984
Ethane	mol-%	max.	35	DIN EN ISO 6975; DIN ISO 6974-4; DIN EN 15984
Propane	mol-%	max.	14	DIN EN ISO 6975; DIN ISO 6974-4; DIN EN 15984

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Characteristics	Unit	Limit	Value	Test method(s)
Butane	mol-%	max.	2	DIN EN ISO 6975; DIN ISO 6974-4; DIN EN 15984
Pentane	mol-%	max.	1	DIN EN ISO 6975; DIN ISO 6974-4; DIN EN 15984
Hexane	mol-%	max.	0.5	DIN EN ISO 6975; DIN ISO 6974-4
Heptane - Decane (sum)	mol-%	max.	0.1	DIN EN ISO 6975
Oxygen	mol-%	max.	3	DIN EN ISO 6975; DIN EN 15984
Hydrogen	mol-% mol-%	max. max.[5]	2 25	DIN EN ISO 6975; DIN EN 15984
Nitrogen and carbon dioxide (sum)	mol-% mol-%	max. max.[6]	22 15	DIN EN ISO 6975; DIN ISO 6974-4; DIN EN 15984
Total sulfur Hydrogen sulfide Mercaptan sulfur	mg/m^3 mg/m^3 mg/m^3	max. max. max.	30 5 6	DIN EN ISO 19739 DIN EN ISO 19739 DIN EN ISO 19739
Water < 10 bar > 10 bar	 mg/m^3 mg/m^3	 max. max.	 200 50	 EN ISO 10101 EN ISO 10101
Chlorine	mg/m^3	max.	10	EN 1911
Fluorine Chlorine and fluorine	mg/m^3 mg/m^3	max. max.	5 10	ISO 15713 EN 1911 and ISO 15713
Ammonia and gaseous amines	mg/m^3	max.	40	NEN 2826:1999; VDI 3496; NF X43-303
Silicon	mg/m^3	max.	1	EN ISO 16017-1; TDS-GC-MS
Particle Concentration Size	 mg/m^3 µm	 max. max.	 50 5	 ISO 8573-4 ISO 8573-4
Vapour, dust, fluid	technically free			ISO 8573-2
Hydrocarbon condensation	must be ruled out			-
Acetone, Ethyl Acetate, Furfural, Ethylene Dichloride, Methyl Ethyl Ketone, 2-Nitropropane, Tar etc. must not be present in the gas. If gas mixture/components deviate from a.m. specification, please contact MAN ES for project specific alignment.				

[1] The gas has to be fit for purpose.

[2] At engine inlet; please note the necessary conversion to standard conditions (0° C and 101.325 kPa) from project specific real conditions.

[3] Engine specific, a load reduction might be required (derating).

[4] Applicable for gas engines only.

[5] For 35/44G TS and 51/60G, G TS for hydrogen application only.

[6] If maximum hydrogen content of 25 mol-% is used, nitrogen and carbon dioxide have to be limited.

Table 175: Gas specification for gaseous fuels[1]

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4.4 Diesel fuel (DMA, DFA) specifications

Diesel oil

General information

Diesel fuel is a middle distillate refined from crude oil. It is also referred to as gas oil, marine gas oil (MGO) and diesel oil. It must not contain any residue from crude oil refining. The fuel may consist of synthetic components (e.g. BtL, CtL, GtL, & HVO).

Selection of suitable diesel fuel

Unsuitable or adulterated fuel generally results in a shortening of the service life of engine parts/ components, damage to these and to catastrophic engine failure. It is therefore important to select the fuel with care in terms of its suitability for the engine and the intended application. Through its combustion, the fuel also influences the emissions behaviour of the engine.

Specifications and approvals

The fuel quality varies regionally and is dependent on climatic conditions. All requirements specified in the current edition of ISO 8217 apply.

The following values must be maintained at the engine inlet:

Property	Unit		Threshold value[1)]	Standard[2)]
Kinematic viscosity at 40 °C[3)]	mm^2/s	Max.	6.000	ISO 3104, ASTM D7042, ASTM D445, DIN EN 16896
		Min.	2.000	
Density at 15°C	kg/m^3	Max.	890.0	ISO 3675, ISO 12185
		Min.	820.0	
Cetane index & cetane number		Min.	40	ISO 4264 & ISO 5165
Sulphur content[4)]	% (m/m)	Max.	1.0	ISO 8754, ISO 14596, ASTM D 4294, DIN 51400-10
Flash point[5)]	°C	Min.	60.0	ISO 2719
Hydrogen sulphide	mg/kg	Max.	2.0	IP 570
Acid number	mg KOH/g	Max.	0.5	ASTM D664
Corrosion on copper	Class	Max.	1	ISO 2160
Oxidation stability[6)]	g/m^3	Max.	25	ISO 12205, EN 15751
	h	Min.	20	
Fatty acid methyl ester (FAME) content[7)]	% (V/V)	Max.	7.0	ASTM D7963, IP 579, EN 14078
Carbon residue[8)]	%(m/m)	Max.	0.30	ISO 10370
Appearance	–	–	Clear & haze free	visually
Water content	% (m/m)	Max.	0.02	DIN 51777, DIN EN 12937, ASTM D6304
Ash content	% (m/m)	Max.	0.010	ISO 6245
Lubricity[9)]	µm	Max.	520	ISO 12156-1, ASTM D6079

Table 176: Requirements for diesel fuel

Remarks:

[1] The fuel must be suitable for the intended application. It must not contain any substance in a concentration that causes additional air pollution, is harmful for personnel, jeopardises ship safety and/or has an adverse effect on machine performance. The fuel must be free from non-ferrous metals according to DIN EN 16476.

[2] Always in relation to the currently applicable edition.

[3] Specific requirements of the injection system must be taken into account.

[4] Independent of the maximum permissible sulphur content, local laws and regulations must be adhered to.

[5] SOLAS specification. A lower flash point is possible for non-SOLAS-regulated applications.

[6] If there is more than 2% (V/V) FAME, an analysis as per EN15751 must additionally be performed.

[7] The FAME must either be in accordance with EN 14214 or with ASTM D6751.

[8] Determined on 10% distillation residue.

[9] Diameter of the corrected wear scar (WSD).

The following fuels are approved for use:

Classes ISO F-DMA & DMZ as per ISO 8217 in the current edition.

Class ISO F-DFA & DFZ as per ISO 8217 in the current edition with additional requirements regarding oxidation stability.

Diesel fuel as per EN 590 in the current edition with additional requirement regarding flash point >60 °C in SOLAS regulated areas.

Diesel fuel no. 2-D as per ASTM D975-15 with additional requirement regarding flash point >60 °C in SOLAS regulated areas.

Synthetic diesel fuel as per EN 15940 in the current edition with additional requirement regarding flash point >60 °C in SOLAS regulated areas. To obtain the full power output from engines with conventional injection systems, the minimum density in the table Requirements for the diesel fuel, Page 267 must be strictly adhered to.

Viscosity

In order to ensure sufficient lubrication, a minimum level of viscosity must be ensured at the fuel injection pump. The specified maximum temperature required to maintain a viscosity of more than 1.9 mm²/s upstream of the fuel injection pump depends on the fuel viscosity. The temperature of the fuel upstream of the fuel injection pump must not exceed 45 °C in any case. The lubricity requirements of the fuel upstream of the engine is a maximum of 520 µm WSD in each case.

Military fuel specification

The fuel types F-75 or F-76 as per NATO STANAG 1385 may be used. The following must be observed when doing so:

- According to the specification, the minimum permissible fuel viscosity for F-75 & F-76 is 1.7 mm²/s at 40 °C. This corresponds to a minimum fuel viscosity of 1.5 mm²/s at 45 °C (upstream of the engine).
- Use of a low-viscosity fuel (1.7 cSt at 40 °C) does not immediately cause the injection system to fail.
- A more severe leakage can trigger a variety of alarms!
- Extended operation of the engine with low-viscosity fuel leads to shortened maintenance intervals for the components of the injection system!
- If permanent operation with low-viscosity fuel is intended, a fuel cooling system should be installed. Contact MAN Energy Solutions for further information.

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- The lubricity requirements of the fuel for the engine are always max. 520 µm WSD as per ISO 12156-1.

Cold suitability

The cold suitability of the fuel is determined by the climatic requirements at the place of installation. It is the responsibility of the operating company to choose a fuel with sufficient cold suitability.

The cold suitability of a fuel may be determined and assessed using the following standards:

- Limit of filterability (CFPP) as per EN 116
- Pour point as per ISO 3016
- Cloud point as per EN 23015

To be able to draw a reliable conclusion, it is recommended to perform all three stated procedures.

Bio-fuel admixture

The DFA fuel may contain up to 7.0% of bio-fuel based on fatty acid methyl ester (FAME). The FAME to be added must comply with either EN14214 or ASTM D 6751. Compared to fuels on mineral oil basis only, fuels containing FAME have an increased tendency to oxidise and age and are more vulnerable to microbiological contamination. Furthermore, the fuel may contain an increased quantity of water. This why it is necessary to check the ageing stability at regular intervals when using this type of fuel. In addition, it is important to regularly check the water content of the fuel.

To minimise microbiological contamination, the tanks must be drained on a regular basis. During standstill periods this is required daily, otherwise weekly.

When first using fuels containing bio-diesel, deposits that have accumulated over a longer period of time may become detached. These deposits can block filters or even cause immediate damage.

Using bio-diesel blends in emergency power generators should be avoided. Bio-diesel fuel should be stored in separate reservoirs. Storing fuel containing bio-diesel for more than 6 months is generally not recommended. MAN Energy Solutions is not liable for damage and any possible consequences resulting from the use of fuel containing bio-diesel.

Analyses

Analysis of fuel oil samples is very important for safe engine operation. We can analyse fuel for customers at MAN Energy Solutions laboratory PrimeServLab.

4.5 Specifications of diesel fuel used as pilot oil for DF engines

General

Diesel fuel is a middle distillate from crude oil processing. Other names are: DMA, gas oil, marine gas oil (MGO), DMB, marine diesel oil (MDO). The fuel is permitted to contain synthetically produced components (e.g. BtL, CtL, GtL & HVO). DMA and DFA must contain no residue of any kind from crude oil processing.

The admixture of up to 7% biodiesel (FAME) is permitted. This results in the DFA and DFB fuel classes. Additional requirements are placed on these mixtures in terms of oxidation stability.

DMB and DFB are treated as a residual fuel in the transport chain. It is therefore possible that the fuel is contaminated with residue from crude oil processing. The fuel must be free of lubricating oil (ULO – used lubricating oil).

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Selection of suitable diesel fuel

Unsuitable or adulterated fuel generally results in a shortening of the service life of engine parts/ components, damage to these and to catastrophic engine failure. It is therefore important to select the fuel with care in terms of its suitability for the engine and the intended application. Through its combustion, the fuel also influences the emissions behaviour of the engine.

Engine	Fuel type
32/40R-DF, 51/60DF, 58/64R-DF	DMB or DFB in addition to DMA/DFA
35/44DF, 49/60DF	DMA or DFA

Table 177: Engine type and suitability of diesel fuel as pilot oil

Specifications and approvals

The fuel quality varies regionally and is dependent on climatic conditions. All requirements specified in the current edition of ISO 8217 apply.

The following values must be complied with at the engine inlet: [a)]

Property	Unit	Limit value	ISO-F class				Test procedure [b)]
			DMA	DFA	DMB	DFB	
Kinematic viscosity at 40 °C [c)]	mm^2/s [d)]	Max.	6.00		11.00		ISO 3104, ASTM D7042
		Min.	2.00		2.00		
Density at 15°C	kg/m^3	Max.	890.0		900.0		ISO 3675, ISO 12185
		Min.	820.0		820.0		
Cetane index and number	–	Min.	40		40 [i)]		ISO 4264 & ISO 5165
Sulphur content [e)]	%(m/m)	Max.	1.00		1.50		ISO 8754, ISO 14596, ASTM D4294, DIN 51400-10
Flash point [f)]	°C	Min.	60.0		60.0		ISO 2719
Hydrogen sulphide (H_2S)	mg/kg	Max.	2.0		2.0		IP 570
Acid number	mg KOH/g	Max.	0.5		0.5		ASTM D664
Corrosion on copper	Class	Max.	1		1		ISO 2160
Total sediment through hot filtration	%(m/m)	Max.	–		0.10 [g)]		ISO 10307-1
Oxidation stability	g/m^3	Max.	25	25	25	25	ISO 12205, EN 15751
	h	Min.	–	20	–	20	
Fatty acid methyl ester content (FAME) [h)]	% (V/V)	Max.	Free from	7.0	Free from	7.0	ASTM D7963, IP 579, EN 14078
Carbon residue	%(m/m)	Max.	0.30 [j)]		0.30		ISO 10370
Total aromatic content	%(m/m)	Max.	45		45		DIN EN 12916
Polycyclic aromatic hydrocarbons	%(m/m)	Max.	20		20		DIN EN 12916
Cloud point [j)]	°C	Winter	must be indicated		–		ISO 3015
	°C	Summer	–		–		

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Property	Unit	Limit value	ISO-F class				Test procedure [b]
			DMA	DFA	DMB	DFB	
Limit of filterability (CFPP) [j]	°C	Winter	must be indicated		–		EN 116, IP 309, IP 612
	°C	Summer	–		–		
Pour point (upper) [j]	°C	Winter	-6		0		ISO 3016
	°C	Summer	0		6		
Appearance	–	–	Clear & haze free [k]		[g], [k]		Visual
Water [k]	%(m/m)	Max.	0.020		0.020		DIN 51777, DIN EN 12937, ASTM D6304
Ash	%(m/m)	Max.	0.010		0.010		ISO 6245
Lubricity of corrected “wear scar diameter” (WSD) at 60 C	µm	Max.	520		520		ISO 12156-1, ASTM D6079

[a] Requirements applying to fuel at engine inlet; includes extra requirements and limit values in addition to ISO 8217.

[b] refers to the latest issue at all times.

[C] the applicable requirements for the injection system must be adhered to

[d] mm^2/s = 1 cSt.

[e] Despite this limit value, the system operator must comply with the regionally applicable sulphur limits in each case.

[f] SOLAS specification. A lower flash point is possible for non-SOLAS-regulated applications.

[g] if the sample is not free of contamination, it is mandatory that the hot filtration is carried out.

[h] The FAME must be in accordance with EN 14214 or ASTM D6751.

[i] conducted at 10% distillate residue.

[j] The pour point alone does not guarantee the pourability of the fuel under the prevailing conditions.

[k] The limit value for water must be adhered to regardless of the colouration of the fuel.

l) Current, commercially available DMB/DFB fuels usually have a Cetane Index (CI) rating > 40. Pursuant to ISO 8217, DMB/DFB has a minimum CI rating of 35. However, the combination of density and viscosity rules out a DMB/DFB with a CI rating < 40.

Table 178: Specification for marine distillate fuels

The following fuels are approved for use (provided they fulfil the requirements listed in the table):

- Classes ISO F-DMA & DMZ as per ISO 8217 in the current edition
- Class ISO F-DFA & DFZ as per ISO 8217 in the current edition
- Class ISO F-DMB as per ISO 8217 in the current edition (51/60 DF & 58/64 RDF only)
- Class ISO F-DFB as per ISO 8217 in the current edition (51/60 DF & 58/64 RDF only)
- Diesel fuel as per EN 590 in the current edition
- Diesel fuel no. 2-D as per ASTM D975-15
- Synthetic diesel fuel as per EN 15940 in the current edition

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Viscosity

To ensure sufficient lubrication, a minimum viscosity must be ensured at the fuel pump. The maximum temperature required to ensure that a viscosity of more than 1.9 mm^2/s is maintained upstream of the fuel pump, depends on the fuel viscosity. In any case, the fuel temperature upstream of the injection pump must not exceed 45 °C.

The lubricity requirements of the fuel upstream of the engine is a maximum of 520 µm WSD in each case.

Cold suitability

The cold suitability of the fuel is determined by the climatic requirements at the place of installation.

The cold suitability of a fuel may be determined and assessed using the following standards:

- Limit of filterability (CFPP) as per EN 116
- Pour point as per ISO 3016
- Cloud point as per ISO 3015

The system operator must be certain that the cold suitability of the fuel (pour point, cloud point, CFPP) is right for the climatic conditions at the place of installation.

Analyses

Analysis of fuel oil samples is very important for safe engine operation. We can analyse fuel for customers at MAN Energy Solutions laboratory PrimeServLab.

Note:

Handling of operating fluids

Handling of operating fluids can cause serious injury and damage to the environment.
Observe safety data sheets of the operating fluid supplier.

4.6 Marine diesel oil (DMB, DFB) specifications

General information

Marine diesel oil as a heavy distillate is available for marine applications only. Alternative designation: marine diesel fuel oil (MDO). It is made from crude oil and may consist of synthetic components (e.g. BtL, CtL, GtL and HVO). The fuel is treated the same as residual fuel in the supply chain. This means that it is possible for the fuel to be blended with high-viscosity residual fuel residue, e.g. in a bunker vessel, and it might therefore contain residue from crude oil processing. This can affect the properties of the fuel.

Selection of suitable diesel fuel

Unsuitable or adulterated fuel generally results in a shortening of the service life of engine parts/components, damage to these and to catastrophic engine failure. It is therefore important to select the fuel with care in terms of its suitability for the engine and the intended application. Through its combustion, the fuel influences the emissions behaviour of the engine.

Specifications and approvals

The fuel quality varies regionally and is dependent on climatic conditions. All requirements specified in the current edition of ISO 8217 apply.

The following values must be maintained at the engine inlet:

Property	Unit		Threshold value[1]	Standard[2]
Kinematic viscosity at 40 °C[3]	mm^2/s	Max.	11.0	ISO 3104, ASTM D7042, ASTM D445, DIN EN 16896
		Min.	2.000	
Density at 15°C	kg/m^3	Max.	900.0	ISO 3675, ISO 12185

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Property	Unit		Threshold value[1]	Standard[2]
		Min.	820.0	
Cetane index & cetane number		Min.	35	ISO 4264 & ISO 5165
Sulphur content[4]	% (m/m)	Max.	1.50	ISO 8754, ISO 14596, ASTM D 4294, DIN 51400-10
Flash point[5]	°C	Min.	60.0	ISO 2719
Hydrogen sulphide	mg/kg	Max.	2.0	IP 570
Acid number	mg KOH/g	Max.	0.5	ASTM D664
Corrosion on copper	Class	Max.	1	ISO 2160
Oxidation stability[6]	g/m³	Max.	25	ISO 12205, EN 15751
	h	Min.	20	
Fatty acid methyl ester (FAME) content[7]	% (V/V)	Max.	7.0	ASTM D7963, IP 579, EN 14078
Carbon residue	% (m/m)	Max.	0.30	ISO 10370
Appearance[8]	–	–	Free from contamination	visually
Water content	% (m/m)	Max.	0.02	DIN 51777, DIN EN 12937, ASTM D6304
Ash content	% (m/m)	Max.	0.010	ISO 6245
Lubricity[9]	µm	Max.	520	ISO 12156-1, ASTM D6079

Table 179: Requirements for diesel fuel

Remarks:

[1] The fuel must be suitable for the intended application. It must not contain any substance in a concentration that causes additional air pollution, is harmful for personnel, jeopardises ship safety and/or has an adverse effect on machine performance. The fuel must be free from non-ferrous metals according to DIN EN 16476. The fuel must not contain any waste oil.

[2] Always in relation to the currently applicable edition.

[3] Specific requirements of the injection system must be taken into account.

[4] Independent of the maximum permissible sulphur content, local laws and regulations must be adhered to.

[5] SOLAS specification. A lower flash point is possible for non-SOLAS-regulated applications.

[6] If there is more than 2% (V/V) FAME, an analysis as per EN15751 must additionally be performed

[7] The FAME must either be in accordance with EN 14214 or with ASTM D6751.

[8] Only possible with clear samples. If the sample is not clear or contains visible contamination, the check must be completed mandatorily for the entire sediment.

[9] Diameter of the corrected wear scar (WSD).

The following fuels are approved for use:

- Class ISO F-DMB according to ISO 8217 in the current edition.
- Class ISO F-DFB as per ISO 8217 in the current edition with additional requirements regarding oxidation stability.

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Viscosity

In order to ensure sufficient lubrication, a minimum level of viscosity must be ensured at the fuel injection pump. The specified maximum temperature required to maintain a viscosity of more than 1.9 mm^2/s upstream of the fuel injection pump depends on the fuel viscosity. The temperature of the fuel upstream of the fuel injection pump must not exceed 45 °C in any case. The lubricity requirements of the fuel upstream of the engine is a maximum of 520 µm WSD in each case.

Cold suitability

The cold suitability of the fuel is determined by the climatic requirements at the place of installation. It is the responsibility of the operating company to choose a fuel with sufficient cold suitability.

The cold suitability of a fuel may be determined and assessed using the following standards:

- Limit of filterability (CFPP) as per EN 116
- Pour point as per ISO 3016
- Cloud point as per EN 23015

To be able to draw a reliable conclusion, it is recommended to perform all three stated procedures.

Contamination

We recommend installing a separator upstream of the fuel filter. Separation temperature 40–50°C. Most solid particles (sand, corrosion and catalytic converter fragments) and water can thus be removed and the cleaning intervals for the filter elements can be significantly extended.

Bio-fuel admixture

The DFA fuel may contain up to 7.0% of bio-fuel based on fatty acid methyl ester (FAME). The FAME to be added must comply with either EN14214 or ASTM D 6751. Compared to fuels on mineral oil basis only, fuels containing FAME have an increased tendency to oxidise and age and are more vulnerable to microbiological contamination. Furthermore, the fuel may contain an increased quantity of water. This why it is necessary to check the ageing stability at regular intervals when using this type of fuel. In addition, it is important to regularly check the water content of the fuel.

To minimise microbiological contamination, the tanks must be drained on a regular basis. During standstill periods this is required daily, otherwise weekly.

When first using fuels containing bio-diesel, deposits that have accumulated over a longer period of time may become detached. These deposits can block filters or even cause immediate damage.

Using bio-diesel blends in emergency power generators should be avoided. Bio-diesel fuel should be stored in separate reservoirs. Storing fuel containing bio-diesel for more than 6 months is generally not recommended. MAN Energy Solutions is not liable for damage and any possible consequences resulting from the use of fuel containing bio-diesel.

Analyses

Analysis of fuel oil samples is very important for safe engine operation. We can analyse fuel for customers at MAN Energy Solutions laboratory PrimeServLab.

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4.7 Residual fuel specification (HFO)

Prerequisites

Four-stroke diesel engines from MAN Energy Solutions can be powered with any residual fuel recovered from crude oil that fulfils the requirements specified in the table Properties of residual fuel, Page 276, provided that the engine and

the fuel management system are designed accordingly. In order to ensure a favourable ratio between fuel costs, spare parts and also repair and maintenance expenditure, we recommend observing the following points.

Residual fuel (HFO)

Origin/refinery process

The quality of the residual fuel depends to a large extent on the quality of the crude oil and the refining process used. For this reason, residual fuels of the same viscosity can have significantly different properties depending on the bunker spaces. Residual fuel is usually a blend of residual oil and distillates. The blend components generally originate from modern refining processes, such as CatCracker or Visbreaker. These processes can have an adverse affect on the stability of the fuel and on the ignition and combustion properties. These factors also have a considerable effect on the preparation of the residual fuel and the operating results of the engine.

The responsibility for selecting suitable residual fuels lies with the engine operator.

Specifications

Fuels that can be used in an engine must satisfy the specifications to ensure adequate quality. The limit values for residual fuels are specified in the table Specifications for residual fuel, Page 276. The entries in the last column of this table contain important background information and must therefore be observed.

The relevant international specification is ISO 8217 in the respectively applicable version. The fuel may only be used if it fully complies with the standard. All qualities in these specifications up to K700 can be used, provided the fuel management system has been designed for these fuels. To use fuels that do not comply with these specifications (e.g. crude oil), consultation with the technical service from MAN Energy Solutions Augsburg is required. Residual fuels with a maximum density of 1,010 kg/m^3 may only be used if up-to-date separators are installed.

Important

Even if they fulfil the aforementioned specifications, the fuel properties specified in the table Specifications for residual fuel, Page 276 may possibly not be adequate to determine the ignition and combustion properties and also the stability of the fuel. This means that the operating behaviour of the engine can depend on properties that are not defined in the specification. This particularly applies to the oil property that causes formation of deposits in the combustion chamber, injection system, gas ducts and exhaust system. A number of fuels have a tendency towards incompatibility with lubricating oil which leads to deposits being formed in the fuel injection pump that can cause a blockage of the pumps. It may therefore be necessary to exclude specific fuels that could cause problems.

Blends

The addition of engine oils (old lubricating oil, ULO – used lubricating oil) and additives that are not manufactured from mineral oils, (coal-tar oil, for example), and residual products of chemical or other processes such as solvents (polymers or chemical waste) is not permitted. Some of the reasons for this are as follows: abrasive and corrosive effects, unfavourable combustion characteristics, poor compatibility with mineral oils and, last but not least, adverse effects on the environment. The order for the fuel must expressly state what is not permitted as the fuel specifications that generally apply do not include this limitation.

If engine oils (old lubricating oil, ULO – used lubricating oil) are added to fuel, this poses a particular danger as the additives in the lubricating oil act as emulsifiers that cause dirt, water and catfines to be transported as fine sus-

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pension. They therefore prevent the necessary cleaning of the fuel. In our experience (and this has also been the experience of other manufacturers), this can severely damage the engine and turbocharger components.

The addition of chemical waste products (solvents, for example) to the fuel is prohibited for environmental protection reasons according to the resolution of the IMO Marine Environment Protection Committee passed on 1st January 1992.

Leak oil collector

Leak oil collectors that act as receptacles for leak oil, and also return and overflow pipes in the lube oil system, must not be connected to the fuel tank. Leak oil lines should be emptied into sludge tanks.

Property	Unit	Threshold value[1]	Standard[2]
Viscosity (at 50°C)[3]	mm^2/s (cSt)	max. 700	ISO 3104, ASTM D7042; ASTM D445, DIN EN 16896
Viscosity (at 100°C)[3]	mm^2/s (cSt)	max. 55	
Density (at 15°C)	kg/m^3	max. 1010	ISO 3675, ISO 12185, DIN 51757
Flash point[4]	°C	min. 60	ISO 2719
Pour point[5]	°C	max. 30	ISO 3016
Acid number	mg KOH/g	max. 2.5	ASTM D664
Aluminium and silicon	mg/kg	Max. 15[6]	IP 501, IP 470, ISO 10478
Total sediment (aged)	%(m/m)	max. 0.10	ISO 10307-2
Carbon residue (Conradson)	%(m/m)	max. 20	DIN EN ISO 10370
Sulphur	%(m/m)	Max. 5.0[7]	ISO 8754, ISO 14596
Ash	%(m/m)	max. 0.15	ISO 6245
Vanadium	mg/kg	max. 450	IP 501, IP 470, ISO 14597, DIN 51790-4
Water	% (v/v)	Max. 0.20[8]	DIN 51777; ASTM D6304
CCAI		870	ISO 8217
Asphalt content	%(m/m)	Max. 2/3 of the carbon residue (Conradson)	factory standard, DIN 51595
Sodium	mg/kg	Max. Na < 1/3 V, Na < 100	IP 501, IP 470, DIN 51399-1
Waste oil[9]	mg/kg	Max. Ca < 30 and Zn < 15 or Ca < 30 and P < 15	IP 501, IP 470, IP 500, DIN 51399-1
Hydrogen sulphide	mg/kg	max. 2	IP 570

[1] Requirement at engine inlet: additional parameters defined for ISO 8217. The entire document ISO 8217 in its current version is mandatory. The fuel mixture at the engine inlet must be homogeneous. The fuel mixture is homogeneous if the p value according to ASTM D7060 is at least 1.20. Other processes (e.g. ASTM D7112 or ASTM D7157) can also be used to check the homogeneity of the fuel mixture. Furthermore, the fuel must be fit for use and must not contain substances in a concentration that contributes to further contamination of the air and/or may impair the safety of personnel or the performance of the machine.

[2] Always reference to the latest edition.

[3] Specific requirements of the injection system must be taken into account.

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Property	Unit	Threshold value[1)]	Standard[2)]
[4)] SOLAS provision: A lower flash point is possible for non-SOLAS-regulated applications.			
[5)] The pour point must be selected by the operating company in accordance with the design of the fuel system and based on the requirements at the place of installation.			
[6)] The bunker product (before cleaning) may contain max. 60 mg/kg Al and Si.			
[7)] Independent of the maximum permissible sulphur content, local laws and regulations must be adhered to. NOTICE: For bore size <400 mm, lower sulphur content for World Bank II - see description D 10 28 0, Emission limits World Bank II.			
[8)] The bunker product (before cleaning) may contain max. 0.50% water.			
[9)] The fuel must be generally free of waste oil. If the threshold values are exceeded, the waste oil will be contaminated.			

Table 180: Properties for residual fuel

Additional information

The following information will clarify the correlation between the quality of the residual fuel, residual fuel preparation, engine operation and the operating results.

Viscosity/injection viscosity

Residual fuels with higher viscosity can be of lower quality. The maximum permissible viscosity depends on the available pre-heating equipment and the capacity (flow rate) of the separator.

The prescribed injection viscosity of 12–14 mm^2/s (for GenSets, L16/24, L21/31, L23/30H, L27/38, L28/32H: 12–18 cSt) and the corresponding fuel temperature upstream of the engine must be complied with. Only in this way can a suitable atomisation and mixture formation be ensured and therefore low-residue combustion. This also prevents mechanical overload of the injection system at the same time. The prescribed injection viscosity and/or the required fuel oil temperature upstream of the motor can be found in the viscosity temperature diagram.

Residual fuel preparation

Fault-free engine operation depends to a considerable extent on the care with which the residual fuel was prepared. Particular attention should be paid to ensuring that inorganic foreign matter with a strongly abrasive effect (catalyst particles, rust, sand) are effectively separated. It has been shown in practice that wear as a result of abrasion in the engine increases considerably if the aluminium and silicon content is higher than 15 mg/kg.

Viscosity and density have an influence on the cleaning effect. This must be taken into account when designing and installing the cleaning system.

Vanadium/Sodium

If the vanadium/sodium ratio is unfavourable, the melting point of the ash may fall in the operating area of the exhaust valve which can lead to high-temperature corrosion. Most of the water and water-soluble sodium compounds it contains can be removed by pre-treating the residual fuel in the settling tank and in the separators.

The risk of high-temperature corrosion is low if the sodium content is one third of the vanadium content or less. It must also be ensured that sodium does not enter the engine in the form of seawater in the intake air.

If the sodium content is higher than 100 mg/kg, this is likely to result in a higher quantity of salt deposits in the combustion chamber and exhaust-gas system. This will impair the function of the engine (including the suction function of the turbocharger).

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	Under certain conditions, high-temperature corrosion can be prevented using a fuel additive that increases the melting point of residual fuel ash (see also Additives for residual fuel oils, Page 281).
Ash	Fuel ash consists for the greater part of vanadium oxide and nickel sulphate (see above chapter for more information). Residual fuels containing a high proportion of ash in the form of foreign matter, e.g. sand, corrosion compounds and catalyst particles, accelerate the mechanical wear in the engine. Catalyst particles produced as a result of the catalytic cracking process may be present in the residual fuels. In most cases, these catalyst particles are aluminium silicates causing a high degree of wear in the injection system and the engine.
Flash point (ASTM D 93)	National and international transportation and storage regulations governing the use of fuels must be complied with in relation to the flash point. In general, a flash point of above 60 °C is prescribed for diesel engine fuels.
Low-temperature behaviour (ASTM D 97)	The pour point is the temperature at which the fuel can no longer flow (but can be pumped). Since many residual fuels with low viscosity have a pour point above 0°C, the bunker facility must also be pre-heated unless fuel in accordance with RMA or RMB is used. The entire fuel system must be designed in such a way that the residual fuel can be pre-heated to around 10°C above the pour point.
Pump characteristics	If the viscosity of the fuel is higher than 1000 mm^2/s (cSt), or the temperature is not at least 10 °C above the pour point, pump problems will occur. For more information, also refer to paragraph Low-temperature behaviour (ASTM D 97, Page 278.
Combustion properties	If the proportion of asphalt is more than two thirds of the coke residue (Conradson), combustion may be delayed which in turn may increase the formation of combustion residues, leading to such as deposits on and in the injection nozzles, large amounts of smoke, low output, increased fuel consumption and a rapid rise in ignition pressure as well as combustion close to the cylinder wall (thermal overloading of lubricating oil film). If the ratio of asphalt to coke residues reaches the limit 0.66, and if the asphalt content exceeds 8%, the risk of deposits forming in the combustion chamber and injection system is higher. These problems can also occur when using unstable residual fuels, or if incompatible residual fuels are blended. This would lead to an increased separation of asphalt (see section Compatibility, Page 281).
Ignition quality	Nowadays, to achieve the prescribed reference viscosity, cracking-process products are used as the low viscosity ingredients of residual fuels although the ignition characteristics of these may also be poor. The cetane number of these compounds should be > 35. If the proportion of aromatic hydrocarbons is high (more than 35%), this also adversely affects the ignition quality. The ignition delay in residual fuels with poor ignition characteristics is longer; the combustion is also delayed which can lead to thermal overloading of the oil film at the cylinder liner and also high cylinder pressures. The ignition delay and accompanying increase in pressure in the cylinder are also influenced by the end temperature and compression pressure, i.e. by the compression ratio, the charge-air pressure and charge-air temperature. The disadvantages of using fuels with poor ignition characteristics can be limited by pre-heating the charge air in partial load operation and reducing the output for a limited period. However, a more effective solution is a high compression ratio and operational adjustment of the injection system to the ignition characteristics of the fuel used, as is the case with piston engines from MAN Energy Solutions.

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The ignition quality is one of the most important properties of the fuel. This value appears as CCAI in ISO 8217. This method is only applicable to 'straight run' residual oils. The increasing complexity of refinery processes has the effect that the CCAI method does not correctly reflect the ignition behaviour for all residual oils.

A testing instrument has been developed based on the constant volume combustion method (Fuel Combustion Analyser FCA), which is used in some fuel testing laboratories (FCA) in conformity with IP 541.
The instrument measures the ignition delay to determine the ignition quality of a fuel and this measurement is converted into an instrument-specific cetane number (ECN: Estimated Cetane Number). It has been determined that residual fuels with a low ECN number cause operating problems and may even lead to damage to the engine. An ECN > 20 can be considered acceptable.

As the liquid components of the residual fuel have a decisive influence on the ignition quality, and flow properties determine the combustion quality, the system operator is responsible for obtaining a fuel that is suitable for the diesel engine. Also see illustration entitled Nomogram for determining the CCAI – assigning the CCAI ranges to engine types, Page 280.

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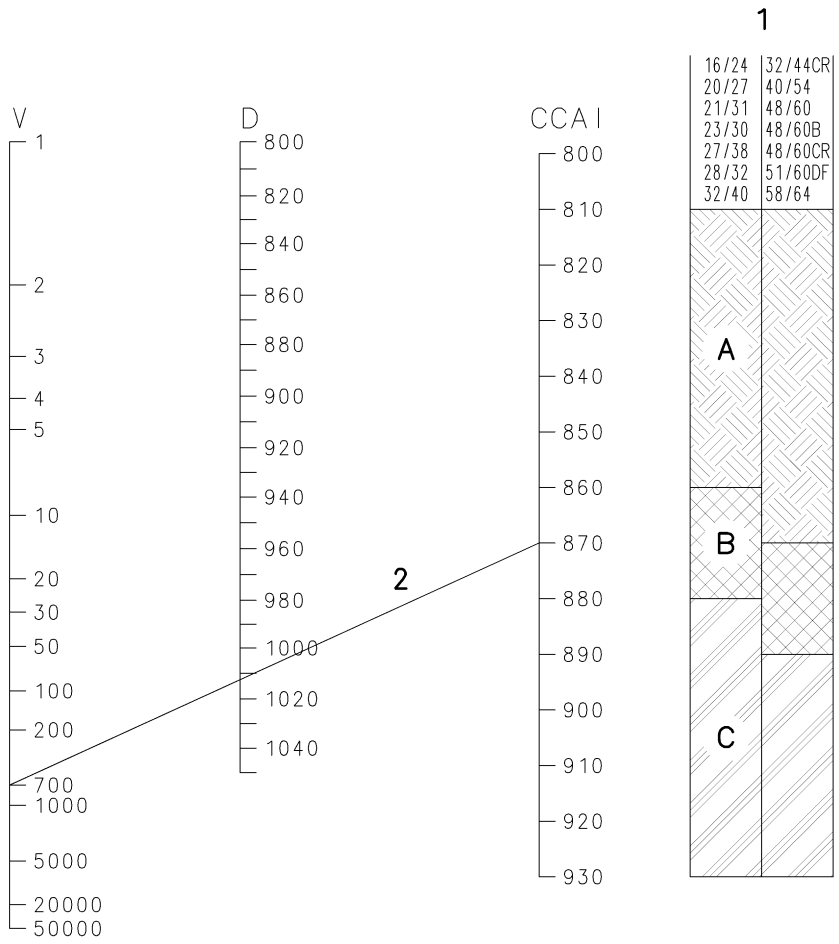

V Viscosity in mm^2/s (cSt) at 50°C

D Density [in kg/m^3] at 15°C

CCAI **C**alculated **C**arbon **A**romaticity **I**ndex

1 Engine type

A Normal operating conditions

B The ignition characteristics can be poor and require an adjustment to the engine or the operating conditions.

C Any problems identified can even lead to engine damage after a short operating period.

2 The CCAI is calculated from the density and viscosity of the residual fuels.

The CCAI can be calculated with the help of the following formula:

$$CCAI = D - 141 \log \log (V+0.85) - 81$$

Figure 106: Nomogram for determining the CCAI and assigning the CCAI ranges to engine types

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Sulphuric acid corrosion

The engine should be operated at the coolant temperatures prescribed in the operating handbook for the relevant load. If the temperature of the components that are exposed to acidic combustion products is below the acid dew point, acid corrosion can no longer be effectively prevented, even if alkaline lube oil is used.

The BN values specified in section Specification of lubricating oil (SAE 40) for residual fuel operation (HFO), Page 258 are sufficient, providing the quality of lubricating oil and the engine's cooling system satisfy the requirements.

Stability

The fuel must be a homogeneous mixture when entering the engine. Precipitation of any fuel components is not permissible! Experience has shown that stability decreases with continued storage and the given conditions. It is hence of great interest to the operator that the fuel has the maximum possible stability reserve so that it can provide a homogeneous fuel mixture at all times when entering the engine (see table Properties for residual fuel, Page 276).

Compatibility

The supplier must guarantee that the residual fuel is homogeneous and remains stable even after the usual storage time. If different bunker oils are mixed, this may lead to separation that is connected with sludge build-up in the fuel system and where large quantities of sludge can be deposited in the separator, clog up the filter, prevent atomisation and lead to residue-rich combustion.

Cases like this can be traced back to incompatibility or instability. The fuel storage tanks should therefore be drained as much as possible before they can be bunkered again, in order to avoid incompatibilities.

Blending residual fuels

If residual fuel for the main engine is blended with distillate fuel (e.g. DMA) or other residual fuels, to obtain the required quality, it is essential that the components are compatible (see section Compatibility, Page 281). The compatibility of the resulting mixture must be tested over the entire mixing range. Reduced long-term stability due to consumption of the stability reserve can be a result. If a mixture of different fuels is planned or unavoidable, the stability reserve of the fuel must be sufficient to ensure that inhomogeneous fuels are not produced when blending.

Additives for residual fuels

MAN Energy Solutions- Engines can also be economically operated without additives. It is up to the customer to decide whether or not the use of additives is beneficial. The supplier of the additive must guarantee that the engine operation will not be impaired by using the product.

As a rule, the use of fuel additives during the warranty period must be avoided.

Residual fuels with low sulphur content

From the perspective of an engine manufacturer, there is no lower threshold for the sulphur content of residual fuels. We have not identified any issues that can be traced to the sulphur content with the low-sulphur residual fuels that are currently commercially available.

If the engine is not constantly operated with low-sulphur residual fuel, the lubricating oil must be selected accordingly for the highest sulphur content of the utilised fuels.

Note:
If operating fluids are improperly handled, this can pose a danger to health, safety and the environment. The relevant safety information by the supplier of operating fluids must be observed.

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Tests

Analysis of samples

To ensure sufficient cleaning of the fuel via the separator, perform regular functional check by sampling up- and downstream of the separator.

Analysis of residual fuel samples is essential for safe engine operation. We can analyse fuel for customers at the MAN Energy Solutions laboratory PrimeServLab.

4.8 Viscosity-temperature diagram (VT diagram)

Explanations of viscosity-temperature diagram

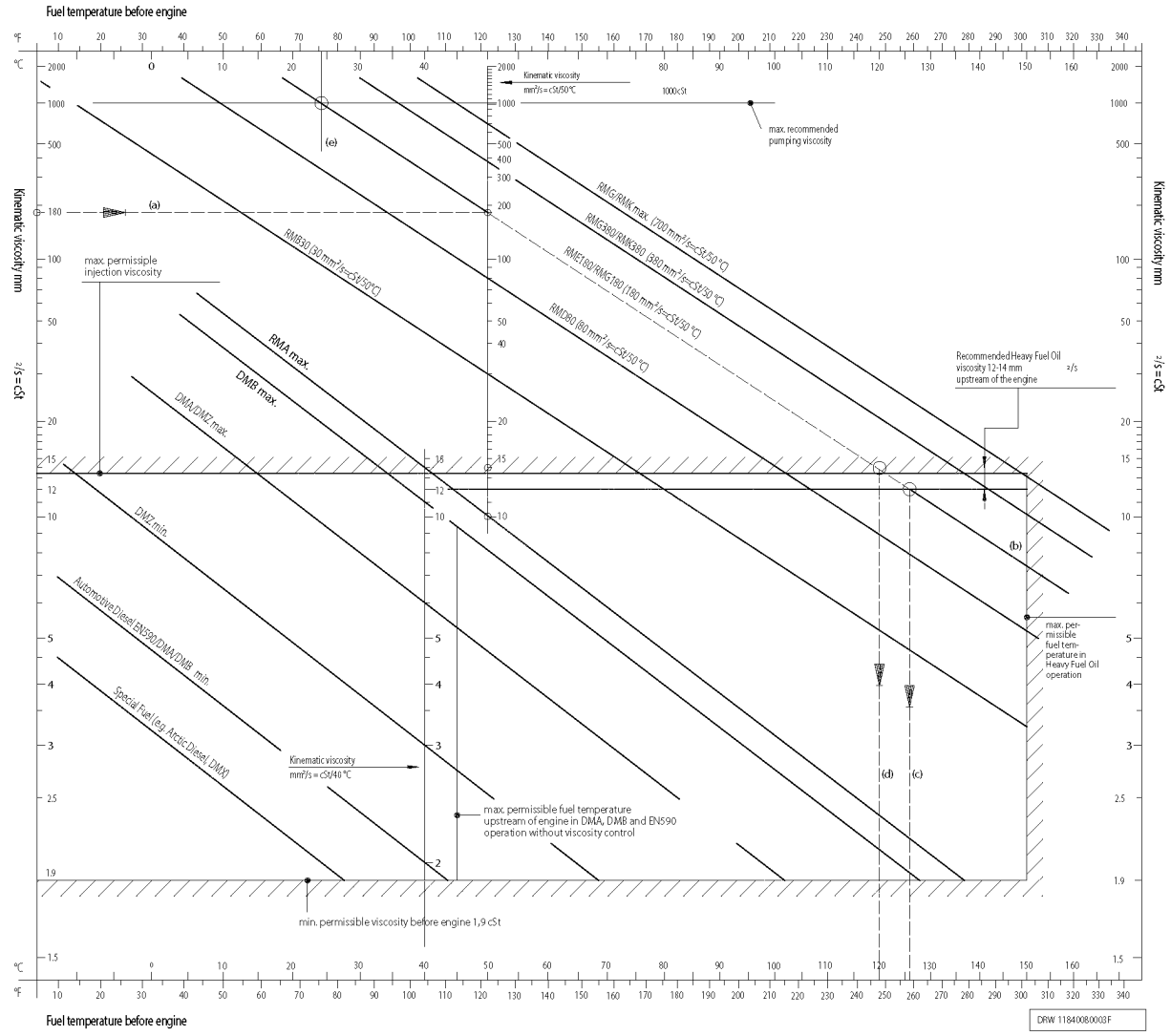

Figure 107: Viscosity-temperature diagram (VT diagram)

In the diagram, the fuel temperatures are shown on the horizontal axis and the viscosity is shown on the vertical axis.

The diagonal lines correspond to viscosity-temperature curves of fuels with different reference viscosities. The vertical viscosity axis in mm^2/s (cSt) applies for 40, 50 or 100 °C.

Determining the viscosity-temperature curve and the required preheating temperature

Example: Heavy fuel oil with 180 mm^2/s at 50 °C

Prescribed injection viscosity in mm^2/s	Required fuel temperature at the engine inlet[1] in °C
≥ 12	126 (line c)
≤ 14	119 (line d)

[1] For these figures, the temperature drop from the last pre-heating device to the fuel injection pump is not taken into account.

Table 181: Determining the viscosity temperature trend and the required pre-heating temperature

A heavy fuel oil with a viscosity of 180 mm^2/s at 50 °C can reach a viscosity of 1,000 mm^2/s at 24 °C (line e) – this is the maximum permissible viscosity of fuel that the pump can deliver.

A heavy fuel oil discharge temperature of 152 °C is reached when using a recent state-of-the-art preheating device with 8 bar saturated steam. At higher temperatures there is a risk of residues forming in the preheating system – this leads to a reduction in heating output and thermal overloading of the heavy fuel oil. Asphalt is also formed in this case, i.e. quality deterioration.

The heavy fuel oil lines between the outlet of the last preheating system and the injection valve must be suitably insulated to limit the maximum drop in temperature to 4 °C. This is the only way to achieve the necessary injection viscosity of 14 mm^2/s for heavy fuel oils with a reference viscosity of 700 mm^2/s at 50 °C (the maximum viscosity as defined in the international specifications such as ISO CIMAC or British Standard). If heavy fuel oil with a low reference viscosity is used, the injection viscosity should ideally be 12 mm^2/s in order to achieve more effective atomisation to reduce the combustion residue.

The delivery pump must be designed for heavy fuel oil with a viscosity of up to 1,000 mm^2/s. The pour point also determines whether the pump is capable of transporting the heavy fuel oil. The bunker facility must be designed so as to allow the heavy fuel oil to be heated to roughly 10 °C above the pour point.

Note:

The viscosity of gas oil or diesel oil (marine diesel oil) upstream of the engine must be at least 1.9 mm^2/s. If the viscosity is too low, this may cause seizing of the pump plunger or nozzle needle valves as a result of insufficient lubrication.

This can be avoided by monitoring the temperature of the fuel. Although the maximum permissible temperature depends on the viscosity of the fuel, it must never exceed the following values:

- 45 °C at the most with MGO (DMA) and MDO (DMB)

A fuel cooler must therefore be installed.

If the viscosity of the fuel is < 2 cSt at 40 °C, consult the technical service of MAN Energy Solutions in Augsburg.

4.9 Specification of engine coolant

Preliminary remarks

An engine coolant is composed as follows: water for heat removal and coolant additive for corrosion protection.

As is also the case with the fuel and lubricating oil, the engine coolant must be carefully selected, handled and checked. If this is not the case, corrosion, erosion and cavitation may occur at the walls of the cooling system in contact with water and deposits may form. Deposits obstruct the transfer of heat and can cause thermal overloading of the cooled parts. The system must be treated with an anticorrosive agent before bringing it into operation for the first time. The concentrations prescribed by the engine manufacturer must always be observed during subsequent operation. The above especially applies if a chemical additive is added.

Requirements

Limit values

The properties of untreated coolant must correspond to the following limit values:

Properties/Characteristic	Properties	Unit
Water type	Distillate or fresh water, free of foreign matter.	–
Total hardness	max. 10	dGH[1)]
pH value	6.5 – 8	–
Chloride ion content	max. 50	mg/l[2)]

Table 182: Properties of coolant that must be complied with

[1)] 1 dGH (German hardness) ≙ 10 mg CaO in 1 litre of water ≙ 17.8 mg $CaCO_3$/l

≙ 0.357 mval/l ≙ 0.178 mmol/l

[2)] 1 mg/l ≙ 1 ppm

Testing equipment

The MAN Energy Solutions water testing equipment incorporates devices that determine the water properties directly related to the above. The manufacturers of anticorrosive agents also supply user-friendly testing equipment.

For information on monitoring cooling water, see section Coolant inspecting, Page 289.

Additional information

Distillate

If distilled water (from a fresh water generator, for example) or fully desalinated water (from ion exchange or reverse osmosis) is available, this should ideally be used as the engine coolant. These waters are free of lime and salts, which means that deposits that could interfere with the transfer of heat to the coolant, and therefore also reduce the cooling effect, cannot form. However, these waters are more corrosive than normal hard water as the thin film of lime scale that would otherwise provide temporary corrosion protection does not form on the walls. This is why distilled water must be handled particularly carefully and the concentration of the additive must be regularly checked.

Hardness

The total hardness of the water is the combined effect of the temporary and permanent hardness. The proportion of calcium and magnesium salts is of overriding importance. The temporary hardness is determined by the carbonate content of the calcium and magnesium salts. The permanent hardness is determined by the amount of remaining calcium and magnesium salts (sulphates). The temporary (carbonate) hardness is the critical factor that determines the extent of limescale deposit in the cooling system.

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Water with a total hardness of > 10°dGH must be mixed with distilled water or softened. Subsequent hardening of extremely soft water is only necessary to prevent foaming if emulsifiable slushing oils are used.

Damage to the cooling water system

Corrosion

Corrosion is an electrochemical process that can widely be avoided by selecting the correct water quality and by carefully handling the water in the engine cooling system.

Flow cavitation

Flow cavitation can occur in areas in which high flow velocities and high turbulence is present. If the steam pressure is reached, steam bubbles form and subsequently collapse in high pressure zones which causes the destruction of materials in constricted areas.

Erosion

Erosion is a mechanical process accompanied by material abrasion and the destruction of protective films by solids that have been drawn in, particularly in areas with high flow velocities or strong turbulence.

Stress corrosion cracking

Stress corrosion cracking is a failure mechanism that occurs as a result of simultaneous dynamic and corrosive stress. This may lead to cracking and rapid crack propagation in water-cooled, mechanically-loaded components if the coolant has not been treated correctly.

Treatment of engine coolant

Formation of a protective film

The purpose of treating the engine coolant using anticorrosive agents is to produce a continuous protective film on the walls of cooling surfaces and therefore prevent the damage referred to above. In order for an anticorrosive agent to be 100 % effective, it is extremely important that untreated water satisfies the requirements in the paragraph Requirements, Page 284.

Protective films can be formed by treating the coolant with anticorrosive chemicals or emulsifiable slushing oil.

Emulsifiable slushing oils are used less and less frequently as their use has been considerably restricted by environmental protection regulations, and because they are rarely available from suppliers for this and other reasons.

Treatment prior to initial commissioning of engine

Treatment with an anticorrosive agent should be carried out before the engine is brought into operation for the first time to prevent irreparable initial damage.

Note:

The engine must not be brought into operation without treating the cooling water first.

Additives for coolants

Only the additives approved by MAN Energy Solutions and listed in the tables under the paragraph entitled Permissible cooling water additives may be used.

Required release

A coolant additive may only be permitted for use if tested and approved as per the latest directives of the ICE Research Association (FVV) “Suitability test of internal combustion engine cooling fluid additives.” The test report must be obtainable on request. The relevant tests can be carried out on request in Germany at the staatliche Materialprüfanstalt (Federal Institute for Materials Research and Testing), Abteilung Oberflächentechnik (Surface Technology Division), Grafenstraße 2 in D-64283 Darmstadt.

Once the coolant additive has been tested by the FVV, the engine must be tested in a second step before the final approval is granted.

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In closed circuits only

Additives may only be used in closed circuits where no significant consumption occurs, apart from leaks or evaporation losses. Observe the applicable environmental protection regulations when disposing of coolant containing additives. For more information, consult the additive supplier.

Chemical additives

Sodium nitrite and sodium borate based additives etc. have a proven track record. Galvanised iron pipes or zinc sacrificial anodes must not be used in cooling systems. This corrosion protection is not required due to the prescribed coolant treatment and electrochemical potential reversal that may occur due to the coolant temperatures which are usual in engines nowadays. If necessary, the pipes must be deplated.

Slushing oil

For MAN Energy Solutions engines, it is not permissible to use corrosion protection oils in the cooling water circuit.

Antifreeze agents

If temperatures below the freezing point of water in the engine cannot be excluded, an antifreeze agent that also prevents corrosion must be added to the cooling system or corresponding parts. Otherwise, the entire system must be heated.

Sufficient corrosion protection can be provided by adding the products listed in the table entitled Antifreeze agent with slushing properties (Military specification: Federal Armed Forces Sy-7025), while observing the prescribed minimum concentration. This concentration prevents freezing at temperatures down to –22 °C and provides sufficient corrosion protection. However, the quantity of antifreeze agent actually required always depends on the lowest temperatures that are to be expected at the place of use.

Antifreeze agents are generally based on ethylene glycol. A suitable chemical anticorrosive agent must be added if the concentration of the antifreeze agent prescribed by the user for a specific application does not provide an appropriate level of corrosion protection, or if the concentration of antifreeze agent used is lower due to less stringent frost protection requirements and does not provide an appropriate level of corrosion protection. Considering that the antifreeze agents listed in the table Antifreeze agents with slushing properties also contain corrosion inhibitors and their compatibility with other anticorrosive agents is generally not given, only pure glycol may be used as antifreeze agent in such cases.

Simultaneous use of anticorrosive agent from the table Nitrite-free chemical additives together with glycol is not permitted, because monitoring the anticorrosive agent concentration in this mixture is no more possible.

Antifreeze agents reduce the capacity of the coolant to absorb heat. In some cases the cooling effect of the coolant may not be sufficient for certain operation conditions. The MAN Energy Solutions standard design is not based on using antifreeze agents. In case it is intended to use anti-freeze agent, consult MAN Energy Solutions beforehand.

Before an antifreeze agent is used, the cooling system must be thoroughly cleaned.

If the coolant contains emulsifiable slushing oil, antifreeze agent may not be added as otherwise the emulsion would break up and oil sludge would form in the cooling system.

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Biocides

If you cannot avoid using a biocide because the coolant has been contaminated by bacteria, observe the following steps:

- You must ensure that the biocide to be used is suitable for the specific application.
- The biocide must be compatible with the sealing materials used in the coolant system and must not react with these.
- The biocide and its decomposition products must not contain corrosion-promoting components. Biocides whose decomposition products contain chloride or sulphate ions are not permitted.
- Biocides that cause foaming of coolant are not permitted.

Prerequisite for effective use of an anticorrosive agent

Clean cooling system

As contamination significantly reduces the effectiveness of the additive, the tanks, pipes, coolers and other parts outside the engine must be free of rust and other deposits before the engine is started up for the first time and after repairs of the pipe system.

The entire system must therefore be cleaned with the engine switched off using a suitable cleaning agent (see section Cooling water system cleaning, Page 290).

Loose solid matter in particular must be removed by flushing the system thoroughly as otherwise erosion may occur in locations where the flow velocity is high.

The cleaning agents must not corrode the seals and materials of the cooling system. In most cases, the supplier of the coolant additive will be able to carry out this work and, if this is not possible, will at least be able to provide suitable products to do this. If this work is carried out by the engine operator, he should use the services of a specialist supplier of cleaning agents. The cooling system must be flushed thoroughly after cleaning. Once this has been done, the engine coolant must be immediately treated with anticorrosive agent. Once the engine has been brought back into operation, the cleaned system must be checked for leaks.

Regular checks of the coolant condition and coolant system

Treated coolant may become contaminated when the engine is in operation, which causes the additive to loose some of its effectiveness. It is therefore advisable to regularly check the cooling system and the coolant condition. To determine leakages in the lube oil system, it is advisable to carry out regular checks of water in the expansion tank. Indications of oil content in water are, e.g. discoloration or a visible oil film on the surface of the water sample.

The additive concentration must be checked at least once a week using the test kits specified by the manufacturer. The results must be documented.

Note:

The chemical additive concentrations shall not be less than the minimum concentrations indicated in the table Nitrite-containing chemical additives.

Excessively low concentrations lead to corrosion and must be avoided. Concentrations that are somewhat higher do not cause damage. Concentrations that are more than twice as high as recommended should be avoided.

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Every 2 to 6 months, a coolant sample must be sent to an independent laboratory or to the engine manufacturer for an integrated analysis.

If chemical additives or antifreeze agents are used, coolant should be replaced after 3 years at the latest.

If there is a high concentration of solids (rust) in the system, the water must be completely replaced and entire system carefully cleaned.

Deposits in the cooling system may be caused by fluids that enter the coolant or by emulsion break-up, corrosion in the system, and limescale deposits if the water is very hard. If the concentration of chloride ions has increased, this generally indicates that seawater has entered the system. The maximum specified concentration of 50 mg chloride ions per kg must not be exceeded as otherwise the risk of corrosion is too high. If exhaust gas enters the coolant, this can lead to a sudden drop in the pH value or to an increase in the sulphate content.

Water losses must be compensated for by filling with untreated water that meets the quality requirements specified in the paragraph Requirements, Page 284. The concentration of anticorrosive agent must subsequently be checked and adjusted if necessary.

Subsequent checks of the coolant are especially required if the coolant had to be drained off in order to carry out repairs or maintenance.

Protective measures

Anticorrosive agents contain chemical compounds that can pose a risk to health or the environment if incorrectly used. Comply with the directions in the manufacturer's material safety data sheets.

Avoid prolonged direct contact with the skin. Wash hands thoroughly after use. If larger quantities spray and/or soak into clothing, remove and wash clothing before wearing it again.

If chemicals come into contact with your eyes, rinse them immediately with plenty of water and seek medical advice.

Anticorrosive agents are generally harmful to the water cycle. Observe the relevant statutory requirements for disposal.

Auxiliary engines

If the same cooling water system used in a MAN Energy Solutions two-stroke main engine is used in a marine engine of type 16/24, 21/ 31, 23/30H, 27/38 or 28/32H, the cooling water recommendations for the main engine must be observed.

Analyses

MAN Energy Solutions can analyse antifreeze agent for their customers in the chemical laboratory PrimeServLab. A 0.25 l sample is required for the test.

Permitted coolant additives

A list of currently approved coolant additives and their concentration can be found at https://corporate.man-es.com/lubrication.

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4.10 Coolant inspecting

Summary

Acquire and check typical values of the operating media to prevent or limit damage.

The freshwater used to fill the cooling water circuits must satisfy the specifications. The cooling water in the system must be checked regularly in accordance with the maintenance schedule.

The following work/steps is/are necessary:

Acquisition of typical values for the operating fluid, evaluation of the operating fluid and checking the concentration of the anticorrosive agent.

Tools/equipment required

Equipment for checking the fresh water quality

The following equipment can be used:

- The MAN Energy Solutions water testing kit, or similar testing kit, with all necessary instruments and chemicals that determine the water hardness, pH value and chloride content (obtainable from MAN Energy Solutions or Mar-Tec Marine, Hamburg).

Equipment for testing the concentration of additives

When using chemical additives:

- Testing equipment in accordance with the supplier's recommendations. Testing kits from the supplier also include equipment that can be used to determine the fresh water quality.

Testing the typical values of water

Short specification

Typical value/property	Water for filling and refilling (without additive)	Circulating water (with additive)
Water type	Fresh water, free of foreign matter	Treated coolant
Total hardness	≤ 10 dGH[1)]	≤ 10 dGH[1)]
pH value	6.5 – 8 at 20 °C	≥ 7.5 at 20 °C
Chloride ion content	≤ 50 mg/l	≤ 50 mg/l[2)]

Table 183: Quality specifications for coolants (short version)

[1)] dGH German hardness

1 dGH = 10 mg/l CaO
= 17.8 mg/l $CaCO_3$
= 0.178 mmol/L

[2)] 1 mg/l = 1 ppm

Testing the concentration of rust inhibitors

Short specification

Anticorrosive agent	Concentration
Chemical additives	According to the quality specification, see section Specification of engine coolant, Page 283.
Anti-freeze agents	

Table 184: Concentration of coolant additives

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Testing the concentration of chemical additives	The concentration should be tested every week, and/or according to the maintenance schedule, using the testing instruments, reagents and instructions of the relevant supplier. Chemical anti-corrosion agents can only provide effective protection if the concentration is precisely maintained. Respectively, the concentrations recommended by MAN Energy Solutions (quality specifications in section Specification of engine coolant, Page 283) must be maintained under all circumstances. These recommended concentrations may deviate from those specified by the manufacturer.
Testing the concentration of anti-freeze agents	The concentration must be checked in accordance with the manufacturer's instructions or the test can be outsourced to a suitable laboratory. If in doubt, consult MAN Energy Solutions.
Regular water samplings	Small quantities of lube oil in coolant can be found by visual check during regular water sampling from the expansion tank.
Testing	Regular analysis of coolant is very important for safe engine operation. We can analyse fuel for customers at MAN Energy Solutions laboratory PrimeServLab.

4.11 Cooling water system cleaning

Summary

Remove contamination/residue from operating fluid systems, ensure/re-establish operating reliability.

Cooling water systems containing deposits or contamination prevent effective cooling of parts. Contamination and deposits must be regularly eliminated.

This comprises the following:

Cleaning the system and, if required removal of limescale deposits, flushing the system.

Cleaning

The coolant system must be checked for contamination at regular intervals. Cleaning is required if the degree of contamination is high. This work should ideally be carried out by a specialist who can provide the right cleaning agents for the type of deposits and materials in the cooling circuit. The cleaning should only be carried out by the engine operator if this cannot be done by a specialist.

Oil sludge

Oil sludge from lubricating oil that has entered the cooling system or a high concentration of anticorrosive agents can be removed by flushing the system with fresh water to which some cleaning agent has been added. Suitable cleaning agents are listed alphabetically in the table entitled Cleaning agents for removing oil sludge., Page 290 Products by other manufacturers can be used providing they have similar properties. The manufacturer's instructions for use must be strictly observed.

Manufacturer	Product	Concentration	Duration of cleaning procedure/temperature
Drew	HDE - 777	4 – 5%	4 h at 50 – 60 °C
Nalfleet	MaxiClean 2	2 – 5%	4 h at 60 °C
Unitor	Aquabreak	0.05 – 0.5%	4 h at ambient temperature

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Manufacturer	Product	Concentration	Duration of cleaning procedure/temperature
Vecom	Ultrasonic Multi Cleaner	4%	12 h at 50 – 60 °C

Table 185: Cleaning agents for removing oil sludge

Lime and rust deposits

Lime and rust deposits can form if the water is especially hard or if the concentration of the anticorrosive agent is too low. A thin lime scale layer can be left on the surface as experience has shown that this protects against corrosion. However, limescale deposits with a thickness of more than 0.5 mm obstruct the transfer of heat and cause thermal overloading of the components being cooled.

Rust that has been flushed out may have an abrasive effect on other parts of the system, such as the sealing elements of the water pumps. Together with the elements that are responsible for water hardness, this forms what is known as ferrous sludge which tends to gather in areas where the flow velocity is low.

Products that remove limescale deposits are generally suitable for removing rust. Suitable cleaning agents are listed alphabetically in the table entitled Cleaning agents for removing limescale and rust deposits., Page 291 Products by other manufacturers can be used providing they have similar properties. The manufacturer's instructions for use must be strictly observed. Prior to cleaning, check whether the cleaning agent is suitable for the materials to be cleaned. The products listed in the table entitled Cleaning agents for removing limescale and rust deposits, Page 291 are also suitable for stainless steel.

Manufacturer	Product	Concentration	Duration of cleaning procedure/temperature
Drew	SAF-Acid Descale-IT Ferroclean	5 – 10 % 5 – 10 % 10 %	4 h at 60 – 70 °C 4 h at 60 – 70 °C 4 – 24 h at 60 – 70 °C
Nalfleet	Nalfleet 9 - 068	5 %	4 h at 60 – 75 °C
Unitor	Descalex	5 – 10 %	4 – 6 h at approx. 60 °C
Vecom	Descalant F	3 – 10 %	ca. 4 h at 50 – 60 °C

Table 186: Cleaning agents for removing lime scale and rust deposits

In emergencies only

Hydrochloric acid diluted in water or aminosulphonic acid may only be used in exceptional cases if a special cleaning agent that removes limescale deposits without causing problems is not available. Observe the following during application:

- Stainless steel heat exchangers must never be treated using diluted hydrochloric acid.
- Cooling systems containing non-ferrous metals (aluminium, red bronze, brass, etc.) must be treated with deactivated aminosulphonic acid. This acid should be added to water in a concentration of 3 – 5 %. The temperature of the solution should be 40 – 50 °C.
- Diluted hydrochloric acid may only be used to clean steel pipes. If hydrochloric acid is used as the cleaning agent, there is always a danger that acid will remain in the system, even when the system has been neutralised and flushed. This residual acid promotes pitting. We therefore recommend you have the cleaning carried out by a specialist.

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The carbon dioxide bubbles that form when limescale deposits are dissolved can prevent the cleaning agent from reaching boiler scale. It is therefore absolutely necessary to circulate the water with the cleaning agent to flush away the gas bubbles and allow them to escape. The length of the cleaning process depends on the thickness and composition of the deposits. Values are provided for orientation in the table entitled Cleaning agents for removing limescale and rust deposits, Page 291.

Following cleaning

The cooling system must be flushed several times once it has been cleaned using cleaning agents. Replace the water during this process. If acids are used to carry out the cleaning, neutralise the cooling system afterwards with suitable chemicals then flush. The system can then be refilled with water that has been prepared accordingly.

Note:

Start the cleaning operation only when the engine has cooled down. Hot engine components must not come into contact with cold water. Open the venting pipes before refilling the cooling water system. Blocked venting pipes prevent air from escaping which can lead to thermal overloading of the engine.

Note:

The products to be used can endanger health and may be harmful to the environment. Follow the manufacturer's handling instructions without fail.

The applicable regulations governing the disposal of cleaning agents or acids must be observed.

4.12 Specification of intake air (combustion air)

General

The quality and condition of intake air (combustion air) have a significant effect on the engine output, wear and emissions of the engine. In this regard, not only are the atmospheric conditions extremely important, but also contamination by solid and gaseous foreign matter.

Mineral dust in the intake air increases wear. Chemicals and gases promote corrosion.

This is why effective cleaning of intake air (combustion air) and regular maintenance of the air filter are required.

When designing the intake air system, the maximum permissible overall pressure drop (filter, silencer, pipe line) of 20 mbar must be taken into consideration.

Exhaust turbochargers for marine engines are equipped with silencers and air filters as a standard.

Requirements

Liquid fuel engines: As minimum, inlet air (combustion air) must be cleaned by an ISO Coarse 45% class filter as per DIN EN ISO 16890, if the combustion air is drawn in from inside (e.g. from the machine room/engine room). If the combustion air is drawn in from outside, in the environment with a risk of higher inlet air contamination (e.g. due to sand storms, due to loading and unloading grain cargo vessels or in the surroundings of cement plants), additional measures must be taken. This includes the use of pre-separators, pulse filter systems and a higher grade of filter efficiency class at least up to ISO ePM10 50% according to DIN EN ISO 16890.

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Gas engines and dual-fuel engines: As minimum, inlet air (combustion air) must be cleaned by an ISO COARSE 45% class filter as per DIN EN ISO 16890, if the combustion air is drawn in from inside (e.g. from machine room/ engine room). Gas engines or dual-fuel engines must be equipped with a dry filter. Oil bath filters are not permitted because they enrich the inlet air with oil mist. This is not permissible for gas operated engines because this may result in engine knocking. If the combustion air is drawn in from outside, in the environment with a risk of higher inlet air contamination (e.g. due to sand storms, due to loading and unloading grain cargo vessels or in the surroundings of cement plants) additional measures must be taken. This includes the use of pre-separators, pulse filter systems and a higher grade of filter efficiency class at least up to ISO ePM10 50% according to DIN EN ISO 16890.

In general, the following applies:

The inlet air path from air filter to engine shall be designed and implemented airtight so that no false air may be drawn in from the outdoor.

The concentration downstream of the air filter and/or upstream of the turbocharger inlet must not exceed the following limit values.

The air must not contain organic or inorganic silicon compounds.

Properties	Limit	Unit [1]
Dust (sand, cement, CaO, Al_2O_3 etc.)	max. 5	mg/Nm^3
Chlorine	max. 1.5	
Sulphur dioxide (SO_2)	max. 1.25	
Hydrogen sulphide (H_2S)	max. 5	
Salt (NaCl)	max. 1	
[1] One Nm^3 corresponds to one cubic meter of gas at 0 °C and 101.32 kPa.		

Table 187: Typical values for intake air (combustion air) that must be complied with

Note:

Intake air shall not contain any flammable gases. Make sure that the combustion air is not explosive and is not drawn in from the ATEX Zone.

4.13 Specification of compressed air

General

For compressed air quality observe the ISO 8573-1. Compressed air must be free of solid particles and oil (acc. to the specification).

Requirements

Compressed air quality of starting air system

The starting air must fulfil at least the following quality requirements according to ISO 8573-1.

Purity regarding solid particles	Quality class 6
Particle size > 40µm	max. concentration < 5 mg/m^3
Purity regarding moisture	Quality class 7

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Residual water content	< 0.5 g/m³
Purity regarding oil	Quality class X

Additional requirements are:

- The air must not contain organic or inorganic silicon compounds.
- The layout of the starting air system must ensure that no corrosion may occur.
- The starting air system and the starting air receiver must be equipped with condensate drain devices.
- By means of devices provided in the starting air system and via maintenance of the system components, it must be ensured that any hazardous formation of an explosive compressed air/lube oil mixture is prevented in a safe manner.

Compressed air quality in the control air system

Please note that control air will be used for the activation of some safety functions on the engine – therefore, the compressed air quality in this system is very important.

Control air must meet at least the following quality requirements according to ISO 8573-1.

- Purity regarding solid particles: Quality class 5
- Purity regarding moisture: Quality class 4
- Purity regarding oil: Quality class 3

For catalysts

The following specifications are valid unless otherwise defined by any other relevant sources:

Compressed air quality for soot blowing

Compressed air for soot blowing must meet at least the following quality requirements according to ISO 8573-1.

- Purity regarding solid particles: Quality class 3
- Purity regarding moisture: Quality class 4
- Purity regarding oil: Quality class 2

Compressed air quality for reducing agent atomisation

Compressed air for atomisation of the reducing agent must fulfil at least the following quality requirements according to ISO 8573-1.

- Purity regarding solid particles: Quality class 3
- Purity regarding moisture: Quality class 4
- Purity regarding oil: Quality class 2

Note:

To prevent clogging of catalyst and catalyst lifetime shortening, the compressed air specification must always be observed.

For gas valve unit control (GVU)

Compressed control air quality for the gas valve unit control (GVU)

Compressed air for the gas valve unit control (GVU) must meet at least the following quality requirements according to ISO 8573-1.

- Purity regarding solid particles: Quality class 2
- Purity regarding moisture: Quality class 3

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▪ Purity regarding oil	Quality class 2

4.14 Specification of inert gas

General

To prevent formation of a hazardous explosive gas mixture, inert gas is used for purging gas pipelines. For inert gas quality, ISO 8573-1:2010 must be observed.

Nitrogen only is permitted as inert gas.

As this gas will finally become part of inlet air, the quality requirements are similarly high.

Requirements

Inert gas quality

Inert gas must fulfil at least the following quality requirements according to ISO 8573-1:2010.

Purity regarding solid particles	Quality class 4
Purity regarding moisture	Quality class 4
Purity regarding oil	Quality class 4

Additional requirement

Only inert gas with a purity of minimum 95% N_2 may be used.

Note:

Functional safety of flushing process

It is imperative that this specification is observed, as the proper function of the flushing process and safety function scope depend on it. Additionally, the service life of the gas valves depends on the purity with respect to solid particles.

4.15 Specification of urea solution

Use of good quality urea solution is essential for the operation of a SCR catalyst.

Note:
The overall SCR system is designed for an aqueous solution having a urea content of 40 % as listed in the table below. This must be taken into account when ordering.

Urea 40% must meet the standard of ISO 18611. MAN Energy Solutions recommends urea according to the specification below.

	Urea solution concentration 39 – 41 [%]	Test method
Density at 20 °C [g/cm³]	1.105 – 1.115	DIN EN ISO 12185
Refractive index at 20 °C	1.3930 – 1.3962	ISO 18611-2 Annex C
Biuret [%]	max. 0.5	ISO 18611-2 Annex E
Alkality as NH_3 [%]	max. 0.5	ISO 18611-2 Annex D
Aldehyde [mg/kg]	max. 10	ISO 18611-2 Annex F
Insolubles [mg/kg]	max. 20	ISO 18611-2 Annex G

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	Urea solution concentration 39 – 41 [%]	Test method
Phosphorus (as PO_4) [mg/kg]	max. 0.5	ISO 18611-2 Annex H
Calcium [mg/kg]	max. 0.5	ISO 18611-2 Annex I
Iron [mg/kg]	max. 0.5	ISO 18611-2 Annex I
Magnesium [mg/kg]	max. 0.5	ISO 18611-2 Annex I
Sodium [mg/kg]	max. 0.5	ISO 18611-2 Annex I
Potassium [mg/kg]	max. 0.5	ISO 18611-2 Annex I
Aluminium [mg/kg]	max. 0.5	ISO 22241-2 Annex I
Nickel [mg/kg]	max. 0.2	ISO 22241-2 Annex I
Copper [mg/kg]	max. 0.2	ISO 22241-2 Annex I
Zinc [mg/kg]	max. 0.2	ISO 22241-2 Annex I
Chromium [mg/kg]	max. 0.2	ISO 22241-2 Annex I

Table 188: Urea 40 % solution specification

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5 Engine room and application planning

5.1 3D Engine Viewer – A support programme to configure the engine room

MAN Energy Solutions offers a free-of-charge online programme for the configuration and provision of installation data required for installation examinations and engine room planning: The 3D Engine Viewer and the GenSet Viewer.

Easy-to-handle selection and navigation masks permit configuration of the required engine type, as necessary for virtual installation in your engine room.

In order to be able to use the 3D Engine, respectively GenSet Viewer, register on our website under:

https://extranet.mandieselturbo.com/Pages/Dashboard.aspx

After successful registration, the 3D Engine and GenSet Viewer is available under:

https://extranet.mandieselturbo.com/content/appengineviewer/Pages/Default.aspx

by clicking onto the requested application.

In only three steps, you will obtain professional engine room data for your further planning:

- Selection

  Select the requested output, respectively the requested type.
- Configuration

  Drop-down menus permit individual design of your engine according to your requirements. Each of your configurations will be presented on the basis of isometric models.
- View

  The models of the 3D Engine Viewer and the GenSet Viewer include all essential geometric and planning-relevant attributes (e.g. connection points, interfering edges, exhaust gas outlets, etc.) required for the integration of the model into your project.

The configuration with the selected engines can now be easily downloaded.

For 2D representation as:

- .pdf
- .dwg
- .dxf

for 3D as:

- .dgn
- .stp
- .sat
- .igs
- 3D-dxf
- and many others

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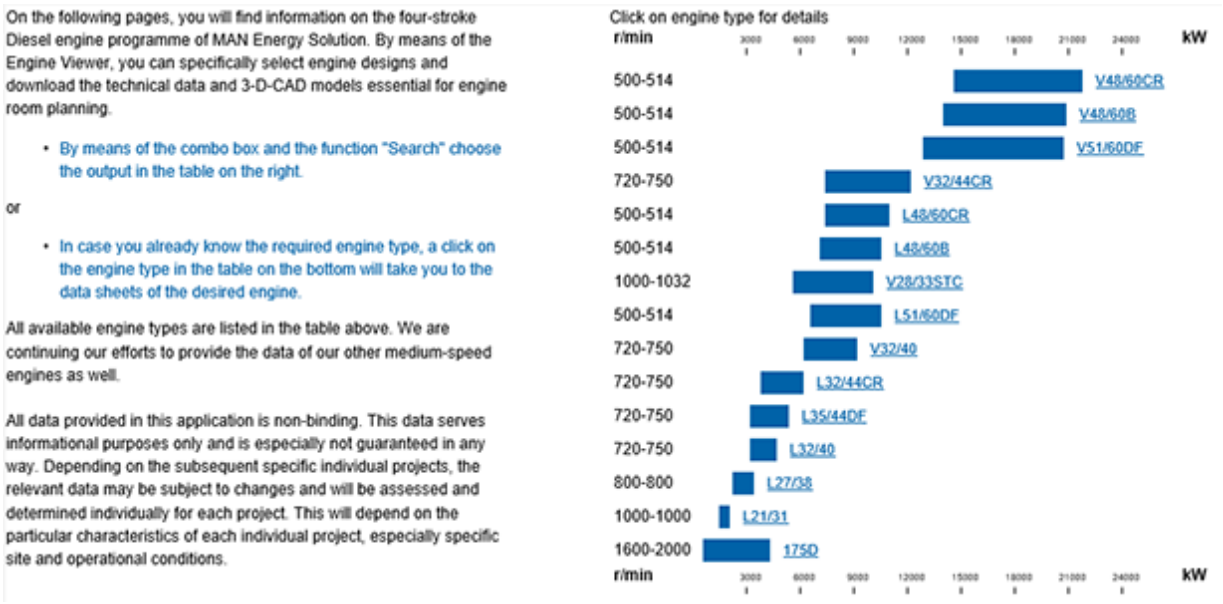

Figure 108: Selection of engine

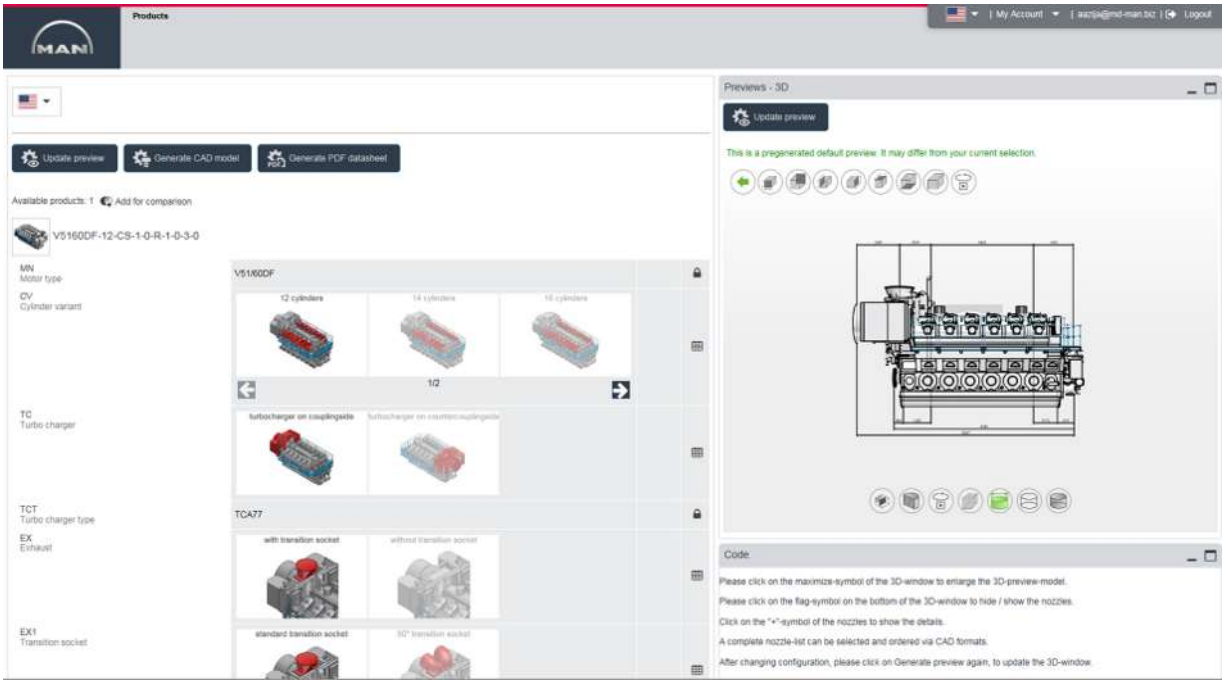

Figure 109: Preselected standard configuration

5.2 Basic principles for pipe selection

5.2.1 External pipe dimensioning

The external piping systems are to be dimensioned, designed, installed and connected to the engine by the shipyard. The pipe systems should to be designed in such a way that the pressure losses are kept within reasonable limits. To achieve this at justifiable cost, it is recommended to maintain the flow

rates as indicated below. Nevertheless, depending on specific conditions of piping systems, it may be necessary in some cases to adopt even lower flow rates. Generally, it is not recommended to use higher flow rates.

For small pipes a low flow speed has to calculated, for bigger pipes (> DN 125) the higher values may be chosen.

	Recommended flow rates (m/s)	
	Suction side	Delivery side
Fresh water (cooling water)	1.0 – 2.0	1.5 – 3.0
Lube oil	0.5 – 1.0	1.5 – 2.5
Seawater	1.0 – 1.5	1.5 – 2.5
Diesel fuel	0.5 – 1.0	1.5 – 2.0
Heavy fuel oil	0.3 – 0.8	1.0 – 1.8
Natural gas (< 5 bar)	-	5 – 10
Natural gas (> 5 bar)	-	10 – 20
Compressed air for control air system	-	2 – 10
Compressed air for starting air system	-	25 – 30 [1]
Intake air	20 – 25	
Exhaust gas	40	
[1] During engine start higher velocities acceptable, depends on total pressure loss of supply system.		

Table 189: Recommended flow rates

5.2.2 Specification of materials for piping

General

- The properties of the piping shall conform to international standards, e.g. DIN EN 10208, DIN EN 10216, DIN EN 10217 or DIN EN 10305, DIN EN 13480-3.
- For piping, black steel pipe should be used; stainless steel shall be used where necessary.
- Outer surface of black steel pipes needs to be primed and painted according to shipyard`s specification.
- The pipes are to be sound, clean and free from all imperfections. The internal surfaces must be thoroughly cleaned and all scale, grit, dirt and sand used in casting or bending has to be removed. No sand is to be used as packing during bending operations.
- In case of pipes with forged bends, care must be taken to ensure that inner surfaces are smooth and that no stray weld metal remains after joining.
- Advices in MAN Energy Solutions work instruction 010.000.001-03. Pipelines cleaning, pickling and preservation. Carry out the pressure test for cleaning of steel pipes before fitting them together should be observed.
- Certain material combinations are sensitive to electro-chemical corrosion, therefore special attention must be paid to the arrangement within a pipe system including all connected components.

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- All information given is to be regarded as indication only; the sole responsibility for the functionality and durability of the external piping system lies with the shipyard.

LT-, HT- and nozzle cooling water pipes

Galvanised steel pipe must not be used for the piping of the system as all additives contained in the engine cooling water attack zinc. Moreover, there is the risk of the formation of local electrolytic element couples where the zinc layer has been worn off, and the risk of aeration corrosion where the zinc layer is not properly bonded to the substrate.

Proposed material (EN)

P235GH, E235, X5CrNiMoTi17-12-2

Fuel oil pipes, lube oil pipes

Galvanised steel pipe must not be used for the piping of the system as acid components of the fuel may attack zinc.

Proposed material (EN)

E235, P235GH, X6CrNiMoTi17-12-2

Urea pipes (for SCR only)

Galvanised steel pipe, brass and copper components must not be used for the piping of the system.

Proposed material (EN)

X6CrNiMoTi17-12-2

Compressed air pipes

Galvanised steel pipe must not be used for the piping of the system.

Proposed material (EN)

E235, P235GH, X6CrNiMoTi17-12-2

Natural gas pipes

Galvanised steel pipe must not be used for the piping of the system.

Proposed material (EN)

E235, P235GH, X6CrNiMoTi17-12-2

Note:
The material for manufacturing the supply gas piping from the GVU to the engine inlet must be stainless steel. Recommended material is X6CrNiMoTi17-12-2.

Seawater pipes

Material depending on required flow speed and mechanical stress.

Proposed material

CuNiFe, glass fiber reinforced plastic, rubber lined steel

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5.2.3 Installation of flexible pipe connections

Arrangement of hoses on engine

Flexible pipe connections are necessary to connect resiliently mounted engines with external piping systems. They are used to compensate the dynamic movements of the engine in relation to the external piping system.

The engine´s movement on its foundation is caused by the engine´s rotation and torque itself as well as by rolling and pitching of the ship. Based on roll angles of +/- 22.5° and pitching of +/- 7.5° (according to prescriptive rules of Classification Societies) the excursions at the exhaust gas outlet can be up to 5 mm in X-, 25 mm in Y- and 6 mm in Z-direction. As the exhaust gas outlet is at the highest point of the engine the excursions at lower positions are smaller respectively. In order to obtain exact data on excursions at certain points, a project-specific calculation of the elastic engine mount is required.

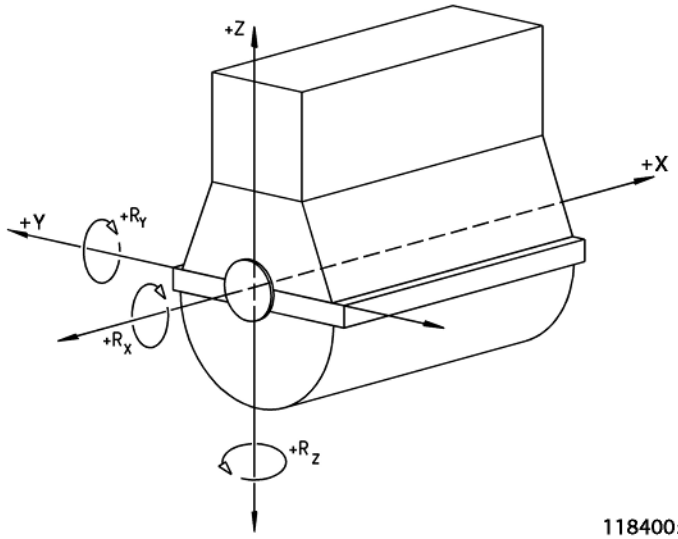

Figure 110: Coordinate system

Generally flexible pipes (rubber hoses with steel inlet, metal hoses, PTFE-corrugated hose-lines, rubber bellows with steel inlet, steel bellows, steel compensators) are nearly unable to compensate twisting movements. Therefore the installation direction of flexible pipes must be vertically (in Z-direction) if ever possible. Torsion on flexible pipe connections must be avoided. Flexible pipe connections which are installed in X-direction are particularly at risk. Therefore the installation of flexible pipe connections in this direction should be avoided. Where the installation of flexible pipe connections in X-direction is nevertheless unavoidable, the continuing pipeline on the plant side must be designed in such a way that the torsional forces can be safely absorbed. An installation in horizontal-lateral (Y-direction) is not recommended.

The media connections (compensators) to and from the engine must be highly flexible whereas the fixations of the compensators on the one hand with the engine and on the other hand with the environment must be realised as stiff as possible.

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Flange and screw connections

Flexible pipes delivered loose by MAN Energy Solutions are fitted with flange connections from DN32 upwards. Smaller sizes are fitted with screw connections. Each flexible pipe is delivered complete with counter flanges or, those smaller than DN32, with weld-on sockets.

Arrangement of the external piping system

Shipyard's pipe system must be exactly arranged so that the flanges or screw connections do fit without lateral or angular offset. Therefore it is recommended to adjust the final position of the pipe connections after engine alignment is completed.

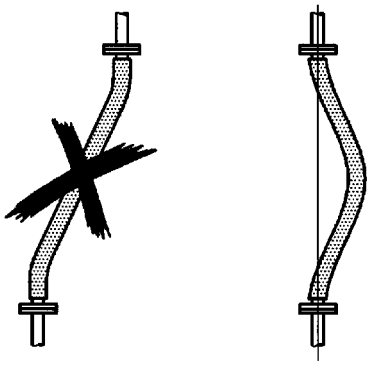

Figure 111: Arrangement of pipes in system

Installation of hoses

In the case of straight-line-vertical installation, a suitable distance between the hose connections has to be chosen, so that the hose is installed with a sag. To satisfy a correct sag in a straight-line-vertically installed hose, the distance between the hose connections (hose installed, engine stopped) has to be approximately 5 % shorter than the same distance of the unconnected hose (without sag). Flexible hoses must not be installed with tensile stress, compression or torsional tension.

In case it is unavoidable (this is not recommended) to connect the hose in lateral-horizontal direction (Y-direction) the hose must preferably be installed with a 90° arc. The minimum bending radii, specified in provided drawings, are to be observed.

Hoses must not be twisted during installation. Turnable lapped flanges on the hoses avoid this.

Where bolted connections are used, hold the hexagon on the hose with a wrench while fitting the nut.

All installation instructions of the hose manufacturer have to be complied with.

Depending on the required application rubber hoses with steel inlet, metal hoses or PTFE-corrugated hose lines are used.

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Installation of steel compensators

Steel compensators are used for hot media, e.g. exhaust gas. They can compensate movements in line and transversal to their centre line, but they are absolutely unable to compensate twisting movements. Compensators are very stiff against torsion. For this reason all kind of steel compensators installed on resilient mounted engines are to be installed in vertical direction.

Note:
Exhaust gas compensators are also used to compensate for thermal expansion. Exhaust gas compensators are therefore required for all type of engine mountings, also for semi-resilient or rigid mounted engines. But in these cases the compensators can be shorter, as they are designed only to compensate the thermal expansions and vibrations, but not other dynamic engine movements.

Supports of pipes

Flexible pipes must be installed as close as possible to the engine connection.

On the shipside, directly after the flexible pipe, the pipe is to be fixed with a sturdy pipe anchor of higher than normal quality. This anchor must be capable to absorb the reaction forces of the flexible pipe, the hydraulic force of the fluid and the dynamic force.

Example of the axial force of a compensator to be absorbed by the pipe anchor:

- Hydraulic force
  = (cross section area of the compensator) x (pressure of the fluid inside)
- Reaction force
  = (spring rate of the compensator) x (displacement of the comp.)
- Axial force
  = (hydraulic force) + (reaction force)

Additionally a sufficient margin has to be included to account for pressure peaks and vibrations.

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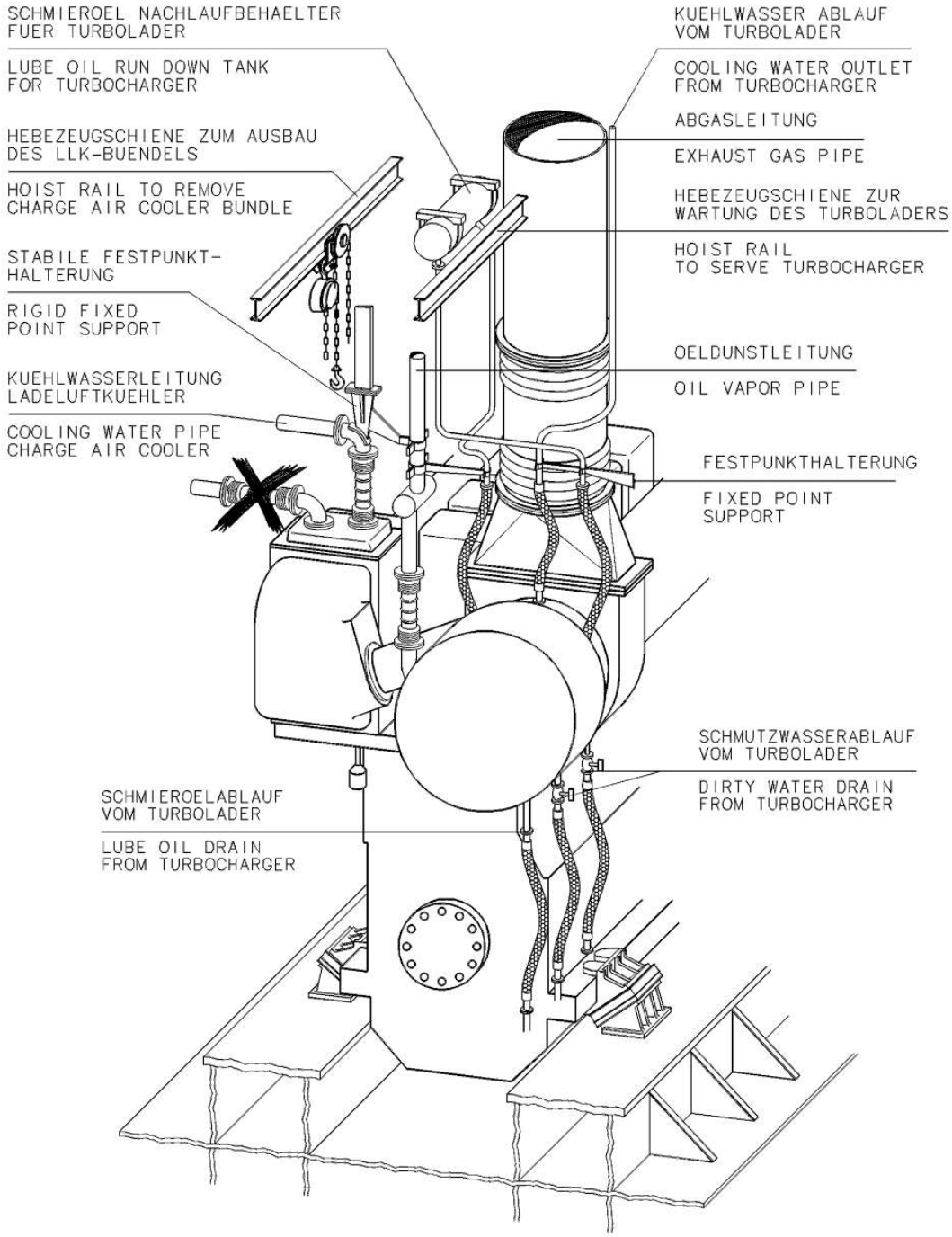

Figure 112: Installation of hoses

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5.2.4 Condensate amount in charge air pipes and air vessels

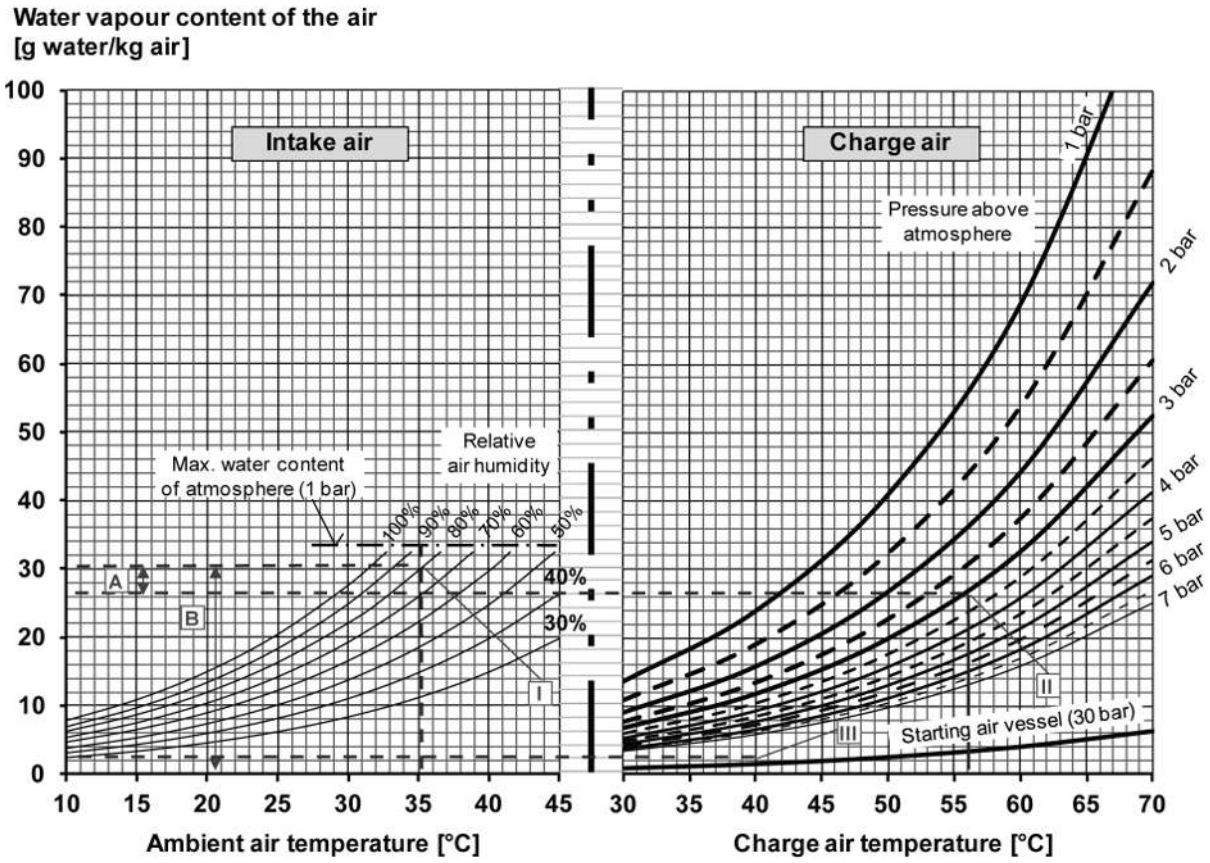

Figure 113: Diagram condensate amount

The amount of condensate precipitated from the air can be considerablly high, particularly in the tropics. It depends on the condition of the intake air (temperature, relative air humidity) in comparison to the charge air after charge air cooler (pressure, temperature).

It is important, that no condensed water of the intake air/charge air will be led to the compressor of the turbocharger, as this may cause damages.

In addition the condensed water quantity in the engine needs to be minimised. This is achieved by controlling the charge air temperature.

How to determine the amount of condensate:

First determine the point I of intersection in the left side of the diagram (intake air), see figure Diagram condensate amount, Page 305 between the corresponding relative air humidity curve and the ambient air temperature.

Secondly determine the point II of intersection in the right side of the diagram (charge air) between the corresponding charge air pressure curve and the charge air temperature. Note that charge air pressure as mentioned in section Planning data, Page 100 is shown as absolute pressure.

At both points of intersection read out the values [g water/kg air] on the vertically axis.

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The intake air water content I minus the charge air water content II is the condensate amount A which will precipitate. If the calculations result is negative no condensate will occur.

For an example see figure Diagram condensate amount, Page 305. Intake air water content 30 g/kg minus 26 g/kg = 4 g of water/kg of air will precipitate.

To calculate the condensate amount during filling of the starting air receiver just use the 30 bar curve (see figure Diagram condensate amount, Page 305) in a similar procedure.

Example how to determine the amount of water accumulating in the charge air pipe

Parameter	Unit	Value
Engine output (P)	kW	9,000
Specific air flow (le)	kg/kWh	6.9
Ambient air condition (I):		
Ambient air temperature	°C	35
Relative air humidity	%	80
Charge air condition (II):		
Charge air temperature after cooler[1)]	°C	56
Charge air pressure (over pressure)[1)]	bar	3.0
Solution according to above diagram		
Water content of air according to point of intersection (I)	kg of water/kg of air	0.030
Maximum water content of air according to point of intersection (II)	kg of water/kg of air	0.026
The difference between (I) and (II) is the condensed water amount (A) A = I – II = 0.030 – 0.026 = 0.004 kg of water/kg of air		
Total amount of condensate Q_A: Q_A = A x le x P Q_A = 0.004 x 6.9 x 9,000 = 248 kg/h		
[1)] In case of two-stage turbocharging choose the values of the high-pressure TC and cooler (second stage of turbocharging system) accordingly.		

Table 190: Example how to determine the amount of water accumulating in the charge air pipe

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Example how to determine the condensate amount in the starting air receiver

Parameter	Unit	Value
Volumetric capacity of tank (V)	litre m^3	3,500 3.5
Temperature of air in starting air receiver (T)	°C K	40 313
Air pressure in starting air receiver ($p_{above\ atmosphere}$) Air pressure in starting air receiver ($p_{absolute}$)	bar bar abs $\frac{N}{m^2}$	30 31 31 x 10^5
Gas constant for air (R)	$\frac{Nm}{kg \times K}$	287
Ambient air temperature	°C	35
Relative air humidity	%	80
Weight of air in the starting air receiver is calculated as follows: $m = \frac{p \times V}{R \times T} = \frac{31 \times 10^5 \times 3.5}{287 \times 313} = 121\,kg$		
Solution according to above diagram		
Water content of air according to point of intersection (I)	kg of water/kg of air	0.030
Maximum water content of air according to point of intersection (III)	kg of water/kg of air	0.002
The difference between (I) and (III) is the condensed water amount (B) B = I – III B = 0.030 – 0.002 = 0.028 kg of water/kg of air		
Total amount of condensate in the vessel (Q_B) Q_B = m x B Q_B = 121 x 0.028 = 3.39 kg		

Table 191: Example how to determine the condensate amount in the starting air receiver

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5.3 Media interfaces

The following presentation of the media connection numbers is for orientation only.

Final drawings will follow as part of the project-specific execution.

Be aware that distinct media connection numbers are linked to optional engine features only.

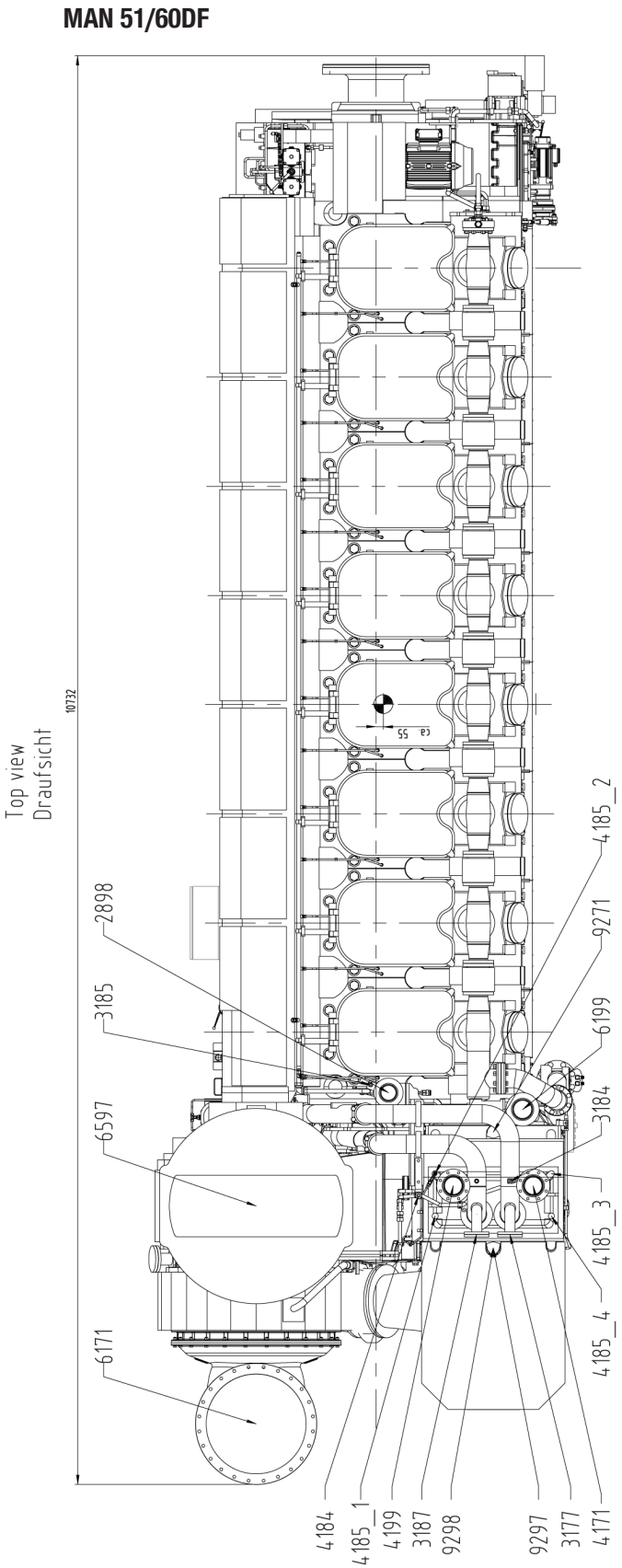

Figure 114: Media interfaces exemplary MAN 51/60DF, top view

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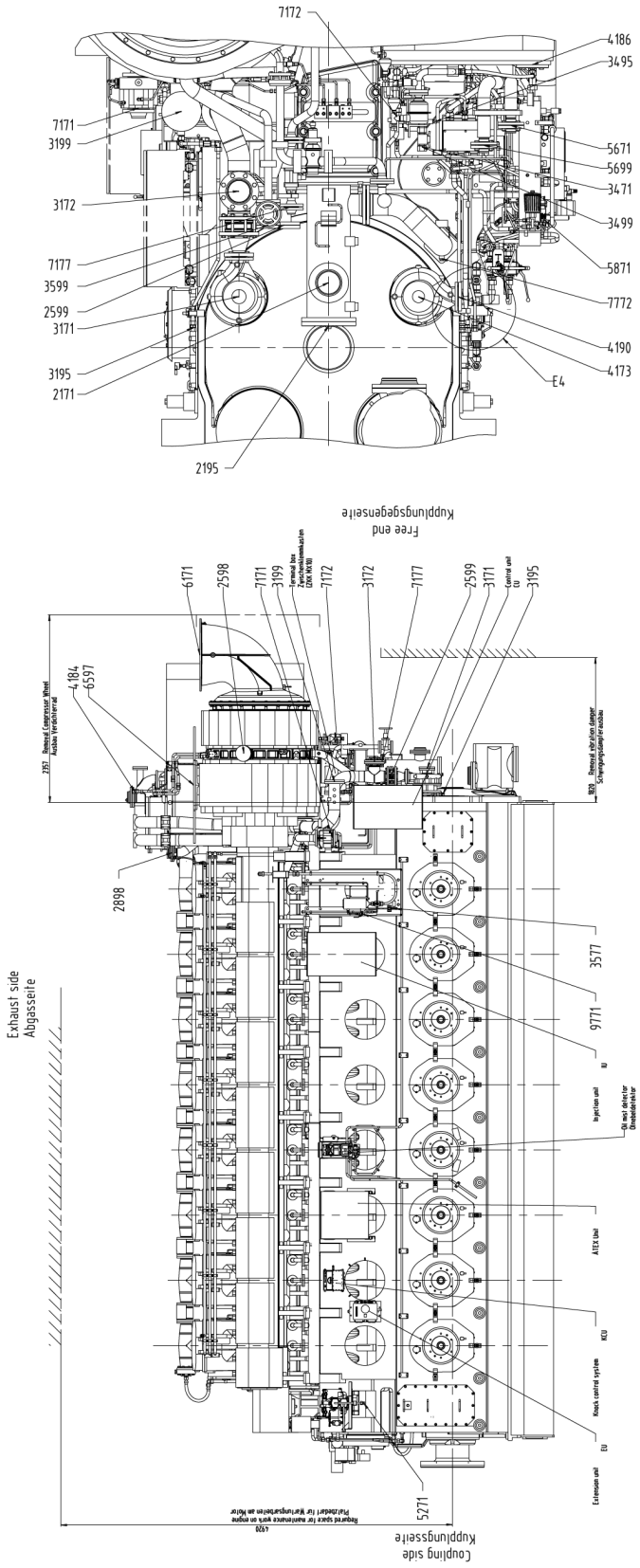

Figure 115: Media interfaces exemplary MAN 51/60DF, exhaust side, free end

Connection No.Anschluss Nr.	X	Y	Z	Description of pipe connection Bennennung des Rohranschlusses	Pipe dimension Rohrabmessungen	Standard code of connection part Normbezeichnung des Anschlussteils	Location: CS or CCS Ort: KS oder KGS	Flow direction Flussrichtung	Detail sketch of connection part Einzelheit des Anschlussteils
2171	0	407	-7465	Oil inlet on the engine Öleintritt am Motor	St.R. 139.7x4	EN 1092-1 11 B1/DN125/PN16 P245GH	Free end Kupplungsgegenseite	Inlet Eintritt	
2173	-443	-797	-7310	Oil pump inlet Ölpumpeneintritt	St.R. 323.9x7.1	EN 1092-1 01 A/DN300/PN10 S235JRG2	Free end Kupplungsgegenseite	Inlet Eintritt	
2175	-443	-257	-7165	Oil pump outlet Ölpumpenaustritt am Motor	St.R. 219.1x6.3	EN 1092-1 11 B1/DN200/PN10 S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
2195	0	136	-7180	**Drain of run in oil filter** **Entleerung des Schmieröl-Einfahrfilters**		Thread/Gewinde G1" DIN 910 - G1A-St	Free end Kupplungsgegenseite	Outlet Austritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
2197	0	-1025	-6830	Drain from oil pan free end Ablauf aus der Motorölwanne KGS		DRW 11 01184 0096	Free end Kupplungsgegenseite	Outlet Austritt	See detail A Siehe Einzelheit A
2199	0	-1025	1150	Drain from oil pan coupling side Ablauf aus der Motorölwanne KS		DRW 11 01184 0096	Coupling side Kupplungsseite	Outlet Austritt	See detail A Siehe Einzelheit A
2598	1658	2624	-7445	Ventilation from turbocharger Entlüftung vom Turbolader	St.R. 76.1x2.9	EN 1092-1 01 A/DN65/PN16-4L S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
2599	281	770	-7180	Drain from turbocharger Ablauf vom Turbolader	St.R. 76.1x2.9	EN 1092-1 01 A/DN65/PN16-4L S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
2898	-93	3520	-6195	Crankcase venting Kurbelraumentlüftung	St.R. 114.3x3.6	EN 1092-1 11 B1/DN100/PN16 S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
3171	564	320	-7255	Cooling water inlet on engine Kühlwassereintritt am Motor	St.R. 88.9x3.2	EN 1092-1 01 A/DN80/PN16 S235JRG2	Free end Kupplungsgegenseite	Inlet Eintritt	
3172	564	982	-7337	Emergency connection (cooling water inlet on the engine) Reserveanschluß (Kühlwassereintritt am Motor)	St.R. 139.7x4	EN 1092-1 11 B1/DN125/PN16 S235JRG2	Free end Kupplungsgegenseite	Inlet Eintritt	
3177	-1005	3925	-7259	HT cooling water inlet reserve connection HT Kühlwassereintritt Reserveanschluss	St.R. 76.1x5	EN 1092-1 05 A/DN65/PN16-4L	Free end Kupplungsgegenseite	Inlet Eintritt	Closed
3184	-1005	4081	-6859	Ventilation of the cooling water pipe Entlüftung Kühlwasserleitung	St.R. 16x2	ISO 8434-1 SDSC - S16xG1/2A-St	Free end Kupplungsgegenseite	Outlet Austritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
3185	41	3515	-6134	Cylinder head venting Zylinderkopf-Entlüftung	St.R. 16x2	ISO 8434-1 BHC - S16-St	Free end Kupplungsgegenseite	Outlet Austritt	
3187	-755	3925	-7259	Charge air preheating water outlet LLK Vorwärmung Kühlwasseraustritt	St.R. 76.1x5	EN 1092-1 05 A/DN65/PN16-4L	Free end Kupplungsgegenseite	Outlet Austritt	Closed
3195	854	122	-6933	Drain of the cooling water pipe Entleerung der Kühlwasserleitung	St.R. 20x2	ISO 8434-1 SC - S20-St	Free end Kupplungsgegenseite	Outlet Austritt	
3199	925	1484	-7087	Cooling water outlet on engine Kühlwasseraustritt aus Motor	St.R. 139.7x4	EN 1092-1 01 A/DN125/PN16 S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
3471	-586	1262	-7330	Cooling water inlet to the needle valve Kühlwassereintritt zum Nadelventil	St.R. 30x2	ISO 8434-1 SDSC - S30xM42x2-St	Free end Kupplungsgegenseite	Inlet Eintritt	
3495	-669	1420	-7480	Drain for needle valve cooling Nadelventilkühlungentleerung	St.R. 10x1.5	ISO 8434-1 SDSC - S10xM16x1.5-St	Free end Kupplungsgegenseite	Outlet Austritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
3499	-586	1262	-7480	Cooling water outlet from needle valve Kühlwasseraustritt am Nadelventil	St.R. 30x2	ISO 8434-1 SDSC - S30xM42x2-St	Free end Kupplungsgegenseite	Outlet Austritt	
3577	1056	718	-5495	Connection for compressor cleaning device wet Anschluss für Nasswascheinrichtung Verdichter			Free end Kupplungsgegenseite	Inlet Eintritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
3599	380	770	-7527	Dirty water drain from turbocharger Schmutzwasserablauf vom Turbolader	St.R. 48.3x2.6	EN 1092-1 01 A/DN40/PN40 S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
4171	-1180	4027	-6894	Cooling water inlet from intercooler step 2 Kühlwassereintritt am Ladeluftkühler Stufe 2	St.R. 139.7x4	EN 1092-1 11 B1/DN125/PN16 S235JRG2	Free end Kupplungsgegenseite	Inlet Eintritt	
4173	-564	320	-7255	Cooling water inlet-pump (suction pipe) Kühlwassereintritt-Pumpe (Saugleitung)	St.R. 88.9x3.2	EN 1092-1 01 A/DN80/PN16 S235JRG2	Free end Kupplungsgegenseite	Inlet Eintritt	
4184	-319	3997	-6968	Compressor wheel cooling Verdichterradkühlung	St.R. 20x2	ISO 8434-1 SC - S20-St	Free end Kupplungsgegenseite	Outlet Austritt	
4185_1	-444	3685	-7121	Ventilation intercooler Entlüftung Ladeluftkühler	St.R. 18x2	ISO 8434-1 SDSC - L18xM22x1.5-St	Free end Kupplungsgegenseite	Outlet Austritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
4185_2	-444	3741	-6801	Ventilation intercooler Entlüftung Ladeluftkühler	St.R. 10x1.5	ISO 8434-1 SDSC - L10xM14x1.5-St	Free end Kupplungsgegenseite	Outlet Austritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
4185_3	-1316	3685	-6803	Ventilation intercooler Entlüftung Ladeluftkühler	St.R. 18x2	ISO 8434-1 SDSC - L18xM22x1.5-St	Free end Kupplungsgegenseite	Outlet Austritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
4185_4	-1316	3685	-7121	Ventilation intercooler Entlüftung Ladeluftkühler	St.R. 18x2	ISO 8434-1 SDSC - L18xM22x1.5-St	Free end Kupplungsgegenseite	Outlet Austritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
4186	-1280	1778	-6878	Drain of the intercooler step 2 Entleerung vom Ladeluftkühler Stufe 2	St.R. 18x2	ISO 8434-1 SDSC - L18xM22x1.5-St	Free end Kupplungsgegenseite	Outlet Austritt	No pipe connection necessary Es muss kein Rohr angeschlossen werden
4190	-814	320	-7155	Cooling water outlet-pump (delivery pipe) Kühlwasseraustritt-Pumpe (Druckleitung)	St.R. 76.1x2.9	EN 1092-1 01 A/DN65/PN16-4L S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
4199	-580	4027	-6894	Cooling water outlet from intercooler step 2 Kühlwasseraustritt am Ladeluftkühler Stufe 2	St.R. 139.7x4	EN 1092-1 11 B1/DN125/PN16 S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
5241	-840	120	-6995	Leakage fuel drain pilot fuel-CR **Leckkraftstoff Ablauf Zündöl-CR**	St.R. 20x2	ISO 8434-1 SC - S20-St	Free end Kupplungsgegenseite	Outlet Austritt	
5271	925	1112	723	Fuel inlet pilot fuel-CR **Kraftstoffeintritt Zündöl-CR**	St.R. 20x2	ISO 8434-1 SC - S20-St	Coupling side Kupplungsseite	Inlet Eintritt	
5645	-860	123	-6936	Fuel oil leakage drain (reuseable) **Kraftstoffleckageablauf (Wiederverwendbar)**	St.R. 30x2	ISO 8434-1 SC - S30-V	Free end Kupplungsgegenseite	Outlet Austritt	
5671	-1145	1397	-7430	Fuel inlet on the engine **Kraftstoffeintritt am Motor**	St.R. 60.3x2.9	EN 1092-1 11 D/DN50/PN40 P245GH	Free end Kupplungsgegenseite	Inlet Eintritt	
5699	-975	1272	-7430	Fuel return pipe from engine **Kraftstoffrücklaufleitung vom Motor**	St.R. 60.3x2.9	EN 1092-1 11 D/DN50/PN40 P245GH	Free end Kupplungsgegenseite	Outlet Austritt	
5871	-1357	802	-6103	Gas inlet on the engine Gaseintritt am Motor	St.R. 114.3x3.6 (St.R. 139.7x4.5)	DRW 11 23484 0104	Free end Kupplungsgegenseite	Inlet Eintritt	See detail B Siehe Einzelheit B
6171	900	3176	-8683	Intake air inlet to turbocharger Ansauglufteintritt in Turbolader		DIN 86044 - 750	Free end Kupplungsgegenseite	Inlet Eintritt	
6199	-1100	3803	-6320	charge air outlet from compressor bypass Ladeluftaustritt aus Verdichterbypass	St.R. 139.7x4	EN 1092-1 11 B1/DN125/PN16 S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	
6597	900	3253	-7046	Exhaust gas outlet Abgasaustritt Lader		DIN 86044 - 1300	Free end Kupplungsgegenseite	Outlet Austritt	
7171	1195	1505	-6380	Air inlet on the starting valve Lufteintritt am Hauptanlaßventil	St.R. 88.9x6.3	DIN 2635-N80x88.9x6.3 C22.8 (PN40)	Free end Kupplungsgegenseite	Inlet Eintritt	
7172	-429	1450	-7560	Starting air for pilot valve Anlaßluft für Steuerschieber	St.R. 25x2	ISO 8434-1 SDSC - S25xM33x2-St	Free end Kupplungsgegenseite	Inlet Eintritt	
7177	235	885	-7350	Air inlet for jet assist Lufteintritt für Jet Assist	St.R. 48.3x2.9	EN 1092-1 11 B1/DN40/PN40 S235JRG2	Free end Kupplungsgegenseite	Inlet Eintritt	
7398	-880	2732	-8576	Air bypass piping Luftabblaseleitung	St.R. 219.1x6.3	EN 1092-1 01 A/DN200/PN6 S235JRG2	Free end Kupplungsgegenseite	Outlet Austritt	**Connection closed - project-specifically defined** **Verbindung geschlossen - projektspezifisch definiert**
7451	-795	-209	1111	Control air from turning gear Steuerluft vom Schaltwerk	St.R. 10x1.5	ISO 8434-1 BHEC - L10-St	Coupling side Kupplungsseite	Inlet Eintritt	
7461	-795	-251	1111	Control air to turning gear Steuerluft zum Schaltwerk	St.R. 10x1.5	ISO 8434-1 BHEC - L10-St	Coupling side Kupplungsseite	Outlet Austritt	
7501	-928	-61	1074	Nitrogen inlet for scavenging of the gas pipe **Stickstoffeintritt zum Spülen der Gasleitung**	St. R. 30x2	ISO 8434-1 SC - S30-V	Coupling side Kupplungsseite	Inlet Eintritt	**Flexible tube and filter are necessary** Schlauch und Filter sind notwendig
7772	-710	559	-6985	Control oil outlet to pressure control valve Steueröl Austritt zum Druckregelventil	St.R. 16x2	ISO 8434-1 SDSC - S16xG1/2A-St	Free end Kupplungsgegenseite	Outlet Austritt	
8171	-919	122	-6932	Heating pipe inlet (free end) Heizleitungseintritt (KGS)	St.R. 16x2	ISO 8434-1 SC - S16-St	Free end Kupplungsgegenseite	Inlet Eintritt	
8172	-743	161	1132	Heating pipe inlet (coupling side) Heizleitungseintritt (KS)	St.R. 16x2	ISO 8434-1 BHC - S16-St	Coupling side Kupplungsseite	Inlet Eintritt	
8198	-798	161	1132	Heating pipe outlet (coupling side) Heizleitungsaustritt (KS)	St.R. 16x2	ISO 8434-1 BHC - S16-St	Coupling side Kupplungsseite	Outlet Austritt	
8199	-919	181	-6858	Heating pipe outlet (free end) Heizleitungsaustritt (KGS)	St.R. 16x2	ISO 8434-1 BHC - S16-St	Free end Kupplungsgegenseite	Outlet Austritt	
8697	-811	122	1187	Condensed water drain from the charging air manifold Kondenswasserablauf von der Ladeluftleitung	St. R. 20x2	ISO 8434-1 SC - S20-V	Coupling side Kupplungsseite	Outlet Austritt	
8699	-411	1386	-7630	Condensed water drain from the intercooler Kondenswasserablauf vom Ladeluftkühler	St.R. 20x2	ISO 8434-1 SDSC - S20xG3/4A-V	Free end Kupplungsgegenseite	Outlet Austritt	
9197	-854	120	1132	Dirty oil drain from covering Schmutzölablauf von der Verschalung	St.R. 20x2	ISO 8434-1 SC - S20-V	Coupling side Kupplungsseite	Outlet Austritt	
9199	-860	179	-6858	Dirty oil drain from covering Schmutzölablauf von der Verschalung	St.R. 20x2	ISO 8431-1 BHC - S20-V	Free end Kupplungsgegenseite	Outlet Austritt	
9271	-880	3568	-6490	Inlet of the detergent on casing after intercooler Zulauf des Reinigungsmittels am Gehäuse nach LLK		R 572-50	Free end Kupplungsgegenseite	Inlet Eintritt	See detail C Siehe Einzelheit C
9297	-880	1892	-7360	Outlet of the detergent on air inlet casing Ablauf des Reinigungsmittels am Gehäuse vor LLK		R 572-50	Free end Kupplungsgegenseite	Outlet Austritt	See detail C Siehe Einzelheit C
9298	-880	3568	-7360	Venting Entlüftung		R 572-50	Free end Kupplungsgegenseite	Outlet Austritt	See detail C Siehe Einzelheit C
9771	1113	1125	-5429	Connection for turbine dry cleaning device Anschluss für Trockenwascheinrichtung Turbine	St.R. 15x2	ISO 8434-1 SDSC - L15xG1/2A-V	Free end Kupplungsgegenseite	Inlet Eintritt	See detail D Siehe Einzelheit D
GRAVITY_POINT	-55	1004	-3276	Point of gravity Schwerpunkt					Theoretically dimension Theoretisches Maß Reference: base coordinate system of the engine Referenz: Basiskoordinatensystem des Motors

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Figure 116: Connections part 1

Connection 3495: No pipe connection necessary.
The pipe can be drain with a collect vessel.
Anschluss 3495: Es muss kein Rohr angeschlossen werden.
Leitung kann mit Auffanggefäß entleert werden.

Connection 3577: The mobile washing device must be connected for cleaning.
Anschluss 3577: Zur Reinigung muss die mobile Reinigungseinheit angeschlossen werden.

Connection 4185: No pipe connection necessary.
Air vent only for filling.
Anschluss 4185: Es muss kein Rohr angeschlossen werden.
Entlüftung nur zum Befüllen.

Connection 9297: Drain of cleaning liquid using flange with slider (international shore connection of DIN 86201) from ultrasonic cleaning of CAC.
Anschluss 9297: Entleerung der Reinigungsflüssigkeit mittels Flansch mit Schieber (Internationaler Landanschluss DIN 86201) aus Ultraschallreinigungsanlage des LLK.

Connection 9771: Only connected during cleaning.
Anschluss 9771: Nur während der Reinigung angeschlossen.

Connection-No. Anschluss Nr.	Typ of medium Art des Medium
21..	Lubrication oil Schmieröl
25..	Lubrication oil turbocharger Schmieröl Turbolader
28..	Oil mist Öldunst
31..	Cooling water Kühlwasser
34..	Cooling water needle valve Kühlwassernadelventil
35..	Sanitary water Sanitärwasser
41..	Cooling water intercooler Kühlwasser Ladeluftkühler
52..	Fuel inlet Kraftstoffeintritt
56..	Diesel oil Kraftstoff Dieselöl
58..	Fuel gas Gas
65..	Exhaust gas Abgas
71..	Starting air and control air Ansaugluft und Steuerluft
74..	Compressd air Druckluft
75..	Nitrogen Stickstoff
77..	Control oil Steueröl
81..	Steam – Thermooil Dampf – Thermoöl
86..	Condensation water Kondenswasser
91..	Dirty oil Schmutzöl
92..	Cleaning intercoler Reingung Ladeluftkühler
97..	Dry cleaning turbocharger Trockenreinigung Turbolader

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Figure 117: Connections part 2

5.4 Lube oil system

5.4.1 Internal lube oil system

As a standard:

- Engine equipment with attached lube oil pump

Internal lube oil system – Exemplary

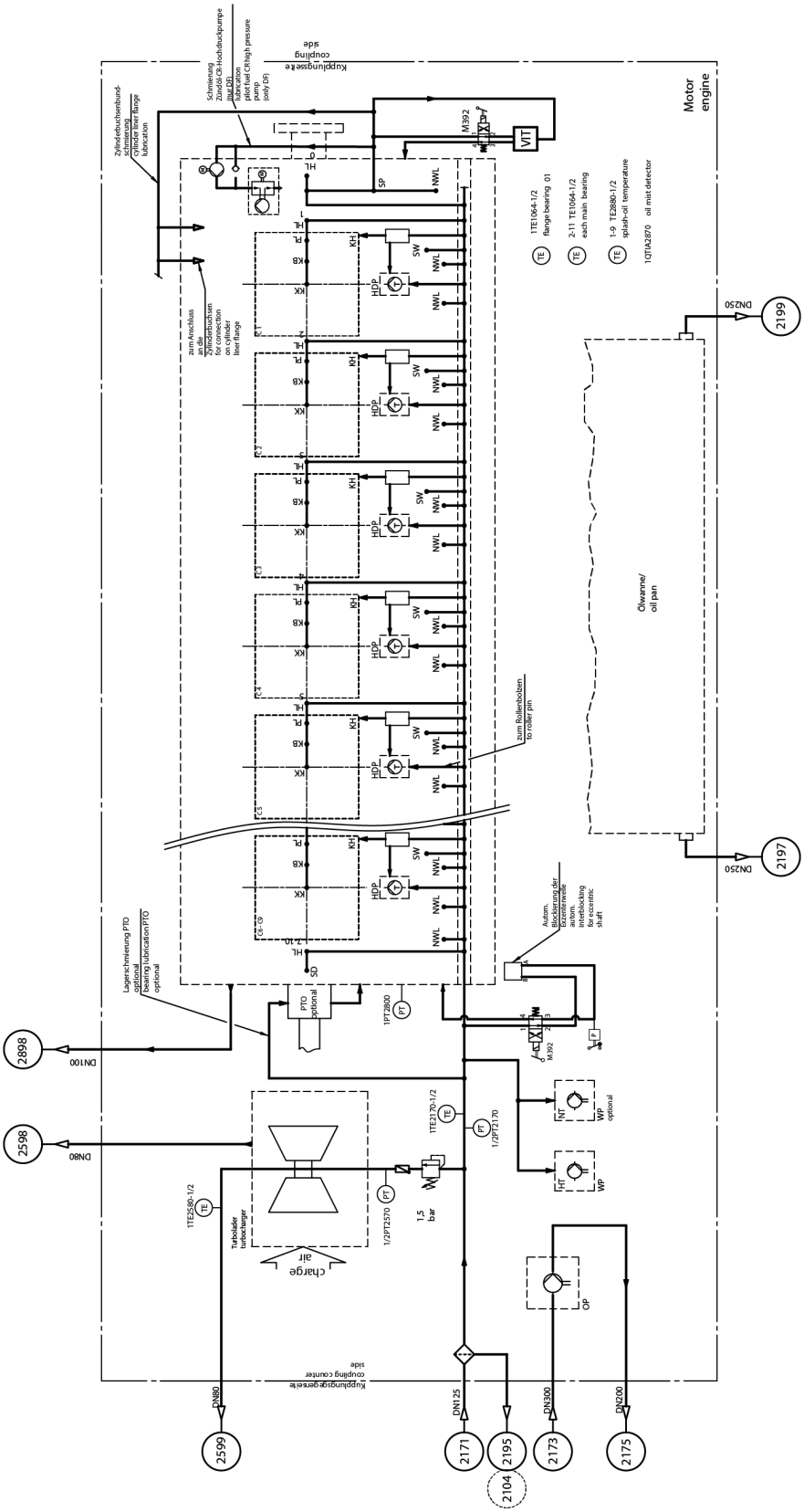

Figure 118: Internal lube oil system MAN 51/60DF – Exemplary

Connection numbers			
2104	Lube oil inlet to engine for provisional lubrication	2197	Drain from oil pan, counter coupling side 1
2171	Lube oil inlet to engine	2199	Drain from oil pan, coupling side 1
2173	Lube oil inlet to lube oil pump 1	2598	Venting of turbocharger 1
2175	Lube oil outlet from lube oil pump 1	2599	Lube oil drain from turbocharger 1
2195	Drain of run in oil filter	2898	Venting of crankcase 1

5.4.2 External lube oil system – Description

The following description refers to the figure(s) Lube oil system diagram(s), Page 322, which represent the standard design of external lube oil service system.

The internal lubrication of the engine and the turbocharger is provided with a force-feed lubrication system.

In multi-engine plants, for each engine a separate lube oil system is required.

According to the required lube oil quality, see table Main fuel/lube oil type, Page 255.

For dual fuel engines (gas-diesel engines) the brochure "Safety Concept – Marine dual fuel engines" will explain additional specific requirements.

Requirements before commissioning of engine

The flushing of the lube oil system in accordance to the MAN Energy Solutions specification (see the relevant working instructions) demands before commissioning of the engine, that all installations within the system are in proper operation. Be aware that special installations for commissioning are required and the lube oil separator must be in operation from the very first phase of commissioning.

Contact MAN Energy Solutions or licensee if any uncertainties occur.

T-001/Lube oil service tank

The main purpose of the lube oil service tank is to separate air and particles from the lube oil, before pumping the lube oil to the engine. For the design of the service tank the class requirements have to be taken in consideration. For design requirements of MAN Energy Solutions see section External lube oil system – Lube oil service tank, Page 329.

H-002/Lube oil preheater

To fulfill the starting conditions (see section Starting conditions, Page 52) preheating of the lube oil in the lube oil service tank is necessary. Therefore the preheater of the separator is often used. The preheater must be enlarged in size if necessary, so that it can heat up the content of the service tank to ≥ 40 °C, within 4 hours. If engines have to be kept in stand-by mode, the lube oil of the corresponding engines always has to be in the temperature range of starting conditions. Means that also the maximum lube oil temperature limit should not be exceeded during engine start.

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Suction pipes

Suction pipes must be installed with a steady slope and dimensioned for the total resistance (incl. pressure drop for suction filter) not exceeding the pump suction head. Before engine starts, venting of suction line must be warranted. Therefore the design of the suction line must be executed accordingly.

PSV-004/Lube oil non-return flap with integrated safety valve

A non-return flap must be installed close to the lube oil tank to prevent lube oil back flow when the engine has been shut off. This non-return flap must be by-passed by a safety valve to protect the pump against high pressure caused by momentary counter-rotation of the engine during shutdown. MAN Energy Solutions solution for these two requirements is a special non-return flap with integrated safety valve. If there is used a normal return flap, the line of the external safety valve should lead back into the lube oil tank submerged. The required opening pressure of the safety valve is approximately 0.4 bar.

FIL-004/Lube oil suction strainer

The lube oil suction strainer protects the lube oil pumps against larger dirt particles that may have accumulated in the tank. It is recommended to use a cone type strainer with a mesh size of 1.5 mm. Two manometers installed before and after the strainer indicate when manual cleaning of filter becomes necessary, which should preferably be done in port.

P-001/P-007/P-074/Lube oil pumps

For ships with more than one main engine additionally to the service pump a prelubrication pump P-007 for pre- and postlubrication is necessary. Dependent on the type of prelubrication pump, an orifice on the discharge side could be necessary, to comply with the required differential pressure over the pump, given by the pump manufacturer. For further information according that pump see section Planning data, Page 100 and paragraph Lube oil, Page 149. A main lube oil pump as spare is required to be on board according to class society.

For ships with a single main engine drive it is preferable to design the lube oil system with a combination of an engine driven lube oil service pump (attached) P-001 and a lube oil stand-by pump (free-standing) P-074 (100 % capacity).

Additionally a prelubrication pump is recommended. If nevertheless the stand-by pump is used for pre- and postlubrication MAN Energy Solutions has to be consulted as there are necessary modifications in the engine automation.

The prelubrication pump must be located as low as possible and close to the lube oil service tank to prevent cavitation. The pressure drop in the piping must not exceed the suction capability of the pump. With adequate diameter, straight lines and short length the pressure drop can be kept low.

Using the stand-by pump for continuous prelubrication is not permissible.

As long as the installed stand-by pump provides 100 % capacity of the operating pump, the class requirement to have a spare part operating pump on board, is fulfilled. Both pumps must be located as low as possible and close to the lube oil service tank to prevent cavitation. The pressure drop in the piping must not exceed the suction capability of the pump. With adequate diameter, straight lines and short length the pressure drop can be kept low.

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For design data of these lube oil pumps see section Planning data, Page 100 and the following.

In case of unintended engine stop (e.g. blackout) the postlubrication must be started as soon as possible (latest within 20 min) after the engine has stopped and must persist for 15 min.

This is required to cool down the bearings of turbocharger and hot inner engine components.

Application	Necessary pumps referred to respective application(s)			
	For operation		For pre- and postlubrication	To keep engine in stand-by
Single main engine	Lube oil service pump (attached) P-001	Lube oil stand-by pump P-074 (100 %)	Prelubrication pump P-007 recommended. If stand-by pump P-074 should be used for pre- and postlubrication, MAN Energy Solutions has to be consulted.	Prelubrication pump P-007 is required
Ships with more than one main engine	Lube oil service pump (attached) P-001	Lube oil stand-by pump P-074 recommended for increased availability (safety). Otherwise pump as spare is requested to be on board according to class requirement.	Prelubrication pump P-007 recommended. If stand-by pump P-074 should be used for pre- and postlubrication, MAN Energy Solutions has to be consulted.	Prelubrication pump P-007 is required

Table 192: Lube oil pumps

HE-002/Lube oil cooler

Dimensioning

Heat data, flow rates and tolerances are indicated in section Planning data, Page 100 and the following.

On the lube oil side, the pressure drop shall not exceed 1.1 bar.

Design/Outfitting

The cooler installation must be designed for easy venting and draining.

TCV-001/Lube oil temperature control valve

The lube oil temperature control valve regulates the inlet oil temperature of the engine. The control valve can be executed with wax-type thermostats.

Set point lube oil inlet temperature	Type of temperature control valve[1)]
55 °C	Thermostatic control valve (wax/copper elements) or electrically actuated control valve (interface to engine control)
[1)] Full open temperature of wax/copper elements must be equal to set point. Control range lube oil inlet temperature: Set point minus 10 K.	

Table 193: Lube oil temperature control valve

Lube oil treatment

The treatment of the circulating lube oil can be divided into two major functions:

- Removal of contaminations to keep up the lube oil performance.
- Retention of dirt to protect the engine.

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The removal of combustion residues, water and other mechanical contaminations is the major task of separators/centrifuges (CF-001) installed in by-pass to the main lube oil service system of the engine. The installation of a lube oil separator per engine is recommended to ensure a continuous separation during engine operation.

The lube oil filters integrated in the system protect the diesel engine in the main circuit retaining all residues which may cause a harm to the engine.

Depending on the filter design, the collected residues are to be removed from the filter mesh by automatic back flushing, manual cleaning or changing the filter cartridge. The retention capacity of the installed filter should be as high as possible.

When selecting an appropriate filter arrangement, the customer request for operation and maintenance, as well as the class requirements, have to be taken in consideration.

Instead of a separator an adequate filtration system (FIL-027) can be used for lube oil treatment. This is only valid for engines which operate with liquid fuels of DM- or DF-class (acc. ISO 8217) exclusively (gas also for dual fuel engines, no heavy fuel oil). By using a filtration system, the used lube oil must be suitable for filtration. A separate heater to preheat the lube oil before engine start has to be foreseen. The filtration system must be approved by MAN Energy Solutions.

FIL-001/FIL-002 Arrangement principles for lube oil filters

Depending on engine type, the number of installed main engines in one plant and on the safety standard demanded by the customer, different arrangement principles for the filters FIL-001/FIL-002 are possible:

	Option 1		Option 2	
	FIL-001 automatic filter continous flushing	**FIL-002 duplex filter as indicator filter**	**FIL-001 automatic filter intermittent flushing**	**FIL-002 duplex filter as indicator filter**
FIL-001 includes second filter stage	Yes	-	No	-
Location	Engine room installed close to engine	Installed upstream of FIL-001	Engine room installed close to engine	Installed upstream of FIL-001
Requirement by-pass	Internal by-pass	-	Required	-
Requirement of FIL-002	To fulfill higher safety concept (optional)		Required	
Mesh width	34 µm first filter stage 80 µm second filter stage	60 µm	34 µm	60 µm
It is always recommended to install one separator in partial flow of each engine. Filter design has to be approved by MAN Energy Solutions.				

Table 194: Arrangement principles for lube oil filters

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FIL-001/Lube oil automatic filter

The lube oil automatic filter is an automatic back washing filter installed as a main filter. The back washing/flushing of the filter elements has to be arranged in a way that lube oil flow and pressure will not be affected. The flushing discharge (oil sludge mixture) is led to the lube oil service tank. The oil will be permanently by-pass cleaned via suction line into a separator. This provides an efficient final removal of deposits (see section External lube oil system – Lube oil service tank, Page 329).

As state-of-the-art, lube oil automatic filter types are recommended to be equipped with an integrated second filtration stage. This second stage protects the engine from particles which may pass the first stage filter elements in case of any malfunction. If the lube oil system is equipped with a two-stage automatic filter, additional lube oil duplex filter FIL-002 can be avoided. As far as the automatic filter is installed without any additional filters downstream before the engine inlet, the filter has to be installed as close as possible to the engine (see table Arrangement principles for lube oil filters, Page 316). In that case the pipe section between filter and engine inlet must be closely inspected before installation. This pipe section must be divided and flanges have to be fitted so that all bends and welding seams can be inspected and cleaned prior to final installation.

Differential pressure gauges have to be installed to protect the filter cartridges and to indicate clogging condition of the filter. A high differential pressure has to be indicated as an alarm.

In case filter stage 1 is not working sufficiently, the engine can run in emergency operation for maximum 72 hours with the second filter stage, but has to be stopped after. This measure ensures that disturbances in backwashing do not result in a complete failure of filtering and that the main stream filter can be cleaned without interrupting filtration.

FIL-002/Lube oil duplex filter as indicator filter

The lube oil duplex filter has the function of an indicator filter and must be cleaned manually. It must be installed downstream of the lube oil automatic filter, as close as possible to the engine. The pipe section between filter and engine inlet must be closely inspected before installation. This pipe section must be divided and flanges have to be fitted so that all bends and welding seams can be inspected and cleaned prior to final installation. In case of a two-stage automatic filter, the installation of a duplex filter can be avoided. Customers who want to fulfil a higher safety level, are free to mount an additional duplex filter close to the engine.

The lube oil duplex filter protects the engine also in case of malfunctions of the lube oil automatic filter. The monitoring system of the automatic filter generates an alarm signal to alert the operating personnel. A maintenance of the automatic filter becomes necessary. For this purpose the lube oil flow through the automatic filter has to be stopped. Single-main engine plants may continue to stay in operation by by-passing the automatic filter. Lube oil can still be filtrated sufficiently in this situation by only using the duplex filter.

In multi-engine plants, where it is not possible to by-pass the lube oil automatic filter without loss of lube oil filtration, the affected engine has to be stopped in this situation.

The design of the lube oil duplex filter must ensure that no parts of the filter can become loose and enter the engine.

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The drain connections equipped with shut-off fittings in the two chambers of the lube oil duplex filter returns into the leakage oil collecting tank (T-006). Draining will remove the dirt accumulated in the casing and prevents contamination of the clean oil side of the filter. Check also table Arrangement principles for lube oil filters, Page 316.

Indication and alarm of filters

The lube oil automatic filter FIL-001 and the lube oil duplex filter FIL-002 are equipped with local visual differential pressure indicators and additionally with differential pressure switches. The switches are used for pre-alarm and main alarm.

Differential pressure between filter inlet and outlet (dp)	Lube oil automatic filter FIL-001		Lube oil duplex filter FIL-002
	Intermittent flushing	Continuous flushing	
dp switch with lower set point is active	This dp switch has to be installed twice if an intermittent flushing filter is used. The first switch is used for the filter control; it will start the automatic flushing procedure. The second switch is adjusted at the identical set point as the first. Once the second switch is activated, and after a time delay of approximately 3 minutes, the dp pre-alarm "filter is polluted" is generated. The time delay becomes necessary to effect the automatic flushing procedure before and to evaluate its effect.	The dp pre-alarm: "Filter is polluted" is generated immediately	
dp switch with higher set point is active	The dp main alarm "filter failure" is generated immediately. If the main alarm is still active after 30 min, the engine output power will be reduced automatically.		

Table 195: Indication and alarm of filters

BL-007/Fan, crankcase venting

To dilute the crankcase atmosphere to a safe level it is necessary to produce a small quantity of additional airflow to the crankcase. This will be achieved by producing a vacuum in the crankcase using a speed controlled venting fan placed within the engine ventilation pipe and regulated via a pressure transmitter placed on the crankcase. Distance between engine and venting fan shall be min. 7 metres and max. 10 metres.

The pressure loss of the piping (including all installed components in the piping) after the fan should not exceed 6 mbar.

Engine operation in gas mode is coupled to a functional check of the venting fan device. If the venting fan is malfunctioning, the engine will be forced to change over to diesel mode via engine control. Quick changeover is not necessary because the volume of the crankcase is large compared to the blow-by amount and accumulation of gases is delayed.

CF-001/Lube oil separator

The lube oil is intensively cleaned by separation in the by-pass thus relieving the filters and allowing an economical design.

The lube oil separator should be of the self-cleaning type. The design is to be based on a lube oil quantity of 1.0 l/kW. This lube oil quantity should be cleaned within 24 hours at:

- HFO-operation 6 – 7 times
- MDO-operation 4 – 5 times

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- Dual fuel engines operating on gas (+ MDO/MGO for ignition only) 4 – 5 times

The formula for determining the separator flow rate (Q) is:

$$Q = \frac{1.0 \times P \times n}{24}$$

Q [l/h]	Separator flow rate
P [kW]	Total engine output
n	HFO = 7 MDO/MGO = 5 Gas (+ MDO/MGO for ignition only) = 5

With the evaluated flow rate the size of separator has to be selected according to the evaluation table of the manufacturer. The separator rating stated by the manufacturer should be higher than the flow rate (Q) calculated according to the formula above.

Separator equipment

The lube oil preheater H-002 must be able to heat the oil to 95 °C and the size is to be selected accordingly. In addition to a PI-temperature control, which avoids a thermal overloading of the oil, silting of the preheater must be prevented by high turbulence of the oil in the preheater.

Control accuracy ±1 °C.

Cruise ships operating in arctic waters require larger lube oil preheaters. In this case the size of the preheater must be calculated with a Δt of 60 K.

The freshwater supplied must be treated as specified by the separator supplier.

The supply pumps shall be of the free-standing type, i.e. not mounted on the separator and are to be installed in the immediate vicinity of the lube oil service tank.

This arrangement has three advantages:

- Suction of lube oil without causing cavitation.
- The lube oil separator does not need to be installed in the vicinity of the service tank but can be mounted in the separator room together with the fuel oil separators.
- Better matching of the capacity to the required separator throughput.

As a reserve for the lube oil separator, the use of the diesel fuel oil separator is admissible. For reserve operation the diesel fuel oil separator must be converted accordingly. This includes the pipe connection to the lube oil system which must not be implemented with valves or spectacle flanges. The connection is to be executed by removable change-over joints that will definitely prevent MDO from getting into the lube oil circuit. See also rules and regulations of classification societies.

FIL-027/Lube oil fine filter

The lube oil is intensively cleaned by fine filtration in the by-pass thus relieving the main filters and allowing an economical design.

The lube oil fine filter is not of self-cleaning type, therefore the filter elements need to be replaced. The design is to be based on a lube oil quantity of 1.0 l/kW. This lube oil quantity should be cleaned within 24 hours at:

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- Liquid fuels of DM- and DF-class operation 4 – 5 times
- Dual fuel engines operating on gas (+MDO/MGO for ignition only) 4 – 5 times

The formula for determining the fine filter flow rate (Q) is:

$$Q = \frac{1.0 \times P \times n}{24}$$

Q [l/h]	Fine filter flow rate
P [kW]	Total engine output
n	MDO/MGO = 5 Gas (+ MDO/MGO for ignition only) = 5

With the evaluated flow rate and the lube oil volume of the engine, the number of filter elements of the fine filter has to be selected according to the evaluation table of the maker. The table per maker is based on a fixed filter life time stated in the table (filter lifetime in engine operating hours). This has to be considered carefully. For a longer filter exchange interval, the number of filter elements has to be increased.

The information regarding released manufacturers and filter types for lube oil fine filters can be found here: https://www.man-es.com/documentation/lube-oil-treatment.

Fine filter equipment

Depending on the size of the fine filter, a lifting possibility above the filter housing can be necessary for filter exchange.

The supply pump shall be of freestanding type, that is not mounted on the same frame as the fine filter. It is to be installed in the immediate vicinity of the lube oil service tank.

This arrangement has three advantages:

- Suction of lube oil without causing cavitation.
- The lube oil fine filter does not need to be installed near of the service tank. It can be mounted in one room together with the fuel oil filters or separators.
- Better accessibility of lube oil fine filter for filter exchange.

As a reserve for the lube oil fine filter, the use of the diesel fuel oil separator is admissible. For reserve operation, the diesel fuel oil separator must be converted accordingly. This includes the pipe connection to the lube oil system which must not be implemented with valves or spectacle flanges. The connection is to be executed by removable changeover joints that will definitely prevent MDO from getting into the lube oil circuit. See also rules and regulations of classification societies.

PCV-007/Pressure relief valve

By use of the pressure relief valve, a constant lube oil pressure before the engine is adjusted.

The pressure relief valve is installed upstream of the lube oil cooler. By spilling off exceeding lube oil quantities upstream of the major components these components can be sized smaller. The return pipe (spilling pipe) from the pressure relief valve returns into the lube oil service tank.

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The control line of the pressure relief valve has to be connected to the engine inlet. In this way the pressure losses of filters, pipes and cooler are compensated.

TR-001/Condensate trap

See section Crankcase vent and lube oil tank vent, Page 333.

T-006/Leakage oil collecting tank

See section External fuel system – Heavy fuel oil (HFO) supply system, Page 384.

Withdrawal points for samples

Points for drawing lube oil samples are to be provided upstream and downstream of the filters and the separator, to verify the effectiveness of these system components.

Piping system

It is recommended to use pipes according to the pressure class PN16.

P-012/Lube oil transfer pump

The lube oil transfer pump supplies fresh oil from the lube oil storage tank to the operating tank. Starting and stopping of the lube oil transfer pump should preferably be done automatically by float switches fitted in the tank.

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Lube oil system diagrams

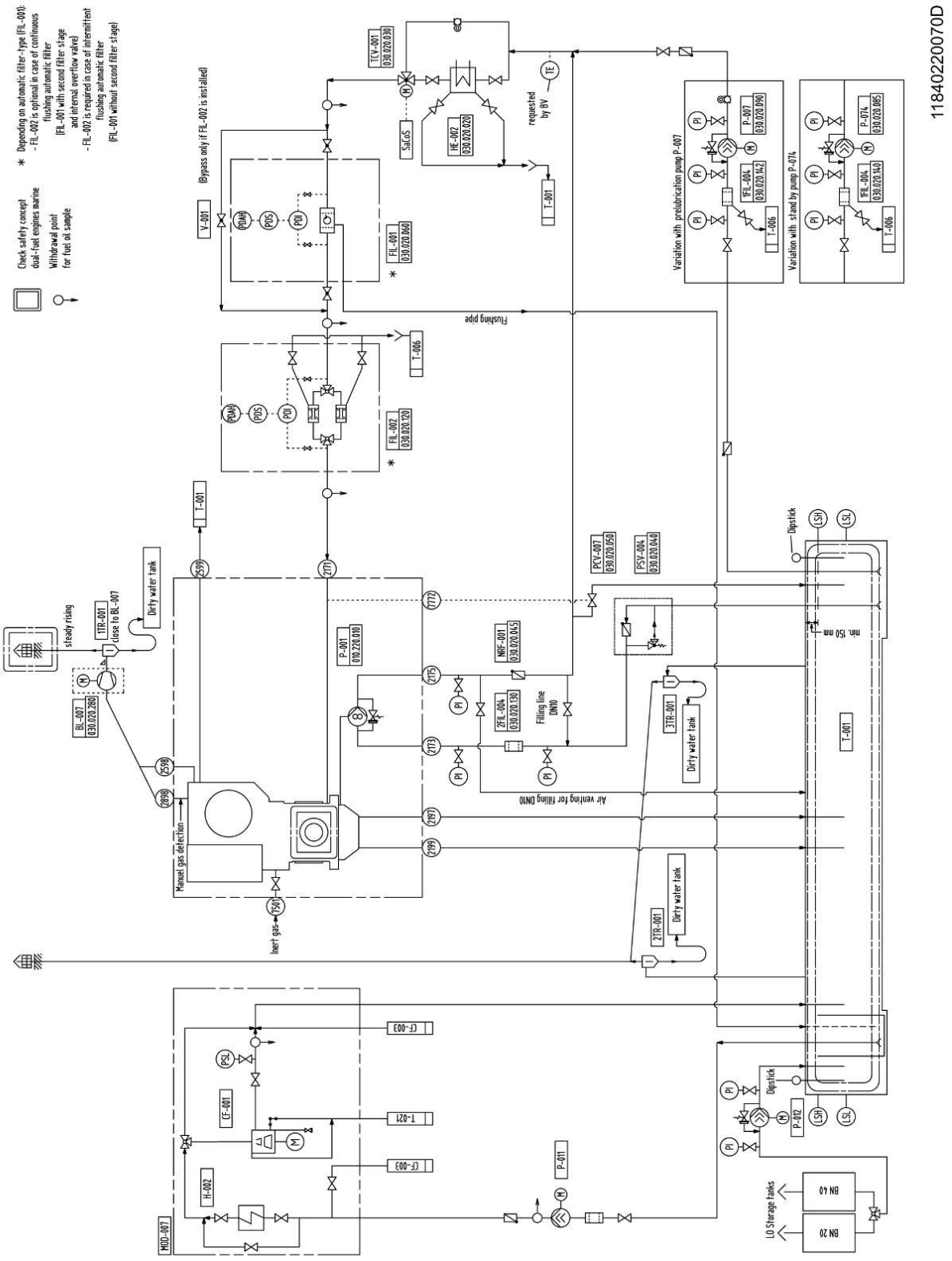

Figure 119: Lube oil system diagram – L engine

Components			
BL-007	Fan, crankcase venting	P-011	Lube oil feed pump separator
CF-001	Lube oil separator	P-012	Lube oil transfer pump
CF-003	Diesel fuel oil separator	P-074	Lube oil stand-by pump, free-standing
FIL-001	Lube oil automatic filter	PCV-007	Lube oil pressure relief valve
FIL-002	Lube oil duplex filter	PSV-004	Lube oil non-return flap with integrated safety valve
1,2 FIL-004	Lube oil suction strainer	T-001	Lube oil service tank
H-002	Lube oil preheating unit	T-006	Leakage oil collecting tank
HE-002	Lube oil cooler	T-021	Sludge tank
MOD-007	Lube oil separator module	TCV-001	Lube oil temperature control valve
NRF-001	Lube oil non-return flap	1,2,3 TR-001	Condensate trap, lube oil system
P-001	Lube oil service pump, attached	V-001	Lead sealed globe valve, bypass to lube oil main filter
P-007	Prelubrication pump		
Connections numbers			
2171	Lube oil inlet on engine	2598	Venting of turbocharger 1
2173	Lube oil inlet to lube oil pump 1	2599	Lube oil drain from turbocharger 1
2175	Lube oil outlet from lube oil pump 1	2898	Venting of crankcase 1
2197	Lube oil drain from oil pan, counter coupling side 1	7501	Inert gas inlet to crankcase 1
2199	Lube oil drain from oil pan, coupling side 1	7772	Control oil outlet to pressure control valve

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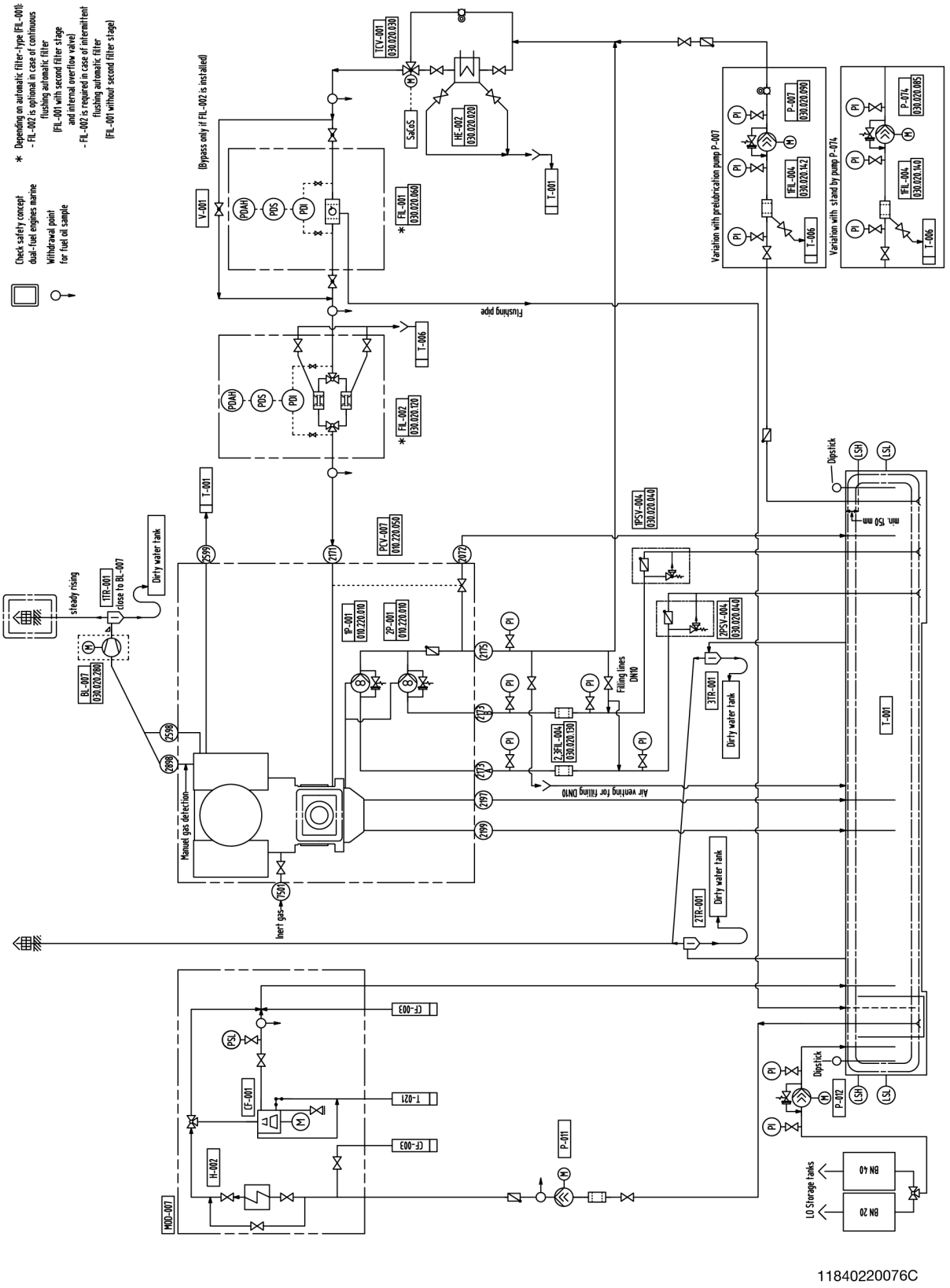

Figure 120: Lube oil system diagram – V engine

Components			
BL-007	Fan, crankcase venting	P-011	Lube oil feed pump separator
CF-001	Lube oil separator	P-012	Lube oil transfer pump
CF-003	Diesel fuel oil separator	P-074	Lube oil stand-by pump, free-standing
FIL-001	Lube oil automatic filter	PCV-007	Lube oil pressure relief valve
FIL-002	Lube oil duplex filter	1,2 PSV-004	Lube oil non-return flap with integrated safety valve
1,2,3 FIL-004	Lube oil suction strainer	T-001	Lube oil service tank
H-002	Lube oil preheating unit	T-006	Leakage oil collecting tank
HE-002	Lube oil cooler	T-021	Sludge tank
MOD-007	Lube oil separator module	TCV-001	Lube oil temperature control valve
1,2 P-001	Lube oil service pump, attached	1,2,3 TR-001	Condensate trap, lube oil system
P-007	Prelubrication pump	V-001	Lead sealed globe valve, bypass to lube oil main filter
Connections numbers			
2072	Lube oil return from pressure control valve	2199	Lube oil drain from oil pan, coupling side 1
2171	Lube oil inlet to engine	2598	Venting of turbocharger 1
2173 A	Lube oil inlet to lube oil pump 1	2599	Lube oil drain from turbocharger 1
2173 B	Lube oil inlet to lube oil pump 1	2898	Venting of crankcase 1
2175	Lube oil outlet from lube oil pump 1	7501	Inert gas inlet to crankcase 1
2197	Lube oil drain from oil pan, counter coupling side 1		

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5.4.3 External lube oil system – Prelubrication/postlubrication

Prelubrication

The prelubrication pump must be switched on at least 5 minutes before engine start. The prelubrication pump serves to assist the engine attached main lube oil pump, until this can provide a sufficient flow rate.

For design data of the prelubrication pump see section Planning data, Page 100 and paragraph Lube oil, Page 149.

During the starting process, the maximal temperature mentioned in section Starting conditions, Page 52 must not be exceeded at engine inlet. Therefore, a small LT cooling waterpump can be necessary if the lube oil cooler is served only by an attached LT pump.

Postlubrication

The prelubrication pump is also to be used for postlubrication after the engine is turned off.

Postlubrication is effected for a period of 15 minutes.

5.4.4 External lube oil system – Connection of lube oil outlets

Lube oil drain

Two connections for oil drain pipes are located on both ends of the engine oil sump, except for L engine with flexible engine mounting – with one drain arranged in the middle of each side.

For an engine installed in the horizontal position, two oil drain pipes are required, one at the coupling end and one at the free end.

If the engine is installed in an inclined position, three oil drain pipes are required, two at the lower end and one at the higher end of the engine oil sump.

The drain pipes must be kept short. The slanted pipe ends must be immersed in the oil, so as to create a liquid seal between crankcase and tank.

Expansion joints

At the connection of the oil drain pipes to the lube oil service tank, expansion joints are required.

Shut-off butterfly valves

If for lack of space, no cofferdam can be provided underneath the lube oil service tank, it is necessary to install shut-off butterfly valves in the drain pipes. If the ship should touch ground, these butterfly valves can be shut via linkages to prevent the ingress of seawater through the engine.

Drain pipes, shut-off butterfly valves with linkages, expansion joints, etc. are not supplied by the engine builder.

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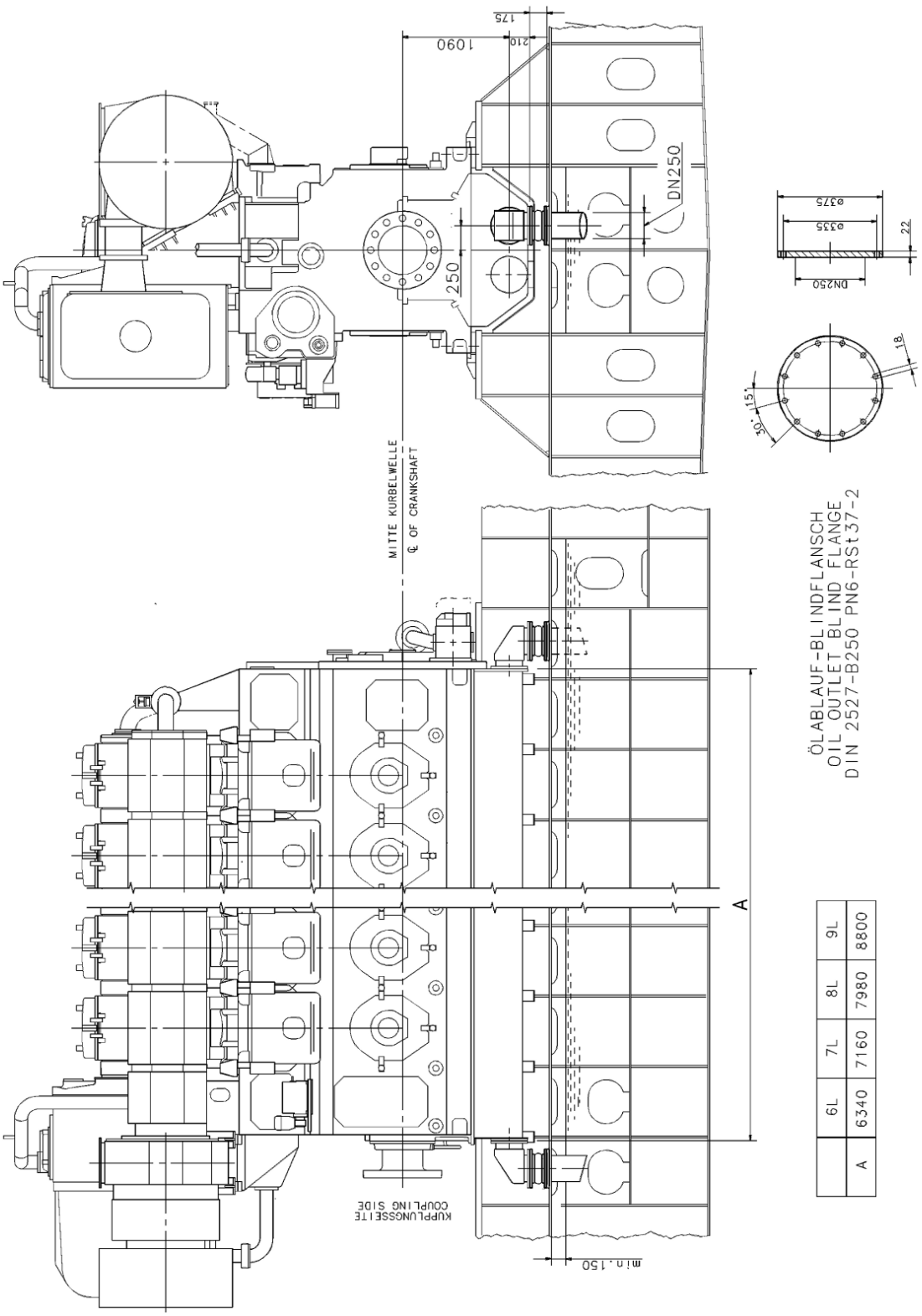
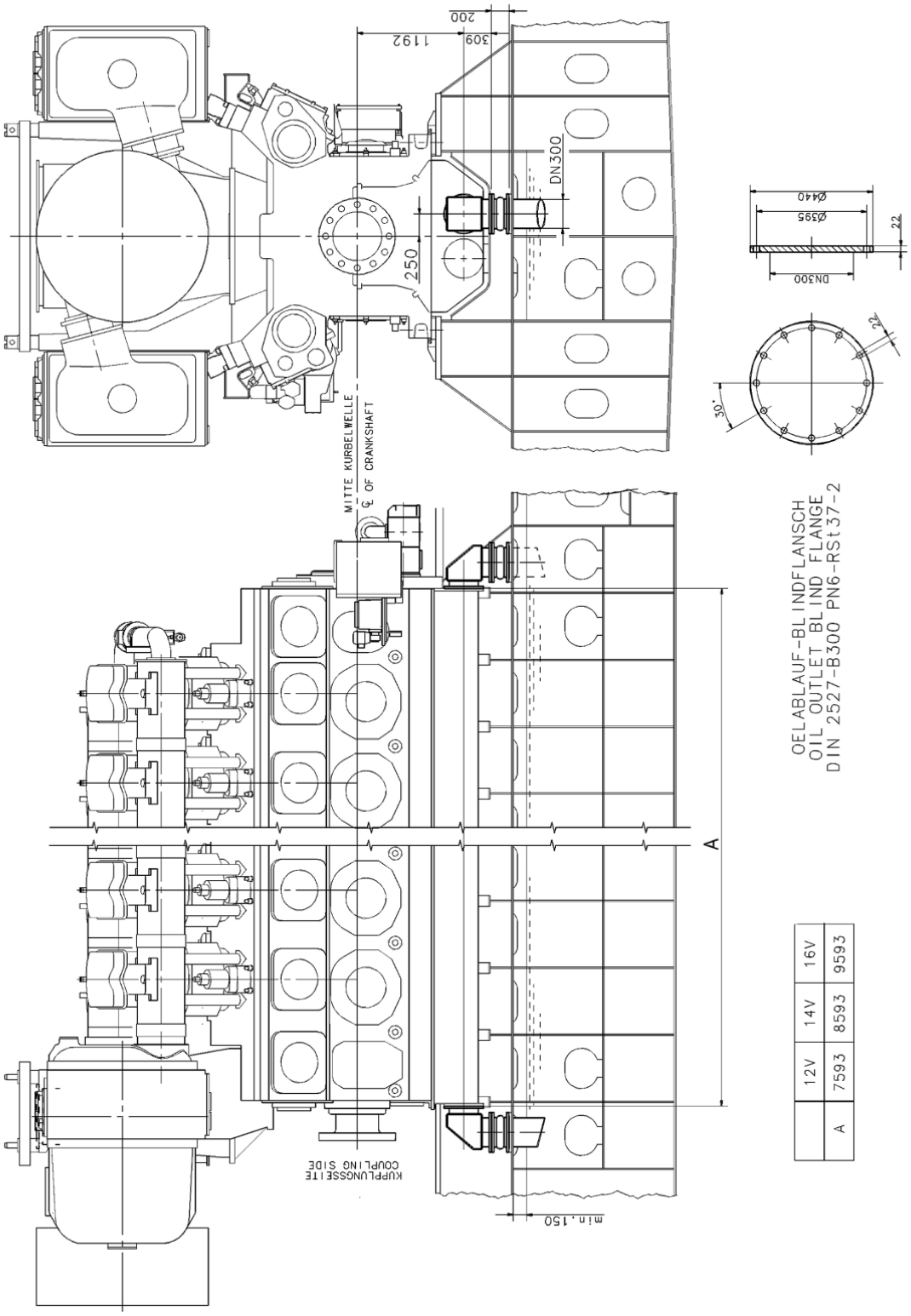
Lube oil outlets – Drawings



	6L	7L	8L	9L
A	6340	7160	7980	8800

Figure 121: Example: Lube oil outlets L engine



	12V	14V	16V
A	7593	8593	9593

Figure 122: Example: Lube oil outlets V engine

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5.4.5 External lube oil system – Lube oil service tank

The lube oil service tank is to be arranged over the entire area below the engine. To ensure uniform vertical thermal expansion of the whole engine foundation.

To provide for adequate degassing, a minimum distance of 150 mm is required between tank top and the highest operating level (high-level alarm). The low oil level (low-level alarm) shall still permit the lube oil to be drawn in free of air, if the ship is moving maximum according to the classification society rules (pitching and rolling).

Depending on the design of the lube oil service tank, a well or a central s is the preferred solution.

The minimum quantity for a proper engine operation is 1.0 l/kW (quantity of lube oil circulation). This is a theoretical factor for permanent lube oil quality control. It is the decisive factor for the design of the by-pass cleaning. The lube oil quantity, which is required during operation, depends on the tank geometry and the volume of the system (piping, system components), and may exceed the theoretical quantity of 1.0 l/kW to be topped up. The low-level alarm in the service tank is to be adjusted to a height, which ensures that the pumps can draw in oil, free of air, at the inclinations according to the classification society but min. 0.7 l/kW.

To ensure an adequate degassing of air out of the lube oil, the arrangement of the suction pipes to the engine and the drain pipes extending from the engine oil sump are to be selected in a way that the oil will remain for the longest possible time in the lube oil service tank. The maximum flow rates must be considered.

The manholes in the floor plates inside the service tank are to be arranged so as to ensure sufficient flow to the suction pipe of the pump also at low lube oil service level.

Vent the tank at both ends, according to section Crankcase vent and lube oil tank vent, Page 333.

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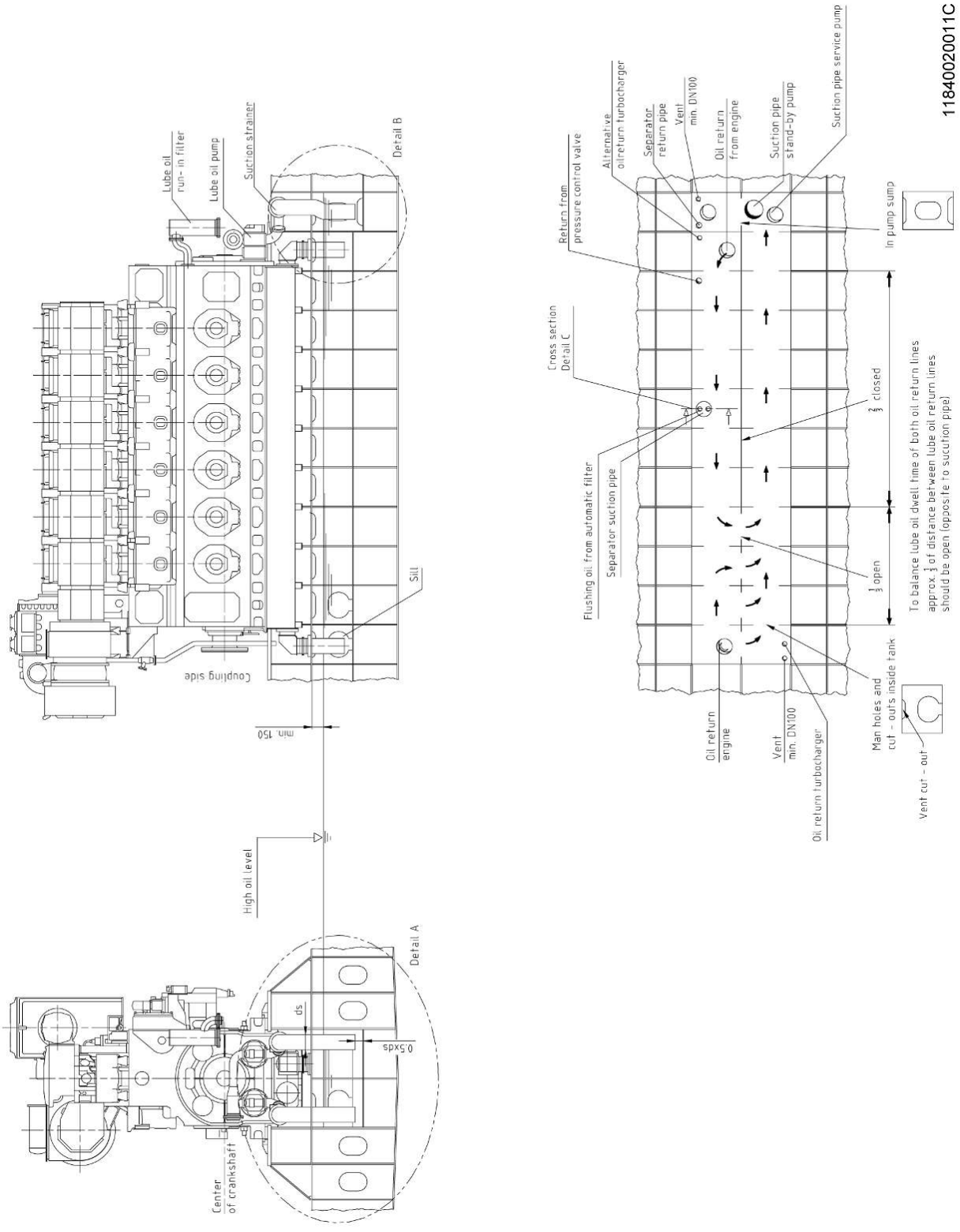

Figure 123: Example: Lube oil service tank

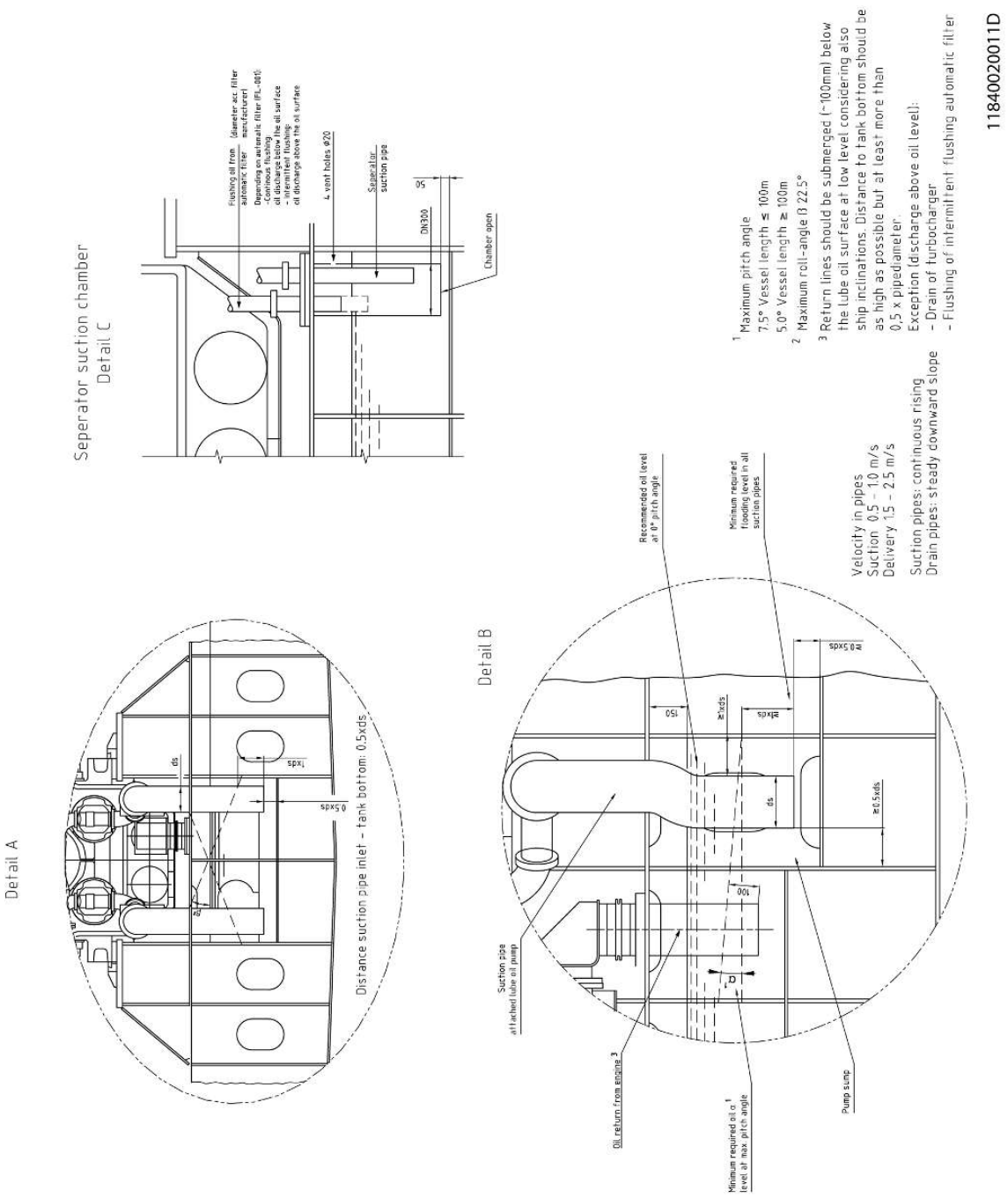

Figure 124: Example: Details lube oil service tank

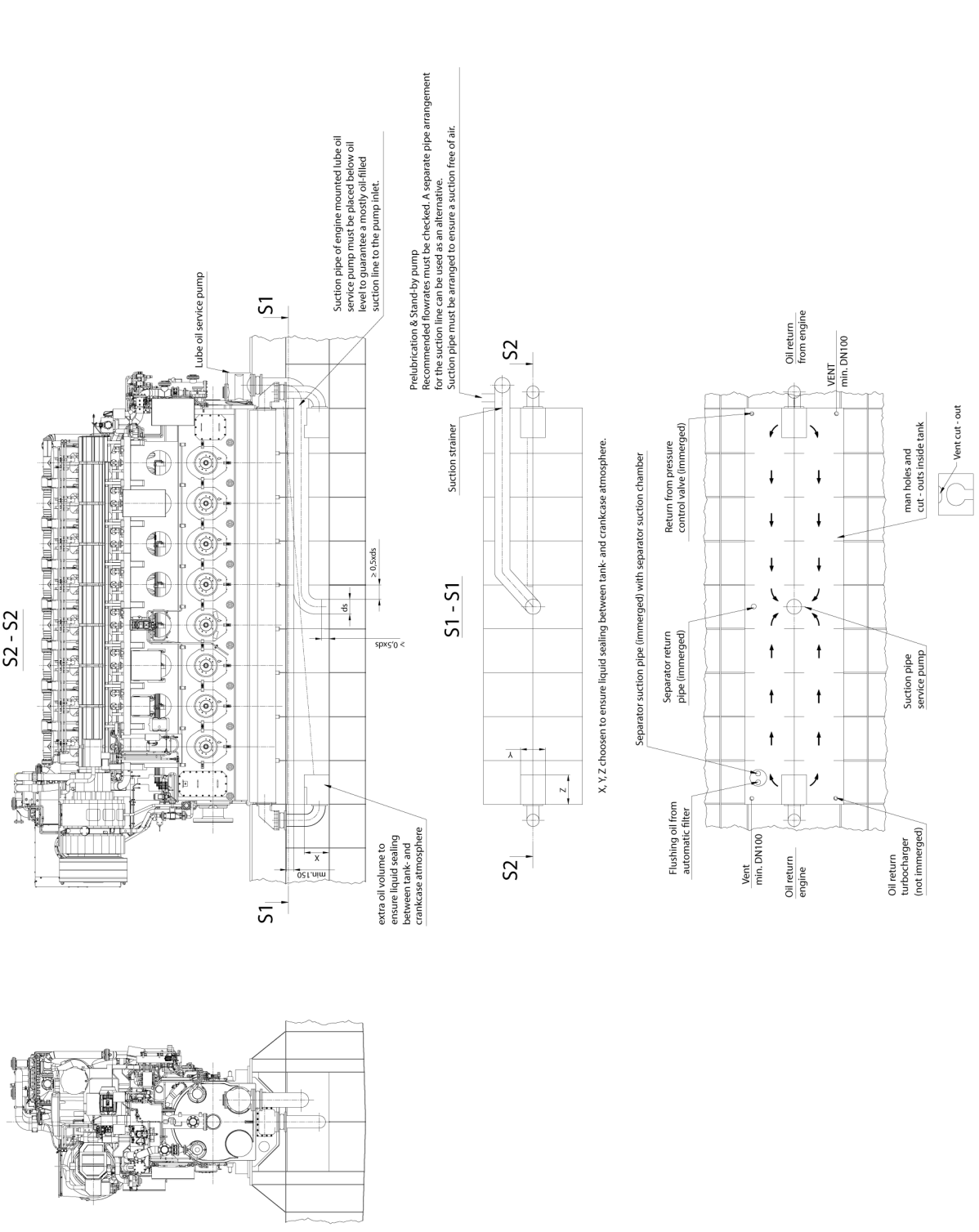

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Figure 125: Example: Lube oil service tank with central suction

5.5 Crankcase vent and lube oil tank vent

Vent pipes

The vent pipes from engine crankcase, turbocharger and lube oil service tank are to be arranged according to the sketch. The engine is equipped with a ventilation opening for crankcase and turbocharger which shall be equipped with a ventilation pipe built steadily ascending to outside. The required nominal diameters ND are stated in the chart following the diagram.

To dilute the crankcase atmosphere to a safe level it is necessary to produce a small quantity of additional airflow to the crankcase. This will be achieved by producing a vacuum in the crankcase using a speed controlled venting fan placed within the engine ventilation pipe and regulated via a pressure transmitter placed on the crankcase. Regarding the venting fan see also paragraph BL-007/Fan, crankcase venting, Page 318. Depending of the relevant environmental legislation a filter has to be installed in this pipe to prevent oil mist emissions to the atmosphere. In this case an additional by-pass has to be installed to prevent an overpressure in the crankcase.

The crankcase ventilation pipe shall lead to a safe location outside the engine room, remote from any source of ignition. The end of the vent pipe has to be equipped with a flame arrester. The crankcase ventilation pipe may not be connected with any other ventilation pipes.

Note:

- In case of multi-engine plants the venting pipework has to be kept separately.
- All venting openings as well as open pipe ends are to be equipped with flame breakers and shall lead to a safe location outside the engine room remote from any source of ignition.
- Condensate trap overflows are to be connected via siphone to drain pipe.
- Specific requirements of the classification societies are to be strictly observed.
- The maximum pressure loss of the arrangement (piping, condensate trap and flame arrester) after the crankcase venting fan depends on the fan type and must not be exceeded to ensure proper and safe engine operation.

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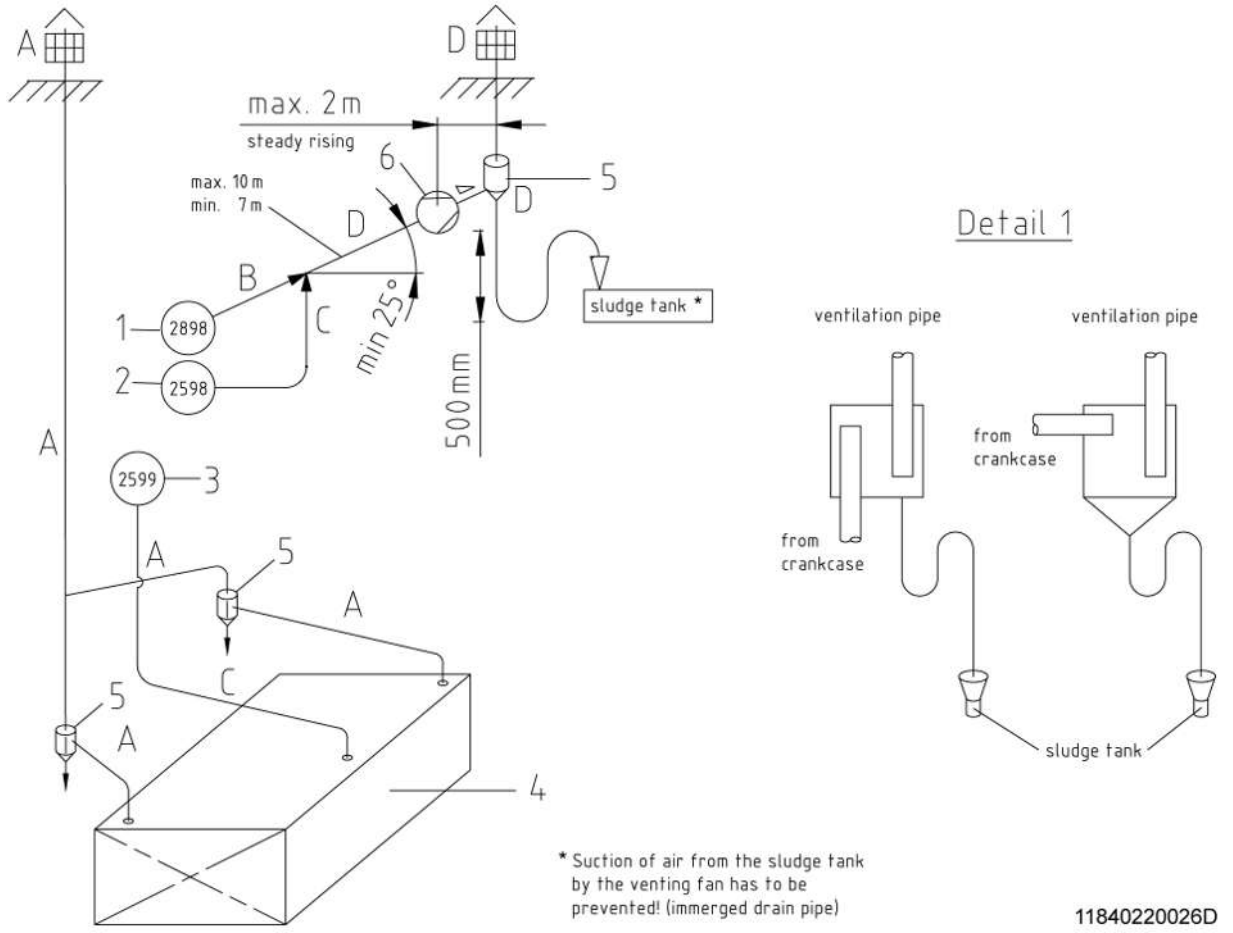

Figure 126: Crankcase vent and tank vent

1	Connection crankcase vent	4	Lubricating oil service tank
2	Connection turbocharger vent	5	Condensate trap, continuously open
3	Connection turbocharger drain	6	Venting fan

Engine	Nominal diameter ND (mm)			
	A	B	C	D
6L, 7L	100	100	65	125
8L, 9L	100	100	80	125
12V, 14V	100	125	100	150
16V	100	125	125	200

Table 196: Nominal Diameter ND (mm)

5.6 Cooling water system

5.6.1 Internal cooling water system

As a standard:

- Engine equipment with attached HT cooling water pump

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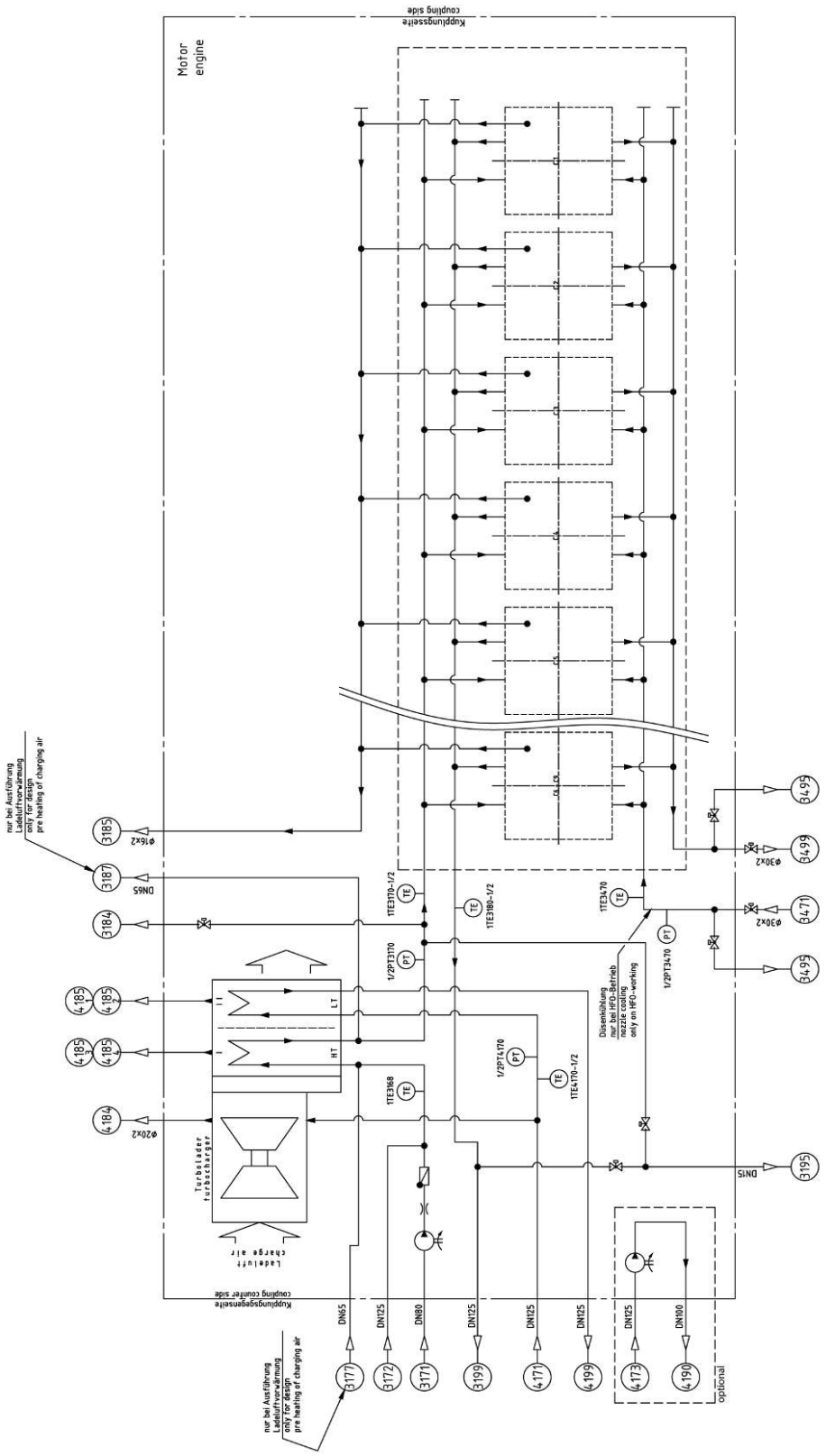

Figure 127: Internal cooling water system MAN 51/60DF – Exemplary

Connection numbers			
3171	HT cooling water inlet to engine	3495	Drain of nozzle cooling water pipe
3172	HT cooling water inlet to engine (reserve connection)	3499	Nozzle cooling water outlet from engine
3177	Charge air preheating water inlet	4171	LT cooling water inlet to engine
3184	Venting of HT cooling water pipe 1	4173	LT cooling water inlet to cooling water pump
3185	Venting of HT cooling water pipe 2	4184	LT cooling water outlet from compressor casing
3187	Charge air preheating water outlet	4185	Venting of LT low pressure charge air cooler
3195	Drain of HT cooling water pipe	4190	LT cooling water outlet from cooling water pump
3199	HT cooling water outlet from engine	4199	LT cooling water outlet from engine
3471	Nozzle cooling water inlet to engine		

5.6.2 External cooling water system – Description

The diagrams showing cooling water systems for main engines comprising the possibility of heat utilisation in a fresh water generator and equipment for preheating of the charge air in a two-stage charge air cooler during part load operation.

Note:
The arrangement of the cooling water system shown here is only one of many possible solutions. It is recommended to inform MAN Energy Solutions in advance in case other arrangements should be desired. In any case two seawater coolers have to be installed to ensure continuous operation while one cooler is shut off (e.g. for cleaning).

For special applications, e.g. GenSets or dual fuel engines, supplements will explain specific necessities and deviations.

For the design data of the system components shown in the diagram see section Planning data, Page 100 and following sections.

There are many possible concepts for the arrangement of cooling water systems and the best solution always depends on the specific application of the engine and ship. The table below shows the main characteristics and schemes of two main concepts. For further information contact MAN Energy Solutions.

Concept	Mixing CW system	Conventional CW system
Arrangement	Physically and thermodynamically connected LT and HT circuit	Physically and thermodynamically separated LT and HT circuit
Features	One central seawater cooler sufficient Reduced piping	No transfer of pressure peaks and pollutants between LT and HT circuits Central cooler possible with HT-cooler recooled by LT-cooling water

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Concept	Mixing CW system	Conventional CW system
Application	Economic solution for vessels with limited space Not recommended for multi-engine plants with attached cooling water pumps or variable engine speed	Vessels which require convenient commissioning and maintenance as well as robust temperature control

Table 197: Characteristics of cooling water concepts

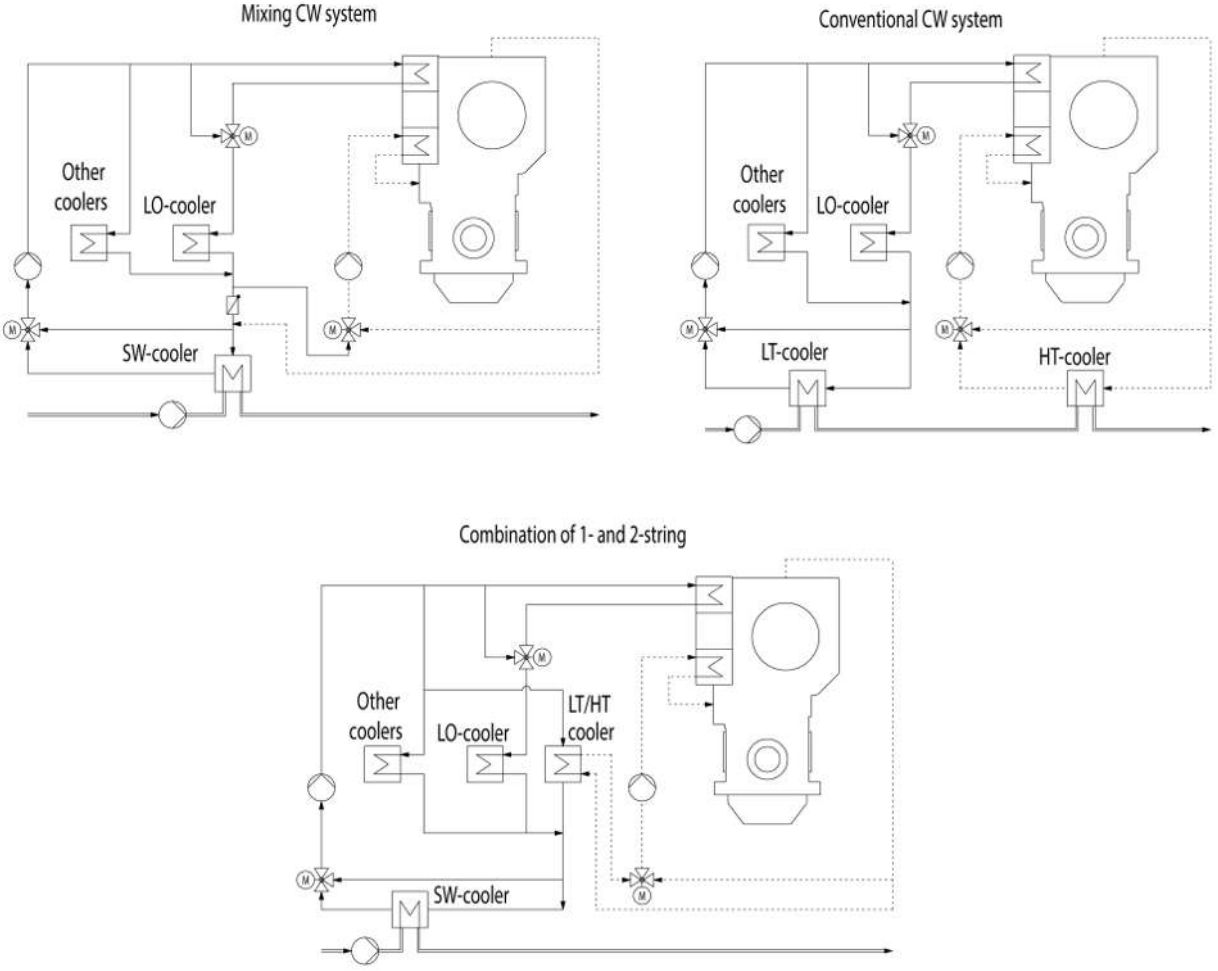

Figure 128: Basic schemes of cooling water concepts

Dual fuel engines may be operated on gas. In case gaskets at the cylinder head are damaged, gas may be blown into the HT cooling water circuit. The gas may accumulate in some areas (e.g. expansion tank) and cause gas dangerous zones. Observe the information given in the "Safety Concept – Marine dual fuel engines" and the relevant P&ID. Check the system with classification surveyor and other authorities (if required). In case the HT cooling water is mixed with LT cooling water, the LT circuit has to be checked with regard to possible accumulation of gas too.

In case the pilot oil system is out of operation, the engine will be running in so called backup mode (see section Engine and operation, Page 23). Depending on the fuel quality used on board (see section Specification for engine supplies, Page 255), we recommend using charge air preheating by HT/LT-switchover. For details please contact MAN.

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The cooling water is to be conditioned by using a corrosion inhibitor, see section Specification of engine coolant, Page 283.

Cooling water that is not treated properly may cause serious corrosion during storage or operation of the engine.

LT = Low temperature

HT = High temperature

Cooler dimensioning, general

For coolers operated by seawater (not treated water), lube oil or MDO/MGO on the primary side and treated freshwater on the secondary side, an additional safety margin of 10 % related to the heat transfer coefficient is to be considered. If treated water is applied on both sides, MAN Energy Solutions does not insist on this margin.

In case antifreeze is added to the cooling water, the corresponding lower heat transfer is to be taken into consideration.

The cooler piping arrangement should include venting and draining facilities for the cooler. In case coolers for lube oil, fuel oil or other environmental hazardous fluids are operated by seawater, we strongly recommend to use double wall plate type coolers. These coolers allow to detect leakage and prevent the seawater from pollution by hazardous fluids.

Open/closed system

Open system

Characterised by "atmospheric pressure" in the expansion tank. Pre-pressure in the system, at the suction side of the cooling water pump is given by the geodetic height of the expansion tank (standard value 6 – 9 m above crankshaft of engine).

Closed system

In a closed system, the expansion tank is pressurised and has no venting connection to open atmosphere. This system is recommended in case the engine will be operated at cooling water temperatures above 100 °C or an open expansion tank may not be placed at the required geodetic height. Use air separators to ensure proper venting of the system.

Venting

Note:
Insufficient venting of the cooling water system prevents air from escaping which can lead to thermal overloading of the engine.

The cooling water system needs to be vented at the highest point in the cooling system. Additional points with venting lines to be installed in the cooling system according to layout and necessity. In case engines may be operated on gas, all venting pipes have to be routed to open atmosphere.

If LT and HT string are separated, make sure that the venting lines are always routed only to the associated expansion tank. The venting pipe must be connected to the expansion tank below the minimum water level, this prevents oxydation of the cooling water caused by "splashing" from the venting pipe. We recommend to connect the venting pipes at the tank bottom. The expansion tank should be equipped with venting pipe and flange for filling of water and inhibitors.

Additional notes regarding venting pipe routing:

- The ventilation pipe should be continuously inclined (min. 5 degrees).
- At the interface to the system (engine connection terminal), the line must be continued directly with an elastic pipe connection (e.g. hose line).

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- No additional masses such as shut-off valves, reducers, extensions, etc. may be attached to the engine-side connection terminal, since they represent an additional oscillating mass that can cause the line to break during operation.
- Venting pipes from several engine circuits may only be connected together if there are no longer any pressure differences between the individual pipes.
- It is to be avoided to merge venting pipes from low-temperature (LT) and high-temperature (HT) circuits, as a separate fault detection (e.g. lubricating oil or fuel in the cooling water) is no longer possible.
- No restrictions, no kinks in the ventilation pipes.
- Merging of ventilation pipes only permitted with appropriate cross-sectional enlargement.

Draining

At the lowest point of the cooling system, a drain has to be provided. Additional points for draining to be provided in the cooling system according to layout and necessity, e.g. for components in the system that will be removed for maintenance.

We recommend using lockable valves or locking caps for sample and draining points to avoid opening by mistake.

LT cooling water system

In general the LT cooling water passes through the following components:

- Stage 2 of the two-stage charge air cooler (HE-008)
- Lube oil cooler, free-standing (HE-002)
- Nozzle cooling water cooler (HE-005)
- Fuel oil cooler (HE-007)
- Gearbox lube oil cooler (HE-023) (or e.g. alternator cooling in case of an electric propulsion plant)
- Cooler for LT cooling water (HE-024)
- Fuel oil cooler, supply circuit (HE-025) (if applicable, see section External fuel system – Heavy fuel oil (HFO) supply system, Page 384)
- Other components such as e.g. auxiliary engines (GenSets)

LT cooling water pumps can be either of engine driven or electrically driven type.

In case an engine driven LT pump is used and no electric driven pump (LT main pump) is installed in the LT circuit, an LT circulation pump has to be installed. We recommend an electric driven pump with a capacity of approximately 8 m^3/h at 1.5 bar pressure head. The pump has to be operated simultaneously to the prelubrication pump. Make sure to keep the pump running for at least 45 minutes after stop of the engine to cool down all connected system components. In case a 100 % lube oil stand-by pump is installed, the circulation pump has to be increased to the size of a 100 % LT stand-by pump to ensure cooling down the lube oil in the cooler during prelubrication before engine start. The system has to be designed, so that the temperatures for lube oil and cooling water, which are given in section Operating/service temperatures and pressures, Page 147, are adhered during operation and stand-by of the engine. For shutdown of the engine the information in section Engine load reduction, Page 68 must be observed. In case no electric stand-by pump will be installed, the engine and lube oil tank have to be cooled down by operating the engine at low load (< 15 % MCR) for at least 4 minutes before shut down. The shipyard has to make sure, that lube oil separators will

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not cause overheating of the oil during standstill of the engine. For max. permissible temperatures see section Starting conditions, Page 52. For details contact MAN Energy Solutions.

The system components of the LT cooling water circuit are designed for a max. LT cooling water temperature of 38 °C with a corresponding seawater temperature of 32 °C (tropical conditions).

However, the capacity of the cooler for LT cooling water (HE-024) is determined by the temperature difference between seawater and LT cooling water. Due to this correlation an LT freshwater temperature of 32 °C can be ensured at a seawater temperature of 25 °C.

To meet the IMO Tier I/IMO Tier II regulations the set point of the LT cooling water temperature control valve (MOV-016) is to be adjusted to 32 °C. However this temperature will fluctuate and reach at most 38 °C with a seawater temperature of 32 °C (tropical conditions). In case other temperatures are required in the LT system, the engine setting has to be adapted accordingly. For details contact MAN Energy Solutions.

The charge air cooler stage 2 (HE-008) and the lube oil cooler (HE-002) are installed in series to obtain a low delivery rate of the LT cooling water pump (P-004 or P-076).

High performing turbochargers lead to a high temperature at the compressor wheel. To limit these temperatures, the compressor wheel casing (HE-034) is cooled by a low LT water flow. The outlet (4184) is to be connected separately to the LT expansion tank in a steady rise.

The pipe diameter must not be reduced and no orifices are allowed in the pipe. In case the pipe is merged with other pipes, the pipe diameter has to be enlarged. Please contact MAN to clarify whether pipes may be merged. Flow rates for cooling water see section Planning data, Page 100.

P-004 or P-076/LT cooling water pump

The delivery rates of the service and stand-by pump are mainly determined by the cooling water required for the charge air cooler stage 2 and the other coolers.

For operating auxiliary engines (GenSets) in port, the installation of an additional smaller pump is recommendable.

MOV-003/Charge air temperature control valve (CHATCO)

This three way valve is to be installed as a mixing valve.

It serves two purposes:

1. In engine part load operation, depending on the ambient temperature, the charge air cooler stage 2 (HE-008) is partially or completely bypassed, so that a higher charge air temperature is maintained.
2. The valve reduces the accumulation of condensed water during engine operation under tropical conditions by regulation of the charge air temperature. Below a certain intake air temperature the charge air temperature is kept constant. When the intake temperature rises, the charge air temperature will be increased accordingly.

The three-way valve is to be designed for a pressure loss of 0.3 – 0.6 bar and is to be equipped with an actuator with high positioning speed. For adjustment of the valve, follow instructions given in MAN Energy Solutions planning documentation. The actuator must permit manual emergency adjustment.

HE-002/Lube oil cooler, free-standing

For the description see section External lube oil system, Page 313. For heat data, flow rates and tolerances see section Planning data, Page 100 and the following. For the description of the principal design criteria see paragraph Cooler dimensioning, general, Page 339.

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HE-024/Cooler for LT cooling water

For heat data, flow rates and tolerances of the heat sources see section Planning data, Page 100 and the following. For the description of the principal design criteria for coolers see paragraph Cooler dimensioning, general, Page 339.

MOV-016/LT cooling water temperature control valve

This is a motor-actuated three-way regulating valve with a linear characteristic. It is to be installed as a mixing valve. It maintains the LT cooling water at set point temperature (32 °C standard).

The three-way valve is to be designed for a pressure loss of 0.3 – 0.6 bar. It is to be equipped with an actuator with low positioning speed. For adjustment of the valve follow instructions given in MAN Energy Solutions planning documentation. The actuator must permit manual emergency adjustment.

The actual LT flow temperature is measured by a temperature sensor, directly downstream of the three-way mixing valve in the supply pipe to charge air cooler stage 1.

This sensor has to be installed by the shipyard. To ensure instantaneous measurement of the mixing temperature of the three-way mixing valve, the distance to the valve should be 10 to 15 times the pipe diameter.

For single engine plants, the control function may be taken over by the SaCoS control unit. For multi engine plants, MAN Energy Solutions can supply a suitable external controller.

Note:
For engine operation with reduced NO_x emission, according to IMO Tier I/IMO Tier II requirement, at 100 % engine load and a seawater temperature of 25 °C (IMO Tier I/IMO Tier II reference temperature), an LT cooling water temperature of 32 °C before charge air cooler stage 2 (HE-008) is to be maintained. For other temperatures, the engine setting has to be adapted. For further details contact MAN Energy Solutions.

FIL-021/Strainer for cooling water

In order to protect the engine and system components, several strainers are to be provided at the places marked in the diagram. We recommend to install Y-type strainers with magnetic inserts and a mesh size of 1 – 2 mm depending on the pipe diameter.

The strainers have to be installed in horizontal pipes or pipes with flow direction downwards.

HE-005/Nozzle cooling water cooler

The nozzle cooling water system is a separate and closed cooling circuit. It is cooled down by LT cooling water via the nozzle cooling water cooler (HE-005).

Heat data, flow rates and tolerances are indicated in section Planning data, Page 100 and the following. The principal design criteria for coolers has been described before in paragraph Cooler dimensioning, general, Page 339. For plants with two main engines only one nozzle cooling water cooler (HE-005) is required. As an option a compact nozzle cooling water module (MOD-005) can be delivered, see section External cooling water system – Nozzle cooling water module, Page 356.

HE-007/Fuel oil cooler

This cooler is required to dissipate the heat of the fuel injection pumps during MDO/MGO operation. For the description of the principal design criteria for coolers see paragraph Cooler dimensioning, general, Page 339. For plants with more than one engine, connected to the same fuel oil system, only one MDO/MGO cooler is required. In case the fuel oil cooler is recooled by seawater, we recommend to use a double-wall type cooler.

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In case fuels with very low viscosity are used (e.g. arctic diesel or military fuels), a chiller system may be necessary to meet the minimum required fuel viscosity (see section Fuel system, Page 364). Contact MAN Energy Solutions in that case.

HE-025/Fuel oil cooler, supply circuit

See section External fuel system – Heavy fuel oil (HFO) supply system, Page 384.

T-075/LT cooling water expansion tank

The effective tank capacity should be high enough to keep approximately 2/3 of the tank content of HT cooling water expansion tank T-002. In case of twin-engine plants with a common cooling water system, the tank capacity should be by approximately 50 % higher. The tanks T-075 and T-002 should be arranged side by side to facilitate installation.

In case HT/LT switchover (charge air preheating) is applied, the tank has to be placed at the same height as the HT expansion tank and a compensating line has to be installed.

In any case the tank bottom must be installed above the highest point of the LT system at any ship inclination.

For the recommended installation height and the diameter of the connecting pipe, see table Service tanks capacities, Page 154.

HT cooling water circuit

General

The HT cooling water system consists of the following coolers and heat exchangers:

- Charge air cooler stage 1 (HE-010)
- Cylinder cooling
- Cooler for HT cooling water (HE-003)
- Heat utilisation, e.g. fresh water generator (HE-026)
- HT cooling water preheating module (MOD-004)

The HT cooling water pumps can be either of engine-driven or electrically-driven type. The outlet temperature of the cylinder cooling water at the engine is to be adjusted to 90 °C.

For HT cooling water systems, where more than one main engine is integrated, each engine should be provided with an individual engine driven HT cooling water pump. Alternatively common electrically-driven HT cooling water pumps may be used for all engines. However, an individual HT temperature control valve is required for each engine. The total cooler and pump capacities are to be adapted accordingly.

The shipyard is responsible for the correct cooling water distribution, ensuring that each engine will be supplied with cooling water at the flow rates required by the individual engines, under all operating conditions. To meet this requirement, orifices, flow regulation valves, by-pass systems etc. are to be installed where necessary. Check total pressure loss in HT circuit. The delivery height of the attached pump must not be exceeded.

HT cooling water preheating module (MOD-004)

Before starting a cold engine, it is necessary to preheat the water jacket up to min. 60 °C at engine outlet.

For the total heating power required for preheating the HT cooling water from 10 °C to 60 °C within 4 hours see table Heating power, Page 343.

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Engine type	L/V engine
Min. heating power (kW/cylinder)	14

Table 198: Heating power

These values include the radiation heat losses from the outer surface of the engine. Also a margin of 20 % for heat losses of the cooling system has been considered.

To prevent a too quick and uneven heating of the engine, the preheating temperature of the HT cooling water must remain mandatory below 90 °C at engine inlet and the circulation amount may not exceed 30 % of the nominal flow. The maximum heating power has to be calculated accordingly.

A secondary function of the preheater is to provide heat capacity in the HT cooling water system during engine part load operation. This is required for marine propulsion plants with a high freshwater requirement, e.g. on passenger vessels, where frequent load changes are common.

It is also required for arrangements with an additional charge air preheating by deviation of HT cooling water to the charge air cooler stage 2 (HE-008). In this case the heat output of the preheater is to be increased by approximately 50 %.

Avoid an installation of the preheater in parallel to the engine driven HT-pump. In this case, the preheater may not be operated while the engine is running. Preheaters operated on steam or thermal oil may cause alarms since a post-cooling of the heat exchanger is not possible after engine start (preheater pump is blocked by counterpressure of the engine driven pump).

An electrically driven pump becomes necessary to circulate the HT cooling water during preheating. For the required minimum flow rate see table below.

No. of cylinders, config.	Minimum flow rate required during preheating and post-cooling
	m^3/h
6L	14 – 21
7L	16 – 24
8L	18 – 27
9L	20 – 30
12V	28 – 42
14V	32 – 48
16V	36 – 54

Table 199: Minimum flow rate during preheating and post-cooling

The preheating of the main engine with cooling water from auxiliary engines is also possible, provided that the cooling water is treated in the same way. In that case, the expansion tanks of the two cooling systems have to be installed at the same level. Furthermore, it must be checked whether the available heat is sufficient to pre-heat the main engine. This depends on the number of auxiliary engines in operation and their load. It is recommended to install a separate preheater for the main engine, as the available heat from the auxiliary engines may be insufficient during operation in port.

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As an option MAN Energy Solutions can supply a compact HT cooling water preheating module (MOD-004). One module for each main engine is recommended. Depending on the plant layout, also two engines can be heated by one module.

Contact MAN Energy Solutions to check the hydraulic circuit and electric connections.

The preheater has to be designed to meet explosion protection requirements, in case gas may accumulate in some components of the module.

HE-003/Cooler for HT cooling water

For heat data, flow rates and tolerances of the heat sources see section Planning data, Page 100 and following sections. For the description of the principal design criteria for coolers see paragraph Cooler dimensioning, general, Page 339.

HE-026/Fresh water generator

The fresh water generator must be switched off automatically when the cooling water temperature at the engine outlet drops below 85 °C continuously.

A binary contact (SaCoS) for the heat consumer release can be used for activation of the fresh water generator. The heat consumer must then be switched off accordingly.

This will prevent operation of the engine at too low temperatures.

HT temperature control

The HT temperature control system consists of the following components:

- 1 electrically activated three-way mixing valve with linear characteristic curve (MOV-002).
- 1 temperature sensor TE, directly downstream of the three-way mixing valve in the supply pipe to charge air cooler stage 1 (for EDS visualisation and control of preheater valve).

  This sensor will be delivered by MAN Energy Solutions and has to be installed by the shipyard.
- 1 temperature sensor TE, directly downstream of the engine outlet.

  This sensor is already installed at the engine by MAN Energy Solutions.

The temperature controllers are available as software functions inside the Gateway Module of SaCoSone. The temperature controllers are operated by the displays at the operating panels as far as it is necessary. From the interface cabinet the relays actuate the control valves.

It serves to maintain the cylinder cooling water temperature constantly at 90 °C at the engine outlet – even in case of frequent load changes – and to protect the engine from excessive thermal load.

For adjusting the outlet water temperature (constantly to 90 °C) to engine load and speed, the cooling water inlet temperature is controlled. The electronic water temperature controller recognises deviations by means of the sensor at the engine outlet and afterwards corrects the reference value accordingly.

- The electronic temperature controller is installed in the switch cabinet of the engine room.

For a stable control mode, the following boundary conditions must be observed when designing the HT freshwater system:

- The temperature sensor is to be installed in the supply pipe to stage 1 of the charge air cooler. To ensure instantaneous measurement of the mixing temperature of the three-way mixing valve, the distance to the valve should be 10 to 15 times the pipe diameter.

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- The three-way valve (MOV-002) is to be installed as a mixing valve. It is to be designed for a pressure loss of 0.3 – 0.6 bar. It is to be equipped with an actuator of high positioning speed. For adjustment of the valve follow instructions given in MAN Energy Solutions planning documentation. The actuator must permit manual emergency adjustment.
- The pipes within the system are to be kept as short as possible in order to reduce the dead times of the system, especially the pipes between the three-way mixing valve and the inlet of the charge air cooler stage 1 which are critical for the control.

The same system is required for each engine, also for multi-engine installations with a common HT fresh water system.

In case of a deviating system layout, MAN Energy Solutions is to be consulted.

P-002/HT cooling water service pump, attached

The engine is normally equipped with a HT cooling water service pump, attached (default solution).

For technical data of the pumps see table HT cooling water – Engine, Page 148.

P-079/HT cooling water stand-by pump, free-standing

The HT cooling water stand-by pump (free-standing) has to be of the electrically driven type.

It is required to cool down the engine for a period of 15 minutes after shut-down. In case that neither an electrically driven HT cooling water pump nor an electrically driven stand-by pump is installed (e.g. multi-engine plants with engine driven HT cooling water pump without electrically driven HT stand-by pump, if applicable by the classification rules), it is possible to cool down the engine by a separate small preheating pump, see table Minimum flow rate during preheating and post-cooling, Page 344. If the optional HT cooling water preheating module (MOD-004) with integrated circulation pump is installed, it is also possible to cool down the engine with this small pump. However, the pump used to cool down the engine, has to be electrically driven and started automatically after engine shut-down.

None of the cooling water pumps is a self-priming centrifugal pump.

Design flow rates should not be exceeded by more than 15 % to avoid cavitation in the engine and its systems. A throttling orifice is fitted at the engine for adjusting the specified operating point.

T-002/HT cooling water expansion tank

The HT cooling water expansion tank compensates changes in system volume and losses due to leakages. It is to be arranged in such a way, that the tank bottom is situated above the highest point of the system at any ship inclination.

The expansion pipe shall connect the tank with the suction side of the pump(s), as close as possible. It is to be installed in a steady rise to the expansion tank, without any air pockets. The minimum required diameter for the pipe is given, see table Service tanks capacities, Page 154 depending on engine size. In case more than one engine is connected to the same tank, the pipe has to be extended accordingly.

For the required volume of the tank and the recommended installation height, see table Service tanks capacities, Page 154.

In case gaskets at the cylinder head are damaged, the cooling water may contain gas. This gas will enter the tank via the venting pipe. Therefore the tank has to be protected according IGF and other applicable standards (see "Safety Concept – Marine dual fuel engines").

Tank equipment:

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- Sight glass for level monitoring or other suitable device for continuous level monitoring
- Low-level alarm switch (electric parts to be placed outside of the tank or ATEX approved)
- Overflow and filling connection
- Inlet for corrosion inhibitor
- Venting to safe area with flame trap
- Inspection opening for manual gas detection device
- Connection for inert gas (flushing with nitrogen gas)

The tank has to be marked as a gas dangerous zone!

Cooling water system diagrams

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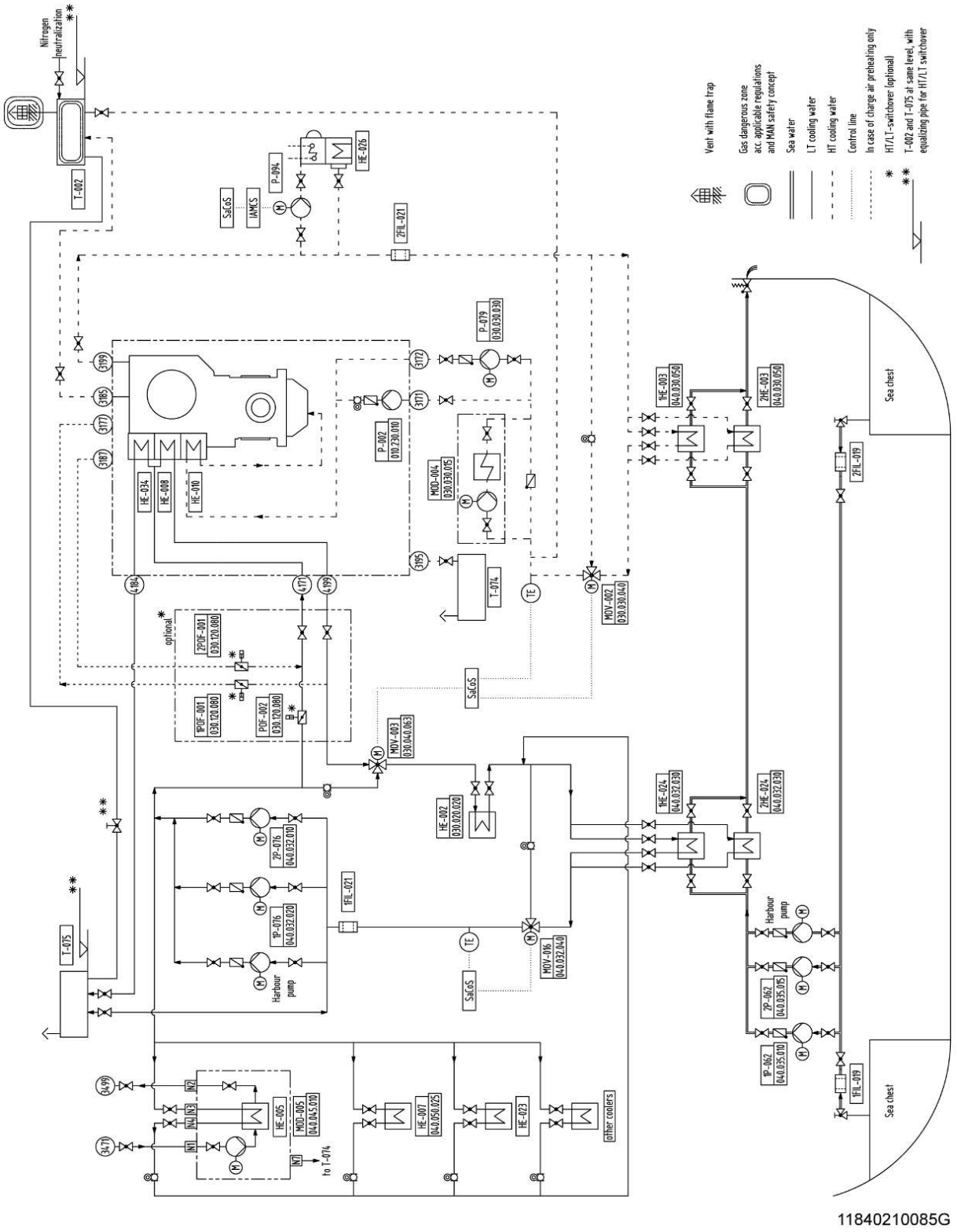

Figure 129: Cooling water system diagram – Single engine plant

Components			
1,2 FIL-019	Seawater filter	MOV-002	HT cooling water temperature control valve

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1,2 FIL-021	Strainer for fresh cooling water	MOV-003	Charge air temperature control valve (CHATCO)
HE-002	Lube oil cooler	MOV-016	LT cooling water temperature control valve
1,2 HE-003	Cooler for HT cooling water	P-002	HT cooling water service pump, attached
HE-005	Nozzle cooling water cooler	1,2 P-062	Seawater pump, free standing
HE-007	Fuel oil cooler	1,2 P-076	LT cooling water pump, free standing
HE-008	Charge air cooler (stage 2)	P-079	HT cooling water pump, free standing
HE-010	Charge air cooler (stage 1)	P-094	HT cooling water pump WHR
HE-023	Gearbox lube oil cooler	1,2 POF-001	Shut-off flap of connection line charge air cooler stage 1 to charge air cooler stage 2 (optional)
1,2 HE-024	Cooler for LT cooling water	1,2 POF-002	Shut-off flap inlet charge air cooler stage 2 (optional)
HE-026	Fresh water generator	T-002	HT cooling water expansion tank
HE-034	Compressor wheel casing (water cooled)	T-074	Fresh water collecting tank
MOD-004	HT cooling water preheating module	T-075	LT cooling water expansion tank
MOD-005	Nozzle cooling water module		
Major engine connections			
3171	HT cooling water inlet to engine	3471	Nozzle cooling water inlet to engine
3172	HT cooling water inlet to engine (reserve connection) 1	3499	Nozzle cooling water outlet from engine
3177	Cooling water inlet charge air preheating	4171	LT cooling water inlet to engine
3185	Venting of HT cooling water pipe 2	4184	LT cooling water outlet from compressor
3187	Cooling water outlet charge air preheating	4195	Drain of charge air cooler 1 (LT)
3195	Drain of charge air cooler 1 (HT)	4199	LT cooling water outlet from engine 1
3199	HT cooling water outlet from engine		
Connections to the nozzle cooling module			
N1	Nozzle cooling water inlet	N3	LT cooling water inlet
N2	Nozzle cooling water outlet	N4	LT cooling water outlet

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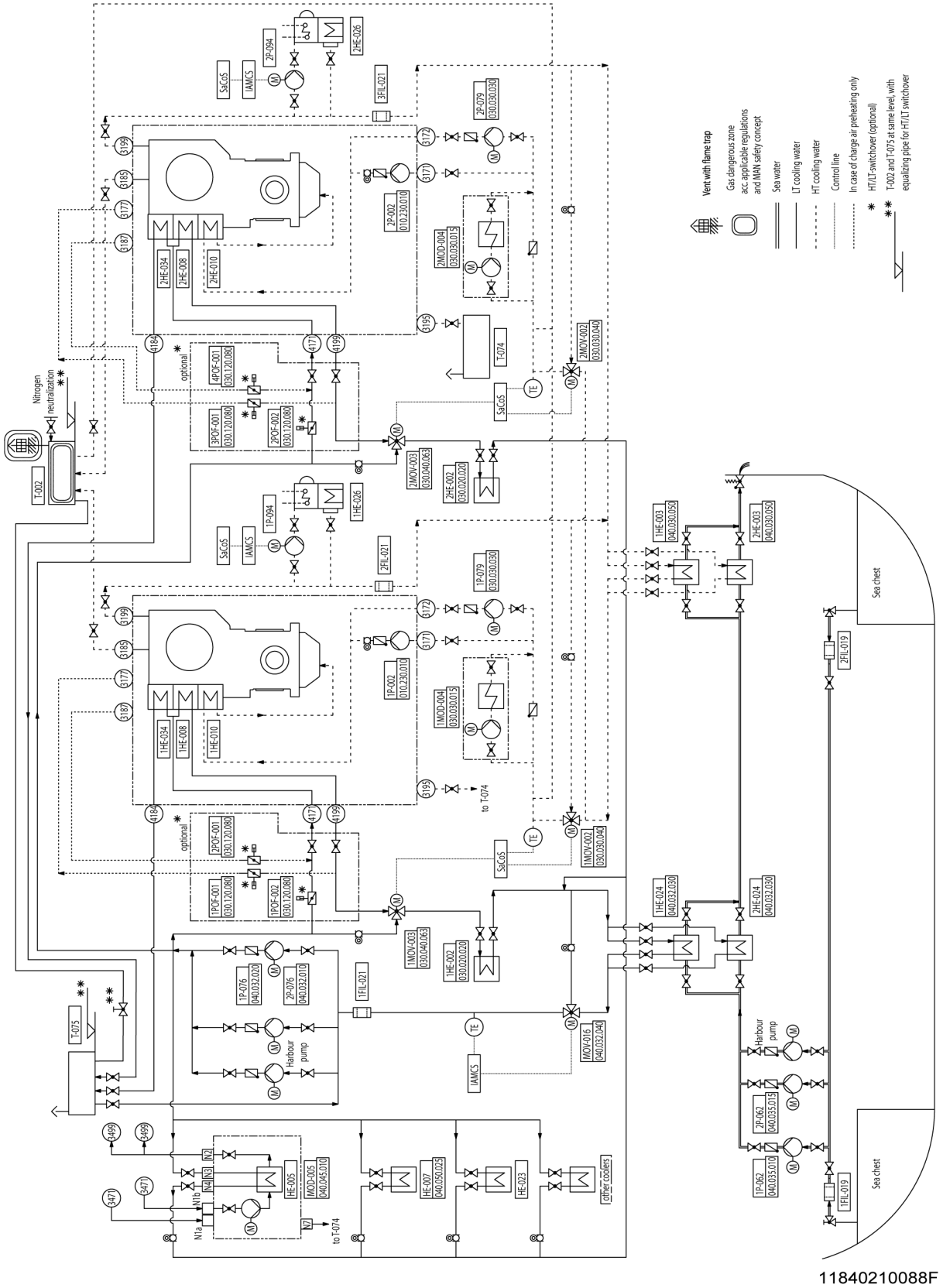

Figure 130: Cooling water system diagram – Twin engine plant

Components			
1,2 FIL-019	Seawater filter	1,2 MOV-002	HT cooling water temperature control valve
1,2,3 FIL-021	Strainer for fresh cooling water	1,2 MOV-003	Charge air temperature control valve (CHATCO)
1,2 HE-002	Lube oil cooler	MOV-016	LT cooling water temperature control valve
1,2 HE-003	Cooler for HT cooling water	1,2 P-002	HT cooling water service pump, attached
HE-005	Nozzle cooling water cooler	1,2 P-062	Seawater pump, free standing
HE-007	Fuel oil cooler	1,2 P-076	LT cooling water pump, free standing
1,2 HE-008	Charge air cooler (stage 2)	1,2 P-079	HT cooling water pump, free standing
1,2 HE-010	Charge air cooler (stage 1)	1,2 P-094	HT cooling water pump WHR
HE-023	Gearbox lube oil cooler	1,2,3,4 POF-001	Shut-off flap of connection line charge air cooler stage 1 to charge air cooler stage 2 (optional)
1,2 HE-024	Cooler for LT cooling water	1,2 POF-002	Shut-off flap inlet charge air cooler stage 2 (optional)
1,2 HE-026	Fresh water generator	T-002	HT cooling water expansion tank
1,2 HE-034	Compressor wheel casing (water cooled)	T-074	Fresh water collecting tank
1,2 MOD-004	HT cooling water preheating module	T-075	LT cooling water expansion tank
MOD-005	Nozzle cooling water module		
Major engine connections			
3171	HT cooling water inlet to engine	3199	HT cooling water outlet from engine
3172	HT cooling water inlet to engine (reserve connection) 1	3471	Nozzle cooling water inlet to engine
3177	Cooling water inlet charge air preheating	3499	Nozzle cooling water outlet from engine
3185	Venting of HT cooling water pipe 2	4171	LT cooling water inlet to engine
3187	Cooling water outlet charge air preheating	4184	LT cooling water outlet from compressor
3195	Drain of charge air cooler 1 (HT)	4199	LT cooling water outlet from engine 1
Connections to the nozzle cooling module			
N1 a,b	Nozzle cooling water inlet	N3	LT cooling water inlet
N2	Nozzle cooling water outlet	N4	LT cooling water outlet

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5.6.3 External cooling water system – Advanced HT cooling water system for increased freshwater generation

Traditional systems

The cooling water systems presented so far, demonstrate a simple and well proven way to cool down the engines internal heat load.

Traditionally, stage 1 charge air cooler and cylinder jackets are connected in sequence, so the HT cooling water circle can work with one pump for both purposes.

Cooling water temperature is limited to 90 °C at the outlet oft the cylinder jackets, the inlet temperature at the charge air cooler is about 55 to 60 °C.

Cooling water flow passing engine block and charge air cooler is the same, defined by the internal design of the cylinder jacket.

As one result of this traditional set-up, the possible heat recovery for fresh water generation is limited.

Advanced systems

To improve the benefit of the HT cooling water circle, this set-up can be changed to an advanced circuit, with two parallel HT pumps.

Cooling water flow through the cylinder jackets and outlet temperature at the engine block is limited as before, but the extra flow through the charge air cooler can be increased.

With two pumps in parallel, the combined cooling water flow can be more than doubled.

Common inlet temperature for both circles is e.g. about 78 °C, the mixed outlet temperature can reach up to 94 °C.

Following this design, the internal heat load of the engine stays the same, but water flow and temperature level of systems in- and outlet will be higher.

This improves considerably the use of heat recovery components at high temperature levels, like e.g. fresh water generators for cruise vessels or other passenger ships.

General requirements, LT system

General requirements for cooling water systems and components concerning the LT system stay the same like for the cooling water systems mentioned before.

Note:
The arrangement of the cooling water system shown here is only one of many possible solutions. It is recommended to inform MAN Energy Solutions in advance in case other arrangements should be desired.

HT cooling water circuit

Following the advanced design, components for the cylinder cooling will not differ from the traditional set-up.

Due to the higher temperature level, the water flow passing the stage 1 charge air cooler has to rise considerably and for some engine types a bigger HT charge air cooler as well as a more powerful HT charge air cooler pump may be necessary.

Note:
The design data of the cooling water system components shown in the following diagram are different from section Planning data, Page 100 and have to be clarified in advance with MAN Energy Solutions.

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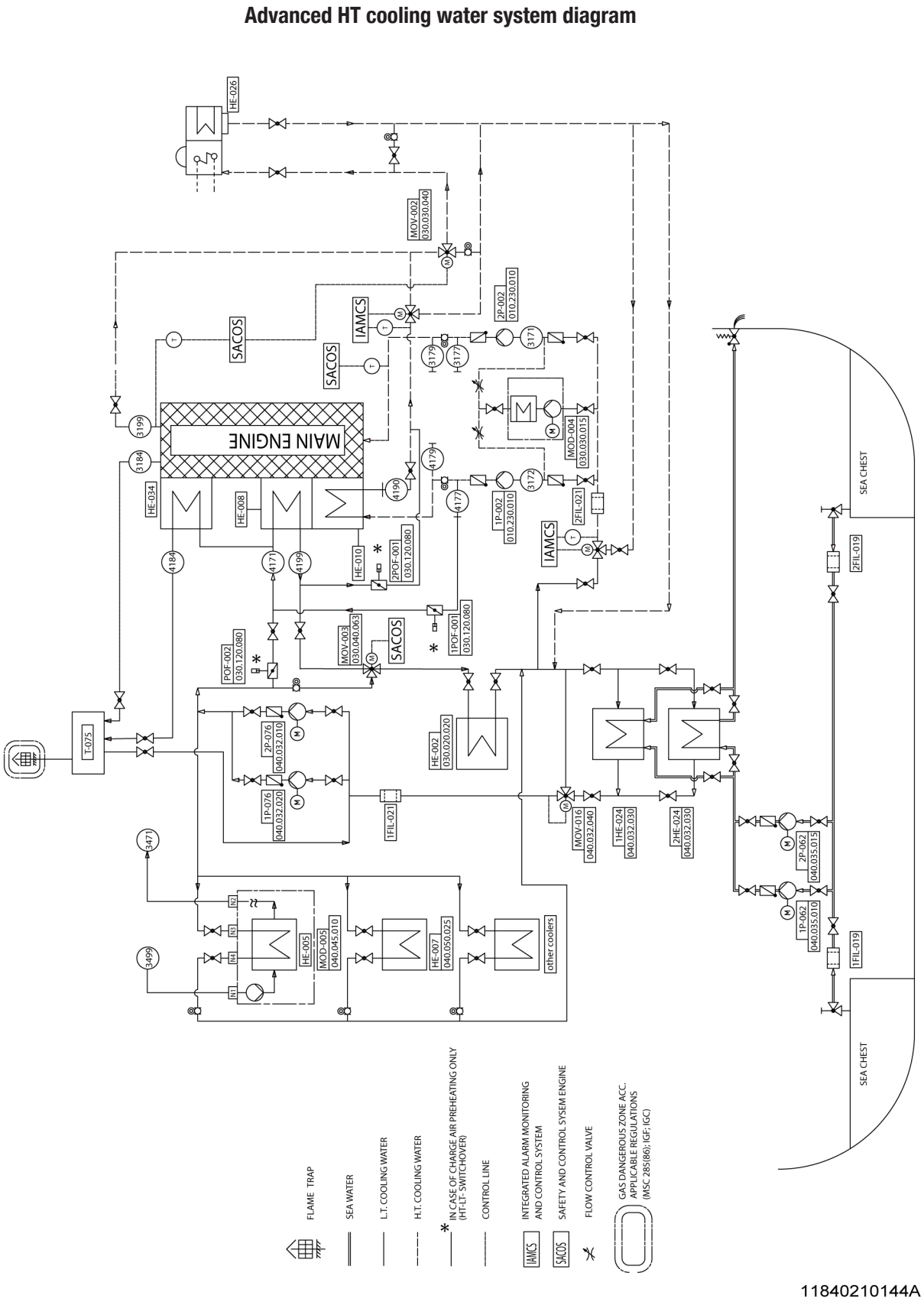

Figure 131: Advanced HT cooling water system diagram for increased fresh water generation

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Components			
1,2 FIL-019	Seawater filter	MOV-002	HT cooling water temperature control valve
1,2 FIL-021	Strainer for commissioning	MOV-003	Charge air temperature control valve (CHATCO)
HE-002	Lube oil cooler	MOV-016	LT cooling water temperature control valve
HE-005	Nozzle cooling water cooler	MOD-004	Preheating module
HE-007	Diesel oil cooler	MOD-005	Nozzle cooling module
HE-008	Charge air cooler (stage 2)	1 P-002	Attached HT cooling water pump
HE-010	Charge air cooler (stage 1)	1,2 P-062	Seawater pump
1,2 HE-024	Cooler for LT cooling water	1,2 P-076	Pump for LT cooling water
HE-026	Fresh water generator	T-075	LT cooling water expansion tank
HE-034	Cooler for compressor wheel casing		
Major engine connections			
3171	HT cooling water inlet	3499	Nozzle cooling water outlet
3172	HT cooling water inlet	4171	LT cooling water inlet
3184	HT cooling water venting	4184	Compressor cooling water outlet
3199	HT cooling water outlet	4199	LT cooling water outlet
3471	Nozzle cooling water inlet		
Connection to the nozzle cooling module			
N1, N2	Return/feeding of engine nozzle cooling water	N3, N4	Inlet/outlet LT cooling water

5.6.4 External cooling water system – Collection and supply system

T-074/Cooling water collecting tank

The tank is to be dimensioned and arranged in such a way that the cooling water content of the circuits of the cylinder, turbocharger and nozzle cooling systems can be drained into it for maintenance purposes.

This is necessary to meet the requirements with regard to environmental protection (water has been treated with chemicals) and corrosion inhibition (re-use of conditioned cooling water).

Volumes for the engine are listed in table Cooling water and oil volume of engine, Page 154.

In case the engine may be operated on gas, the tank has to be protected according IGF and other applicable standards (see "Safety Concept – Marine dual fuel engines").

Tank equipment:

- Sight glass for level monitoring or other suitable device for continuous level monitoring

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- Low-level alarm switch (electric parts to be placed outside of the tank or ATEX approved)
- Overflow and filling connection
- Inlet for corrosion inhibitor
- Venting to safe area with flame trap
- Inspection opening for manual gas detection device
- Connection for inert gas (flushing with nitrogen gas)

The tank has to be marked as a gas dangerous zone!

P-031/Cooling water filling pump (not indicated in the diagram)

To refill the HT- and LT cooling water system after maintenance, we recommend to install a cooling water filling pump. The pump shall be installed below the cooling water collecting tank with pipe connections to the related systems.

5.6.5 External cooling water system – Miscellaneous items

Piping

Coolant additives may attack a zinc layer. It is therefore imperative to avoid using galvanised steel pipes. Treatment of cooling water as specified by MAN Energy Solutions will safely protect the inner pipe walls against corrosion.

Moreover, there is the risk of the formation of local electrolytic element couples where the zinc layer has been worn off, and the risk of aeration corrosion where the zinc layer is not properly bonded to the substrate.

See the working instructions 6682 000.16-01E for cleaning of steel pipes before fitting.

Pipes shall be manufactured and assembled in a way that ensures a proper draining of all segments. Venting is to be provided at each high point of the pipe system and drain openings at each low point. Make sure to use lockable ball valves or locking caps to prevent hot water leaving the system in case the valves are opened by mistake.

Cooling water pipes are to be designed according to pressure values and flow rates stated in section Planning data, Page 100 and the following sections. Consider the maximal possible pressure level, that can be produced by the pumps (zero delivery head) integrated in the system and static pressure by expansion tanks or other equipment. The engine cooling water connections have to be designed according to PN10/PN16.

5.6.6 External cooling water system – Nozzle cooling system

General

In HFO and gas operation, the nozzles of the fuel injection valves are cooled by freshwater circulation, therefore a nozzle cooling water system is required. It is a separate and closed system re-cooled by the LT cooling water system, but not directly in contact with the LT cooling water. The separate nozzle cooling water system ensures easy detection of damages at the nozzles. Even small fuel leakages are visible via the sight glass. The closed system also prevents the engine and other parts of the cooling water system from pollution by fuel oil. Cleaning of the system is quite easy and only a small amount of contaminated water has to be discharged to the sludge tank. In case the nozzle cooling water is not contaminated, it may be drained to the cooling water col-

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lecting tank T-074. The nozzle cooling water is to be treated with corrosion inhibitor according to MAN Energy Solutions specification. For further information see section Specification of engine coolant, Page 283.

Note:
In diesel engines designed to operate prevalently on HFO the injection valves are to be cooled during operation on HFO. In the case of MGO or MDO operation exceeding 72 h, the nozzle cooling is to be switched off and the supply line is to be closed. The return pipe has to remain open.
In diesel engines designed to operate exclusively on MGO or MDO (no HFO operation possible), nozzle cooling is not required. The nozzle cooling system is omitted.

For operation on HFO or gas, the nozzle cooling system has to be activated.

P-005/Nozzle cooling water pump

The centrifugal (non self-priming) pump discharges cooling water via the nozzle cooling water cooler (HE-005) and the strainer for cooling water (FIL-021) to the header pipe on the engine and then to the individual injection valves.

From here, it is pumped through a manifold into the nozzle cooling water service tank from where it returns to the pump.

One system can be installed for up to three engines.

T-076/Nozzle cooling water service tank

The nozzle cooling water service tank (T-076) is used for deaeration of the nozzle cooling water.

HE-005/Nozzle cooling water cooler

The nozzle cooling water cooler is to be connected in the LT cooling water circuit according to schematic diagram. Cooling of the nozzle cooling water is effected by the LT cooling water.

If an antifreeze is added to the cooling water, the resulting lower heat transfer rate must be taken into consideration. The cooler is to be provided with venting and draining facilities.

TCV-005/Nozzle cooling water temperature control valve

The nozzle cooling water temperature control valve with thermal-expansion elements regulates the flow through the cooler to reach the required inlet temperature of the nozzle cooling water. It has a regulating range from approximately 55 °C (valve begins to open the pipe from the cooler) to 65 °C (pipe from the cooler completely open).

FIL-021/Strainer for cooling water

To protect the nozzles for the first commissioning of the engine a strainer for cooling water has to be provided.

We recommend to install Y-type strainers. For further operation of the engine, we recommend to use a cartridge with mesh size of about 1 mm to catch particles.

The strainers have to be installed in horizontal pipes or pipes with flow direction downwards.

5.6.7 External cooling water system – Nozzle cooling water module

Design

The nozzle cooling water module consists of a storage tank, on which all components required for nozzle cooling are mounted.

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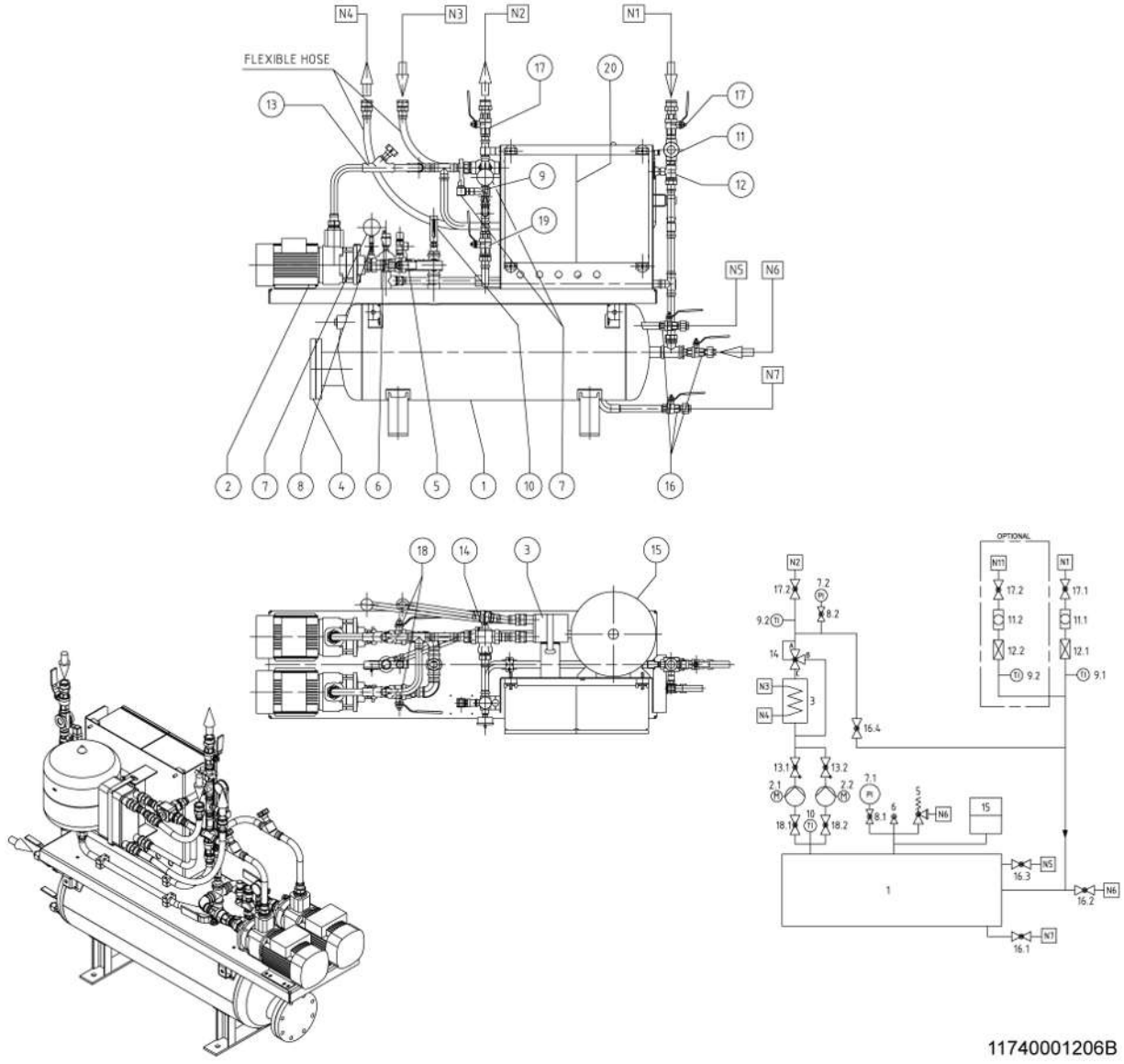

Figure 132: Example: Compact nozzle cooling water module

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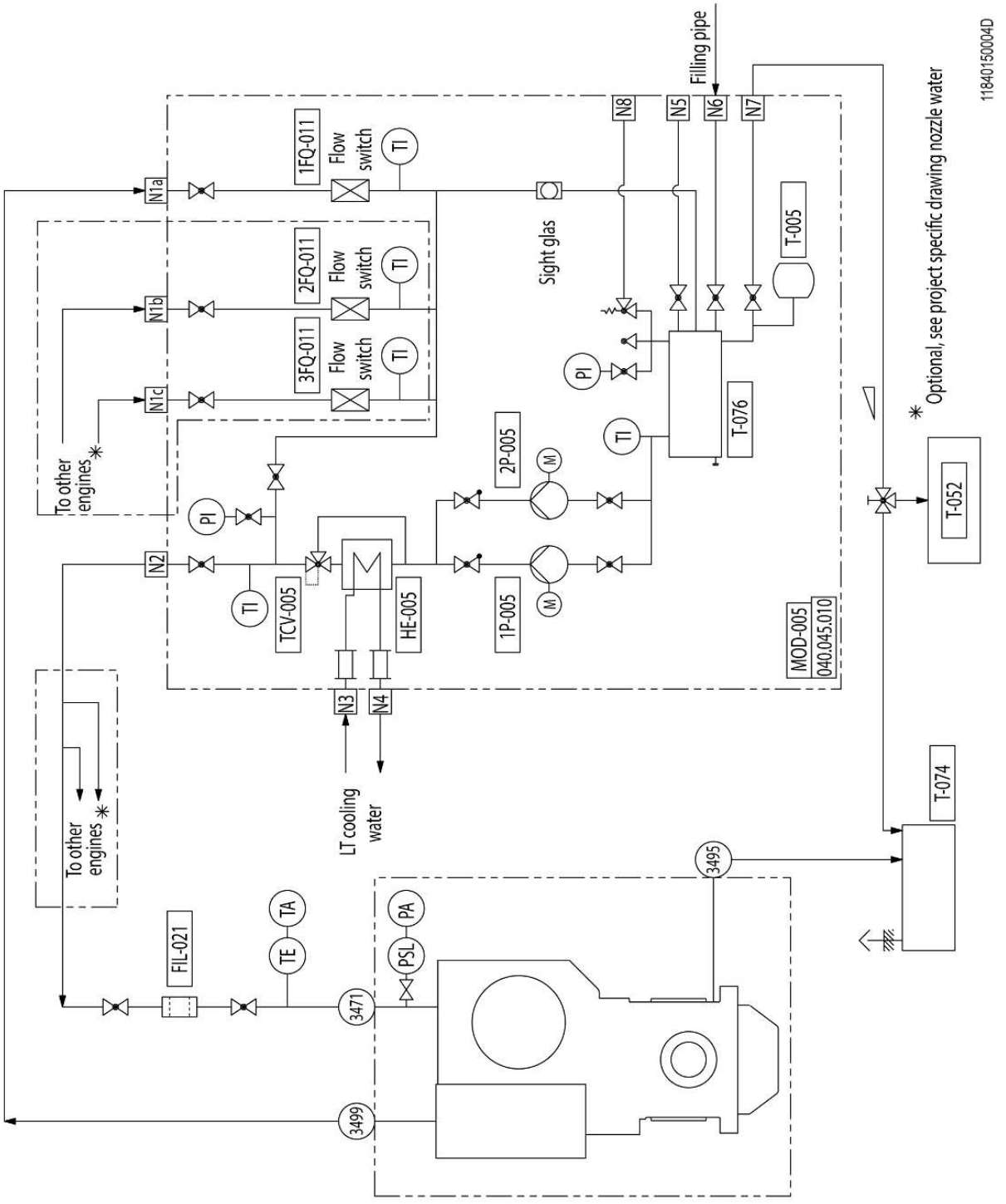

Figure 133: Nozzle cooling water module diagram

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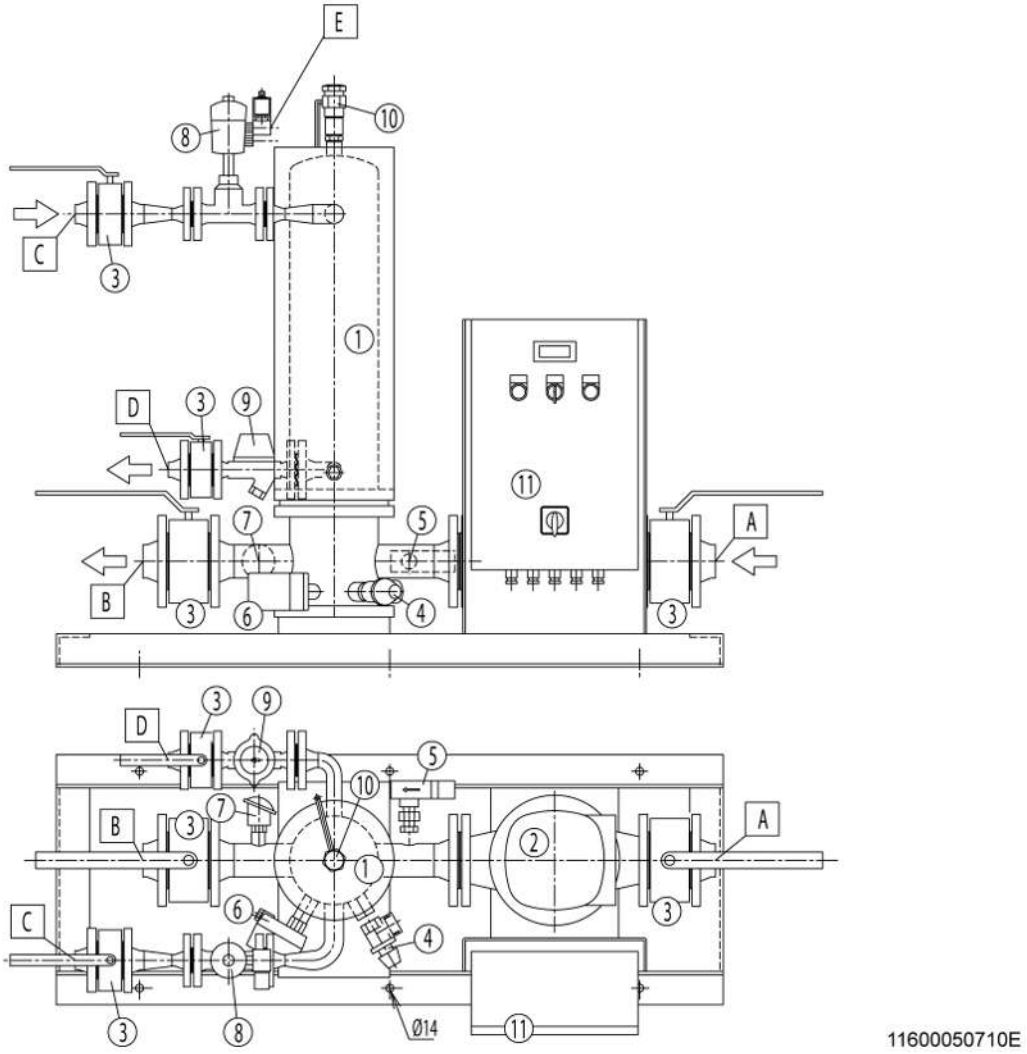
5.6.8 External cooling water system – HT cooling water preheating module



Components

1 Preheater
2 Circulating pump
3 Valve
4 Safety valve
5 Flow switch
6 Temp. limiter
7 Temp. sensor
8 Pneumatic valve
9 Condensate water discharger
10 Automatic ventilation
11 Switch cabinet

Connections

A Cooling water inlet, PN16/40
B Cooling water outlet, PN16/40
C Steam inlet, PN40
D Condensate outlet PN40
E Pilot solenoid valve

Figure 134: Example – Compact HT cooling water preheating module

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5.7 Bilge water/oily water

5.7.1 Introduction

For operation and maintenance of the engine, several pipes and tanks for supply of fresh water and disposal of bilge/oily water are to be provided.

A water cleaning system for bilge water in accordance with MARPOL requirements and rules of classification societies and local authorities must be installed. Oily water or water treated with chemical additives must not be discharged into the sea.

5.7.2 Turbocharger washing equipment

The turbocharger of engines operating on heavy fuel oil or MDO must be cleaned at regular intervals. This requires the installation of a freshwater supply line from the sanitary fresh water system to the turbine washing equipment and dirty-water drain pipes via a funnel (for visual inspection) to the bilge system. A fresh water connection DN 25 with shut-off valve, pressure reducing device (2 – 4 bar) with integrated filter and pressure gauge (0 – 6 bar) is to be provided.

The water lance must be removed after every washing process. This is a precautionary measure, which serves to prevent an inadvertent admission of water to the turbocharger.

The turbocharger washing equipment is completely mounted on the turbocharger.

For further information contact MAN Energy Solutions.

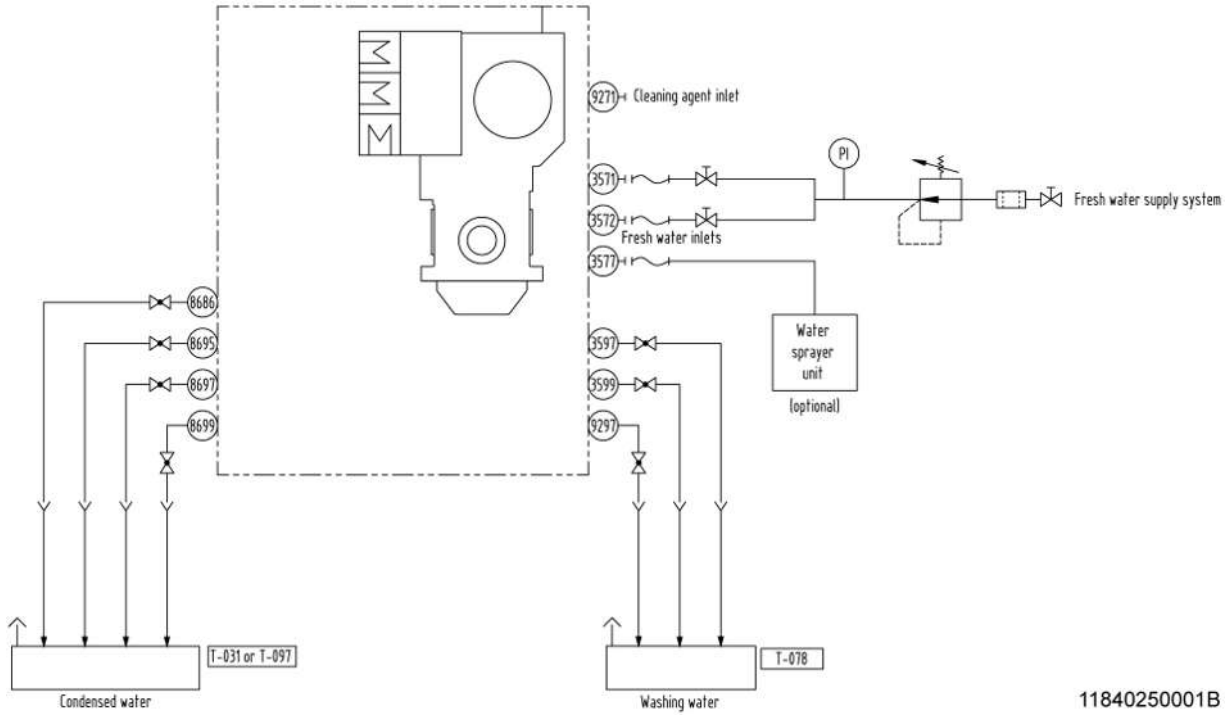

Figure 135: P&ID washing water/condensed water

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To be defined

Figure 136: Cleaning turbine

5.7.3 Cleaning of charge air cooler

The cooler bundle can be cleaned without being removed. Prior to filling with cleaning solvent, the charge air cooler and its adjacent housings must be separated from the turbocharger and charge air pipe using blind flanges.

- The casing must be filled and drained with a firehose with shut-off valve.
- If the cooler bundle is contaminated with oil, the charge air cooler casing is to be filled with freshwater and a liquid detergent additive.
- Insert the ultrasonic cleaning device after addition of the cleaning agent in default dosing portion.
- Flush with freshwater at least two times.

The contaminated water must be removed after every sequence and drained into the bilge or an oily/dirty water tank.

Recommended cleaning medium:

"PrimeServClean MAN C 0186"

Note:
When using cleaning agents:
The instructions of the manufacturers must be observed. In particular the data sheets with safety relevance must be followed. The temperature of these products has (due to the fact that some of them are inflammable) to be at least 10 °C lower than the respective flash point. The waste disposal instructions of the manufacturers must be observed. All terms and conditions of the classification societies as well as MARPOL requirements must be fulfilled.

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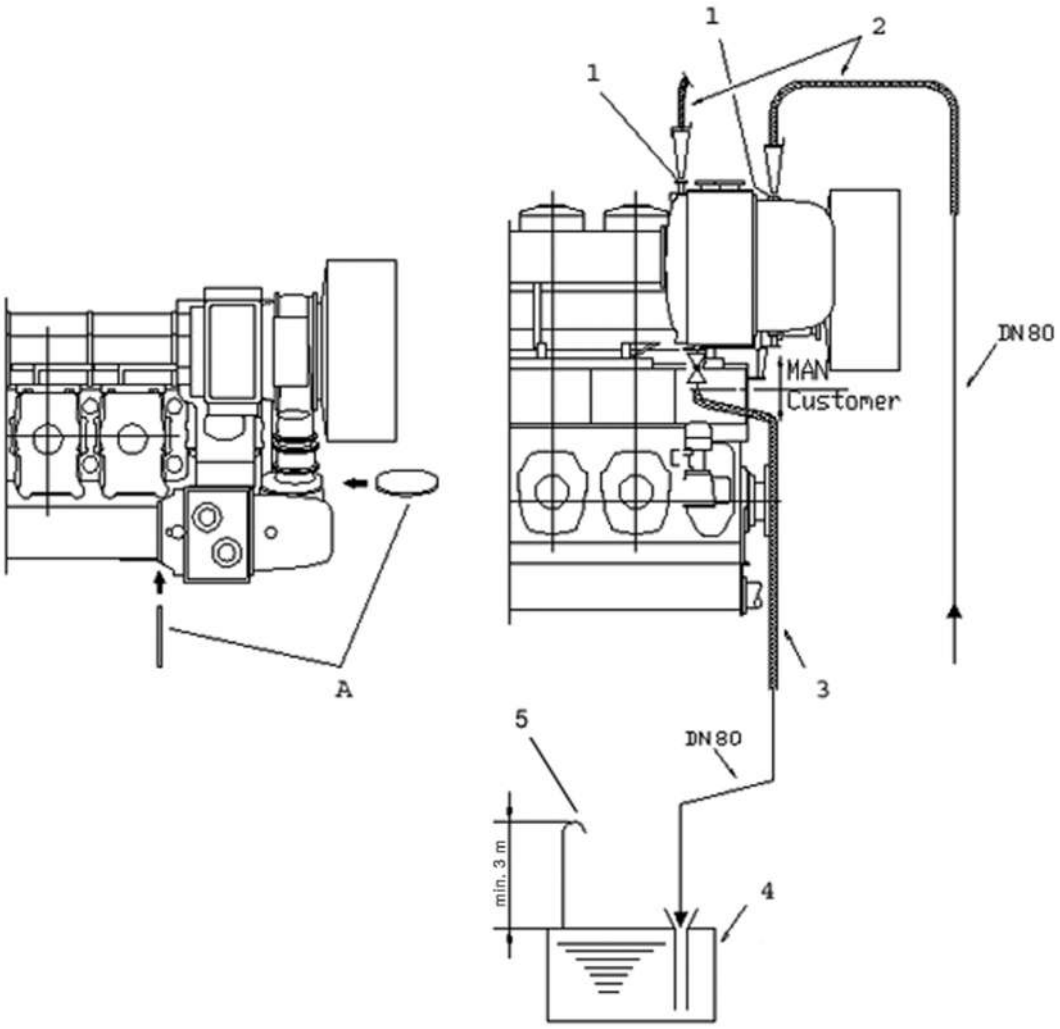

1 Installation ultrasonic cleaning
2 Firehose with sprag nozzle
3 Firehose
4 Dirty water collecting tank.
Required size of dirty water collecting tank:
Volume at the least 4-multiple charge air cooler volume.
5 Ventilation
A Isolation with blind flanges

Figure 137: Principle layout

5.7.4 Condensate monitoring tank drain

The condensate deposition in the charge air cooler is drained via the condensate monitoring tank. The condensate can be drained to the bilge.

If sailing mainly in tropical areas with high amount of condensate, the installation of a separate condensate holding tank might be useful to avoid the unnecessary use of an oily water separator. This condensate might be directly discharged to the sea. Nevertheless, oil monitoring of the condensate is mandatory.

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5.7.5 Nozzle cooling water drain

It might occur, that the nozzle cooling water becomes contaminated with fuel oil due to a nozzle failure.

For such an event, a draining possibility to the sludge tank should be provided. As the cooling water volume is quite low, the draining can be done manually (e.g. with a bucket or a temporary hose).

5.7.6 Condensate drain starting air system

The condensed water from condensate traps related to the starting air system may be led to the bilge. The function of the drains should be monitorable (e.g. via funnels, sight glasses or other devices).

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5.8 Fuel system

5.8.1 Internal fuel system

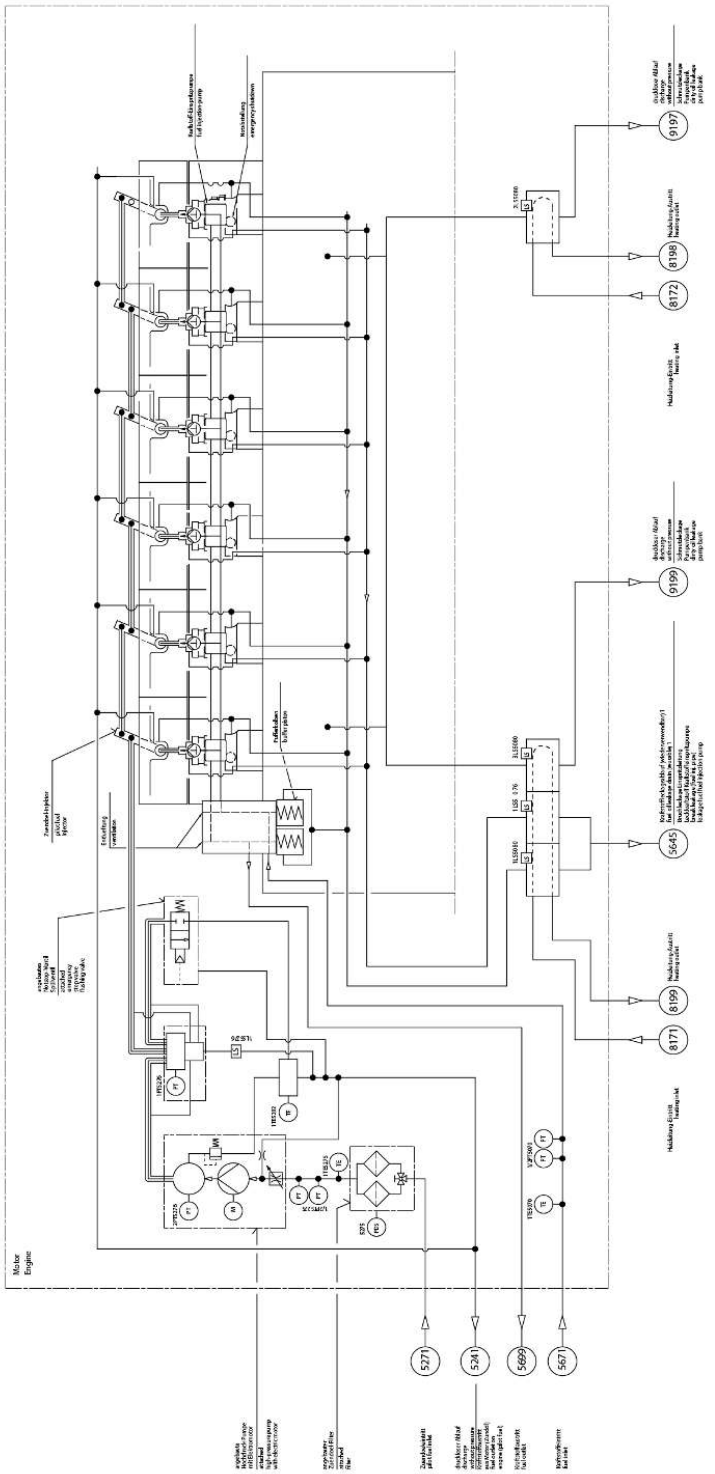

Figure 138: Internal fuel system MAN 51/60DF – Exemplary

Connection numbers			
5101	Fuel oil inlet to engine (main injection system)	5211	Fuel oil outlet from engine
5111	Fuel oil outlet from engine	9141	Dirty oil drain from pump bank, counter coupling side 1
5145	Fuel oil leakage drain (reusable) 1	9142	Dirty oil drain from pump bank, counter coupling side 2
5146	Fuel oil leakage drain (reusable) 2	9143	Dirty oil drain from pump bank, coupling side 1
5201	Fuel oil inlet to engine (pilot injection system)	9144	Dirty oil drain from pump bank, coupling side 2

5.8.2 General introduction of liquid fuel oil system for dual fuel engines (designed to burn HFO, MDO and MGO)

Each cylinder of the engine is equipped with two injection nozzles, the pilot fuel oil nozzle and the main fuel oil nozzle.

Pilot fuel oil

The pilot fuel oil nozzles are part of the pilot fuel oil common rail system. In gas mode this system is used to ignite the gaseous fuel. For this purpose MGO (DMA, DFA) or MDO (DMB, DFB) is used. Pilot fuel oil nozzles are designed to operate with very small fuel oil quantities in order to minimise the pilot fuel oil consumption.

Also in liquid fuel oil mode pilot fuel oil is injected for cooling the nozzles of the pilot fuel oil injectors.

As a safety function, in case of a failure on the pilot fuel oil system, the engine can be operated in liquid fuel oil mode without pilot fuel oil (back up mode).

Without further pilot fuel oil injection, cooling of the pilot fuel oil nozzles is missing. With the low pilot fuel oil pressure, there is a danger that the combustion pressure could flow back into the injector. In both cases the injector will be damaged after a few operating hours. Back up mode should only be used at emergency conditions and as short as possible.

Main fuel oil injection system

The main fuel oil nozzles are designed to ensure full load operation of the engine in liquid fuel oil mode.

Only if the engine is operated in liquid fuel oil mode, fuel oil is injected through the main fuel oil nozzles and burned. Nevertheless, to ensure the lubrication and cooling of the injection pumps and to be prepared to switch the engine automatically and immediately from gas mode to liquid fuel oil mode for safety reasons, main fuel oil has to be supplied to the engine, also when operated in gas mode. In gas mode there is no main fuel oil consumption, the complete main fuel oil quantity will circulate.

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The engine is equipped with two main fuel oil connections, one for inlet and one for outlet, both under pressure. The required main fuel oil flow at engine inlet is equal to 3 times the max. fuel oil consumption of the engine. Non-burned fuel oil will circulate via the main fuel oil outlet connection back to the external fuel oil system.

As main fuel oil HFO or MDO (DMB, DFB) can be used. In case HFO is used, it must be heated up to meet a viscosity of 11 cSt (max. 14 cSt for very high fuel oil viscosity) at engine inlet.

When MDO is used, it is normally not necessary to heat up the fuel. It must be ensured that the MDO temperature at engine inlet does not become to warm. Therefore a fuel oil cooler must be installed in the external fuel oil system.

External fuel oil system

The external fuel oil system has to feed the engine with pilot fuel oil and with main fuel oil and it has to ensure safety aspects in order to enable the engine to be switched from gas mode to liquid fuel oil mode automatically and immediately. Also transient conditions, like conditions during fuel changing from HFO to MDO, must be considered.

Normally two or three engines (one engine group) are served by one fuel oil system in common.

Standard main fuel oil flexibility for the engine group means that all engines connected to the same external fuel oil system can operate simultaneously on the same main fuel oil only. For example, engine No. 1 and No. 2 are operating together and at the same time on HFO as main fuel oil. It is possible to switch the main fuel oil from HFO to MDO, but this can be done for the whole engine group only. It is not possible to select for each single engine of the group a different main fuel oil.

Each engine can be operated in gas mode or liquid fuel oil mode individually and at any time. Dual fuel engines are operated frequently and for long time periods in gas mode or in stand-by mode. In these cases no main fuel oil is burned, but it is circulated. HFO is subject to alteration if circulated in the fuel oil system without being consumed. It becomes necessary to avoid circulation of the same HFO content for a period longer than 12 hours. Therefore the external main fuel oil system must be designed to ensure that the HFO content of the fuel system is completely exchanged with "fresh" HFO every 12 hours. This can be done by a return pipe from the booster system into the heavy fuel oil settling tank. Alternatively HFO can be substituted by MDO, which is not so sensitive to alterations if circulated for long time.

Other limitations for long term operation on gas, MDO or HFO can be given by the selected lube oil (base number) and by the minimum admissible load.

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5.8.3 External fuel system – Marine diesel oil (MDO) treatment system

A prerequisite for safe and reliable engine operation with a minimum of servicing is a properly designed and well-functioning fuel oil treatment system.

The schematic diagram, see figure MDO treatment system diagram, Page 369 shows the system components required for fuel oil treatment for marine diesel oil (MDO).

T-015/Diesel fuel oil storage tank

The minimum effective capacity of the tank should be sufficient for the operation of the propulsion plant, as well as for the operation of the auxiliary diesels for the maximum duration of voyage including the resulting sediments and water. Regarding the tank design, the requirements of the respective classification society are to be observed.

Tank heating

The tank heater must be designed so that the MDO in it is at a temperature of at least 10 °C minimum above the pour point. The supply of the heating medium must be automatically controlled as a function of the MDO temperature.

Fuel with biodiesel

In case fuel oils with up to 7 % of biodiesel (FAME) are used, there is an increased risk of degradation especially due to microbial activity which can threaten engine performance. In order to minimise this risk, long storage periods of this fuel have to be avoided. Furthermore all distillate tanks are to be supplied with a drainage system to prevent bacterial growth by water accumulation. To increase the water separation a coalescer unit could be installed as a parallel installation on the most critical tanks (not indicated in the diagram).

Tank venting

In case of longer storage of diesel fuel oil, a technical device (e.g. air dryer) should be installed to prevent condensation of water in the fuel oil tanks due to humidity.

T-021/Sludge tank

If disposal by an incinerator plant is not planned, the tank has to be dimensioned so that it is capable of absorbing all residues which accumulate during the operation in the course of a maximum duration of voyage. The content of this tank must not be added to the engine fuel oil. In order to enable the emptying of the tank, it must be heated.

The heating is to be dimensioned so that the content of the tank can be heated to approximately 40 °C.

P-073/Diesel fuel oil separator feed pump

The diesel fuel oil separator feed pump should always be electrically driven, i.e. not mounted on the separator, as the delivery volume can be matched better to the required throughput.

H-019/Fuel oil preheater

In order to achieve the separating temperature, a separator adapted to suit the fuel oil viscosity should be fitted.

The preheater must be able to heat the diesel oil up to 40 °C and the size must be selected according to the maximum throughput. However the medium temperature prescribed in the separator manual must be observed and adjusted.

CF-003/Diesel fuel oil separator

A self-cleaning separator must be provided. The diesel fuel oil separator is dimensioned in accordance with the separator manufacturers' guidelines.

The required flow rate (Q) can be roughly determined by the following equation:

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$$Q = \frac{P \times b_e}{\rho}$$

Q [l/h]	Separator flow rate
P [kW]	Total engine output
b_e [g/kWh]	Fuel oil consumption
ρ [g/l]	Density at separating temp approximately 870 kg/m^3 = [g/l]

With the evaluated flow rate, the size of the separator has to be selected according to the evaluation table of the manufacturer. The separator rating stated by the manufacturer should be higher than the flow rate (Q) calculated according to the above formula.

For the first estimation of the maximum fuel oil consumption (b_e), increase the specific table value by 15 %, see section Planning data, Page 100.

For project-specific values contact MAN Energy Solutions.

In the following, characteristics affecting the fuel oil consumption are listed exemplary:

- Tropical conditions
- The engine-mounted pumps
- Fluctuations of the calorific value
- The consumption tolerance

Regarding required limits on water and particles in the fuel after separation refer to table Requirements for diesel fuel, Page 272.

Note:
If a homogeniser is used, it must never be installed between the settling tank and separator as otherwise it will not be possible to ensure satisfactory separation of harmful contaminants, particularly seawater.

Withdrawal points for samples

Fuel oil sampling points are to be provided upstream and downstream of each separator, to verify the effectiveness of these system components.

T-003/Diesel fuel oil service tank

See description in section External fuel system – Heavy fuel oil (HFO) supply system, Page 384.

T-071/Clean leakage fuel oil tank

See description in section External fuel system – Marine diesel oil (MDO) supply system, Page 370.

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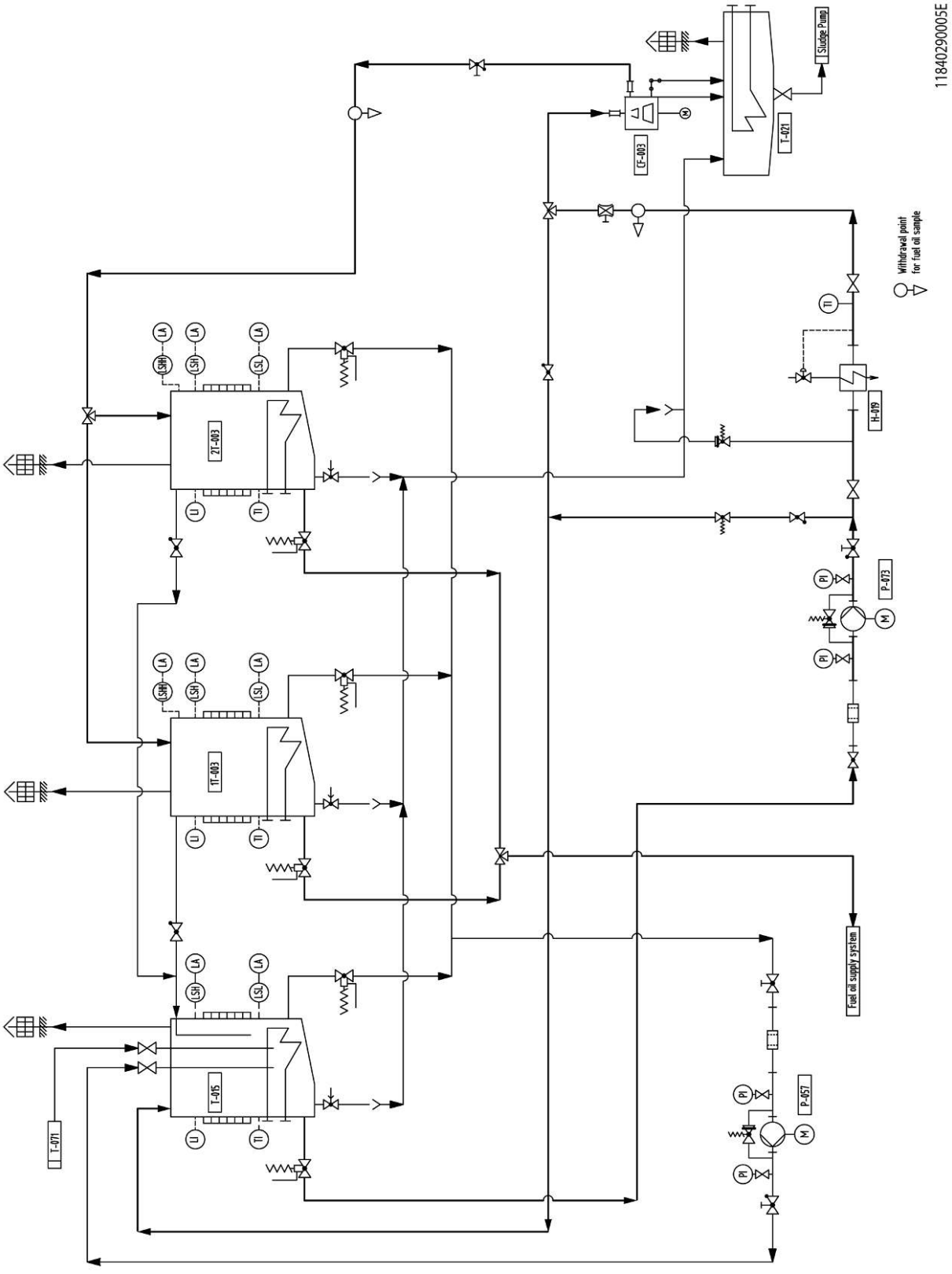
MDO treatment system diagram



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	Components		
CF-003	Diesel fuel oil separator	T-015	Diesel fuel oil storage tank
H-019	Fuel oil preheater	T-021	Sludge tank
P-057	Diesel fuel oil transfer pump	1,2 T-003	Diesel fuel oil service tank
P-073	Diesel fuel oil separator feed pump	T-071	Clean leakage fuel oil tank

Figure 139: Fuel oil treatment with separator CF-003

5.8.4 External fuel system – Marine diesel oil (MDO) supply system for dual fuel engines

General

The MDO supply system is an open system with open deaeration service tank. Usually one or two main engines are connected to one fuel system. If required auxiliary engines can be connected to the same fuel system as well (not indicated in the diagram).

MDO fuel oil viscosity

MDO-DMB, DFB with a max. nominal viscosity of 11 cSt (at 40 °C), or lighter MDO qualities, can be used.

At engine inlet the fuel oil viscosity should be 11 cSt or less. The fuel temperature has to be adapted accordingly. It is also to ensure, that the MDO fuel temperature of max. 45 °C at engine inlet (for all MDO qualities) is not exceeded. Therefore, a tank heating and a cooler in the fuel return pipe are required.

T-003/Diesel fuel oil service tank

The classification societies specify that at least two service tanks are to be installed on board. The minimum tank capacity of each tank should, in addition to the MDO consumption of other consumers, enable a full load operation of min. 8 operating hours for all engines under all conditions.

The tank should be provided with a sludge space with a tank bottom inclination of preferably 10° and sludge drain valves at the lowest point. An overflow pipe from the diesel fuel oil service tank T-003 to the diesel fuel oil storage tank T-015 including heating coils and insulation is to be installed.

If DMB, DFB fuel with 11 cSt (at 40 °C) is used, the tank heating is to be designed to keep the tank temperature at min. 40 °C.

For lighter types of fuel oil it is recommended to adjust the tank temperature in order to ensure a fuel oil viscosity of 11 cSt or less. Rules and regulations for tanks issued by the classification societies must be observed.

The required minimum MDO capacity of each service tank is:

$V_{MDOST} = (Q_p \times t_o \times M_s)/(3 \times 1{,}000\ l/m^3)$		
Required min. volume of one diesel fuel oil service tank	V_{MDOST}	m^3
Required supply pump capacity, MDO 45 °C See paragraph P-008/Diesel fuel oil supply pump, Page 371	Q_p	l/h
Operating time t_o = 8 h	t_o	h
Margin for sludge M_s = 1.05	M_s	-

Table 200: Required minimum MDO capacity

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In case more than one engine or different engines are connected to the same fuel oil system, the service tank capacity has to be increased accordingly.

STR-010/Suction strainer

To protect the fuel oil supply pumps, an approximately 0.5 mm gauge (sphere-passing mesh) strainer is to be installed at the suction side of each supply pump.

P-008/Diesel fuel oil supply pump

The supply pump shall keep sufficient fuel pressure before the engine.

The volumetric capacity must be at least 300 % of the maximum fuel oil consumption of the engine, including margins for:

- Tropical conditions
- Realistic heating value and
- Tolerance

To reach this, the diesel fuel oil supply pump has to be designed according to the following formula:

$Q_p = P_1 \times br_{ISO1} \times f_3$		
Required supply pump capacity with MDO 45 °C	Q_p	l/h
Engine output power at 100 % MCR	P_1	kW
Specific engine fuel oil consumption (ISO) at 100 % MCR	br_{ISO1}	g/kWh
Factor for pump dimensioning: $f_3 = 3.75 \times 10^{-3}$	f_3	l/g

Table 201: Formula to design the diesel fuel oil supply pump

In case more than one engine or different engines are connected to the same fuel oil system, the pump capacity has to be increased accordingly.

The discharge pressure shall be selected with reference to the system losses and the pressure required before the engine (see section Planning data, Page 100 and the following). Normally the required discharge pressure is 10 bar.

FIL-003/Fuel oil automatic filter, supply circuit

The automatic filter should be a type that causes no significant pressure drop during flushing sequence. As a reference an acceptable value for a pressure decrease during back flushing is 0.3 – 0.5 bar. The filter mesh size shall be 0.010 mm (absolute) for common rail injection and 0.034 mm (absolute) for conventional injection.

The automatic filter must be equipped with differential pressure indication and switches.

The design criterion relies on the filter surface load, specified by the filter manufacturer.

A by-pass pipe in parallel to the automatic filter is required. A stand-by filter in the by-pass is not required. In case of maintenance on the automatic filter, the by-pass is to be opened; the fuel is then filtered by the fuel oil duplex filter FIL-013.

FIL-013/Fuel oil duplex filter

See description in paragraph FIL-013/Fuel oil duplex filter, Page 390.

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FQ-003/Fuel oil flowmeter

For a fuel oil consumption measurement (not mentioned in the diagram), flowmeters have to be installed upstream and downstream of the engine. The measured difference of these flows equals the consumption.

The upstream flow rate of each engine should be displayed in the ship automation and SaCoS system. Each flowmeter requires a by-pass to ensure a continuous fuel oil flow in case of maintenance or malfunction. A coriolis type flowmeter is recommended for fuel oil flow measuring. The by-pass of the coriolis type flowmeter needs a shut-off valve. If a positive displacement type flowmeter is used the by-pass needs to be equipped with a spring loaded overflow valve which opens automatically in case of blocking. The pressure resistance of the fuel oil flowmeter should be as low as possible and considered during the system design stage.

FBV-010/Flow balancing valve

MDO supply system for only one main engine and without auxiliary engines

The flow balancing valve FBV-010 is not required.

MDO supply system for more than one main engine or/and additional auxiliary engines

The flow balancing valve (1,2 FBV-010) is required at the fuel outlet of each engine. It is used to adjust the individual fuel flow for each engine. It will compensate the influence (flow distribution due to pressure losses) of the piping system. Once these valves are adjusted, they have to be blocked and must not be manipulated later.

PCV-011/Fuel oil spill valve

MDO supply systems for only one main engine and without auxiliary engines

Fuel oil spill valve PCV-011 is not required.

MDO supply systems for more than one main engine or/and additional auxiliary engines

In case two engines are operated with one fuel module, it has to be possible to separate one engine at a time from the fuel circuit for maintenance purposes. In order to avoid a pressure increase in the pressurised system, the fuel, which cannot circulate through the shut-off engine, has to be rerouted via this valve into the return pipe.

This valve is to be adjusted so that rerouting is effected only when the pressure, in comparison to normal operation (multi-engine operation), is exceeded. This valve should be designed as a pressure relief valve, not as a safety valve.

V-002/Shut-off cock

MDO supply systems for only one main engine and without auxiliary engines

Shut-off cock V-002 is not required.

MDO supply systems for more than one main engine or/and additional auxiliary engines

The shut-off cock is closed during normal operation (multi-engine operation). When one engine is separated from the fuel circuit for maintenance purposes, this cock has to be opened manually.

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HE-007/Fuel oil cooler

The fuel oil cooler is required to cool down the fuel, which was heated up while circulating through the injection pumps. The cooler is normally connected to the LT cooling water system and should be dimensioned so that the MDO does not exceed a temperature of max. 45 °C.

Only for very light MDO fuel types this temperature has to be even lower in order to preserve the minimum admissible fuel oil viscosity on engine inlet.

Depending on the fuel oil type a chiller unit might be required to decrease the fuel oil temperature to reach the required injection system operation viscosity, see section Viscosity-temperature diagram (VT diagram), Page 282.

Engine type	Cooler capacity
L/V engine	7.0 kW/cyl.
The max. MDO/MGO throughput is approximately identical to the engine inlet fuel flow (=delivery quantity of the installed fuel oil booster pump).	

Table 202: Dimensioning of the fuel oil cooler for common rail engines

The recommended pressure class of the fuel oil cooler is PN16.

PCV-008/Pressure retaining valve

In open fuel oil supply systems (fuel loop with circulation through the diesel fuel oil service tank; service tank under atmospheric pressure) this pressure-retaining valve is required to keep the system pressure to a certain value against the diesel fuel oil service tank. It is to be adjusted so that the pressure before engine inlet can be maintained in the required range (see section Operating/service temperatures and pressures, Page 147).

FSH-001/Leakage fuel oil monitoring tank

High pressure pump overflow and escaping fuel oil from burst control pipes is carried to the monitoring tanks from which it is drained into the clean leakage fuel oil collecting tank. The engine interface connection of the monitoring tank is a pressureless drain. Leakage fuel oil flows by gravity only from these tank into collecting tanks (to be installed below the engine connections). To prevent pressure resistance at the engine interface connection the pipework to the collecting tank has to have a sufficient downward slope. The float switch mounted in the tanks must be connected to the alarm system. The classification societies require the installation of monitoring tanks for unmanned engine rooms. Lloyd's Register specifies tank monitoring for manned engine rooms as well.

T-006/Leakage oil collecting tank

Dirty leak fuel and leak oil are collected in the leakage oil collecting tank. It must be emptied into the sludge tank. The content of the leakage oil collecting tank T-006 must not be added to the engine fuel. It can be burned for instance in a waste oil boiler.

The leakage oil collecting tank T-006 is only necessary if the leakages can not flow directly into the sludge tank T-021 by gravity.

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T-071/Clean leakage fuel oil tank

When only MDO is used, the high pressure pump overflow and other, clean fuel oil that escapes from the conventional injection system is lead pressureless (by gravity only) to an extra clean leakage fuel oil collecting tank. From there it can be emptied into the diesel fuel oil storage tank. Clean leakage fuel oil from T-071 can be used again after passing the separator. For additional information see description in section External fuel system – Heavy fuel oil (HFO) supply system, Page 384.

Withdrawal points for samples

Fuel oil sampling points are to be provided upstream and downstream of each filter, to verify the effectiveness of these system components.

T-015/Diesel fuel oil storage tank

See description in paragraph T-015/Diesel fuel oil storage tank, Page 367.

T-021/Sludge tank

See description in paragraph T-021/Sludge tank, Page 367.

CF-003/Diesel fuel oil separator

See description in paragraph CF-003/Diesel fuel oil separator, Page 367.

CV-004/Pilot fuel oil service tank filling valve

See description in section External fuel system – Pilot fuel supply system, Page 399.

T-101/Pilot fuel oil service tank

See description in section External fuel system – Pilot fuel supply system, Page 399.

FIL-033/Pilot fuel oil duplex filter

See description in section External fuel system – Pilot fuel supply system, Page 399.

Piping

The fuel oil system design pressure is PN16 (see section External pipe dimensioning, Page 298). The system is to be protected from higher pressure levels by corresponding safety valves. Additional safety valves, not displayed in the P&ID, might become necessary depending on the actual fuel oil system design.

Material

The casing material of pumps and filters should be EN-GJS (nodular cast iron), in accordance to the requirements of the classification societies.

General notes

The arrangement of the final fuel filter directly upstream of the engine inlet (depending on the plant design the final filter could be either the fuel oil duplex filter FIL-013 or the fuel oil automatic filter (supply circuit) FIL-003) has to ensure that no parts of the filter itself can be loosen.

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The pipe between the final filter and the engine inlet has to be done as short as possible and is to be cleaned and treated with particular care to prevent damages (loosen objects/parts) to the engine. Valves or components shall not be installed in this pipe. It is required to dismantle this pipe completely in presents of our commissioning personnel for a complete visual inspection of all internal parts before the first engine start. Therefore, flange pairs have to be provided on eventually installed bends.

The recommended pressure class for the fuel oil pipes is PN16.

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MDO supply system diagrams

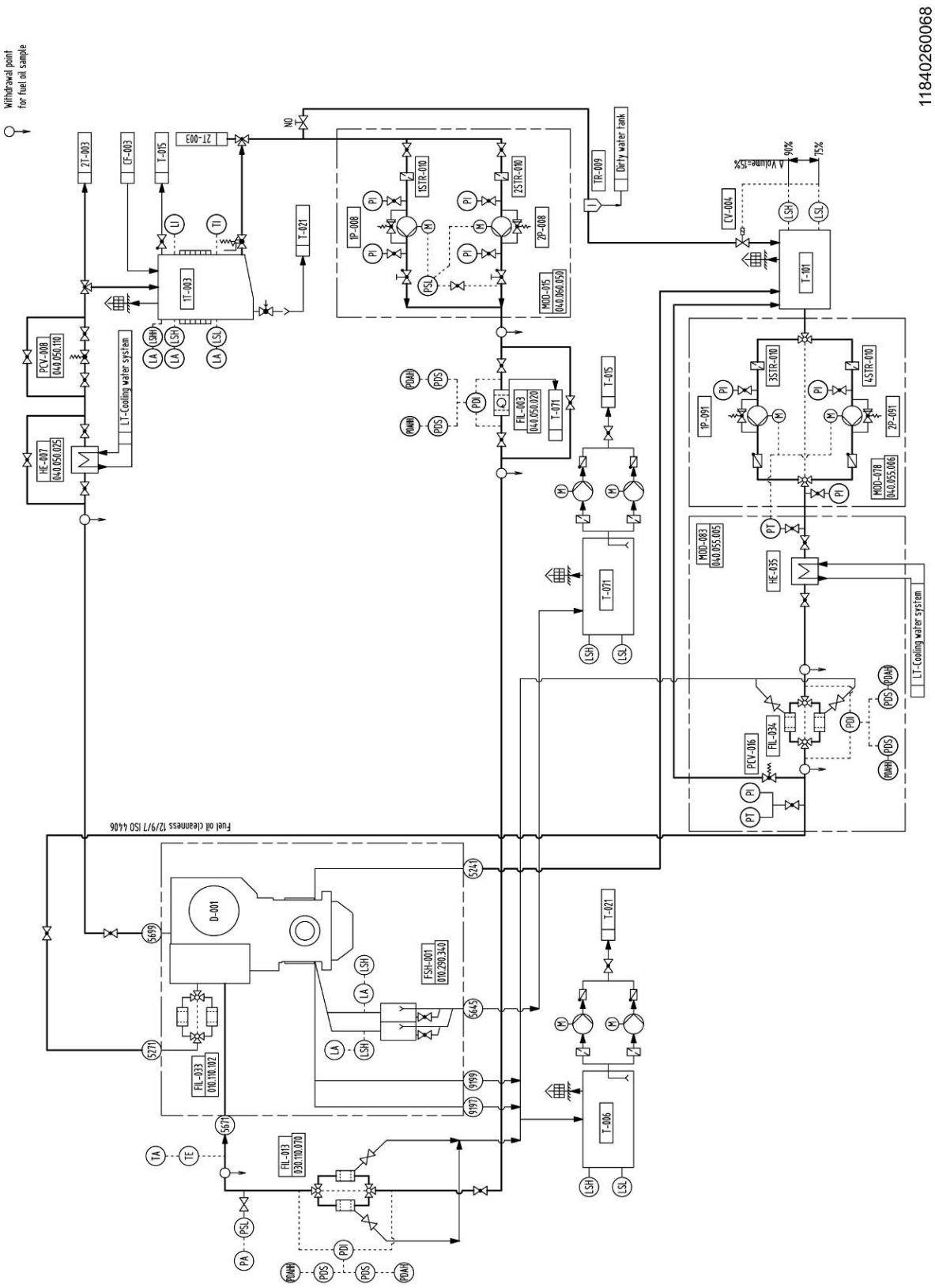

Figure 140: MDO supply system diagram – Single engine plant

Components			
CF-003	Diesel fuel oil separator	1,2 P-008	Diesel fuel oil supply pump
CV-004	Pilot fuel oil service tank filling valve	1,2 P-091	Pilot fuel oil supply pump
D-001	Diesel engine	PCV-008	Pressure retaining valve
FIL-003	Fuel oil automatic filter, supply circuit	PCV-016	Pilot fuel oil spill valve
FIL-013	Fuel oil duplex filter	1,2,3,4 STR-010	Suction strainer
FIL-033	Pilot fuel oil duplex filter	1,2 T-003	Diesel fuel oil service tank
FIL-034	Pilot fuel oil duplex filter	T-006	Leakage oil collecting tank
FSH-001	Leakage fuel oil monitoring tank	T-015	Diesel fuel oil storage tank
HE-007	Fuel oil cooler	T-021	Sludge tank
HE-035	Pilot fuel oil cooler	T-071	Clean leakage fuel oil tank
MOD-015	Fuel oil supply pump unit	T-101	Pilot fuel oil service tank
MOD-078	Pilot fuel oil supply pump module	TR-009	Coalescer (water trap)
MOD-083	Pilot fuel oil filter module		
Major engine connections			
5241	Leakage fuel oil drain pilot fuel-CR	5699	Fuel oil return pipe from engine
5271	Fuel oil inlet pilot fuel-CR	9197	Dirty oil drain from covering, coupling side
5645	Fuel oil break leakage drain (reusable) 1	9199	Dirty oil drain from covering, counter coupling side
5671	Fuel oil inlet on the engine		

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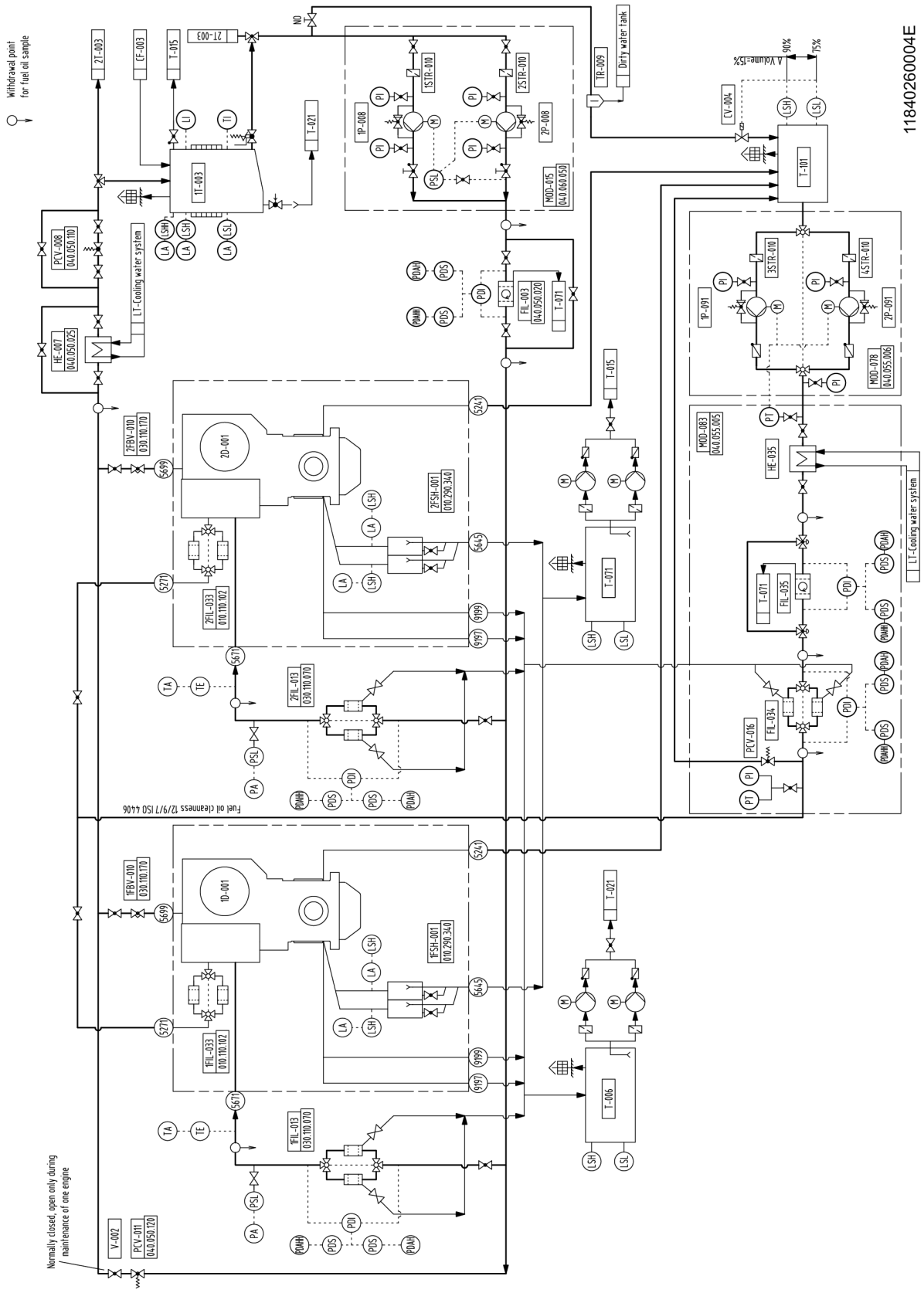

Figure 141: MDO supply system diagram – Twin engine plant

Components			
CF-003	Diesel fuel oil separator	1,2 P-008	Diesel fuel oil supply pump
CV-004	Pilot fuel oil service tank filling valve	1,2 P-091	Pilot fuel oil supply pump
1,2 D-001	Diesel engine	PCV-008	Pressure retaining valve
1,2 FBV-010	Flow balancing valve	PCV-011	Fuel oil spill valve
FIL-003	Fuel oil automatic filter, supply circuit	PCV-016	Pilot fuel oil spill valve
1,2 FIL-013	Fuel oil duplex filter	1,2,3,4 STR-010	Suction strainer
1,2 FIL-033	Pilot fuel oil duplex filter	1,2 T-003	Diesel fuel oil service tank
FIL-034	Pilot fuel oil duplex filter	T-006	Leakage oil collecting tank
FIL-035	Pilot fuel oil automatic filter	T-015	Diesel fuel oil storage tank
1,2 FSH-001	Leakage fuel oil monitoring tank	T-021	Sludge tank
HE-007	Fuel oil cooler	T-071	Clean leakage fuel oil tank
HE-035	Pilot fuel oil cooler	T-101	Pilot fuel oil service tank
MOD-015	Fuel oil supply pump unit	TR-009	Coalescer (water trap)
MOD-078	Pilot fuel oil supply pump module	V-002	Shut-off cock
MOD-083	Pilot fuel oil filter module		
Major engine connections			
5241	Leakage fuel oil drain pilot fuel-CR	5699	Fuel oil return pipe from engine
5271	Fuel oil inlet pilot fuel-CR	9197	Dirty oil drain from covering, coupling side
5645	Fuel oil break leakage drain (reusable) 1	9199	Dirty oil drain from covering, counter coupling side
5671	Fuel oil inlet on the engine		

5.8.5 External fuel system – Heavy fuel oil (HFO) treatment system

A prerequisite for safe and reliable engine operation with a minimum of servicing is a properly designed and well-functioning fuel oil treatment system.

The schematic diagram, see figure HFO treatment system diagram, Page 383 shows the system components required for fuel treatment of heavy fuel oil (HFO).

Bunker fuel oil

Fuel compatibility problems are avoidable if mixing of newly bunkered fuel oil with remaining fuel oil can be prevented by a suitable number of bunkers. Moreover the overall fuel oil system (all fuel oil tanks, piping etc.) must be designed to limit mixing of different bunker batches to an absolute minimum. Heating coils in bunkers need to be designed so that the HFO in it is at a temperature of at least 10 °C minimum above the pour point.

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P-038/Heavy fuel oil transfer pump

The heavy fuel oil transfer pump discharges fuel from the bunkers into the heavy fuel oil settling tanks. Being a screw pump, it handles the fuel oil gently, thus prevent water being emulsified in the fuel oil. Its capacity must be sized to fill the complete heavy fuel oil settling tank within ≤ 2 hours.

T-016/Heavy fuel oil settling tank

Two heavy fuel oil settling tanks should be installed, in order to obtain thorough pre-cleaning and to allow fuels of different origin to be kept separate. When using RM-fuels we recommend two heavy fuel oil settling tanks for each fuel type (high sulphur HFO, low sulphur HFO).

Size

Pre-cleaning by settling is the more effective the longer the solid material is given time to settle. The storage capacity of the heavy fuel oil settling tank should be designed to hold at least a 24-hour supply of fuel oil at full load operation, including sediments and water the fuel oil contains.

The minimum volume (V) to be provided is:

$$V = \frac{5.7 \times P}{1{,}000}$$

V [m^3]	Minimum volume
P [kW]	Engine rating

Tank heating

The heating surfaces should be dimensioned that the heavy fuel oil settling tank content can be evenly heated to 75 °C within 6 to 8 hours. The heating should be automatically controlled, depending on the fuel oil temperature.

In order to avoid:

- Agitation of the sludge due to heating, the heating coils should be arranged at a sufficient distance from the tank bottom.
- The formation of asphaltene, the fuel oil temperature should not be permissible to exceed 75 °C.
- The formation of carbon deposits on the heating surfaces, the heat transferred per unit surface must not exceed 1.1 W/cm^2.

Design

The heavy fuel oil settling tank is to be fitted with baffle plates in longitudinal and transverse direction in order to reduce agitation of the fuel oil in the tank in rough seas as far as possible. The suction pipe of the heavy fuel oil separator must not reach into the sludge space. One or more sludge drain valves, depending on the slant of the tank bottom (preferably 10°), are to be provided at the lowest point. The heavy fuel oil settling tank is to be insulated against thermal losses.

Sludge must be removed from the heavy fuel oil settling tank before the separators draw fuel oil from it.

T-021/Sludge tank

If disposal by an incinerator plant is not planned, the tank has to be dimensioned so that it is capable of absorbing all residues which accumulate during the operation in the course of a maximum duration of voyage. The content of this tank must not be added to the engine fuel oil. In order to enable the emptying of the tank, it must be heated.

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The heating is to be dimensioned so that the content of the tank can be heated to approximately 60 °C.

P-015/Heavy fuel oil separator feed pump

The heavy fuel oil separator feed pump should preferably be of the free-standing type, i.e. not mounted on the heavy fuel oil separator, as the delivery volume can be matched better to the required throughput.

H-008/Heavy fuel oil preheater

To reach the separating temperature a heavy fuel oil preheater matched to the fuel oil viscosity has to be installed.

CF-002/Heavy fuel oil separator

As a rule, poor quality, high viscosity fuel oil is used. Two new generation separators must therefore be installed.

The separators shall be capable to separate water from heavy fuel oil with a density of 1,010 kg/m³ @ 15 °C.

Heavy fuel oil separators must always be provided in sets of 2 of the same type:

- 1 service separator
- 1 stand-by separator

of self-cleaning type.

As a matter of principle, all separators are to be equipped with an automatic programme control for continuous desludging and monitoring.

Mode of operation

The stand-by separator is always to be put into service, to achieve the best possible fuel oil cleaning effect with the separator plant as installed.

The piping of both heavy fuel oil separators is to be arranged in accordance with the manufacturer´s advice, preferably for both parallel and series operation.

The discharge flow of the free-standing dirty oil pump is to be split up equally between the two separators in parallel operation.

The freshwater supplied must be treated as specified by the separator supplier.

Size

The heavy fuel oil separators are dimensioned in accordance with the separator manufacturers' guidelines. The required design flow rate (Q) can be roughly determined by the following equation:

$$Q = \frac{P \times b_e}{\rho}$$

Q [l/h]	Separator flow rate
P [kW]	Total engine output
b_e [g/kWh]	Fuel oil consumption
ρ [g/l]	Density at separating temp approximately 930 kg/m³ = [g/l]

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With the evaluated flow rate, the size of the separator has to be selected according to the evaluation table of the manufacturer. The separator rating stated by the manufacturer should be higher than the flow rate (Q) calculated according to the above formula.

For the first estimation of the maximum fuel oil consumption (b_e), increase the specific table value by 15 %, see section Planning data, Page 100.

For project-specific values contact MAN Energy Solutions.

In the following, characteristics affecting the fuel oil consumption are listed exemplary:

- Tropical conditions
- The engine-mounted pumps
- Fluctuations of the calorific value
- The consumption tolerance

Regarding required limits on water and particles in the fuel after separation refer to table Requirements for diesel fuel, Page 272.

Note:
If a homogeniser is used, it must never be installed between the settling tank and separator as otherwise it will not be possible to ensure satisfactory separation of harmful contaminants, particularly seawater.

Withdrawal points for samples

Fuel oil sampling points are to be provided upstream and downstream of each separator, to verify the effectiveness of these system components.

MOD-008/Fuel oil module

See description in figure(s) HFO supply system diagram(s), Page 395.

T-022/Heavy fuel oil service tank

See description in paragraph T-022/Heavy fuel oil service tank, Page 384.

T-071/Clean leakage fuel oil tank

See description in paragraph T-071/Clean leakage fuel oil tank, Page 391.

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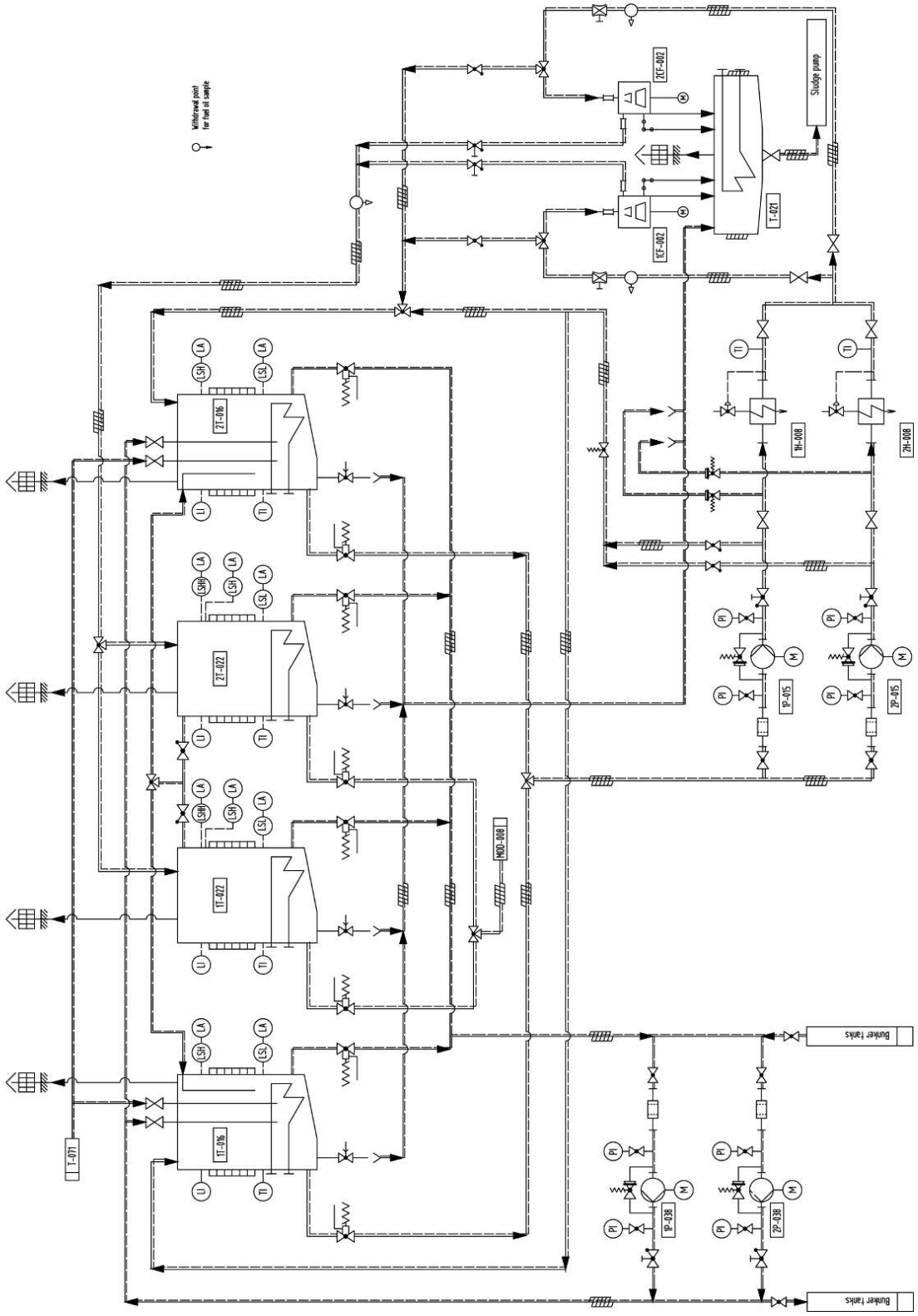
HFO treatment system diagram



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	Components		
1,2 CF-002	HFO separator	1,2 T-016	HFO settling tank
1,2 H-008	HFO preheater	T-021	Sludge tank
MOD-008	Fuel oil module	1,2 T-022	HFO service tank
1,2 P-015	HFO separator feed pump	T-071	Clean leakage fuel oil tank
1,2 P-038	HFO transfer pump		

Figure 142: HFO treatment system diagram

5.8.6 External fuel system – Heavy fuel oil (HFO) supply system

General

The HFO supply system is a pressurised closed loop system. Normally one or two main engines are connected to one fuel system. If required, auxiliary engines can be connected to the same fuel system as well (not indicated in the diagram).

To ensure that high-viscosity fuel oils achieve the specified injection viscosity, high operation temperature is necessary, which may cause degassing problems in conventional, pressureless systems.

A remedial measure is adopting a pressurised system in which the required system pressure is 1 bar above the evaporation pressure of water.

Fuel	Injection viscosity[1)]	Temperature after final heater HFO	Evaporation pressure	Min. required system pressure
mm²/50 °C	mm²/s	°C	bar	bar
180	12	126	1.4	2.4
320	12	138	2.4	3.4
380	12	142	2.7	3.7
420	12	144	2.9	3.9
500	14	141	2.7	3.7
700	14	147	3.2	4.2

[1)] For fuel oil viscosity depending on fuel temperature see section Viscosity-temperature diagram (VT diagram), Page 282.

Table 203: Injection viscosity and temperature after final heater heavy fuel oil

The indicated pressures are minimum requirements due to the fuel characteristic. Nevertheless, to meet the required fuel pressure at the engine inlet (see section Planning data, Page 100 and the following), the pressure in the fuel oil mixing tank and booster circuit becomes significant higher than indicated in this table.

T-022/Heavy fuel oil service tank

The heavy fuel oil cleaned in the heavy fuel oil separator is passed to the service tank, and as the separators are in continuous operation, the tank is always kept filled.

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To fulfil this requirement it is necessary to fit the heavy fuel oil service tank T-022 with overflow pipes, which are connected with the heavy fuel oil settling tanks T-016. The tank capacity is to be designed for at least eight-hours' fuel supply at full load so as to provide for a sufficient period of time for separator maintenance.

The tank should have a sludge space with a tank bottom inclination of preferably 10° with sludge drain valves at the lowest point and it is to be equipped with heating coils.

The sludge must be drained from the service tank at regular intervals.

The heating coils are to be designed for a tank temperature of 75 °C.

The rules and regulations for tanks issued by the classification societies must be observed.

HFO with high and low sulphur content must be stored in separate service tanks.

T-003/Diesel fuel oil service tank

The classification societies specify that at least two service tanks are to be installed on board. The minimum volume of each tank should, in addition to the MDO/MGO consumption of the generating sets, enable an eight-hour full load operation of the main engine.

Cleaning of the MDO/MGO by an additional separator should, in the first place, be designed to meet the requirements of the diesel alternator sets on board. Just like the heavy fuel oil service tank, the diesel fuel oil service tank is to be provided with a sludge space with sludge drain valve and with an overflow pipe from the diesel fuel oil service tank T-003 to the diesel fuel oil storage tank T-015. For more detailed information see section External fuel system – Marine diesel oil (MDO) supply system, Page 370.

CK-002/Three-way valve for fuel oil changeover

This valve is used for changing over from MDO/MGO operation to heavy fuel operation and vice versa. This valve could be operated manually or automatically. It is equipped with two limit switches for remote indication and suppression of alarms from the viscosity measuring and control system during MDO/MGO operation.

STR-010/Suction strainer

To protect the fuel oil supply pumps, an approximately 0.5 mm gauge (sphere-passing mesh) strainer is to be installed at the suction side of each supply pump.

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P-018/Fuel oil supply pump

The volumetric capacity must be at least 160 % of max. fuel oil consumption.

$Q_{P1} = P_1 \times br_{ISO} \times f_4$		
Required supply pump delivery capacity with HFO at 90 °C	Q_{P1}	l/h
Engine output at 100 % MCR	P_1	kW
Specific engine fuel oil consumption (ISO) at 100 % MCR	br_{ISO}	g/kWh
Factor for pump dimensioning ▪ For diesel engines operating on main fuel HFO: f_4 = 2.00 x 10^{-3}	f_4	l/g

Note:
The factor f_4 includes the following parameters:

- 160 % fuel oil flow
- Main fuel: HFO 380 $mm^2/50$ °C
- Attached lube oil and cooling water pumps
- Tropical conditions
- Realistic lower heating value
- Specific fuel oil weight at pumping temperature
- Tolerance

In case more than one engine is connected to the same fuel oil system, the pump capacity has to be increased accordingly.

Table 204: Simplified fuel oil supply pump dimensioning

The delivery height of the fuel oil supply pump shall be selected according to the required system pressure (see table Injection viscosity and temperature after final heater heavy fuel oil, Page 384), the required pressure in the mixing tank and the resistance of the automatic filter, flowmeter and piping system.

		Injection system
		bar
Positive pressure at the fuel module inlet due to tank level above fuel module level	–	0.10
Pressure loss of the pipes between fuel module inlet and mixing tank inlet	+	0.20
Pressure loss of the automatic filter	+	0.80
Pressure loss of the fuel flow measuring device	+	0.10
Pressure in the fuel oil mixing tank	+	5.70
Operating delivery height of the supply pump	=	6.70

Table 205: Example for the determination of the expected operating delivery height of the fuel oil supply pump

It is recommended to install fuel oil supply pumps designed for the following pressures:

Engines with conventional fuel oil injection system: Design delivery height 7.0 bar, design output pressure 7.0 bar.

Engines with common rail injection system: Design delivery height 8.0 bar, design output pressure 8.0 bar.

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HE-025/Fuel oil cooler, supply circuit

If no fuel is consumed in the system while the pump is in operation, the finned-tube cooler prevents excessive heating of the fuel. Its cooling surface must be adequate to dissipate the heat that is produced by the pump to the ambient air.

In case of continuos MDO/MGO operation, a water cooled fuel oil cooler is required to keep the fuel oil temperature below 45 °C.

PCV-009/Pressure limiting valve

This valve is used for setting the required system pressure and keeping it constant. It returns in the case of

- engine shutdown 100 %, and of
- engine full load 37.5 % of the quantity delivered by the fuel oil supply pump back to the pump suction side.

FIL-003/Fuel oil automatic filter, supply circuit

The automatic filter should be a type that causes no significant pressure drop during flushing sequence. As a reference an acceptable value for a pressure decrease during back flushing is 0.3 – 0.5 bar.

The automatic filter must be equipped with differential pressure indication and switches.

Design criterion is the filter area load specified by the filter manufacturer. The fuel oil automatic filter (supply circuit) has to be installed in the plant. It is not attached to the engine.

	Conventional fuel injection system
Filter mesh width (mm)	0.034
Design pressure	PN10

Table 206: Required filter mesh width (sphere passing mesh)

FQ-003/Fuel oil flowmeter

For a fuel oil consumption measurement (not mentioned in the diagram), flowmeters have to be installed upstream and downstream of the engine. The measured difference of these flows equals the consumption.

The upstream flow rate of each engine should be displayed in the ship automation and SaCoS system. Each flowmeter requires a by-pass to ensure a continuous fuel oil flow in case of maintenance or malfunction. A coriolis type flowmeter is recommended for fuel oil flow measuring. The by-pass of the coriolis type flowmeter needs a shut-off valve. If a positive displacement type flowmeter is used the by-pass needs to be equipped with a spring loaded overflow valve which opens automatically in case of blocking. The pressure resistance of the fuel oil flowmeter should be as low as possible and considered during the system design stage.

One fuel oil flowmeter upstream of the fuel oil mixing tank inlet and an additional flowmeter downstream the minimum flow valve is required. Alternatively a flowmeter upstream and downstream the engine can be installed. The measured difference of these flows equals the consumption.

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T-011/Fuel oil mixing tank

The mixing tank compensates pressure surges which occur in the pressurised part of the fuel system.

For this purpose, there has to be an air cushion in the tank. As this air cushion is exhausted during operation, compressed air (max. 10 bar) has to be refilled via the control air connection from time to time.

Before prolonged shutdowns the system is changed over to MDO/MGO operation.

The tank volume shall be designed keep the engine running at 50 % load for 5 minutes. At 50 % load the fuel oil consumption is approx. 1/2 x fuel oil consumption at full load.

The tank shall be designed for the maximum possible service pressure, usually approximately 10 bar and is to be accepted by the classification society in question.

The expected operating pressure in the fuel oil mixing tank depends on the required fuel oil pressure at the inlet (see section Planning data, Page 100) and the pressure losses of the installed components and pipes.

		Injection system
		bar
Required max. fuel oil pressure at engine inlet	+	8.00
Pressure difference between fuel oil inlet and outlet engine	–	2.00
Pressure loss of the fuel oil return pipe between engine outlet and mixing tank inlet, e.g.	–	0.30
Pressure loss of the flow balancing valve (to be installed only in multi-engine plants, pressure loss approximately 0.5 bar)	–	0.00
Operating pressure in the fuel oil mixing tank	=	5.70

Table 207: Example for the determination of the expected operating pressure of the fuel oil mixing tank

This example demonstrates, that the calculated operating pressure in the fuel oil mixing tank is (for all HFO viscosities) higher than the min. required fuel oil pressure (see table Injection viscosity and temperature after final heater heavy fuel oil, Page 384).

P-003/Fuel oil booster pump

To cool the engine mounted injection pumps, the capacity of the booster pump has to be at least 300 % of maximum fuel oil consumption at injection viscosity.

$Q_{P2} = P_1 \times br_{ISO} \times f_5$		
Required booster pump delivery capacity with HFO at 145 °C	Q_{P2}	l/h
Engine output at 100 % MCR	P_1	kW
Specific engine fuel oil consumption (ISO) at 100 % MCR	br_{ISO}	g/kWh
Factor for pump dimensioning ▪ For diesel engines operating on main fuel HFO: $f_5 = 3.90 \times 10^{-3}$	f_5	l/g

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Note:
The factor f_s includes the following parameters:

- 300 % fuel oil flow at 100 % MCR
- Main fuel: HFO 380 $mm^2/50$ °C
- Attached lube oil and cooling water pumps
- Tropical conditions
- Realistic lower heating value
- Specific fuel oil weight at pumping temperature
- Tolerance

In case more than one engine is connected to the same fuel oil system, the pump capacity has to be increased accordingly.

Table 208: Simplified fuel oil booster pump dimensioning

The delivery height of the fuel oil booster pump is to be adjusted to the total resistance of the booster system.

		Injection system
		bar
Pressure difference between fuel inlet and outlet engine	+	2.00
Pressure loss of the flow balancing valve (to be installed only in multi-engine plants, pressure loss approximately 0.5 bar)	+	0.00
Pressure loss of the pipes, mixing tank – Engine mixing tank, e.g.	+	0.50
Pressure loss of the final heater heavy fuel oil max.	+	0.80
Pressure loss of the indicator filter	+	0.80
Operating delivery height of the booster pump	=	4.10

Table 209: Example for the determination of the expected operating delivery height of the fuel oil booster pump

It is recommended to install booster pumps designed for the following pressures:

Engines with conventional fuel oil injection system: Design delivery height 7.0 bar, design output pressure 10.0 bar.

Engines common rail injection system: Design delivery height 10.0 bar, design output pressure 14.0 bar.

VI-001/Viscosimeter

This device regulates automatically the heating of the final heater heavy fuel oil depending on the viscosity of the circulating fuel oil, to reach the viscosity required for injection.

H-004/Final heater heavy fuel oil

The capacity of the final heater shall be determined on the basis of the injection temperature at the nozzle, to which at least 4 K must be added to compensate for heat losses in the piping. The piping for both heaters shall be arranged for single and series operation.

Parallel operation with half the throughput must be avoided due to the risk of sludge deposits.

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T-008/Fuel oil damper tank

The injection nozzles cause pressure peaks in the pressurised part of the fuel oil system. In order to protect the viscosity measuring and control unit, these pressure peaks have to be equalised by a compensation tank. The volume of the pressure peaks compensation tank is 20 l.

Alternatively a metal bellow damper can be used in combination with an air cushion in the fuel oil mixing tank.

FIL-013/Fuel oil duplex filter

This filter is to be installed upstream of the engine, as close as possible to the engine.

The emptying port of each filter chamber should be fitted with a valve and a pipe to the sludge tank. If the filter elements are removed for cleaning, the filter chamber must be emptied. This prevents the dirt particles remaining in the filter casing from migrating to the clean oil side of the filter.

After changing the filter cartridge, the reconditioned filter chamber must be vented manually.

Design criterion is the filter area load specified by the filter manufacturer.

	Injection system
Filter mesh width (mm)	0.034
Design pressure	PN16

Table 210: Required filter mesh width (sphere passing mesh) – Injection system

FBV-010/Flow balancing valve

Heavy fuel oil supply system for only one main engine, without auxiliary engines

The flow balancing valve FBV-010 is not required.

Heavy fuel oil supply system for more than one main engine or/and additional auxiliary engines

The flow balancing valve at engine outlet is to be installed only (one per engine) in multi-engine arrangements connected to the same fuel system. It is used to balance the fuel flow through the engines. Each engine has to be fed with its correct, individual fuel flow.

FSH-001/Leakage fuel oil monitoring tank

High pressure pump overflow and escaping fuel oil from burst control pipes is carried to the monitoring tanks from which it is drained into the clean leakage fuel oil collecting tank. The engine interface connection of the monitoring tank is a pressureless drain. Leakage fuel oil flows by gravity only from these tank into collecting tanks (to be installed below the engine connections). To prevent pressure resistance at the engine interface connection the pipework to the collecting tank has to have a sufficient downward slope. The float switch mounted in the tanks must be connected to the alarm system. The classification societies require the installation of monitoring tanks for unmanned engine rooms. Lloyd's Register specifies tank monitoring for manned engine rooms as well.

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T-006/Leakage oil collecting tank

Dirty leak fuel and leak oil are collected in the leakage oil collecting tank. It must be emptied into the sludge tank. The content of the leakage oil collecting tank T-006 must not be added to the engine fuel. It can be burned for instance in a waste oil boiler.

The leakage oil collecting tank T-006 is only necessary if the leakages can not flow directly into the sludge tank T-021 by gravity.

MOV-017/Leakage fuel oil switch-over valve

Depending on the supplied fuel oil type, the operation leakage of the high-pressure injection system can be drained into the HFO or distillate clean leakage fuel oil tank by switching over this valve.

Note:
It must be ensured that no HFO contamination is led into the distillate clean leakage fuel oil tank.

T-071/Clean leakage fuel oil tank

High pressure pump overflow and other clean fuel oil that escapes from the injection system is led to the clean leakage fuel oil tanks.

From there the content of the distillate clean leakage fuel oil tank (1 T-071) must be emptied into the diesel fuel oil storage tank (T-015). The content of the HFO clean leakage fuel oil tank (2 T-071) must be emptied into the heavy fuel oil settling tank (T-016). The installation of these two clean leakage fuel oil tanks enables an effective separation of different fuel oil types.

The amount of clean operation leakage differs in a broad range, depending on the wear of the injection pumps, the type of fuel oil and the operating temperatures.

For data regarding the leak rate, see table(s) Leakage rate, Page 153.

A high flow of leakage fuel oil will occur in case of a pipe break, for short time only (< 1 min). Engine will give an alarm after a pipe break alarm.

Clean leakage fuel oil from the clean leakage fuel oil tanks 1,2 T-071 can be used again after passing the fuel oil separator.

Leakage fuel oil flows pressureless (by gravity only) from the engine into these tanks (to be installed below the engine connections). Pipe clogging must be avoided by trace heating and by a sufficient downward slope.

It must be ensured that the leakage fuel oil is well diluted with fresh fuel before entering the engine again. Nevertheless, leakage oil collecting tank T-006 is still required to collect lube oil leakages from lube oil drains (and other).

In case the described clean leakage fuel oil tank T-071 is installed, leakages from the following engine connections are to be conducted into this tank:

Engine type	Connection
L engine	5645
V engine	5645, 5646

Table 211: Connections clean leakage fuel oil tank

Withdrawal points for samples

Fuel oil sampling points are to be provided upstream and downstream of each filter, to verify the effectiveness of these system components.

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CV-003/Cooling water control valve for fuel oil cooler

The purpose of this valve is to allow a smooth temperature/viscosity adaption to avoid sludge formation due to a harsh temperature change during a fuel oil change over. This is achieved by controlling the cooling water flow through the fuel oil cooler.

HE-007/Fuel oil cooler
CK-003/Three-way valve (fuel oil cooler/by-pass)

The propose of the fuel oil cooler is to ensure that the viscosity of MDO/MGO will not become too fluid in engine inlet.

With the three-way valve (fuel oil cooler/by-pass) CK-003, the fuel oil cooler HE-007 has to be opened when the engine is switched from HFO to MDO/MGO operation.

That way, the MDO/MGO, which was heated while circulating via the injection pumps, is re-cooled before it is returned to the fuel oil mixing tank T-011. Switching on the fuel oil cooler may be effected only after flushing the pipes with MDO/MGO.

The cooler is cooled by LT cooling water.

Depending on the fuel oil type a chiller unit might be required to decrease the fuel oil temperature to reach the required injection system operation viscosity, see section Viscosity-temperature diagram (VT diagram), Page 282.

The thermal design of the cooler is based on the following data:

$P_c = P_1 \times br_{ISO} \times f_1$		
$Q_c = P_1 \times br_{ISO} \times f_2$		
Cooler outlet temperature MDO/MGO[1] T_{out} = 45 °C	T_{out}	°C
Dissipated heat of the cooler	P_c	kW
MDO flow for thermal dimensioning of the cooler[2]	Q_c	l/h
Engine output power at 100 % MCR	P_1	kW
Specific engine fuel oil consumption (ISO) at 100 % MCR	br_{ISO}	g/kWh
Factor for heat dissipation: f_1 = 2.68 x 10^{-5}	f_1	kWh/g
Factor for MDO/MGO flow: f_2 = 2.80 x 10^{-3}	f_2	l/g
Note: In case more than one engine, or different engines are connected to the same fuel system, the cooler capacity has to be increased accordingly.		
[1] This temperature has to be normally maximum 45 °C. Only for very light MGO fuel types this temperature has to be even lower in order to preserve the minimum admissible fuel oil viscosity in engine inlet (see section Viscosity-temperature diagram (VT diagram), Page 282). [2] The maximum MDO/MGO throughput is identical to the delivery quantity of the installed fuel oil booster pump.		

Table 212: Simplified fuel oil cooler dimensioning for engines without common rail (MAN 32/40, MAN 48/60B, MAN 51/60DF)

The recommended pressure class of the fuel oil cooler is PN16.

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FBV-013/Minimum flow valve

The minimum flow valve has to be installed in the plant. This valve is used to adjust the flushing flow to exchange the HFO supply system with fresh HFO every 12 hours.

It becomes necessary to avoid circulation of the same HFO content for a long period. Therefore the external main fuel oil system must be designed to ensure that the HFO content of the fuel system is completely exchanged with "fresh" HFO every 12 hours. This can be done by a return pipe from the booster system into the heavy fuel oil service tank.

In case the fuel supply system is filled with MGO, no fuel exchange is required. For that the manual valve prior the FBV-013 can be closed.

V-002/Shut-off cock

HFO supply system for only one main engine, without auxiliary engines

Shut-off cock V-002 is not required.

HFO supply system for more than one main engine or/and additional auxiliary engines

The stop cock is closed during normal operation (multi-engine operation). When one engine is separated from the fuel oil circuit for maintenance purposes, this cock has to be opened manually.

PCV-011/Fuel oil spill valve

HFO supply system for only one main engine, without auxiliary engines

Fuel oil spill valve PCV-011 is not required.

HFO supply system for more than one main engine or/and additional auxiliary engines

In case two engines are operated with one fuel oil module, it has to be possible to separate one engine at a time from the fuel oil circuit for maintenance purposes. In order to avoid a pressure increase in the pressurised system, the fuel oil, which cannot circulate through the shut-off engine, has to be rerouted via this valve into the return pipe. This valve is to be adjusted so that rerouting is effected only when the pressure, in comparison to normal operation (multi-engine operation), is exceeded. This valve should be designed as a pressure relief valve, not as a safety valve.

CF-002/Heavy fuel oil separator

See description in paragraph CF-002/Heavy fuel oil separator, Page 381.

CF-003/Diesel fuel oil separator

See description in paragraph CF-003/Diesel fuel oil separator, Page 367.

T-015/Diesel fuel oil storage tank

See description in paragraph T-015/Diesel fuel oil storage tank, Page 367.

T-016/Heavy fuel oil settling tank

See description in paragraph T-016/Heavy fuel oil settling tank, Page 380.

T-021/Sludge tank

See description in paragraph T-021/Sludge tank, Page 380.

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CV-004/Pilot fuel oil service tank filling valve

See description in section External fuel system – Pilot fuel supply system, Page 399.

T-101/Pilot fuel oil service tank

See description in section External fuel system – Pilot fuel supply system, Page 399.

FIL-033/Pilot fuel oil duplex filter

See description in section External fuel system – Pilot fuel supply system, Page 399.

Piping

The fuel oil system design pressure is PN16 (see section External pipe dimensioning, Page 298). The system is to be protected from higher pressure levels by corresponding safety valves. Additional safety valves, not displayed in the P&ID, might become necessary depending on the actual fuel oil system design.

Material

The casing material of pumps and filters should be EN-GJS (nodular cast iron), in accordance to the requirements of the classification societies.

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HFO supply system diagrams

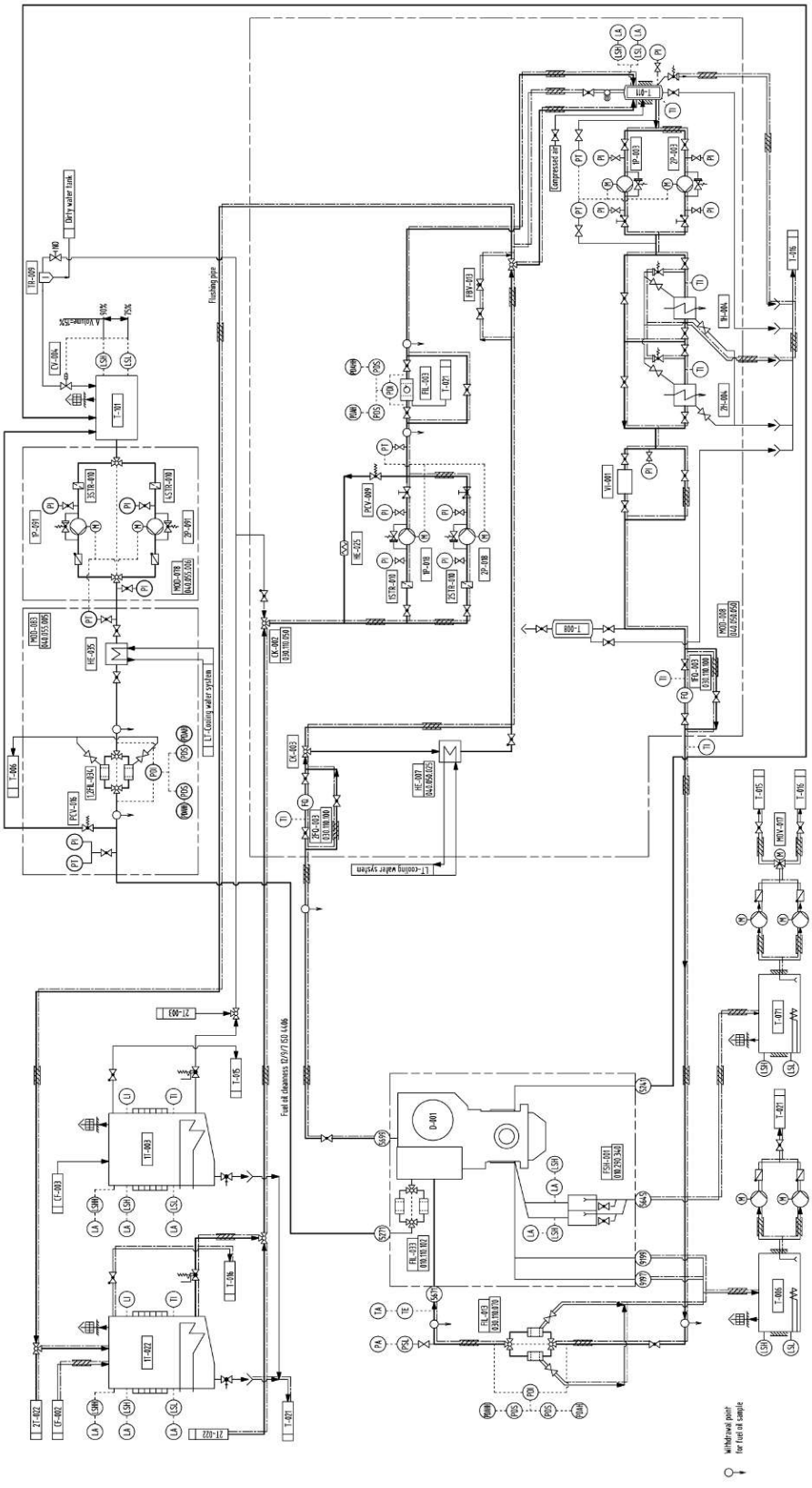

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Figure 143: HFO supply system diagram – Single engine plant

Components			
CF-002	Heavy fuel oil separator	MOV-017	Leakage fuel oil switch-over valve
CF-003	Diesel fuel oil separator	1,2 P-003	Fuel oil booster pump
CK-002	Three-way valve for fuel oil changeover	1,2 P-018	Fuel oil supply pump
CK-003	Three-way valve (fuel oil cooler/by-pass)	1,2 P-091	Pilot fuel oil supply pump
CV-004	Pilot fuel oil service tank filling valve	PCV-009	Pressure limiting valve
D-001	Diesel engine	PCV-016	Pilot fuel oil spill valve
FBV-013	Minimum flow valve	1,2,3,4 STR-010	Suction strainer
FIL-003	Fuel oil automatic filter, supply circuit	1,2 T-003	Diesel fuel oil service tank
FIL-013	Fuel oil duplex filter	T-006	Leakage oil collecting tank
FIL-033	Pilot fuel oil duplex filter	T-008	Fuel oil damper tank
FIL-034	Pilot fuel oil duplex filter	T-011	Fuel oil mixing tank
1,2 FQ-003	Fuel oil flowmeter	T-015	Diesel fuel oil storage tank
FSH-001	Leakage fuel oil monitoring tank	T-016	Heavy fuel oil settling tank
1,2 H-004	Final heater heavy fuel oil	T-021	Sludge tank
HE-007	Fuel oil cooler	1,2 T-022	Heavy fuel oil service tank
HE-025	Fuel oil cooler, supply circuit	T-071	Clean leakage fuel oil tank
HE-035	Pilot fuel oil cooler	T-101	Pilot fuel oil service tank
MOD-008	Fuel oil module	TR-009	Coalescer (water trap)
MOD-078	Pilot fuel oil supply pump module	VI-001	Viscosimeter
MOD-083	Pilot fuel oil filter module		
Major engine connections			
5241	Leakage fuel oil drain pilot fuel-CR	5699	Fuel oil return pipe from engine
5271	Fuel oil inlet pilot fuel-CR	9197	Dirty oil drain from covering, coupling side
5645	Fuel oil break leakage drain (reusable) 1	9199	Dirty oil drain from covering, counter coupling side
5671	Fuel oil inlet on the engine		

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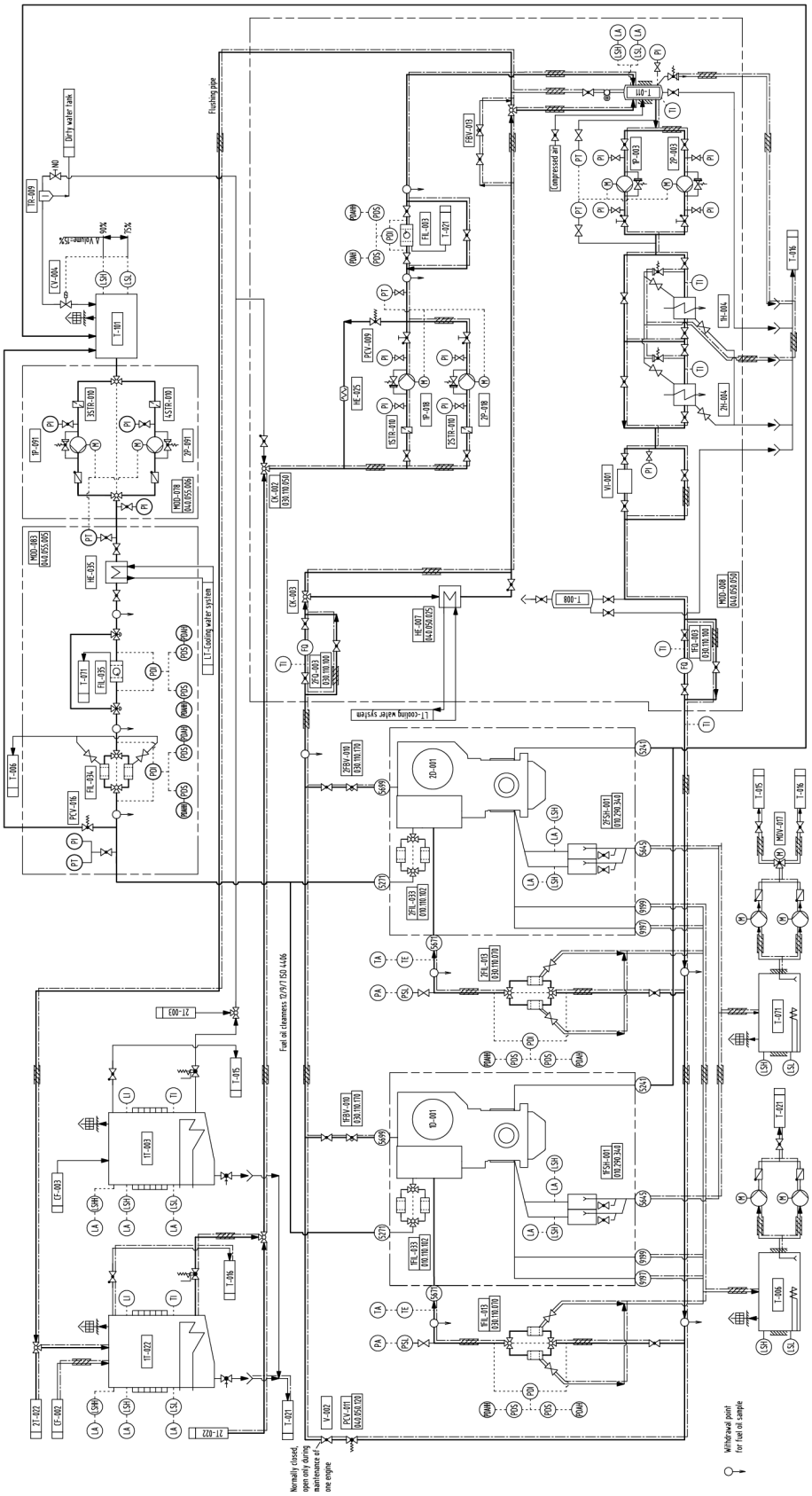

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Figure 144: HFO supply system diagram – Twin engine plant

Components			
CF-002	Heavy fuel oil separator	MOV-017	Leakage fuel oil switch-over valve
CF-003	Diesel fuel oil separator	1,2 P-003	Fuel oil booster pump
CK-002	Three-way valve for fuel oil changeover	1,2 P-018	Fuel oil supply pump
CK-003	Three-way valve (fuel oil cooler/by-pass)	1,2 P-091	Pilot fuel oil supply pump
CV-004	Pilot fuel oil service tank filling valve	PCV-009	Pressure limiting valve
1,2 D-001	Diesel engine	PCV-011	Fuel oil spill valve
1,2 FBV-010	Flow balancing valve	PCV-016	Pilot fuel oil spill valve
FBV-013	Minimum flow valve	1,2,3,4 STR-010	Suction strainer
FIL-003	Fuel oil automatic filter, supply circuit	1,2 T-003	Diesel fuel oil service tank
1,2 FIL-013	Fuel oil duplex filter	T-006	Leakage oil collecting tank
1,2 FIL-033	Pilot fuel oil duplex filter	T-008	Fuel oil damper tank
FIL-034	Pilot fuel oil duplex filter	T-011	Fuel oil mixing tank
FIL-035	Pilot fuel oil automatic filter	T-015	Diesel fuel oil storage tank
1,2 FQ-003	Fuel oil flowmeter	T-016	Heavy fuel oil settling tank
1,2 FSH-001	Leakage fuel oil monitoring tank	T-021	Sludge tank
1,2 H-004	Final heater heavy fuel oil	1,2 T-022	Heavy fuel oil service tank
HE-007	Fuel oil cooler	T-071	Clean leakage fuel oil tank
HE-025	Fuel oil cooler, supply circuit	T-101	Pilot fuel oil service tank
HE-035	Pilot fuel oil cooler	TR-009	Coalescer (water trap)
MOD-008	Fuel oil module	V-002	Shut-off cock
MOD-078	Pilot fuel oil supply pump module	VI-001	Viscosimeter
MOD-083	Pilot fuel oil filter module		
Major engine connections			
5241	Leakage fuel oil drain pilot fuel-CR	5699	Fuel oil return pipe from engine
5271	Fuel oil inlet pilot fuel-CR	9197	Dirty oil drain from covering, coupling side
5645	Fuel oil break leakage drain (reusable) 1	9199	Dirty oil drain from covering, counter coupling side
5671	Fuel oil inlet on the engine		

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5.8.7 External fuel system – Pilot fuel supply system

General

The pilot fuel oil supply system is an open system with open deaeration pilot fuel oil service tank. Usually one or two engines are connected to one pilot fuel oil supply system, see figure(s) HFO supply system diagram(s), Page 395. Each cylinder of the engine is equipped with two injection nozzles, the pilot fuel oil nozzle and the main fuel oil nozzle.

Diesel fuel oil viscosity

As pilot fuel oil only MGO (DMA, DFA) or MDO (DMB, DFB) according to ISO 8217 is permissible (see section Specification of diesel fuel used as pilot oil for DF engines, Page 269).

Pilot fuel oil

The pilot fuel oil nozzles are part of the pilot fuel oil common rail system. In gas mode this system is used to ignite the gaseous fuel. For this purpose MGO (DMA, DFA) or MDO (DMB, DFB) is used. Pilot fuel oil nozzles are designed to operate with very small fuel oil quantities in order to minimise the pilot fuel oil consumption. Also in liquid fuel oil mode pilot fuel oil is injected for cooling the nozzles of the pilot fuel oil injectors. As a safety function, in case of a failure of the pilot fuel oil system, the engine can be operated in liquid fuel oil mode without pilot fuel oil (back up mode).

Without further pilot fuel oil injection, cooling of the pilot fuel oil nozzles is missing. With the low pilot fuel oil pressure, there is a danger that the combustion pressure could flow back into the injector. In both cases the injector will be damaged after a few operating hours. Back up mode should only be used at emergency conditions and as short as possible.

The engine has two pilot fuel oil connections, the pressurised pilot fuel oil inlet and the pressureless pilot fuel oil outlet. Non-burned fuel oil and leakage fuel oil from the pilot fuel oil nozzles is circulated via the pilot fuel oil outlet connection to the pilot fuel oil service tank.

The leakage fuel oil flows pressure less (by gravity only) from the engine into the pilot fuel oil service tank (to be installed below the engine connections). Pipe resistance and clogging must be avoided by a sufficient downward slope.

T-101/Pilot fuel oil service tank

The pilot fuel oil service tank, installed on the pilot fuel oil return pipe, has to be designed for a content of min. 200 l for each connected engine. At the engine outlet the pilot fuel oil is pressureless. Therefore the pilot fuel oil return pipe between the engine and the pilot fuel oil collecting tank has to be installed with a downward slope. Filling of the tank is to be governed by fuel level switches. A difference of 15% of the total tank volume between filling start and stop is to be established. The filling of the pilot fuel oil service tank should be done with well separated fuel from the diesel fuel oil service tank. The tank is filled by hydrostatic pressure of the diesel fuel oil service tank or a supply pump.

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CV-004/Pilot fuel oil service tank filling valve

The valve is installed if hydrostatic filling of the pilot fuel oil service tank is applied. The valve must be operated automatically depending on the filling level of the pilot fuel oil service tank.

STR-010/Suction strainer

To protect the fuel supply pumps, an approximately 0.25 mm gauge (sphere-passing mesh) strainer is to be installed at the suction side of each supply pump.

P-091/Pilot fuel oil supply pump

The pilot fuel oil supply pump shall keep sufficient fuel pressure before the engine mounted pilot fuel oil high pressure pump.

The pilot fuel oil supply pump has to be designed for a flow of 130 l/h for each connected L engine and 250 l/h for each connected V engine, including margins for:

- Tropical conditions
- Realistic heating value
- Tolerance

In case more than one engine is connected to the same pilot fuel oil system, the pump capacity has to be increased accordingly. The delivery height shall be selected with reference to the system losses and the pressure required before the engine (see section Planning data, Page 100). Normally the required delivery height is 10 bar.

The pilot fuel oil pumps have to be installed on the same level as the pilot oil service tank T-101. Furthermore the pumps shall be installed as close as possible to the pilot oil service tank T-101.

HE-035/Pilot fuel oil cooler

The pilot fuel oil cooler is required to cool down the fuel oil, which was heated up while circulating through the high pressure pilot fuel oil injection system. The pilot fuel oil cooler is normally connected to the LT cooling water system and should be dimensioned so that the MGO does not exceed a temperature of max. 45 °C.

The thermal design of the pilot fuel oil cooler is based on the following data:

Pilot fuel outlet temperature	≤ 45 °C
Dissipated heat of the pilot fuel oil cooler	
Standard operation	min. 0.05 kW/cyl.
Only in case of an activation of the pressure limiting valve of the pilot fuel system by the safety system	L engine: 1.6 kW per engine V engine: 3.5 kW per engine

Table 213: Dimensioning of the pilot fuel oil cooler

The max. MGO volume flow is identical to the delivery quantity of the installed pilot fuel oil supply pump P-091. The recommended pressure class of the pilot fuel oil cooler is PN16.

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FIL-035/Pilot fuel oil automatic filter

The pilot fuel oil automatic filter must be equipped with differential pressure indication and switches. The filter must be designed for automatic cleaning in case of exceeding a specific differential pressure. The back flushing oil and particles shall leave the filter by a separate pipe.

Pilot fuel oil injection system	
Filter mesh width (mm)	0.015
Design pressure	PN10

Table 214: Required filter mesh width (sphere passing mesh) – Pilot fuel oil automatic filter

Design criterion is the filter area load specified by the filter manufacturer. The pilot fuel oil automatic filter has to be installed in the plant. It is not attached to the engine.

FIL-034/Pilot fuel oil duplex filter

To ensure high fuel oil quality (see section Pilot fuel, Page 257) this filter has to be designed as a depth filter. The pilot fuel oil duplex filter is to be installed upstream of the engine, the emptying port of each filter chamber is to be fitted with a valve and a pipe to the leakage oil collecting tank (T-006).

If the filter elements are removed for replacing, the filter chamber must be emptied. This prevents the dirt particles remaining in the filter casing from migrating to the clean oil side of the filter. Design criterion is the filter area load specified by the filter manufacturer.

Pilot fuel oil injection system	
Filter mesh width (mm)	0.010
Design pressure	PN16

Table 215: Required filter mesh width (sphere passing mesh)

TR-009/Coalescer

The coalescer is equipped with a water level sensor. If the water reaches a certain level an alarm will be triggered and the coalescer can be emptied to the leakage oil collecting tank T-006 by switching to the redundant coalescer and opening the manual drain valve.

The coalescer is equipped with a fine filter stage. To ensure a high fuel quality at the engine inlet the filter is designed as a depth filter. Before changing the filter element, the housing has to be drained analog the water draining.

Pilot fuel oil injection system	
Filter mesh width (mm)	0.001
Design pressure	PN16

Table 216: Required filter mesh width (sphere passing mesh)

PCV-016/Pilot fuel oil spill valve

This valve is used for setting the required pressure before the pilot fuel oil high pressure pump. It should be designed as a pressure relief valve, not as a safety valve.

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FIL-033/Pilot fuel oil duplex filter

This filter (designed as a depth filter) is attached on the engine. The emptying port of each filter chamber is to be fitted with a valve and a pipe to the sludge tank. If the filter elements are removed for replacing, the filter chamber must be emptied. This prevents the dirt particles remaining in the filter casing from migrating to the clean oil side of the filter. Design criterion is the filter area load specified by the filter manufacturer.

Pilot fuel oil injection system	
Filter mesh width (mm)	0.001
Design pressure	PN40

Table 217: Required filter mesh width (sphere passing mesh)

Piping

The fuel oil system design pressure is PN16 (see section External pipe dimensioning, Page 298). The system is to be protected from higher pressure levels by corresponding safety valves. Additional safety valves, not displayed in the P&ID, might become necessary depending on the actual fuel oil system design.

Material

The casing material of pumps and filters should be EN-GJS (nodular cast iron), in accordance to the requirements of the classification societies.

5.8.8 External fuel system – Fuel oil supply at blackout conditions

As the main electrical grid is not available during a blackout, an alternative energy source has to guarantee fuel oil supply. If a sufficient uninterruptible power supply (UPS) system is available, it can be connected to the regular fuel oil supply pumps and run them in spite of blackout.

Alternatively an additional pneumatic pump can be installed. If this pump is connected to a working air system, it must be ensured that this system can always deliver sufficient compressed air required to outlast the blackout operation.

Also the starting air system can be used, if the additional air is considered for design of starting air receivers and the adequate control of the blackout pump is implemented in the ship automation system. Background is that the amount of compressed air required by class societies for engine starts must not be affected. MAN Energy Solutions can design a suitable pneumatic pump and calculate its compressed air consumption.

For a short time the engines can also run by use of a gravity fuel oil tank (MDO/MGO) or in a HFO system by the air pressure cushion in the fuel oil mixing tank (see required pressure in section Operating/service temperatures and pressures, Page 147.

Duration of blackout operation

Duration of the blackout pump operation should last till the regular fuel supply is recovered:

- Duration of the emergency GenSet for connecting to the main electrical grid
- Start-up time of the fuel oil module after main grid is restored

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- Buffer time

On the other hand, the duration of the blackout pump operation should be limited by the ship automation system due to:

- Reduction of UPS or compressed air consumption
- Consideration of engine related systems without power supply (e.g. cooling water system might overheat)

Depending on engine load it can be advisable to schedule blackout operation to maximum 90 seconds.

Integration in fuel oil system

In a diesel fuel oil supply system it is recommended to integrate the blackout pump parallel to the regular fuel oil supply pumps. In order to reduce compressed air consumption, it is possible to choose a downsized pump and operate the engine in part load.

For a heavy fuel oil supply system a pneumatic pump delivers fuel oil from MDO service tank into the mixing tank to guarantee low load operation. For high-load operation contact MAN Energy Solutions.

Note:
A fuel oil supply with cold MDO/MGO shortly after HFO-operation will lead to temperature shocks in the injection system and has to be avoided under any circumstances.

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5.9 Fuel gas system

5.9.1 Internal fuel gas system

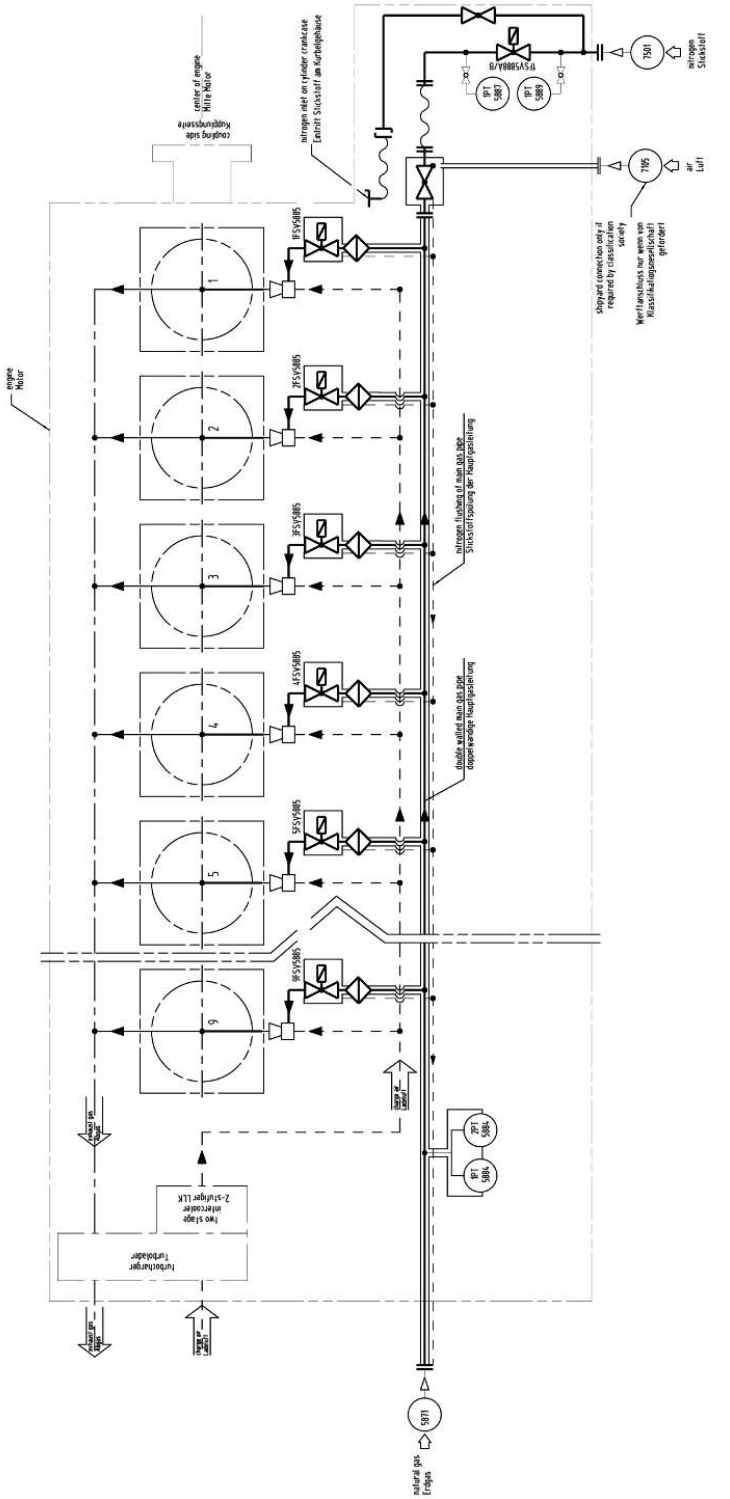


Figure 145: Internal fuel gas system MAN 51/60DF – Exemplary

5.9.2 External fuel gas system – Fuel gas supply system

Note:
The Gas Valve Unit is a crucial part of the MAN Energy Solutions safety concept. Therefore it needs to be sourced via MAN Energy Solutions only in order to ensure the accordance to the technical requirements of a gas valve unit as defined by MAN Energy Solutions incl. project specific boundary conditions.

The external gas supply system is necessary to feed the dual fuel engine with fuel gas according to the requirements of the engine. It consists of:

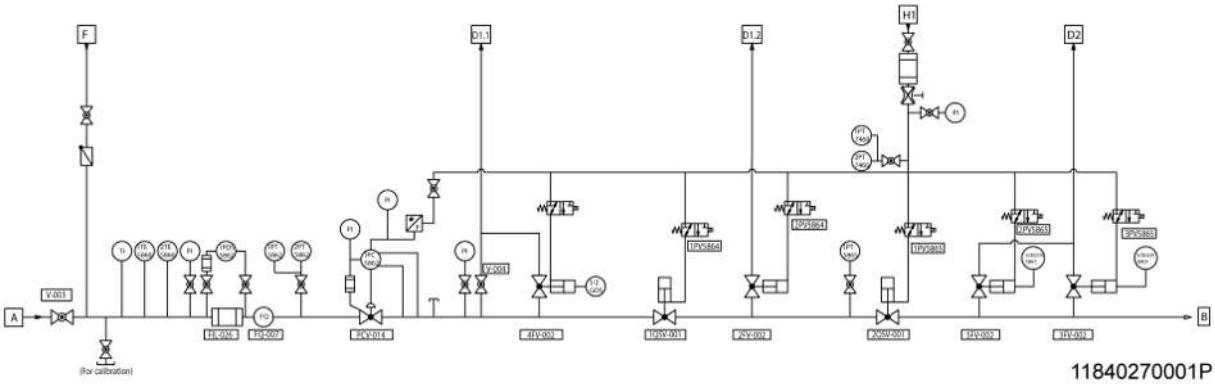
- The plant related fuel gas supply system
- The gas valve unit with connection pipes

The plant related fuel gas supply system provides gas with the correct conditions at the inlet of the gas valve unit.

The pressure and the temperature of the fuel gas supplied to the GVU shall be in the range as specified in section Specifications and requirements for the gas supply of the engine, Page 154. The temperature- and pressure-dependent dew point of natural gas must be exceeded to prevent condensation.

If the pressure of the fuel gas supplied to the GVU exceeds the permissible range as stated in section Specifications and requirements for the gas supply of the engine, Page 154, a safety valve has to be installed on the GVU to protect the engine against excessive pressure. In any case the maximum design pressure of the GVU system shall not be exceeded.

MOD-052/Gas valve unit



Components

FIL-026	Gas filter
FQ-007	Gas flow meter
2,3,5 FV-002	Automatic venting valve
4 FV-002	Automatic venting valve (optional)
MOD-052	Gas valve unit (GVU)
PCV-014	Gas pressure control valve
1,2 QSV-001	Quick action stop valve
V-003	Gas shut-off valve, manual

Connections

A	Gas inlet
B	Gas outlet
D1.1/D1.2/ D2	Gas venting
F	Inert gas
H1	Compressed air

Figure 146: Gas valve unit (GVU)

The gas valve unit (MOD-052) is a regulating and safety device permitting the engine to be safely operated in the gas mode. The unit is equipped with block and bleed valves (quick-acting stop valves and venting valves) and a gas pressure regulating device.

The gas valve unit fulfils the following functions:

- Gas leakage test by engine control system before engine start
- Control of the pressure of the gas fed into the dual fuel engine
- Quick stop of the gas supply at the end of the DF-operation mode
- Quick stop of the gas supply in case of an emergency stop
- Purging of the gas distribution system and the feed pipe with N_2 after DF-operation
- Purging with N_2 for maintenance reasons

In order to keep impurities away from the downstream control and safety equipment, a gas filter (FIL-026) is installed after the hand-stop valve (V-003). The maximum mesh width (absolute, sphere-passing mesh) of the gas filter (FIL-026) must be 0.005 mm. The pressure loss at the filter is monitored by a differential pressure gauge.

The gas pressure control device (PCV-014) adjusts the pressure of the gas fed into the engine. The control devices include a regulating valve with pressure regulator and an IP transducer.

In accordance with the engine load, the pressure control device maintains a differential gas over pressure to the charge air pressure. This ensures that the gas feed pressure is correct at all operating points.

At the outlet of the gas control line, quick-acting stop valves (1, 2 QSV-001) and automatic venting valves (2, 3, 4, 5 FV-002) are mounted. The quick-acting stop valves will interrupt the gas supply to engine on request. The automatic venting valve (2 FV-002) relieves the pressurised gas trapped between the two closed quick-acting stop valves (1, 2 QSV-001). The automatic venting valves (3, 5 FV-002) relieves the pressurised gas trapped between the quick-acting stop valves (2 QSV-001) and the engine and is used to purge the gas distribution system and pipe with N_2 in inverse direction.

A venting valve (4 FV-002) for purging the gas supply pipe in front of the GVU can be supplied optional. This valve is not controlled by the engine control system and has to be controlled by ship automation system or similar.

For safety reasons, the working principle of the quick-acting stop valves (1, 2 QSV-001) ensures that the valves are normally closed (closed in case there is no signal) while the venting valves (2, 3, 5 FV-002) are normally open.

The gas valve unit includes pressure transmitters/gauges and a thermocouple. The output of these sensors is transmitted to the engine management system. The control logic meets MAN Energy Solutions requirements and controls the opening and closing of the block and bleed valves as well as the gas-control-line leak test.

See also section Media interfaces, Page 307 for details on gas system on engine.

Gas valve unit room/enclosure

The gas valve unit is to be installed in a separate room meeting the following requirements:

- Gas tight compartment Installation of a fire detection and fire fighting system

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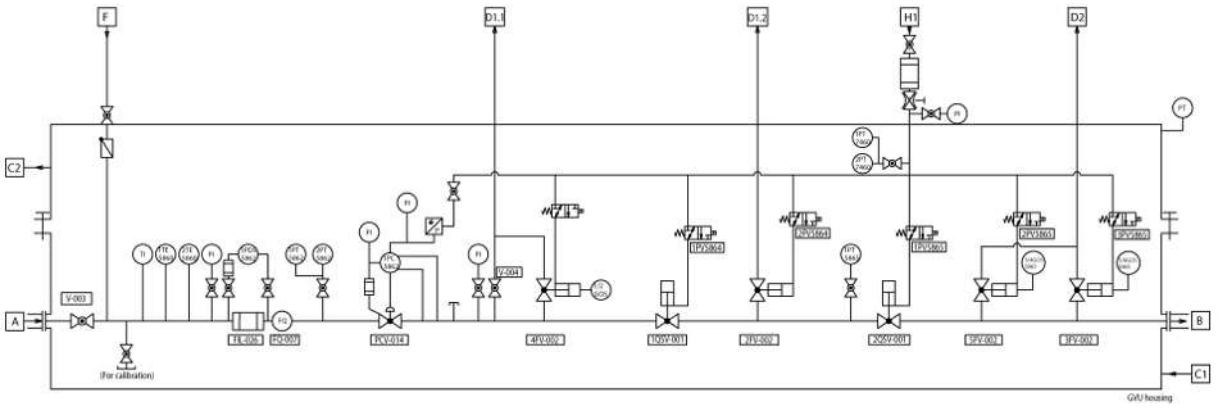
- Installed room ventilation system with exhaust air fan to outside area. This ensures that there is always a lower pressure in this room in comparison to the engine room
- Installation of a gas detection system
- Installation of a fire detection and fire fighting system

As an alternative for an installation in a GVU room the GVU can be equipped with a dedicated enclosure/housing. In this way the GVU can be installed directly beside the engine in the machinery space. The safety and operation principal is analog to the GVU room installation.



	Components		
FIL-026	Gas filter	MOD-052	Gas valve unit (GVU)
FQ-007	Gas flow meter (optional)	PCV-014	Pressure control valve
2,3,5 FV-002	Automatic venting valve	1,2 QSV-001	Quick action stop valve
4 FV-002	Automatic venting valve (optional)	V-003	Gas shut-off valve, manual
	Connections		
A	Gas inlet	D1.1/D1.2/ D2	Gas venting
B	Gas outlet	F	Inert gas
C	Ventilation	H1	Compressed air

Figure 147: Gas valve unit with housing

Gas piping and ventilation

The GVU shall be located as close as possible to the engine to achieve optimal control behaviour. Therefore the maximum length of the piping between GVU and engine inlet is limited to 20 metres. The material for manufacturing the supply gas piping from the GVU to the engine inlet must be stainless steel.

The inner diameter of the gas piping between GVU and engine shall be equal to the piping diameter of the GVU outlet to minimise pressure losses.

The gas supply pipe between the gas valve unit and the engine gas inlet connection is to be of double-wall design or a pipe in a separate duct. The interspace between the two pipes (or between pipe and duct) is to be connected to the gas valve unit room/housing. A gas detection for the interspace is to be installed, and a ventilation system ensuring that the air is exchanged at least 30 times per hour is required.

If for integration reasons the double wall supply piping presents low points (siphons), particular construction attention shall be paid for avoiding eventual accumulation of condensation water between the internal and external piping which might obstruct the ventilation.

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To achieve a 30 times air exchange inside the GVU room/enclosure in generally a lower difference pressure is required compared to 30 times ventilation of the double-walled piping. Therefore it is recommended to install a adjustable flow restrictor at the ventilation air inlet of the GVU housing. With this flow restrictor the negative pressure inside the housing can be set for appropriate ventilation of the double-walled piping. In case of a GVU room with a big volume it is recommended to install separate ventilation equipment for piping and GVU compartment. Contact MAN Energy Solutions for support.

In the following it is shown how to determine the necessary air flow for a min. 30 times air exchange of the double walled piping and the GVU compartment. Furthermore it is shown how to estimate the dedicated necessary negative pressure within the GVU compartment.

Necessary min. air flow through double walled piping and GVU compartment

Air flow through piping min.:
$V_{dw} = 30 \times (V_{dw\ engine} + V_{dw\ plant})/h$

No. of cylinders, config.	6L	7L	8L	9L	12V	14V	16V
Volume of annular space on engine piping $V_{dw\ engine}$ in litres	72	80	88	96	171	191	211

Table 218: Volume of annular space on engine piping $V_{dw\ engine}$ in litres

Volume of annular space on plant piping between GVU compartment and engine $V_{dw\ plant}$ has to be calculated by customer.

The volume of the air suction line has not be considered for the 30-times air exchange rate.

Air flow through GVU compartment min.:
$V_{GVU} = 30 \times V_{GVU}/h$

Necessary min. negative pressure at double walled piping on engine inlet

To generate the necessary min. 30 times air exchange, a certain negative pressure has to be adjusted by the ventilation system inside the GVU compartment.

The pressure loss over the engine (incl. expansion bellow) is shown in figure Relation of double walled space and pressure loss over the engine, Page 409. It is depending on the annular space volume of the plant piping $V_{dw\ plant}$.

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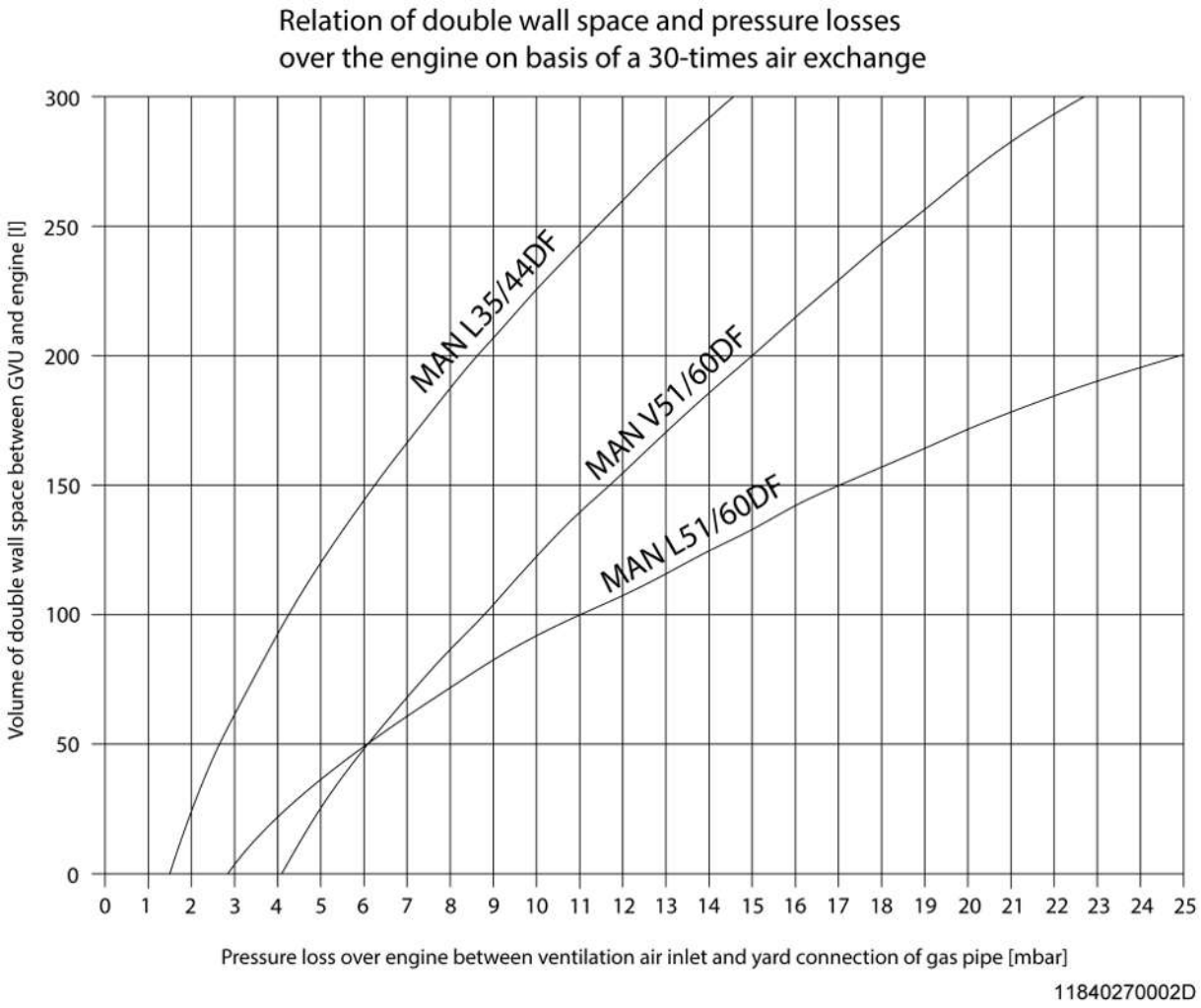

Figure 148: Relation of double walled space and pressure loss over the engine

For estimating the negative pressure to be set inside the GVU compartment the pressure loss on the plant side piping has to be added. If the air is taken from outside an additional pressure loss has to be considered for the air inlet piping. The pressure losses can be calculated by the costumer with the air flow through the piping V_{dw}.

With the air flow through GVU compartment and the necessary negative pressure inside GVU compartment, the ventilation equipment can be designed. (Ventilation of double walled inlet piping to GVU not considered). To adjust the correct pressure inside the GVU compartment an additional ventilation inlet piping for GVU compartment with an appropriate throttle valve will be necessary.

Appropriate equipment for monitoring the air exchange on the engine has to be installed, e.g. pressure monitoring before and after engine in annular space.

If not achieving the correct pressure and flow conditions within the annular space and the GVU compartment an alarm shall occur and after a delay an external QCO has to be initiated by the ventilation control or ship automation system.

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If an air suction line is connected to the annular space on the engine, the line has to be made of stainless steel to avoid intake of rusty particles into the system. The line has to be steadily ascending to ensure free pass of the air flow (no condensate trapped). For further requirements refer to relevant classification rules.

Ventilation of double walled pipes – Condensation has to be prevented

For marine applications, the air for ventilation of the double wall pipes is drawn in from a zone-free atmosphere. This is subject to various conditions with regard to temperature and absolute humidity.

It must be prevented, that the moisture of the ventilation air condenses in between the double wall of the pipes. This danger exists especially in areas, where the air hits the surface of the cold, inner fuel gas pipe. If the fuel gas (and thus the inner fuel gas line) is colder than the dew point of the ventilation air, water will condense on the surface of the inner fuel gas line and accumulate inside the double-walled pipes. This will lead to component damages.

Therefore it has to be ensured, that the temperature of the fuel gas according to figure Minimum gas temperatures for humid air, Page 410 does not fall below the dew point of the corresponding ventilation air. The expected moisture content of the ventilation air has to be estimated. Then the system (evaporator and heater capacities) must be designed for the fuel gas temperature in accordance with the expected conditions, so that the minimum fuel gas temperature is always above the dew point of the ventilation air. If this is not possible and the fuel gas temperature is below the curve shown, the ventilation air must be conditioned, that under no circumstances water condensation will occur inside the double-walled pipes (moisture reduction).

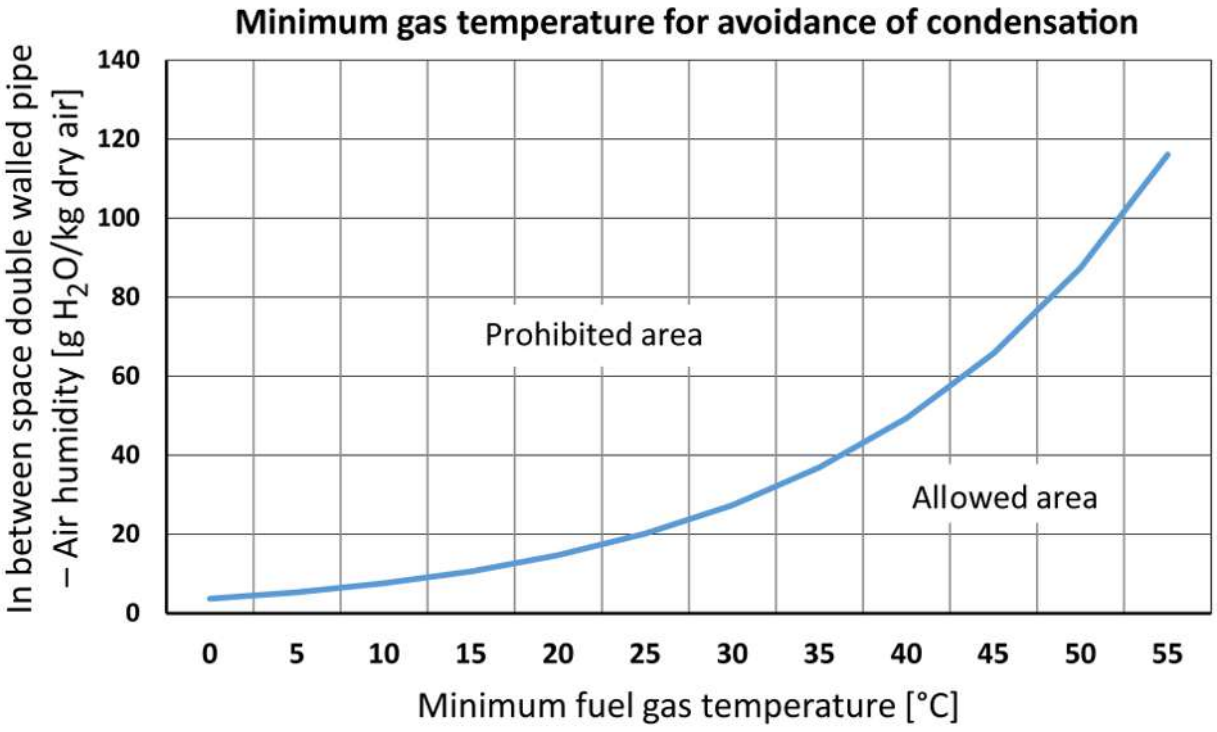

Figure 149: Minimum gas temperatures for humid air

Safety concept

For further information on the installation of the gas supply system and the gas valve unit refer to our brochure "*Safety Concept – Marine dual fuel engines*".

Inert gas system

To secure the gas supply line on the DF engine from an explosive atmosphere up to the block valve 2 QSV-001 of the GVU, the piping will be automatically purged with inert gas after each normal or quick change over from gas mode to liquid fuel mode and each emergency shutdown from gas mode. Therefore an inert gas purge valve is installed on the DF engine.

In common during purge mode, the purge valve on the engine and the venting valves 3- and 5 FV-001 will be opened and inert gas can purge the remaining gas downstream out to the gas venting installation.

After the purge time (depending on the length of the gas piping from GVU to DF engine manifold), 3 FV-001, 5 FV-001 and the inert gas purge valve will be closed. During preparation to gas mode, the purge mode has to start once again to secure an air-free gas piping.

In case depressurising of the gas pipe between GVU and engine is not possible due to a malfunction (e.g. sticking venting valve), the limited fuel gas volume trapped in the pipe could slowly pass through the not completely tight gas admission valves and can so reach the charge air manifold and from here, if the engine is not in operation, the engine room or, trough open cylinder inlet and outlet valves, the exhaust gas system. To minimise this risk, redundant venting valves 3 FV-001 and in parallel 5 FV-001 are installed on the GVU.

In case the nitrogen purging system fails, the gas pipe is once purged with charge air and a gas blocking is set. Thus an explosive atmosphere can only occur seldom and for short periods. Operation in gas mode is only possible if Nitrogen pressure is available and gas blocking alarm has been reset by operator.

The crankcase has to be also purged with nitrogen in case of doing maintenance in this space. For further information check the dedicated working sheet. If using nitrogen as inert gas the purity has to be minimum 95% nitrogen to ensure to be safely below the limiting oxygen concentration if mixed with fuel gas.

The nitrogen supply pressure shall be min. 5.5 bar and max. 6.5 bar at engine inlet. The nitrogen supply pressure shall be stable in any operating condition. Also during purging, the nitrogen supply pressure shall not fall below the required minimum value of 5.5 bar.

For supplying nitrogen to the engine either high-pressure bottle batteries or nitrogen generators with buffer vessels can be used. The design of this equipment is depending on the number of switch-overs from gas to diesel mode and vice versa as the gas piping is purged with every switch-over. For a guidance value of the necessary nitrogen amount which shall be held available in a pressurised vessel before the engine following formula can be used:

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$V_{N2} = 2 \times N_{SO} \times N_{EX} \times (V_{engine} + V_{plant}) \times P_{LL\ max}/bar$

V_{N2}	Nitrogen volume in Nm^3 per engine
Factor 2	Post- and pre-purging operation
N_{SO}	Number of switch-over operations to be operated without waiting time in between
N_{EX}	Number of volume exchanges in piping (3 to 5 is recommended)
V_{engine}	Volume of inner engine piping
V_{plant}	Volume of inner plant piping between GVU and engine
$P_{LL\ max}$	Maximum relative charge air pressure

The design of the nitrogen equipment e.g. a nitrogen generator is depending on the required filling time of the pressurised nitrogen vessel in front of the engine and other nitrogen consumers.
The max. allowable temperature range of the nitrogen at the interface (or shipyard connection) to the engine is 5 – 50 °C.

No. of cylinders, config.	**6L**	**7L**	**8L**	**9L**	**12V**	**14V**	**16V**
Volumes of inner engine piping in litres	79	89	98	107	187	209	231

Table 219: Volumes of inner engine piping in litres

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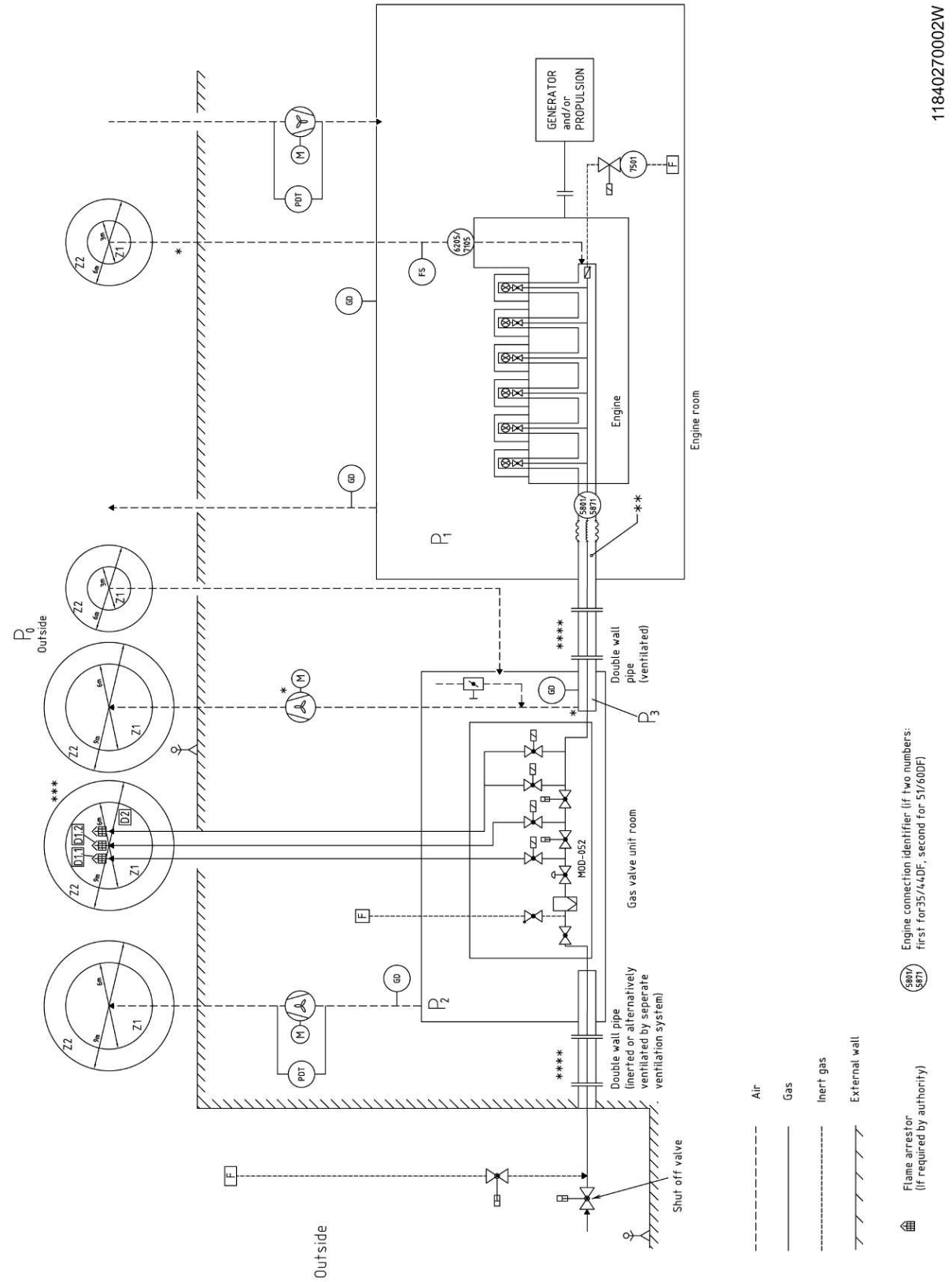

Figure 150: Fuel gas supply system diagram – Engine room arrangement

Components			
MOD-052	Gas valve unit (details see figure Gas valve unit (GVU), Page 405)	D1.1, D1.2, D2	Gas venting
F	Inter gas inlet	GD	Gas detector: Exact number, position, type and set point of gas detectors to be agreed with the authority and according local surrounding conditions.
Additional information			
$P_1 > P_0 > P_2 > P_3$	Ambient pressures		

* Air inlet to annular space from areas which would be non-hazardous in absence of the air inlet. Air inlet inside or outside according requirements of classification society. If inside gas detector has to be installed at air inlet opening. MAN Energy Solutions recommends to install separate fans for the ventilation of GVU room and piping as the pressure loss on the engine side double-walled space will be significantly higher than the required negative pressure to achieve a 30-times air exchange inside the GVU room. Therefore double-walled space has to be closed after entry into the GVU room.
The monitoring of the 30-times air exchange rate on the engine is done by dedicated pressure transmitters. For design of the ventilation equipment refer to the dedicated Project Guide or contact MAN Energy Solutions.

** Gas pipe between gas valve unit and engine to be made of stainless steel. Length of this pipe to be as short as possible, maximum 20 m. The volume of the double wall space has to be designed as small as possible.

*** Hazardous area on open deck at venting pipe outlet as well as at outlets and inlets for ventilation pipes of GVU room and annular space (spherical). Gas venting pipes shall not be merged according MAN Energy Solutions. In any case a backflow or back-pressure in any venting pipe during a venting or purging process over a connected pipe is not permitted. Therefore the collector pipe has to be sized sufficiently. Single non-return devices installed in venting pipes or other fittings which could block the venting pipe in case for a failure are not allowed. Gas venting pipes can be merged if accepted by authority.

**** There have to be installed short, removable gas pipe segments directly upstream and downstream of the gas valve unit to ensure visual inspection and pressure testing capability.

5.10 Compressed air system

General

The engine requires compressed air for starting, for activation of control devices and for activation of the optional jet assist.

Compressed air quality

The compressed air, supplied to the engine, must meet the requirements given in section Specification of compressed air, Page 293.

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5.10.1 Internal compressed air system

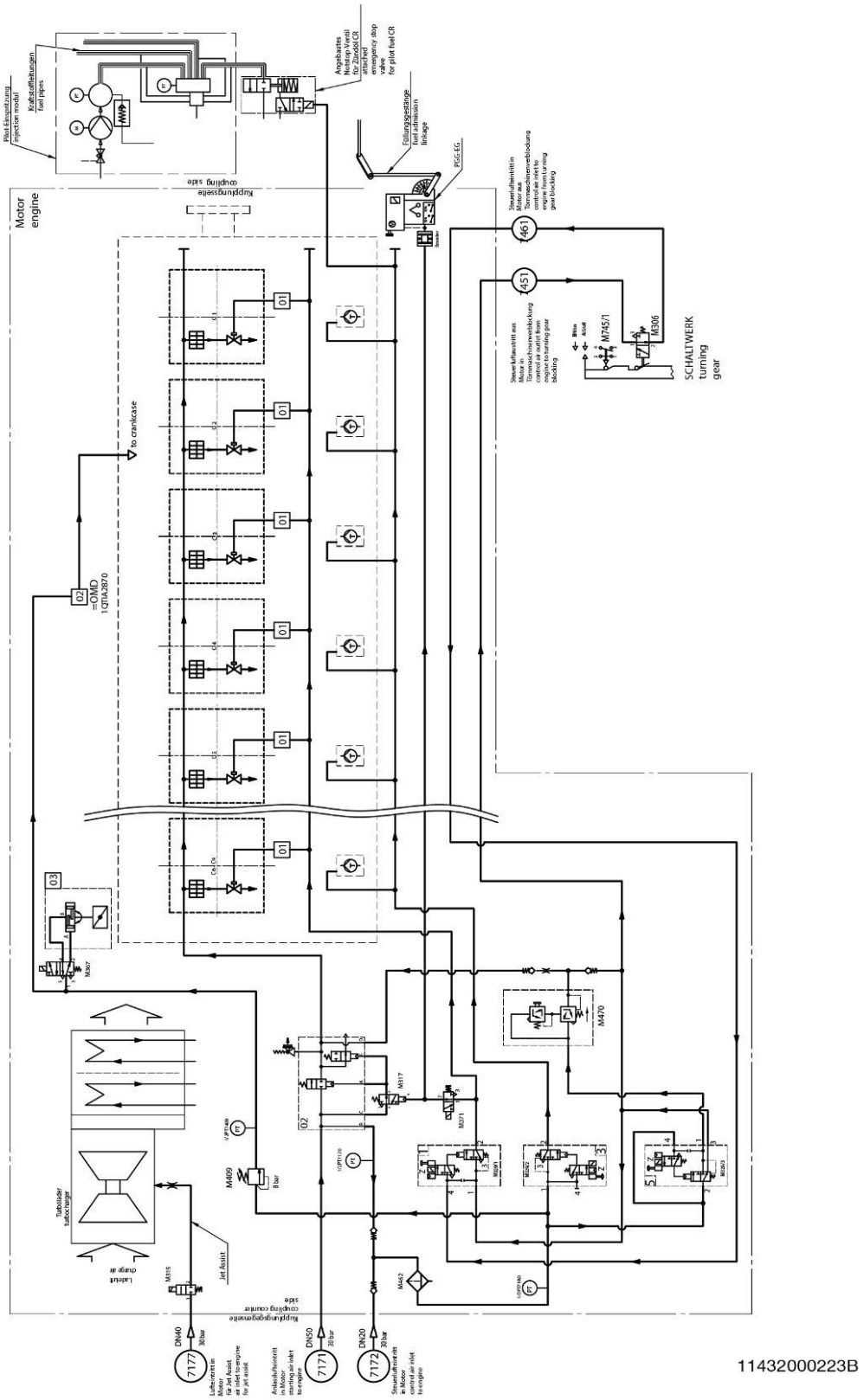

Figure 151: Internal compressed air system MAN 51/60DF – Exemplary

5.10.2 External compressed air system – Description

The compressed air supply to the engine plant requires starting air receivers and starting air compressors of a capacity and air delivery rating which will meet the requirements of the relevant classification society.

Piping

The main starting pipe (engine connection 7171), connected to both air receivers, leads to the main starting valve (MSV-001) of the engine.

A second 30 bar pressure line (engine connection 7172) with separate connections to both air receivers supplies the engine with control air. This does not require larger air receivers.

A line branches from the aforementioned control air pipe to supply other air-consuming engine accessories (e.g. fuel oil automatic filter) with compressed air.

A third 30 bar pipe is required for engines with jet assist (engine connection 7177). Depending on the air receiver arrangement, this pipe can be branched off from the starting air pipe near engine or must be connected separately to the air receiver for jet assist.

The pipes to be connected by the shipyard have to be supported immediately behind their connection to the engine. Further supports are required at sufficiently short distance.

Flexible connections for starting air (steel tube type) have to be installed with elastic fixation. The elastic mounting is intended to prevent the hose from oscillating. For detail information refer to planning and final documentation and manufacturer manual.

Galvanised steel pipes must not be used for the piping of the system.

1 T-007, 2 T-007/Starting air receivers

The installation situation of the air receivers must ensure a good drainage of condensed water. Air receiver must be installed with a downward slope sufficiently to ensure a good drainage of accumulated condensate water.

The installation also has to ensure that during emergency discharging of the safety valve no persons can be compromised.

It is not permissible to weld supports (or other) on the air receivers. The original design must not be altered. Air receivers are to be bedded and fixed by use of external supporting structures.

A max. service pressure of 30 bar is required. The standard design pressure of the starting air receivers is 30 bar and the design temperature is 50 °C.

1 C-001, 2 C-001/Air compressor

These are multi-stage compressor sets with safety valves, cooler for compressed air and condensate traps.

The operational compressor is switched on by the pressure control at low pressure then switched off when maximum service pressure is attained.

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Starting air system diagram

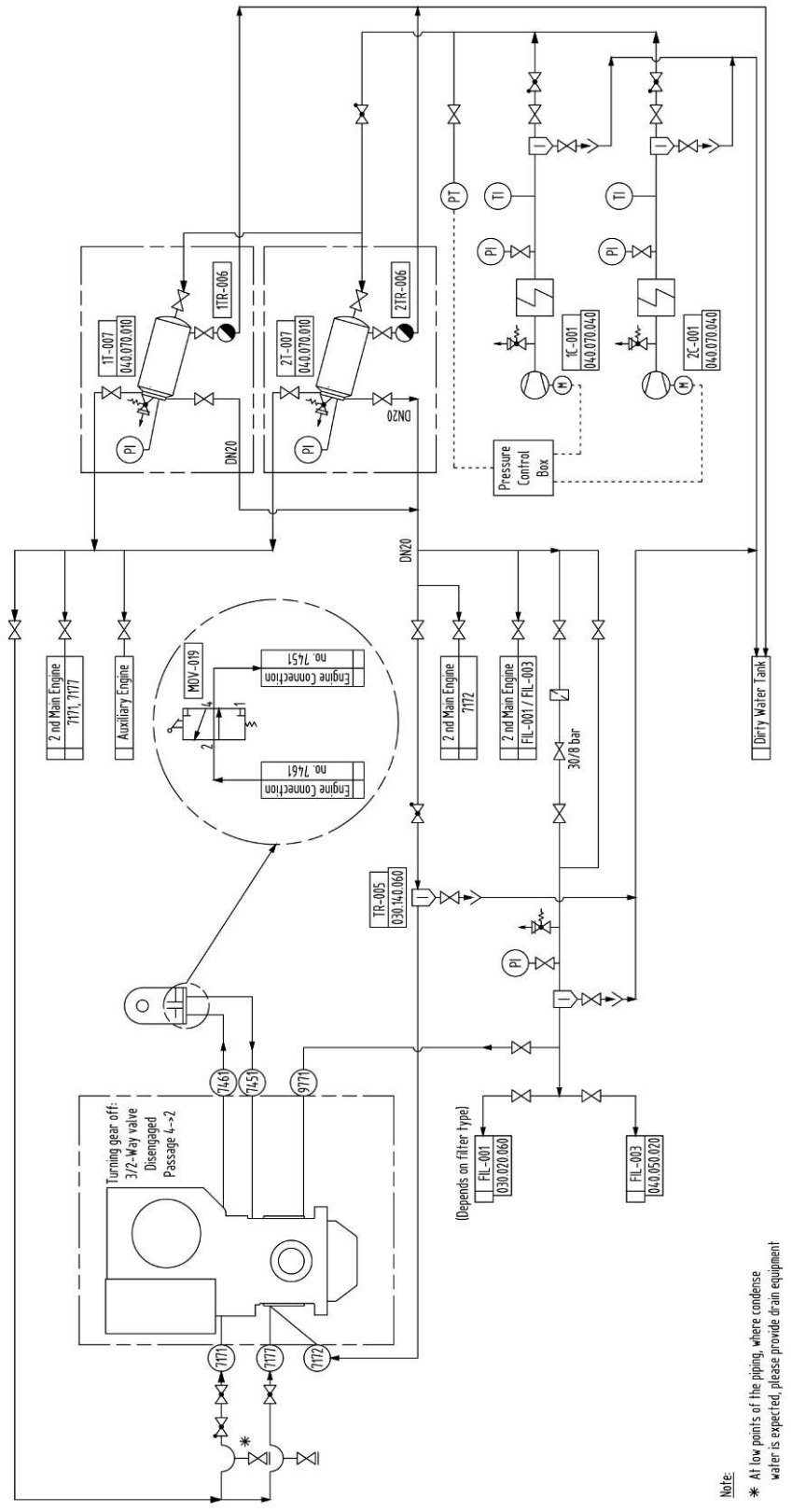

Figure 152: Starting air system diagram

5.10.3 External compressed air system – Dimensioning starting air receivers, compressors

Starting air receivers

The starting air supply is to be split up into not less than two starting air receivers of nominally the same size, which can be used independently of each another.

The engine requires compressed air for starting, start-turning, for the jet assist function as well as several pneumatic controls. The design of the pressure air receiver directly depends on the air consumption and the requirements of the classification societies.

For dimensioning the air receivers according to the classification rules, the air consumption of the liquid fuel mode start is used as standard. The air receivers will get larger, if the increased air consumption of the gas start is demanded (e.g. by classification society).

For air consumption see section Starting air and control air consumption, Page 96.

The air consumption per starting manoeuvre and per Slow Turn activation depends on the inertia moment of the unit.

In case of diesel-mechanical drive without shifting clutch but with shaft driven alternator consult MAN Energy Solutions.

For more information concerning jet assist see section External compressed air system – Jet assist, Page 420.

Calculation for starting air receiver of engines without jet assist and Slow Turn:

$$V = V_{st} \times f_{Drive} \times (z_{st} + z_{Safe}) \Big/ (p_{max} - p_{min})$$

Calculation for starting air receiver of engines with jet assist and Slow Turn:

$$V = \left(V_{st} \times f_{Drive} \times (z_{st} + z_{Safe}) + \frac{V_{Jet}}{5_{sec}} \times z_{Jet} \times t_{Jet} + V_{sl} \times z_{sl} \times f_{Drive} \right) \Big/ (p_{max} - p_{min})$$

V [litre]	Required receiver capacity
V_{st} [litre]	Air consumption per nominal start[1)]
f_{Drive}	Factor for drive type (1.0 = diesel-mechanic, 1.5 = alternator drive)
z_{st}	Number of starts required by the classification society
z_{Safe}	Number of starts as safety margin
V_{Jet} [litre]	Assist air consumption per jet assist[1)]
z_{Jet}	Number of jet assist procedures[2)]
t_{Jet} [sec]	Duration of jet assist procedures
V_{sl}	Air consumption per Slow Turn litre[1)]
z_{sl}	Number of Slow Turn manoeuvres

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p_{max} [bar]	Maximum starting air pressure (normally 30 bar)
p_{min} [bar]	Minimum starting air pressure (10 bar)

[1] Tabulated values see section Starting air and control air consumption, Page 96.
[2] The required number of jet manoeuvres has to be checked with yard or ship owner. To make a decision, consider the information in section External compressed air system – Jet assist, Page 420.

If other consumers (i.e. auxiliary engines, ship air etc.) which are not listed in the formula are connected to the starting air receiver, the capacity of starting air receiver must be increased accordingly, or an additional separate air receiver has to be installed.

Compressors

According to most classification societies, two or more air compressors must be provided. At least one of the air compressors must be driven independently of the main engine and must supply at least 50 % of the required total capacity.

The total capacity of the air compressors has to be capable to charge the receivers from the atmospheric pressure to full pressure of 30 bar within one hour.

The compressor capacities are calculated as follows:

$$P = \frac{V \times 30}{1000}$$

P [Nm^3/h]	Total volumetric delivery capacity of the compressors
V [litres]	Total volume of the starting air receivers at 30 bar service pressure

As a rule, compressors of identical ratings should be provided. An emergency compressor, if provided, is to be disregarded in this respect.

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5.10.4 External compressed air system – Jet assist

General

Air consumption

The air consumption for jet assist is, to a great extent, dependent on the load profile of the ship. In case of frequently and quickly changing load steps, jet assist will be actuated more often than this will be the case during long routes at largely constant load.

Layout of starting air vessels and compressor – Guiding values for consideration of jet assist manoeuvres

For the layout of starting air vessels and compressor add to the air consumption of the considered starts and slow turns also the air consumption of these jet assist manoeuvres.

The data in following table is not binding. The required number of jet manoeuvres has to be checked with yard or ship owner. For decision see also section Start-up and load application, Page 54.

The values shown in the following table are based on diesel oil mode.

Application	Recommended no. of jet assist with average duration	
	Per hour	In rapid succession
General drive	None[1)]	None[1)]
Diesel-mechanical drive without shifting clutch	None[1)]	None[1)]
Diesel-mechanical drive with shifting clutch	3 x 5 sec	2 x 5 sec
Diesel-mechanical drive with shaft-driven alternator (> 50 % MCR)	5 x 5 sec	3 x 5 sec
Electric propulsion	10 x 5 sec	5 x 5 sec
Electric propulsion offshore applications – Semisub production/drilling applications and drillships[2)]	(20 x 5 sec)	(10 x 5 sec)
Ships with frequent load changes (e.g. ferries)	10 x 5 sec	5 x 5 sec

[1)] According the necessity of the application "jet assist" check figures Load application dependent on base load (engine condition hot), Page 60. If the curve "without jet assist" is sufficient, jet assist can be omitted.

[2)] For these applications contact MAN Energy Solutions for a project specific estimation.

Table 220: Guiding values for the number of jet assist manoeuvres dependent on application

Dynamic positioning for drilling vessels, cable-laying vessels, off-shore applications

When applying dynamic positioning, pulsating load application of > 25 % may occur frequently, up to 30 times per hour. In these cases, the possibility of a specially adapted, separate compressed air system has always to be checked.

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Air supply

Generally, larger air receivers are to be provided for the air supply of the jet assist.

For the design of the jet assist air supply the temporal distribution of events needs to be considered, if there might be an accumulation of events.

In each case the delivery capacity of the compressors is to be adapted to the expected jet assist requirement per unit of time.

5.11 Engine room ventilation and intake air

5.11.1 General information

Engine room ventilation system

Its purpose is:

- Supplying the engines and auxiliary boilers with combustion air.
- Carrying off the radiant heat from all installed engines and auxiliaries.

Intake air engine

The intake air of the engine must be free from spray water, snow, dust and oil mist.

This is achieved by:

- Louvres, protected against the head wind, with baffles in the back and optimally dimensioned suction space so as to reduce the air flow velocity to 1 – 1.5 m/s.
- Self-cleaning air filter in the suction space (required for dust-laden air, e.g. cement, ore or grain carrier).
- Sufficient space between the intake point and the openings of exhaust air ducts from the engine and separator room as well as vent pipes from lube oil and fuel oil tanks and the air intake louvres (the influence of winds must be taken into consideration).
- Positioning of engine room doors on the ship's deck so that no oil-laden air and warm engine room air will be drawn in when the doors are open.
- Arranging the separator station at a sufficiently large distance from the turbochargers.

As a standard, the engines are equipped with turbochargers with air intake silencers and the intake air is normally drawn in from the engine room.

In tropical service a sufficient volume of air must be supplied to the turbocharger(s) at outside air temperature. For this purpose there must be an air duct installed for each turbocharger, with the outlet of the duct facing the respective intake air silencer, separated from the latter by a space of approximately 1.5 m (see figure Example: Exhaust gas ducting arrangement, Page 430). No water of condensation from the air duct must be permissible to be drawn in by the turbocharger. The air stream must not be directed onto the exhaust manifold.

If the ship operates at arctic conditions, an air preheater must be applied to maintain the engine room temperature above 5° C. In order to reduce power for air preheating, the engines can be supplied by a separate system directly from outside, see section Intake air ducting in case of arctic conditions, Page 422.

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Air fans are to be designed so as to maintain a positive air pressure of 50 Pa (5 mm WC) in the engine room.

Radiant heat

The heat radiated from the main and auxiliary engines, from the exhaust manifolds, waste heat boilers, silencers, alternators, compressors, electrical equipment, steam and condensate pipes, heated tanks and other auxiliaries is absorbed by the engine room air.

The amount of air V required to carry off this radiant heat can be calculated as follows:

$$V = \frac{Q}{\Delta t \times cp \times \rho t}$$

V [m^3/h]	Air required
Q [kJ/h]	Heat to be dissipated
Δt [°C]	Air temperature rise in engine room (10 – 12.5)
cp [kJ/kg*k]	Specific heat capacity of air (1.01)
ρt [kg/m^3]	Air density at 35 °C (1.15)

Ventilator capacity

The capacity of the air ventilators (without separator room) must be large enough to cover at least the sum of the following tasks:

- The combustion air requirements of all consumers.
- The air required for carrying off the radiant heat.

A rule-of-thumb applicable to plants operating on heavy fuel oil is 20 – 24 m^3/kWh.

Moreover it is recommended to apply variable ventilator speed to regulate the air flow. This prevents excessive energy consumption and cooling down of engines in stand-by.

5.11.2 Intake air ducting in case of arctic conditions

General recommendations for external intake air system of vessels operating in arctic conditions

The design of the intake air system ducting is crucial for reliable operation of the engine. The following points need to be considered:

- Every single engine must be provided with a dedicated intake air system. It is not allowed to combine air intake systems of different engines.
- According to classification rules it may be required to install two air inlets from the exterior, one at starboard and one at portside.
- It must be prevented that exhaust gas and oil dust is sucked into the intake air duct as fast filter blocking might occur.
- Suitable corrosion and low temperature resistant materials should be applied. Stainless steel S316 L might be suitable.

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- Inside the duct, there must not be any parts (e.g. bolts, nuts, stiffening, etc.) that could fall off and move towards the engine. Installations, that are absolutely necessary (e.g. light behind filter wall) must be specially secured (self-locking nuts, screwed covers instead of clamped covers etc.).
- Due to the air flow, load changes and other external forces, (especially during ice breaking, if applicable) the intake air pipe is subject to heavy vibrations. Additionally engine and propeller exciting frequencies have to be taken into account. This has to be considered within the overall layout and the intake air duct needs to be reinforced sufficiently.
- Thermal expansion has to be considered for the layout and foundation of the duct (e.g. flexible mounting, additional compensators).
- Suitable drainage arrangements to remove any water from the intake air ducting should be provided. Backflow of air through drains has to be avoided (e.g. by syphons) and regularly checked for proper functioning. Adequate heating is required to prevent icing of drains.
- The air duct and its components need to be insulated properly. Especially a vapor barrier has to be applied to prevent atmospheric moisture freezing in the insulation material.
- An (automatic) shut-off flap should be installed to prevent a chimney effect and cooling down of engine during stand-still (maintenance or stand-by of engine). This flap is to be monitored and engine start should only be allowed in fully-open position. As an alternative, the intake system can be closed by a roller shutter or tarpaulin in front of the filter.
- The overall pressure drop of the intake air system ducting and its components is to be limited to 20 mbar. Moreover the differential pressure of the intake air filter must be monitored to keep this requirement. For additional safety, other components as the droplet separator and the weather hood can be monitored by differential pressure devices. During commissioning and maintenance work, checking of the air intake system back pressure by means of a temporarily connected measuring device may become necessary. For this purpose, a measuring socket is to be provided approximately 1 to 2 metres before the turbocharger, in a straight length of pipe at an easily accessible position. Standard pressure measuring devices usually require a measuring socket size of 1/2".
- The turbocharger as a flow machine is dependent on a uniform inflow. Therefore, the ducting must enable an air flow without disturbances or constrictions. For this, multiple deflections with an angle > 45° within the ducting must be avoided.
- The intake air must not flow against the direction of the compressor rotation, otherwise stalling could occur.
- It is recommended to verify the layout of the intake air piping by CFD calculations up to the entry of the compressor of the turbocharger.
- The maximum specified air flow speed of 20 m/s should not be exceeded at any location of the pipe.
- A silencer is recommended to reduce the noise emissions from engine inlet and charge air blow-off. Sound power levels can be found in the relevant section of the Project Guide. Care must be taken, that no insulation material can escape from the silencer, which can fuse into glass spheres in the combustion chamber.

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Components of intake air ducting

The whole system and its components must be designed suitably robust to withstand pressure peaks occurring from turbocharger surge. This will not happen during normal operation, but it could occur at fast load changes of the engine. This can happen 2 – 3 times consecutively, until the turbocharger comes back to its normal working range.

The table below shows values at engine inlet connection with a suitable intake air ducting. An unfavorable intake air duct design can also lead to higher values.

Type	Variation	Frequency	Comment
Pressure oscillation	± 40 mbar, 5 – 10 Hz	Permanent	Normal operation/ constant load
Peak pressure (shock wave)	± 300 mbar	Sporadically	Engine emergency stop/ turbocharger surge

The ambient air, which is led to engine by the intake air duct, needs to be conditioned by several components as shown in figure External intake air supply system for arctic conditions, Page 426. It needs to be cleaned according to the requirements in section Specification of intake air (combustion air), Page 292. This could be done by the following components:

- Section for cleaning of intake air (1 – 4)

  A weather hood (1) in combination with a snow trap (2) removes coarse dirt, snow and rain. A heated droplet separator (3) subsequently separates remaining water droplets or snow from the air. An appropriate filter cleans the intake air from particles. (4). As a minimum, inlet air must be cleaned by an ISO coarse 45 % class filter as per DIN EN ISO 16890. If there is a risk of high inlet air contamination, filter efficiency should be at least ISO ePM10 50 % according to DIN EN ISO 16890. See figure External intake air supply system for arctic conditions, Page 426.

- Combustion air silencer (5)

  Noise emissions of engine inlet and charge air blow-off can be reduced by a silencer in the intake air duct. It is recommended to apply a mesh (5a) at the outlet of the silencer to protect the turbocharger against any loose parts (e.g. insulation material of silencer, rust etc.) from the intake air duct. This mesh is to be applied even if the silencer will not be supplied. A drain close to the turbocharger is required to separate condensate water. See section Noise, Page 162.

- Overpressure flap (6) (optional)

  Depending on the system volume and chosen components it might be necessary to install a overpressure flap between silencer and engine. Peak pressure pulses (e.g. during emergency stop) are conducted into the engine room via this flap, preventing possible damage to the filter and silencer.

- Shut-off flap/blind plate (7)

  It is recommended to install a shut-off flap to prevent cooling down of the engine during longer standstills under arctic conditions. This flap should be monitored by the engine automation system to prevent engine start with closed flap.
  As an alternative, the intake system can be closed by a roller shutter or tarpaulin in front of the filter.

- Compensator (8) and transition piece (9)

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A steel compensator (rubber might also be considered) has to be installed direct vertically upstream of the 90° transition piece behind turbocharger. A rigid support must be provided as close as possible upstream of the compensator. It has to be noted, that this compensator is solely foreseen to compensate engine-borne movements. Additional compensators might be necessary to cope for thermal expansion.

- Strainer for commissioning phase (9a)

  To prevent residues from installation phase entering the engine during commissioning, it is recommended to install a strainer or protective mesh as close as possible to the turbocharger. After running-in is finished, the strainer must be removed and exchanged by an intermediate pipe.

- Charge air blow-off or recirculation

  For arctic conditions (see section Engine operation under arctic conditions, Page 70) an increased firing pressure, which is caused by higher density of cold air, is prevented by an additional valve, which blows off charge air (11). A compensator (10) connects the engine with the charge air blow-off piping. The blown-off air is taken after (cold blow-off) the charge air cooler or before the charge air cooler (hot blow-off) and is circulated (12) back in the intake air duct or blown out via an additional silencer. A homogenous temperature profile and a correct measurement of intake air temperature in front of compressor has to be achieved. For this a minimum distance of five times the diameter of the intake air duct between inlet of blown-off air and the measuring point must be kept.

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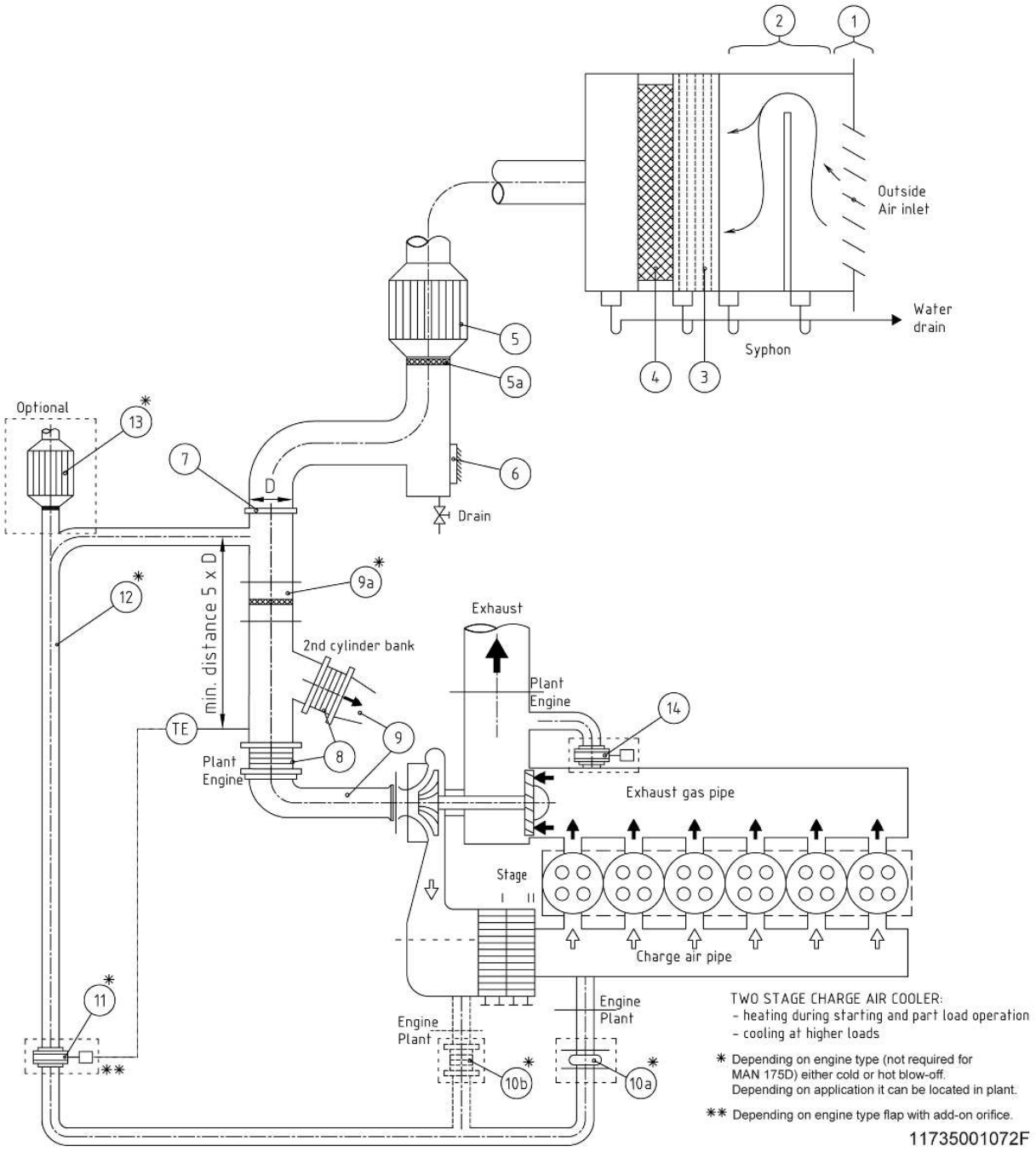
2
1
Outside
Air inlet
Water
drain
Syphon
4
3
5
5a
6
Drain
Optional
13
7
D
12
9a
min. distance 5 x D
2nd cylinder bank
Exhaust
Plant
Engine
14
TE
9
8
Plant
Engine
Exhaust gas pipe
Stage
I
II
Charge air pipe
11
Engine
Plant
Engine
Plant
10b
10a
TWO STAGE CHARGE AIR COOLER:
- heating during starting and part load operation
- cooling at higher loads
* Depending on engine type (not required for MAN 175D) either cold or hot blow-off. Depending on application it can be located in plant.
** Depending on engine type flap with add-on orifice.
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1 Weather hood
2 Snow trap
3 Heated droplet separator
4 Air intake filter 030.120.010
5 Combustion air silencer 030.130.040
5a Protective mesh
6 Overpressure flap (optional)
7 Blind plate/shut-off flap (for maintenance case)
8 Metal bellow expansion joint combustion air (rubber might be considered)
9 Transition piece
9a (Optional) intermediate pipe with protective grid for running-in phase (to be removed afterwards)
10a Rubber bellow expansion joint – Cold blow-off
10b Metal bellow expansion joint – Hot blow-off
11 Charge air blow-off valve
12 Charge air blow-off pipe
13 Charge air blow-off silencer
14 Waste gate (if required for relevant engine type)

Figure 153: External intake air supply system for arctic conditionsx

5.12 Exhaust gas system

5.12.1 General

Layout

The flow resistance in the exhaust system has a very large influence on the fuel consumption and the thermal load of the engine. The values given in this document are based on an exhaust gas system which flow resistance does not exceed 50 mbar. If the flow resistance of the exhaust gas system is higher than 50 mbar, contact MAN Energy Solutions for project-specific engine data.

Note:
20 mbar resistance for the MAN SCR-LPC or 37 mbar for the MAN SCR-LPS as part of the total resistance have to be considered.

The pipe diameter selection depends on the engine output, the exhaust gas volume and the system back pressure, including silencer and SCR (if fitted). The back pressure also being dependent on the length and arrangement of the piping as well as the number of bends. Sharp bends result in very high flow resistance and should therefore be avoided. If necessary, pipe bends must be provided with guide vanes.

It is recommended not to exceed a maximum exhaust gas velocity of approximately 40 m/s.

For the installation of exhaust gas systems in dual fuel engines plants, in ships and offshore applications, several rules and requirements from IMO Tier II, classification societies, port and other authorities have to be applied. For each individual plant the design of the exhaust gas system has to be approved by one ore more of the above mentioned parties.

The design of the exhaust gas system of dual fuel engines has to ensure that unburned gas fuel cannot gather anywhere in the system. This case may occur, if the exhaust gas contains unburned gas fuel due to incomplete combustion or other malfunctions.

The exhaust gas system shall be designed and build sloping upwards in order to avoid formations of gas fuel pockets in the system. Only very short horizontal lengths of exhaust gas pipe can be permissible.

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In addition the design of other main components, like exhaust gas boiler and silencer, has to ensure that no accumulation of gas fuel can occur inside. For the exhaust gas system in particular this reflects to following design details:

- Design requirements for the exhaust system installation
- Installation of adequate purging device
- Installation of explosion venting devices (rupture discs, or similar)

Note:
For further information refer to our brochure "Safety Concept – Marine dual fuel engines".

Installation

When installing the exhaust system, the following points must be observed:

- The exhaust pipes of two or more engines must not be joined.
- Because of the high temperatures involved, the exhaust pipes must be able to expand. The expansion joints to be provided for this purpose are to be mounted between fixed-point pipe supports installed in suitable positions. One compensator is required just after the outlet casing of the turbocharger (see section Position of the outlet casing of the turbocharger, Page 432) in order to prevent the transmission of forces to the turbocharger itself. These forces include those resulting from the weight, thermal expansion or lateral displacement of the exhaust piping. For this compensator/expansion joint one sturdy fixed-point support must be provided.
- The exhaust piping should be elastically hung or supported by means of dampers in order to prevent the transmission of sound to other parts of the vessel.
- The exhaust piping is to be provided with water drains, which are to be regularly checked to drain any condensation water or possible leak water from exhaust gas boilers if fitted.
- During commissioning and maintenance work, checking of the exhaust gas system back pressure by means of a temporarily connected measuring device may become necessary. For this purpose, a measuring socket is to be provided approximately 1 to 2 metres after the exhaust gas outlet of the turbocharger, in a straight length of pipe at an easily accessed position. Standard pressure measuring devices usually require a measuring socket size of 1/2". This measuring socket is to be provided to ensure back pressure can be measured without any damage to the exhaust gas pipe insulation.

5.12.2 Components and assemblies of the exhaust gas system

Exhaust gas silencer and exhaust gas boiler

Mode of operation

The silencer operates on the absorption and resonance principle so it is effective in a wide frequency band. The flow path, which runs through the silencer in a straight line, ensures optimum noise reduction with minimum flow resistance.

A spark arrestor should be provided in the exhaust gas system (e.g. integrated in the silencer).

Note:
Spark arrestors are mandatory for certain ship types.

Installation

If possible, the silencer should be installed towards the end of the exhaust line. A vertical installation situation is to be preferred, but at least it has to build steadily ascending to avoid any accumulation of explosive gas concentration. The cleaning ports of the spark arrestor are to be easily accessible.

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Note:
Water entry into the silencer and/or boiler must be avoided, as this can cause damages of the components (e.g. forming of deposits) in the duct.

Exhaust gas boiler

To utilise the thermal energy from the exhaust, an exhaust gas boiler producing steam or hot water may be installed.

Insulation

The exhaust gas system (from outlet of turbocharger, boiler, silencer to the outlet stack) is to be insulated to reduce the external surface temperature to the required level.

The relevant provisions concerning accident prevention and those of the classification societies must be observed.

The insulation is also required to avoid temperatures below the dew point on the interior side. In case of insufficient insulation intensified corrosion and soot deposits on the interior surface are the consequence. During fast load changes, such deposits might flake off and be entrained by exhaust in the form of soot flakes.

Insulation and covering of the compensator must not restrict its free movement.

Explosion venting devices/rupture disc

The external exhaust gas system of a dual fuel engine installation is to be equipped with explosion venting devices (rupture discs, or similar) to relief the excess pressure in case of explosion. The number and location of explosion venting devices is to be approved by the classification societies.

Purging device/fan

The external exhaust gas system of dual fuel engine installations is to be equipped with a purging device to ventilate the exhaust system after an engine stop or emergency shut down. The design and the capacity of the ventilation system is to be approved by the classification societies.

Safety concept

For further information refer to our brochure "Safety Concept – Marine dual fuel engines".

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5.12.3 Example: Ducting arrangement

Note:
Number and position of rupture discs or automatic valves have to be designed according to real ducting geometrie. The shown arrangement is only a principle example.

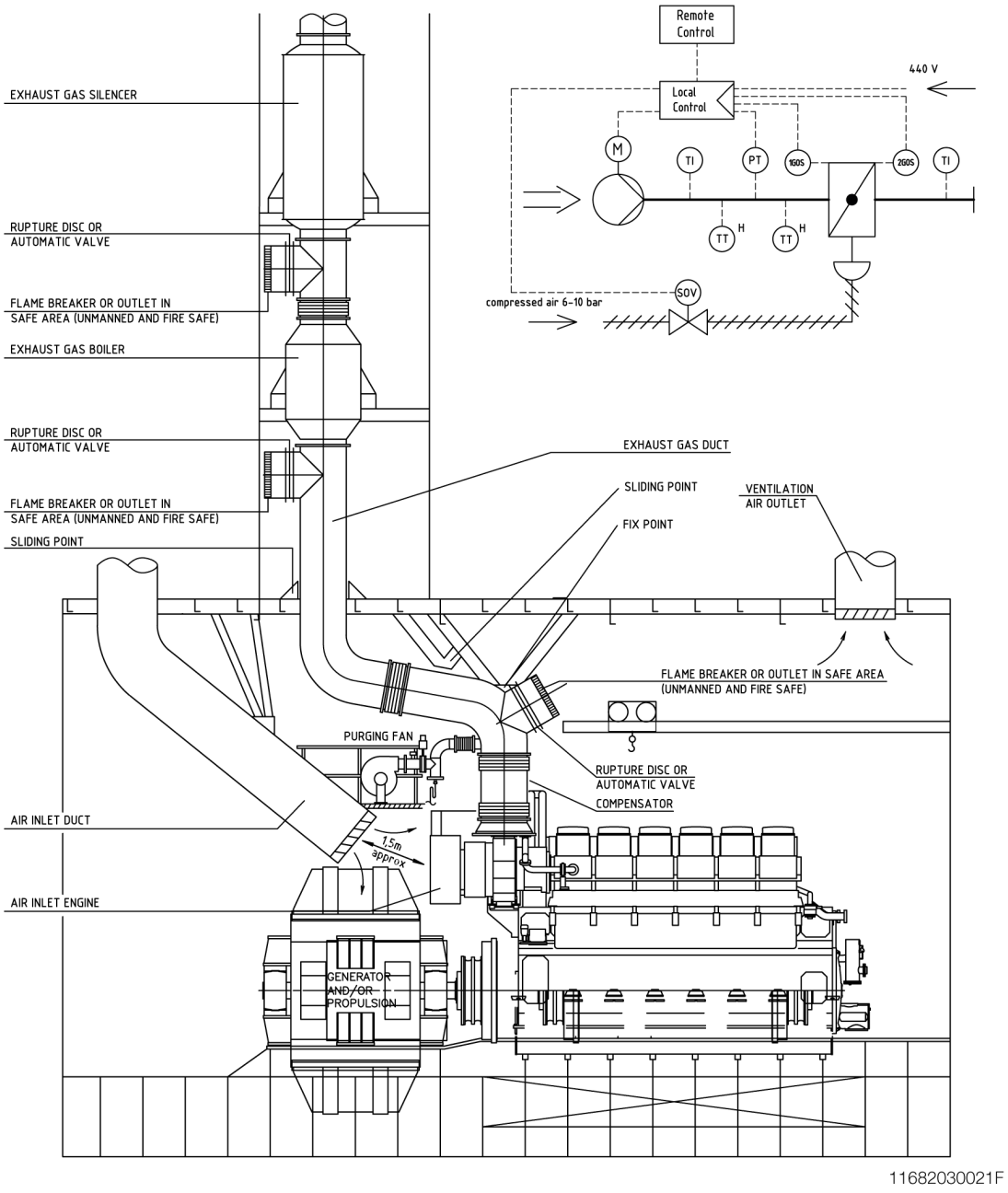

Figure 154: Example: Exhaust gas ducting arrangement

Fix point support

The engine related compensator has to be connected directly to the exhaust gas outlet of the turbocharger (installation of compensator vertically or max. 45° position after turbocharger). In case that the compensator cannot be directly connected to the exhaust gas outlet of the turbocharger, contact MAN Energy Solutions. Immediately downstream of the engine related compensator, it is required to install a strong and rigid fix point to support the exhaust gas pipe. It is not permitted to compensate with the engine related compensator movements or vibrations coming from components or systems installed downstream of this compensator.

5.12.4 General details for Tier III SCR system duct arrangement

MAN Energy Solutions recommends that the SCR reactor housing should be mounted before all other components (e.g. boiler, silencer) in the exhaust duct, coming from the engine side. A painting on the inside wall of the exhaust duct in front of the SCR system is not permissible.

All of the spaces/openings for cleaning and maintenance on the entire unit, including air reservoir module, dosing unit and reactor housing with soot-blowers must be accessible.

We strongly recommend that in front of the reactor housing sufficient space for the maintenance personal and/or for the temporary storage of the catalyst honeycombs has to be foreseen (see section SCR system, Page 441).

Catalyst elements could reach a weight of 25 kg, the reactor openings could reach a total weight of about 70 kg, MAN Energy Solutions strongly recommends a lifting capability above the reactors.

A very important point is the transportation way and storage space of the catalyst honeycombs within the funnel for supply of the SCR reactor during maintenance or catalyst refreshment, one reactor could contain more than 100 elements.

To avoid time-consuming or implementation of a scaffolding, MAN Energy Solutions strongly recommends at minimum a lifting device in the funnel or any kind of material elevator. A porthole from outside rooms on level with the reactor housing is also a possibility, as far as those rooms could be supplied with the catalyst honeycombs.

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5.12.5 Position of the outlet casing of the turbocharger

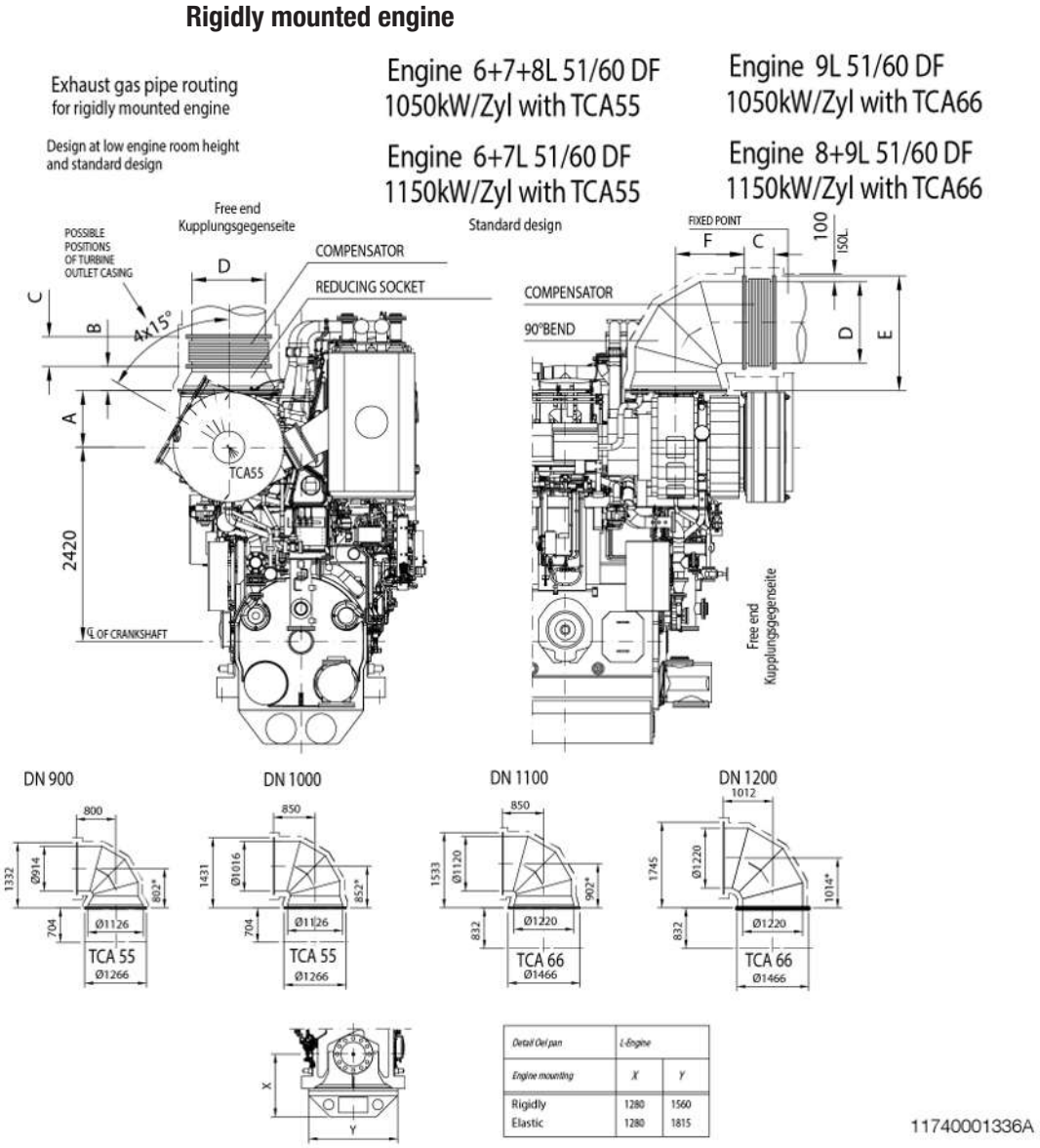

Figure 155: Design at low engine room height and standard design, part 1

No. of cylinders, config.		6L	7L	8L	9L
Turbocharger/cyl. power (kW/cyl.)		TCA 55/1,150		TCA 66/1,150	
A	mm	704	704	832	832
B*		302	302	302	302
C*		372	387	432	432
D		DN 1,000	DN 1,000	DN 1,100	DN 1,200
E		1,431	1,431	1,533	1,745
F		850	850	850	990

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No. of cylinders, config.	6L	7L	8L	9L
Turbocharger/cyl. power (kW/cyl.)	TCA 55/1,150		TCA 66/1,150	
* incl. 2 mm sealing				

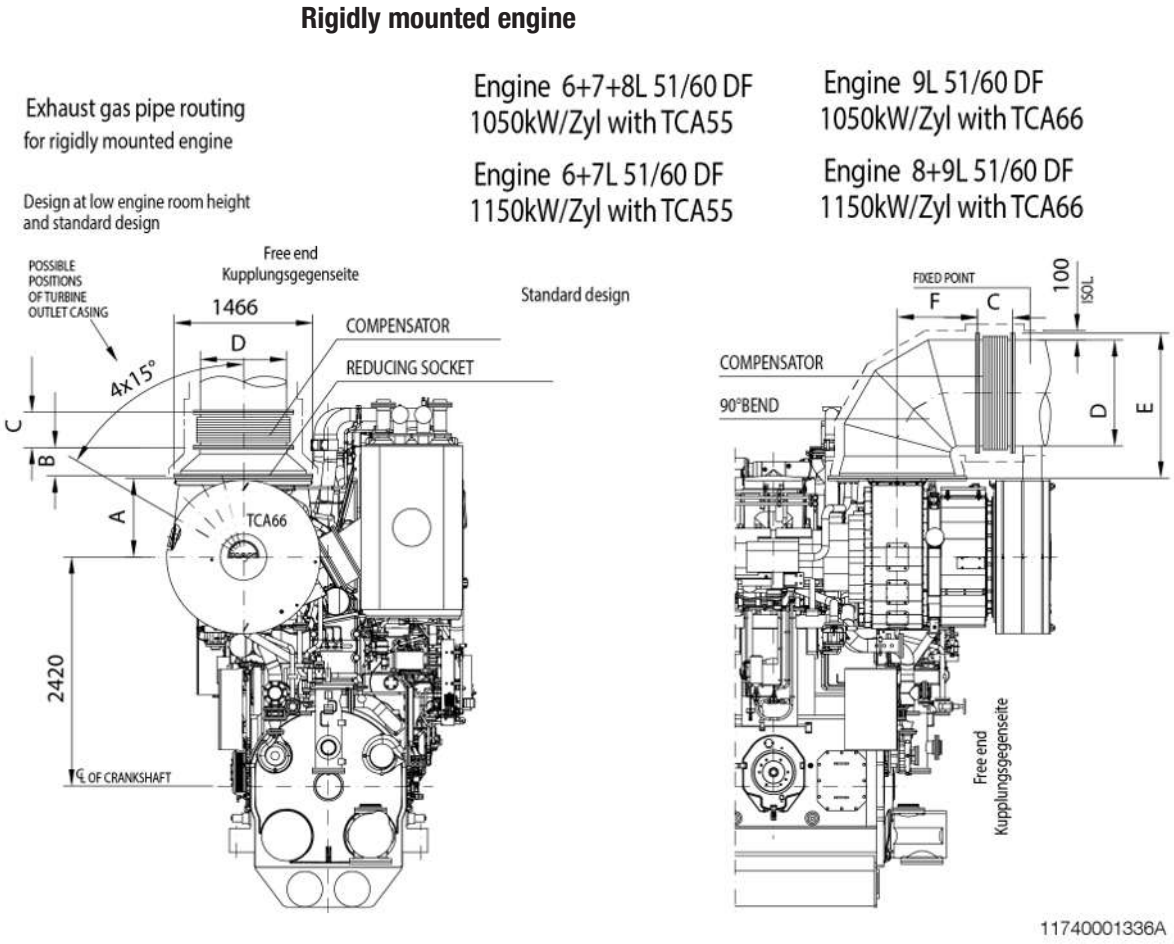

Figure 156: Design at low engine room height and standard design, part 2

No. of cylinders, config.		6L	7L	8L	9L
Turbocharger/cyl. power (kW/cyl.)		TCA 55/1,050	TCA 55/1,050	TCA 55/1,050	TCA 66/1,050
A	mm	704	704	704	832
B*		302	302	302	302
C*		372	387	387	432
D		DN 900	DN 1,000	DN 1,000	DN 1,100
E		1,332	1,431	1,431	1,533
F		800	850	850	850
* incl. 2 mm sealing					

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Rigidly mounted engine

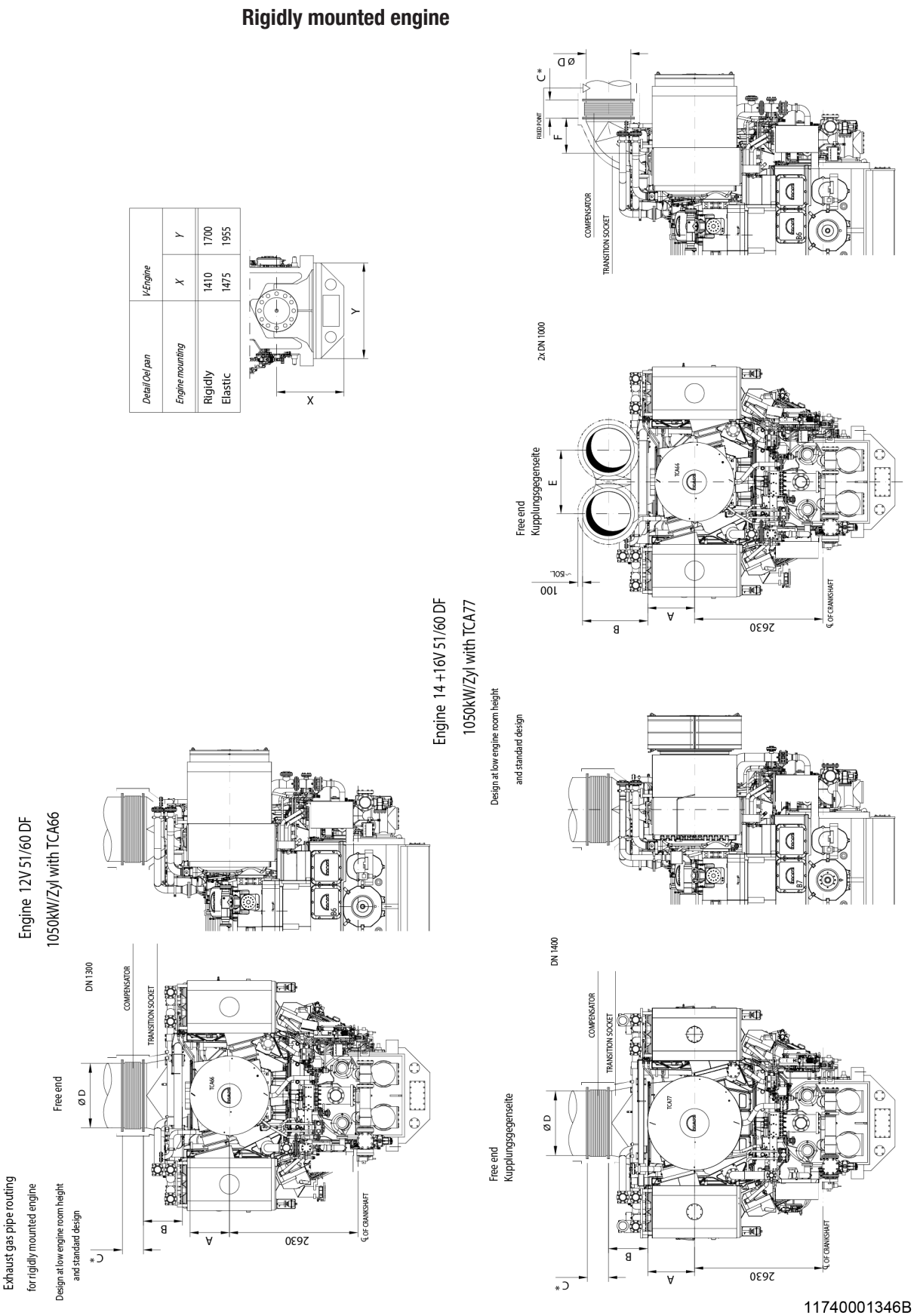

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Figure 157: Design at low engine room height and standard design, part 1

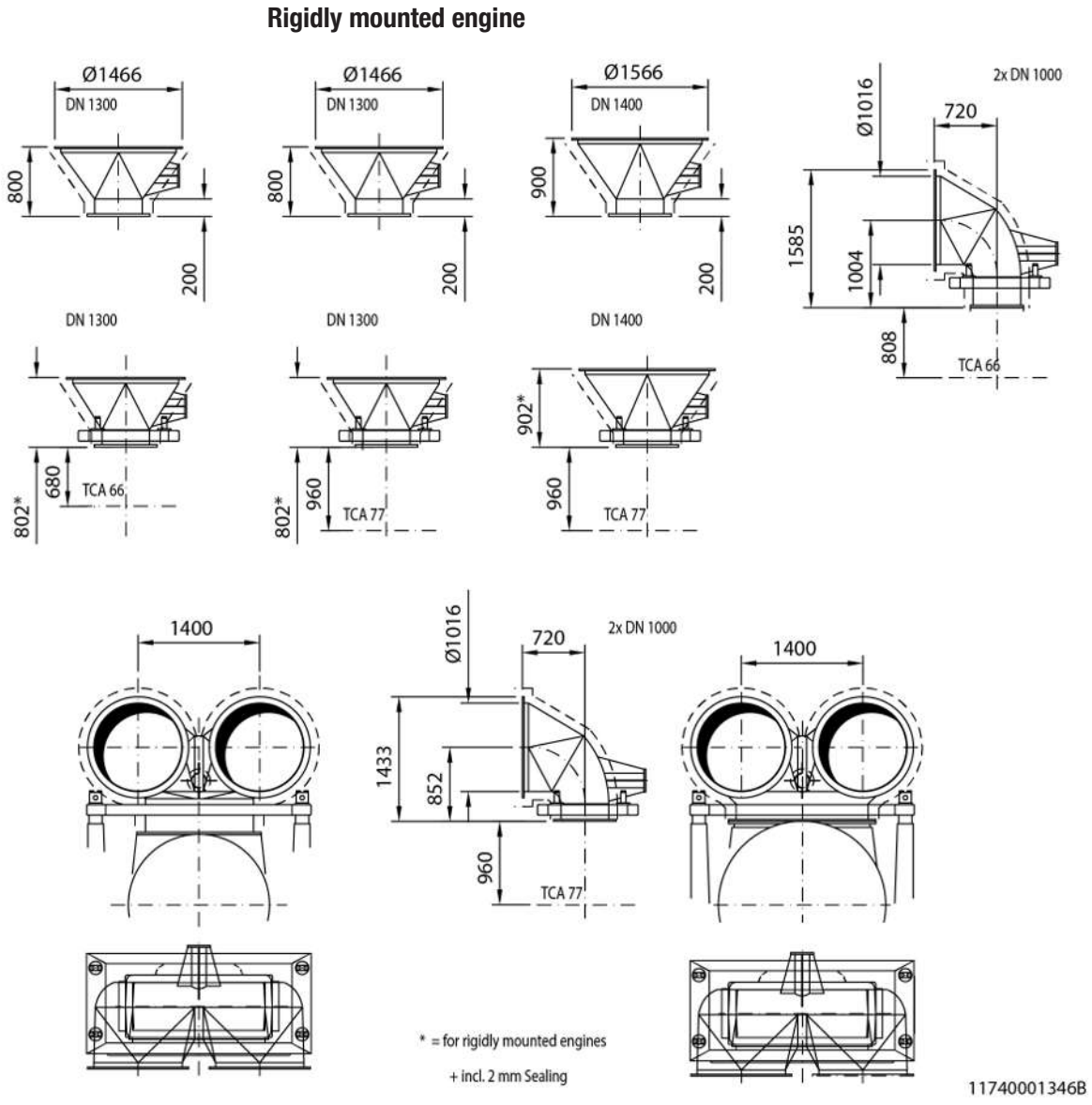

Figure 158: Design at low engine room height and standard design, part 2

No. of cylinders, config.		12V	14V	16V
Turbocharger/cyl. power (kW/cyl.)		TCA 66/1,050	TCA 77/1,050	TCA 77/1,050
A	mm	808	960	960
+B		802	802	902
B* (devided)		1,585	1,433	1,433
+C*		432	424	424
D		DN 1,300	DN 1,400	DN 1,400
D (devided)		2 x DN 1,000	2 x DN 1,000	2 x DN 1,000
E		1,400	1,400	1,400
F		720	720	720

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No. of cylinders, config.	12V	14V	16V
Turbocharger/cyl. power (kW/cyl.)	**TCA 66/1,050**	**TCA 77/1,050**	**TCA 77/1,050**
* = for rigidly mounted engines + = incl. 2 mm sealing			

No. of cylinders, config.		12V	14V
Turbocharger/cyl. power (kW/cyl.)		**TCA 77/1,150**	**TCA 77/1,150**
A	mm	960	960
+B		802	802
B* (devided)		1,433	1,433
+C*		432	424
D		DN 1,300	DN 1,400
D (devided)		2 x DN 1,000	2 x DN 1,000
E		1,400	1,400
F		720	720
* = for rigidly mounted engines + = incl. 2 mm sealing			

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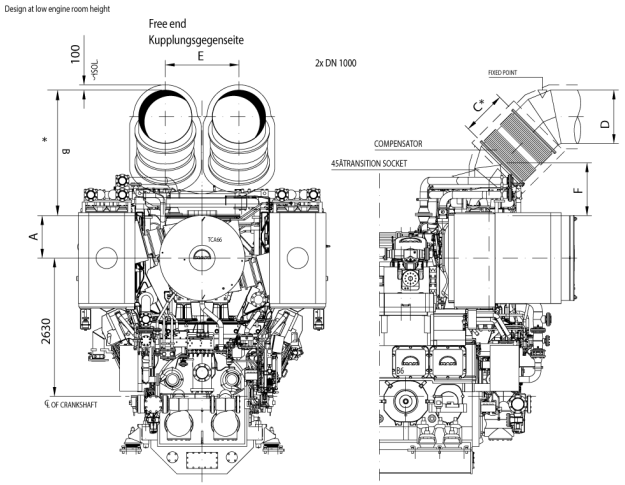
Exhaust gas pipe routing
for elastic mounted engine

Design at low engine room height
and standard design

Engine 12V 51/60 DF
1050kW/Zyl with TCA66

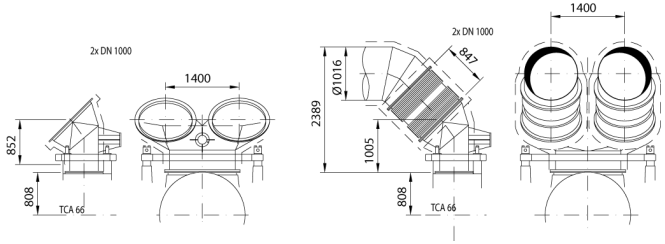
Engine 14 +16V 51/60 DF
1050kW/Zyl with TCA77

Engine 12+14V 51/60 DF
1150kW/Zyl with TCA77



No. of cylinders		12V	14V	16V
Turbocharger / Cyl. power (kW/Cyl.)		TCA 66 1050	TCA 77 1050	TCA 77 1050
A	Measurement mm	808	960	960
B *		2389	2237	2237
C *		847	847	847
D		2xDN 1000	2x DN 1000	2x DN 1000
E		1400	1400	1400
F		1005	852	852

* incl. 2 mm Sealing

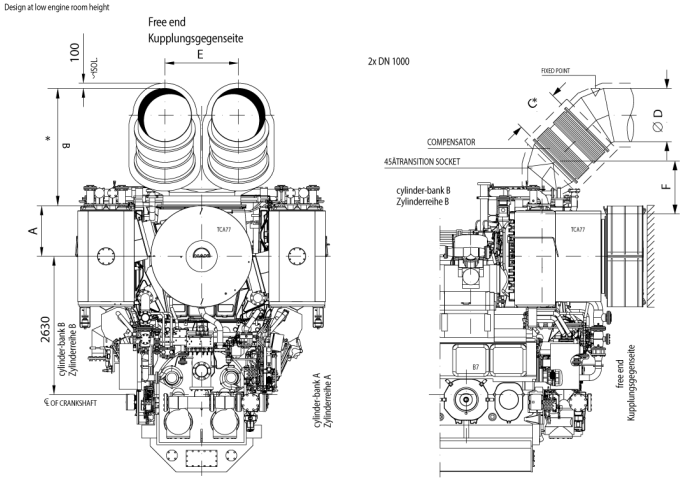


Exhaust gas pipe routing
for elastic mounted engine

Design at low engine room height
and standard design

Engine 14 + 16V 51/60 DF
1050kW/Zyl with TCA77

Engine 12 + 14V 51/60 DF
1150kW/Zyl with TCA77



No. of cylinders		12V	14V
Turbocharger / Cyl. power (kW/Cyl.)		TCA 77 1150	TCA 77 1150
A	Measurement mm	960	960
B *		2237	2237
C *		847	847
D		2x DN 1000	2x DN 1000
E		1400	1400
F		852	852

* incl. 2 mm Sealing

Detail Oel pan	V-Engine	
Engine mounting	X	Y
Rigidly	1410	1700
Elastic	1475	1955

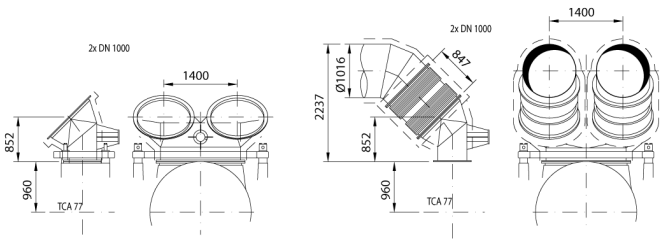


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Figure 159: Design at low engine room height and standard design

No. of cylinders, config.		12V	14V	16V
Turbocharger/cyl. power (kW/cyl.)		TCA 66/1,050	TCA 77/1,050	TCA 77/1,050
A	mm	808	960	960
B*		2,389	2,237	2,237
C*		847	847	847
D		2 x DN 1,000	2 x DN 1,000	2 x DN 1,000
E		1,400	1,400	1,400
F		1,005	852	852
* incl. 2 mm sealing				

No. of cylinders, config.		12V	14V
Turbocharger/cyl. power (kW/cyl.)		TCA 77/1,150	TCA 77/1,150
A	mm	960	960
B*		2,237	2,237
C*		847	847
D		2 x DN 1,000	2 x DN 1,000
E		1,400	1,400
F		852	852
* incl. 2 mm sealing			

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Resiliently mounted engine

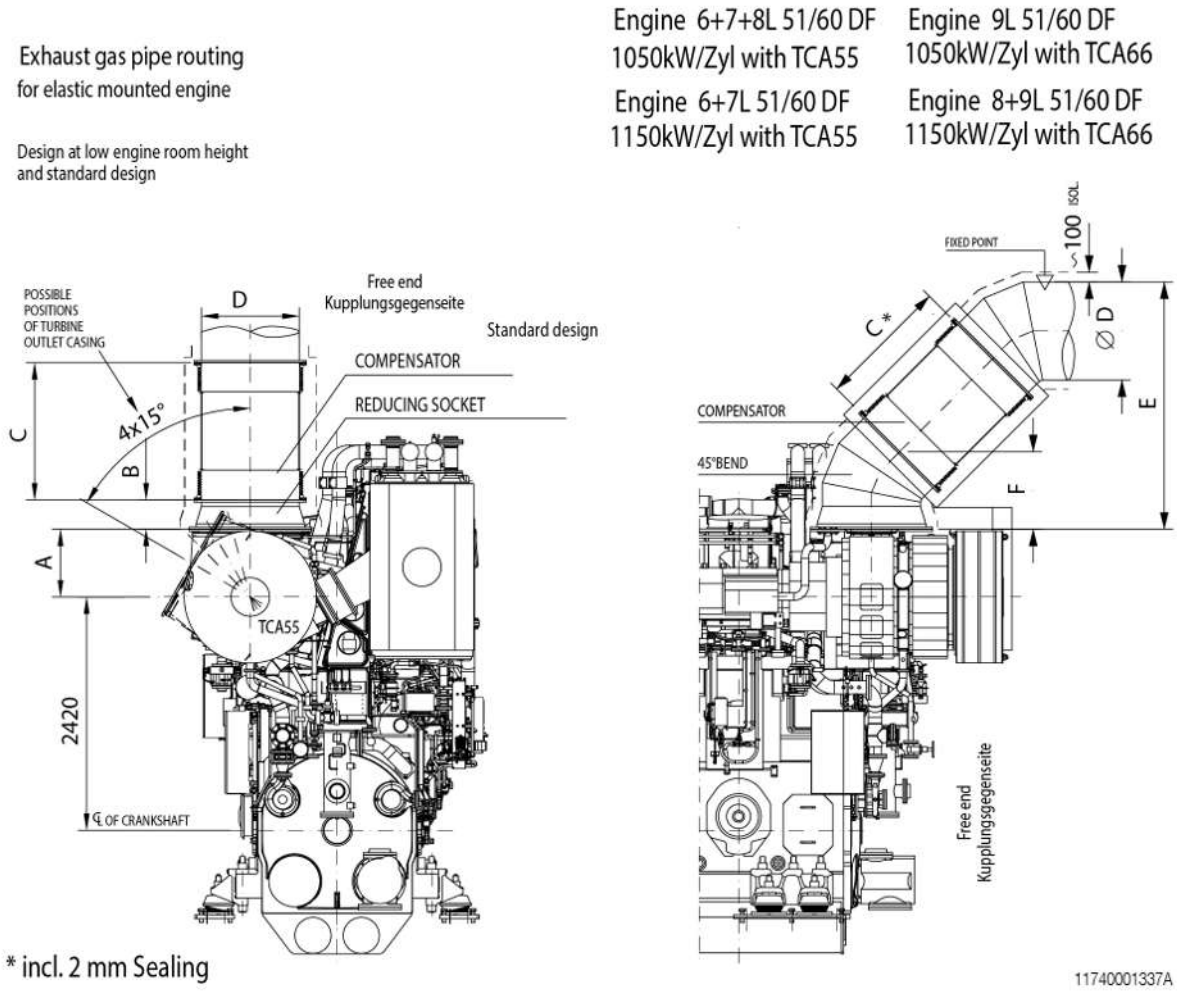

Figure 160: Design at low engine room height – Resiliently mounted engine, part 1

No. of cylinders, config.		6L	7L	8L	9L
Turbocharger/cyl. power (kW/cyl.)		TCA 55/1,150		TCA 66/1,150	
A	mm	704	704	832	832
B*		302	302	302	302
C*		1,216	1,359	1,358	1,423
D		DN 900	DN 1,000	DN 1,100	DN 1,200
E		2,340	2,564	2,650	3,160
F		762	802	842	1,140
* incl. 2 mm sealing					

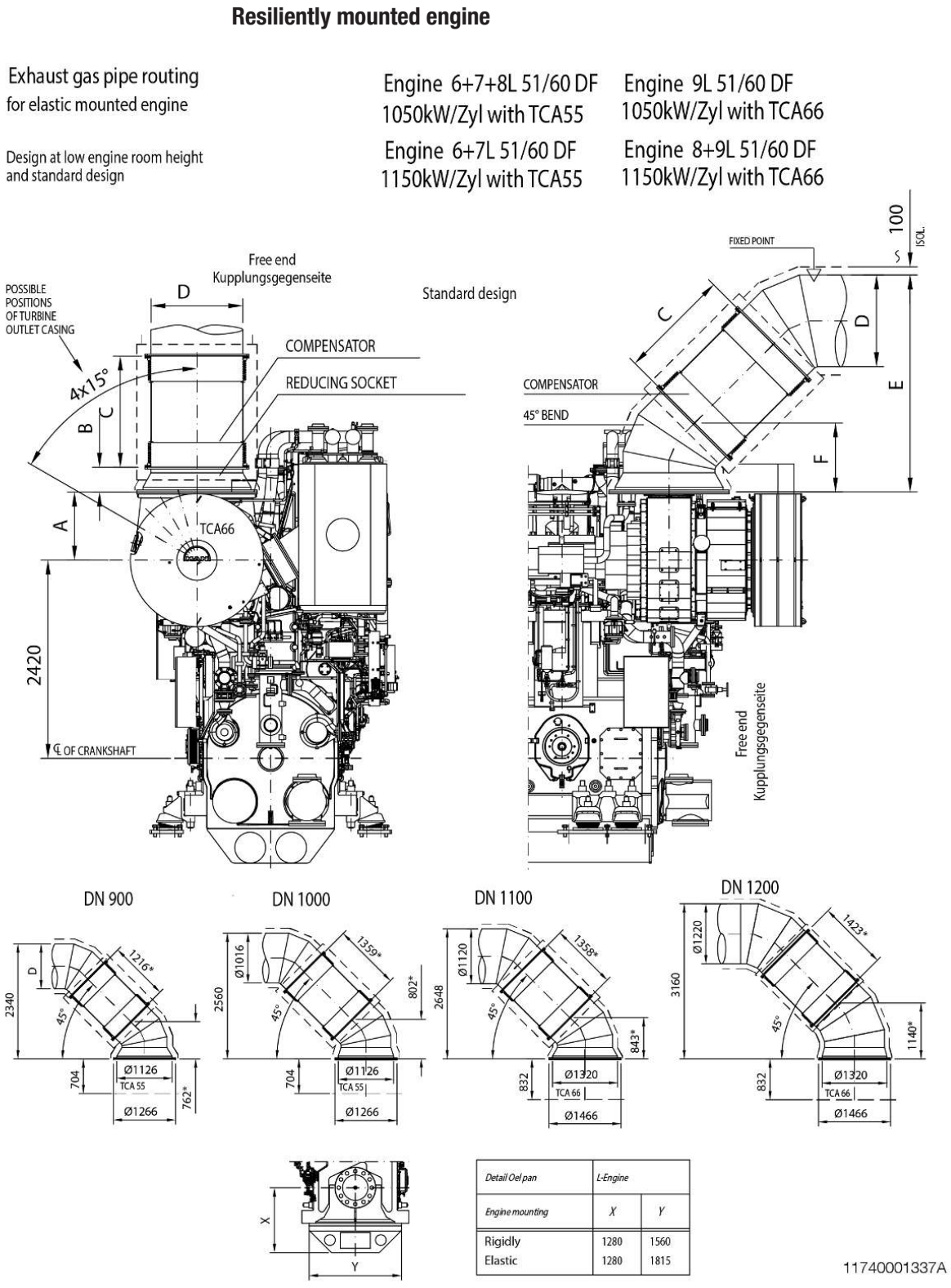

Figure 161: Design at low engine room height – Resiliently mounted engine, part 2

No. of cylinders, config.		6L	7L	8L	9L
Turbocharger/cyl. power (kW/cyl.)		TCA 55/1,050	TCA 55/1,050	TCA 55/1,050	TCA 66/1,050
A	mm	704	704	704	832
B*		302	302	302	302
C*		1,216	1,359	1,359	1,358
D		DN 900	DN 1,000	DN 1,000	DN 1,100
E		2,340	2,564	2,564	2,650
F		762	802	802	842
* incl. 2 mm sealing					

5.13 SCR system

5.13.1 General

The SCR system uses aqueous urea solution and a catalyst material to transform the pollutant nitrogen oxides into harmless nitrogen and water vapor. The main components of the SCR system are described in the following section.

For further information read section SCR – Special notes, Page 25.

5.13.2 As-delivered conditions and packaging

All components will be delivered and packaged in a seaworthy way (with dry agent, wooden boxing, shrink wrapped). Black carbon steel components will be coated with an anti-corrosive painting. Stainless steel components will not be coated.

The original packaging should not be removed until the date of installation.

The physical integrity of the packaging must be checked at the date of delivery.

5.13.3 Transportation and handling

Compressed air reservoir module (MOD-085)

Transport of the compressed air reservoir module can be organised by crane, via installed metal eyelets on the top side or fork-lifter.

Urea pump module (MOD-084)

Transport of the urea pump module can be organised by crane, via installed metal eyelets on the top side.

Dosing unit (MOD-082)

Transport of the dosing unit can be organised by crane, via installed metal eyelets on the top side.

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Urea injection lance and mixing unit (MOD-087)

Transport of the mixing unit can be organised by crane, via two installed metal eyelets. For horizontal lifting it is sufficient using one of the metal eyelets.

Using a vertical way, the two cables each fixed on one metal eyelet have to be stabilised by a transversal bar.

Note:
The metal eyelets are designed to carry only the segments of the mixing unit, further weights are not allowed (e.g. complete welded mixing pipe).

SCR reactor (R-001)

Transport of the reactor can be organised by crane, via installed metal eyelets on the top side.

SCR cabinet

Transport of the SCR cabinet can be organised by crane, via installed metal eyelets on the top side.

5.13.4 Storage

Compressed air reservoir module (MOD-085), urea pump module (MOD-084), dosing unit (MOD-082), SCR cabinet and sensor elements have to be stored in dry and weather-resistant conditions.

Catalyst elements shall be handled free from shocks and vibrations. Furthermore, catalyst elements have to be stored in dry and weather-resistant conditions. Keep oils or chemicals away from catalyst elements. Seaworthy packaging is only a temporary protection.

5.13.5 Components and assemblies of the SCR system

Catalyst elements

The catalyst elements are placed in metallic frames, so called modules. Due to the honeycomb structure of the catalyst elements, the catalytic surface is increased. The active component vanadium pentoxide (V_2O_5) in the surface supports the reduction of NO_x to harmless nitrogen.

The effectivity of the catalytic material decreases over time because of poisoning via fuel oil components or thermal impact. The durability depends on the fuel type and conditions of operation.

The status of catalyst deactivation is monitored continuously and the amount of urea injected is adapted according to the current status of the catalyst.

Compressed air reservoir module (MOD-085) and soot blowing system (MOD-086)

The compressed air required for the operation of the SCR system is provided by the compressed air reservoir module. It receives its compressed air via the ship´s compressed air grid. For the quality requirements read section Specification of compressed air, Page 293. The main supply line feeds the compressed air reservoir module, where a compressed air tank is installed. This high-pressure tank is a reservoir with enough capacity to ensure the supply of the dosing unit and the air consumption for the periodically cleaning of the catalysts´ surface, by avoiding fluctuations in the soot blowing system. In case

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of black out the volume of the tank will be used for flushing the urea line and nozzle. The module has to be positioned close to the reactor and the dosing unit. The maximum length of the compressed air line to the soot blowing system is 10 m.

The soot blower valves are positioned upstream each catalyst layer in order to clean the complete surface of the catalyst elements by periodical air flushing. The soot blowing always has to be in operation while engine running.

Urea pump module (MOD-084)

The urea pump module boosts urea to the dosing unit and maintains an adequate pressure in the urea lines. Upstream of the supply pump, a filter is installed for protection of solid pollutants. Downstream, the module is equipped with a return line to the urea storage tank with a pressure relief valve to ensure the required urea flow.

The urea pump module has to be positioned close to the urea storage tank. The pump accepts a suction height of 3.5 m and a maximum pressure loss of 2 bar. One urea pump module can supply up to four SCR systems.

For requirements regarding redundancy, contact MAN Energy Solutions.

Note:
Urea quality according section Specification of urea solution, Page 295 is required. For urea consumption calculation for Tier III read section Urea consumption for emission standard IMO Tier III, Page 95.

Dosing unit (MOD-082)

The dosing unit controls the flow of urea to the injection nozzle based on the operation of the engine. Furthermore it regulates the compressed air flow to the injector.

In order to avoid clogging due to the evaporation of urea in the urea pipe and in the nozzle, a line between compressed air line and urea line is installed. An installed solenoid valve will open to flush and cool the urea line and nozzle with compressed air before and after injecting urea into the exhaust gas.

The dosing unit has to be installed close to the urea injection lance and mixing unit (maximum pipe length 5 m).

Urea injection lance and mixing unit (MOD-087)

The urea solution will be injected into the exhaust gas using a two-phase nozzle. The urea will be atomised with compressed air. The evaporation of the urea occurs immediately when the urea solution gets in contact with the hot exhaust gas.

The urea injection and the mixing unit have to be positioned according to MAN Energy Solutions requirements. In general, the mixing section is between 3.0 – 4.5 m long and of DN 500 to DN 2,300. The mixing duct is a straight pipe upstream of the reactor. The exact length has to be calculated.

SCR reactor (R-001)

Each engine is equipped with its own SCR reactor and it is fitted in the exhaust gas piping without a by-pass. The SCR reactor housing is a steel structure. The reactor configuration is vertical and consists of several layers of catalysts. For horizontal installation, contact MAN Energy Solutions. The reactor is equipped with differential pressure and temperature monitoring, a maintenance door for service and the soot blowing system for each layer.

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The maximum temperature of the exhaust gas is 450 °C and a minimum exhaust gas temperature is required to ensure a reliable operation. Therefore temperature indicators are installed in the inlet and outlet of the reactor in order to monitor and control the optimum operating range.

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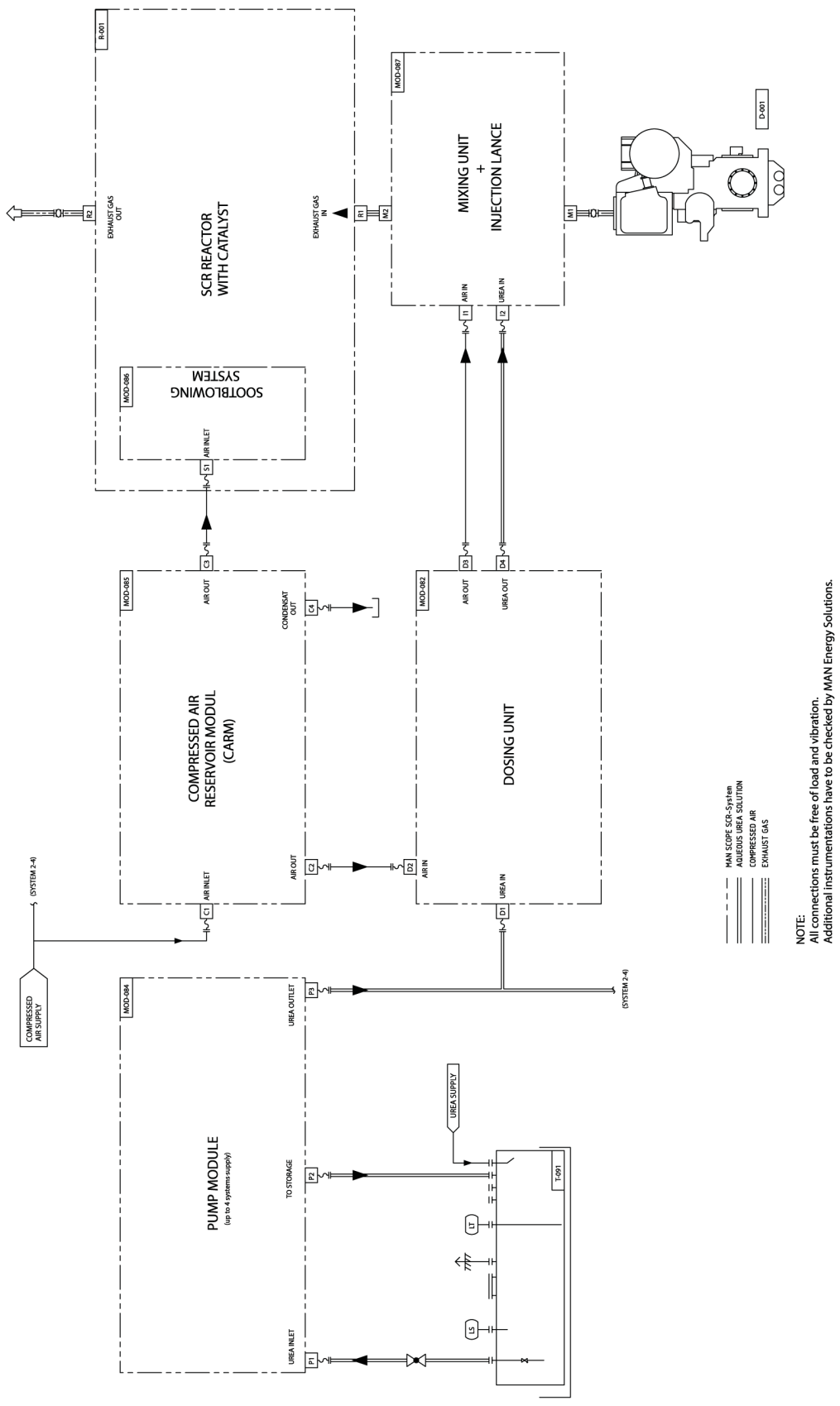

Figure 162: P&ID SCR system diagram

5.13.6 Installation of the SCR system

All modules are checked regarding pressure and tightness.

Catalyst elements

For handling the catalyst elements sufficient space and supply tracks have to be foreseen. Depending on the amount of catalyst elements transport devices like carriages, pulleys, fork lifter or elevators are required.

Reactor and soot blowing system

A service space of recommended 760 mm in front of the inspection doors of the reactor has to be foreseen for mounting and dismounting the catalyst elements, as well as for service and maintenance of the soot blower equipment and sensors.

Reactor and piping

In case of a bend before the reactor inlet, a bend radius of minimum one time of the exhaust gas pipe diameter has to be considered.

Mixing unit

The mixing unit is designed for vertical or horizontal installation. Bend on the downstream side has to be in accordance to above mentioned “Reactor and Piping”. Upstream of the mixing unit a bend can be installed according the MAN Energy Solutions requirements mentioned on the planning drawing.

5.13.7 Recommendations

All parts mentioned in this paragraph are not MAN Energy Solutions scope of supply.

Piping in general

All piping's have to be in accordance to section Specification of materials for piping, Page 299. Piping for fluids shall be mounted in an increasing/decreasing way. Siphons should be avoided, drainage system be foreseen.

Exhaust gas piping

The complete inside wall of the exhaust gas piping between engine outlet and SCR reactor inlet should not be coated by any protection material. Poisoning of the catalyst honeycombs could occur.

Intake air equipment, compressed air supply

Silicates in exhaust gas can cause capital damage of the catalyst elements of the exhaust gas after treatment system (SCR).

Therefore, it has to be ensured that no silicates can reach and poison the catalyst by related systems.

Possible sources for silicates could be e.g.:

- Intake air silencer of low quality (absorption material might get loose).
- Filters in compressed air system for SCR (adsorption filters containing silica gel).

Preferred materials

All materials used for the construction of tanks and containers including tubes, valves and fittings for storage, transportation and handling must be compatible with urea 40 % solution to avoid any contamination of urea and corrosion of device used. In order to guarantee the urea quality the following materials for tank, pipes and fittings are compatible: Stainless steel (1.4301 or 1.4509) or urea-resistant plastics (e.g. PA12). For gaskets EPDM or FKM. Piping for compressed air see section Specification of materials for piping, Page 299.

Unsuitable materials

Unsuitable materials for tank, pipes and fittings are among others: Aluminum, unalloyed steel, galvanised steel, copper and brass.

In case incompatible material is used, clogging of urea filter inside the pump module may occur, or even worse, the catalyst elements may be damaged by catalyst poisons derived from this material. In this case, exchanging the catalyst modules may be necessary.

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Urea tank

Store this material in cool, dry, well-ventilated areas. Regarding the urea storage temperature, the requirements of the respective manufacturer information or applicable standards (e.g. ISO 18611-3) are to be observed. The storage capacity of the urea tank should be designed depending on ship load profile and bunker cycle.

The urea supply line should be provided with a strainer and a non-return valve in order to assure a correct performance for the suction of the urea pump, which is installed downstream the tank. A level switch with the possibility to read out the signal will protect the pump of a dry run. A return line from the urea pump module over a pressure relief valve is entering the tank.

Urea solution quality

Use of good quality urea is essential for the operation of an SCR catalyst. Using urea not complying with the specification below e.g. agricultural urea, can either cause direct operational problems or long term problems like deactivation of the catalyst. For quality requirements, see section Specification of urea solution, Page 295.

Insulation

The quality of the insulation has to be in accordance with the safety requirements. All insulations for service and maintenance spaces have to be dismountable. The delivered modules have no fixations, if fixations are necessary take care about the permissible material combination. Regarding max. permissible thermal loss see section Boundary conditions for SCR operation, Page 27.

Water trap

Water entry into the reactor housing must be avoided, as this can cause damage and clogging of the catalyst. Therefore a water trap has to be installed, if the exhaust pipe downstream of the SCR reactor is facing upwards.

5.14 Maintenance space and requirements

5.14.1 General details

Apart from a functional arrangement of the components, the shipyard is to provide for an engine room layout ensuring good accessibility of the components for servicing.

The cleaning of the cooler tube bundle, the emptying of filter chambers and subsequent cleaning of the strainer elements, and the emptying and cleaning of tanks must be possible without any problem whenever required.

All of the openings for cleaning on the entire unit, including those of the exhaust silencers, must be accessible.

There should be sufficient free space for temporary storage of pistons, camshafts, turbocharger etc. dismounted from the engine. Additional space is required for the maintenance personnel. The panels on the engine sides for inspection of the bearings and removal of components must be accessible without taking up floor plates or disconnecting supply lines and piping. Free space for installation of a torsional vibration meter should be provided at the crankshaft end.

A very important point is that there should be enough room for storing and handling vital spare parts so that replacements can be made without loss of time.

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In planning marine installations with two or more engines driving one propeller shaft through a multi-engine transmission gear, provision must be made for a minimum clearance between the engines because the crankcase panels of each engine must be accessible. Moreover, there must be free space on both sides of each engine for removing pistons or cylinder liners.

Note:
MAN Energy Solutions delivered scope of supply is to be arranged and fixed by proven technical experiences as per state of the art. Therefore the technical requirements have to be taken in consideration as described in the following documents subsequential:

- Order related engineering documents.
- Installation documents of our sub-suppliers for vendor specified equipment.
- Operating manuals for diesel engines and auxiliaries.
- Project Guides of MAN Energy Solutions.

Any deviations from the principles specified in the aforementioned documents require a previous approval by MAN Energy Solutions.

Arrangements for fixation and/or supporting of plant related equipment deviating from the scope of supply delivered by MAN Energy Solutions, not described in the aforementioned documents and not agreed with us are not permissible.

For damages due to such arrangements we will not take over any responsibility nor give any warranty.

5.14.2 Installation drawings

Note:
Specific requirements to the passageway e.g. of the classification societies or flag state authority may result in a higher space demand.

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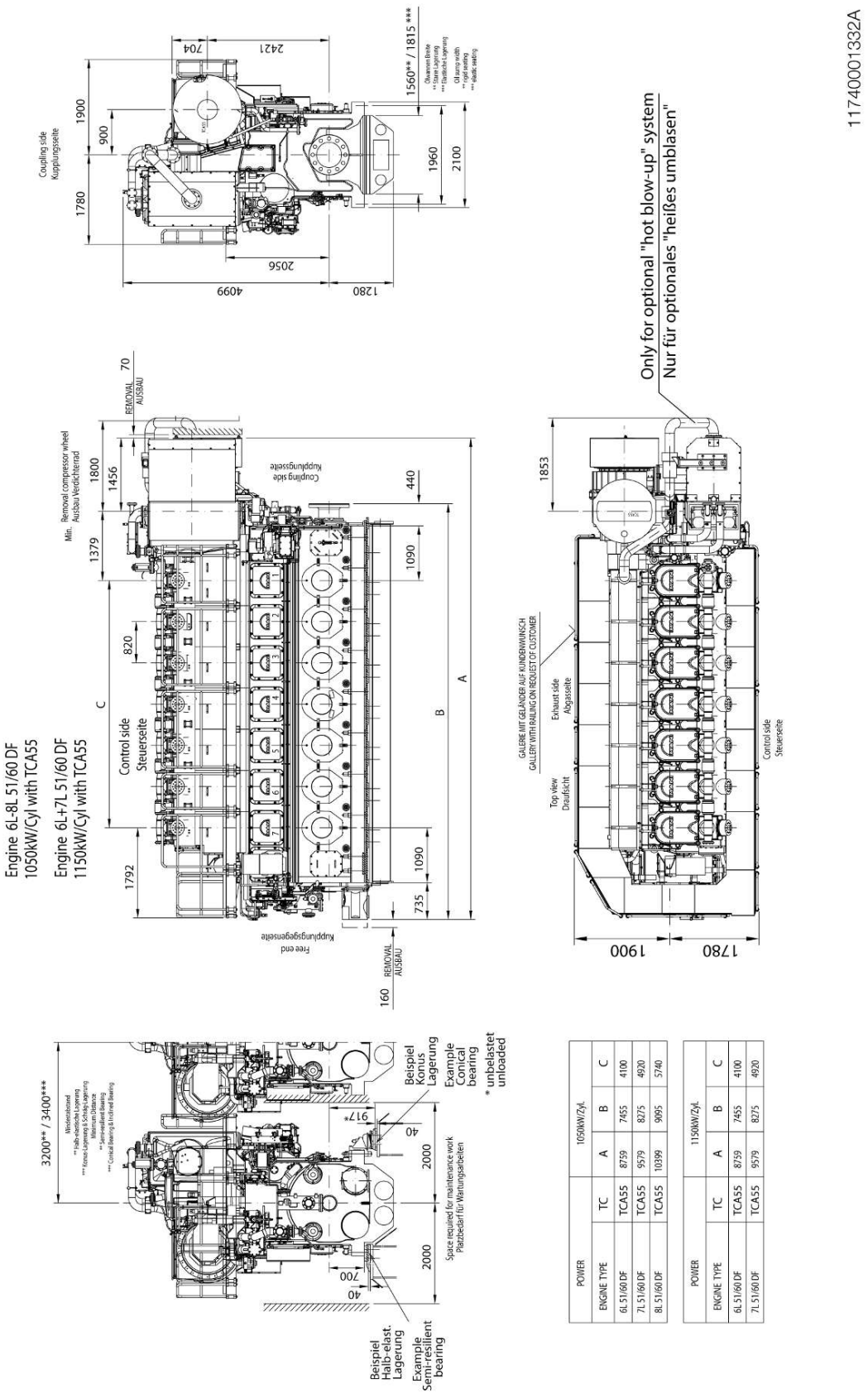
6L, 7L, 8L engine



POWER	1050kW/Zyl.			
ENGINE TYPE	TC	A	B	C
6L 51/60 DF	TCA55	8739	7455	4100
7L 51/60 DF	TCA55	9579	8275	4920
8L 51/60 DF	TCA55	10399	9095	5740

POWER	1150kW/Zyl.			
ENGINE TYPE	TC	A	B	C
6L 51/60 DF	TCA55	8739	7455	4100
7L 51/60 DF	TCA55	9579	8275	4920

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Figure 163: Installation drawing 6L, 7L, 8L engine – Turbocharger on coupling side

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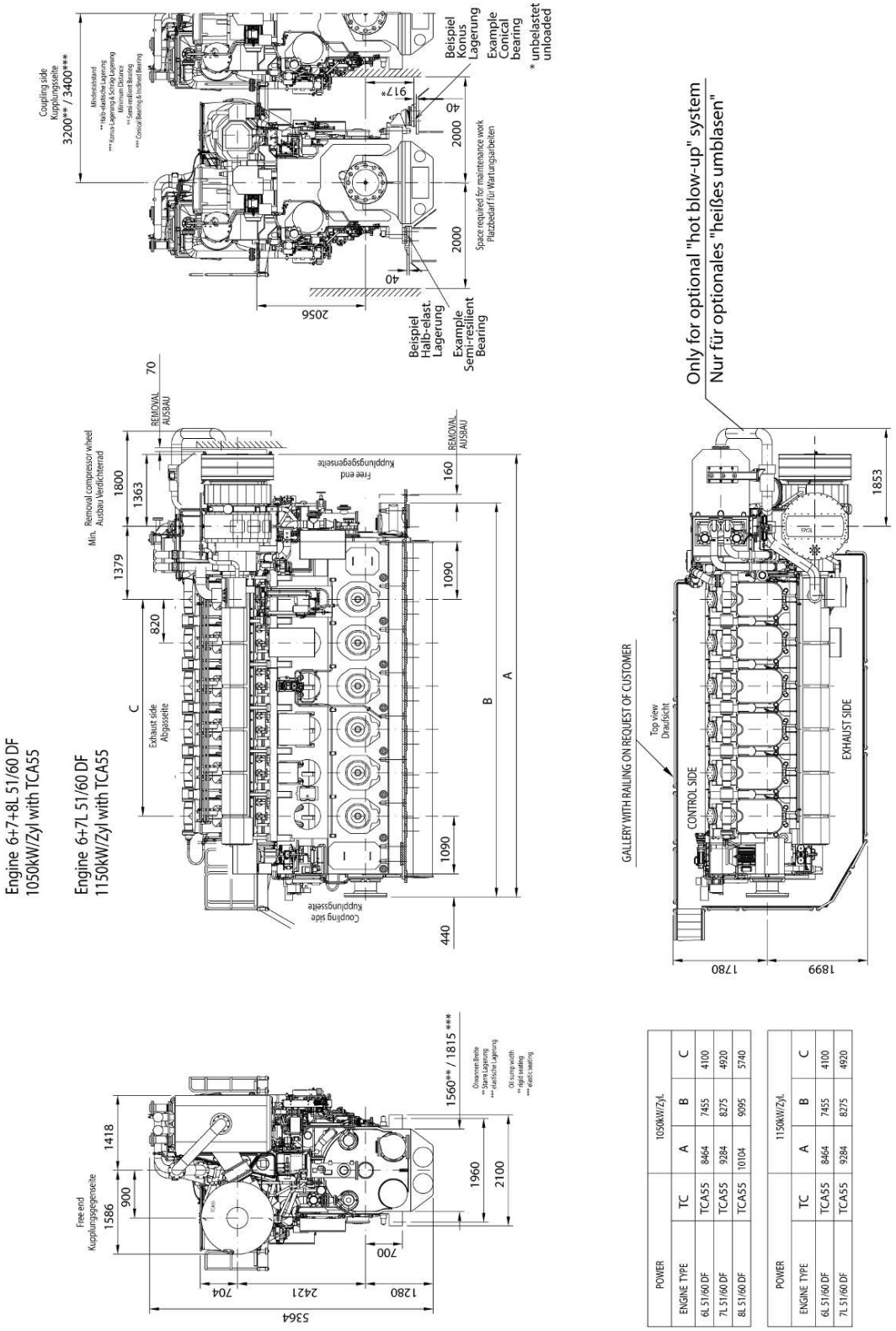
6L, 7L, 8L engine



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POWER	1050kW/Zyl.			
ENGINE TYPE	TC	A	B	C
6L 51/60 DF	TCA55	8464	7455	4100
7L 51/60 DF	TCA55	9284	8275	4920
8L 51/60 DF	TCA55	10104	9095	5740

POWER	1150kW/Zyl.			
ENGINE TYPE	TC	A	B	C
6L 51/60 DF	TCA55	8464	7455	4100
7L 51/60 DF	TCA55	9284	8275	4920

Figure 164: Installation drawing 6L, 7L, 8L engine – Turbocharger on counter coupling side

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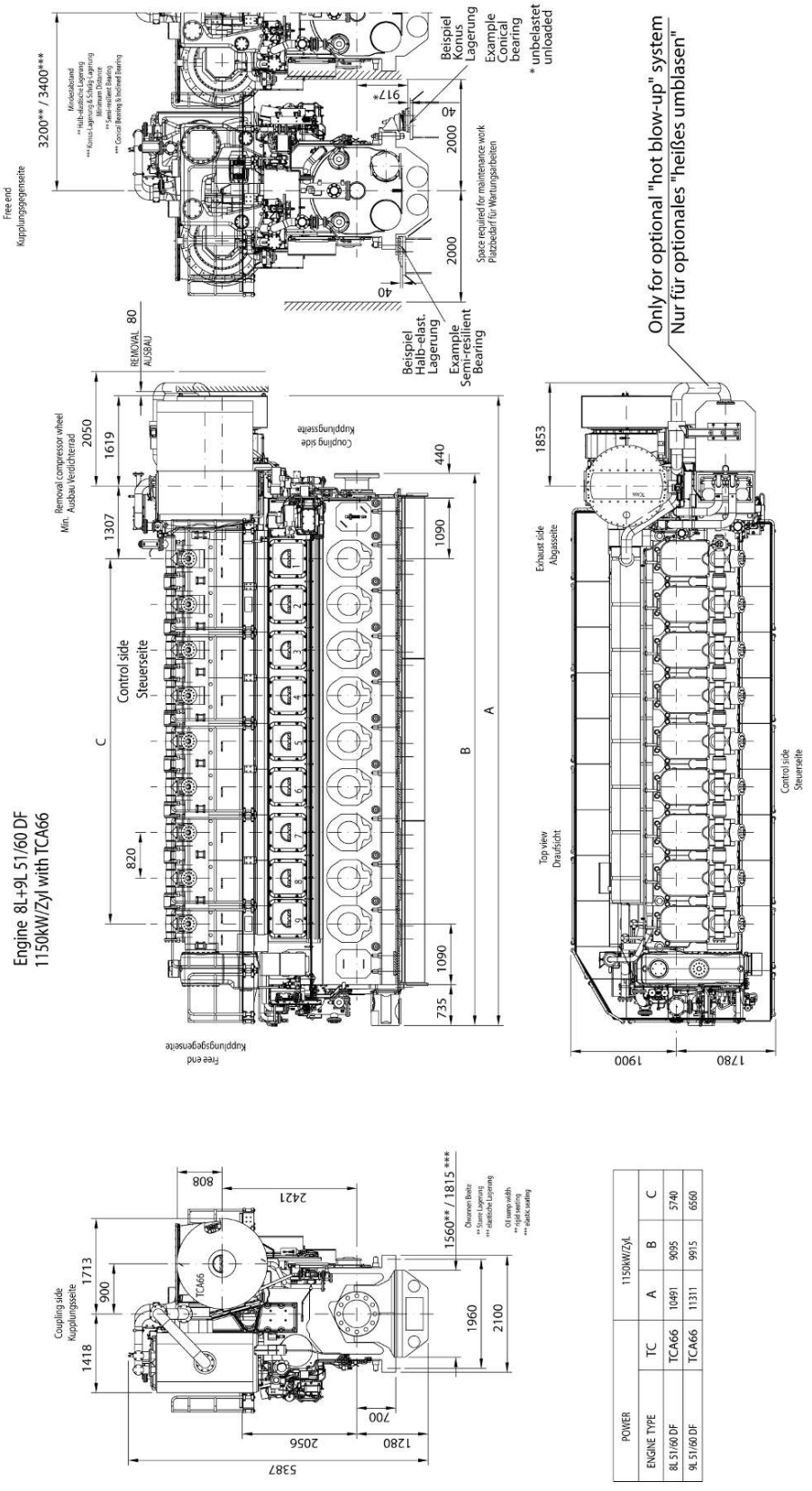
8L, 9L engine



POWER		1150kW/Zyl.		
ENGINE TYPE	TC	A	B	C
8L 51/60 DF	TCA66	10491	9095	5740
9L 51/60 DF	TCA66	11311	9915	6560

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Figure 165: Installation drawing 8L, 9L engine – Turbocharger on coupling side

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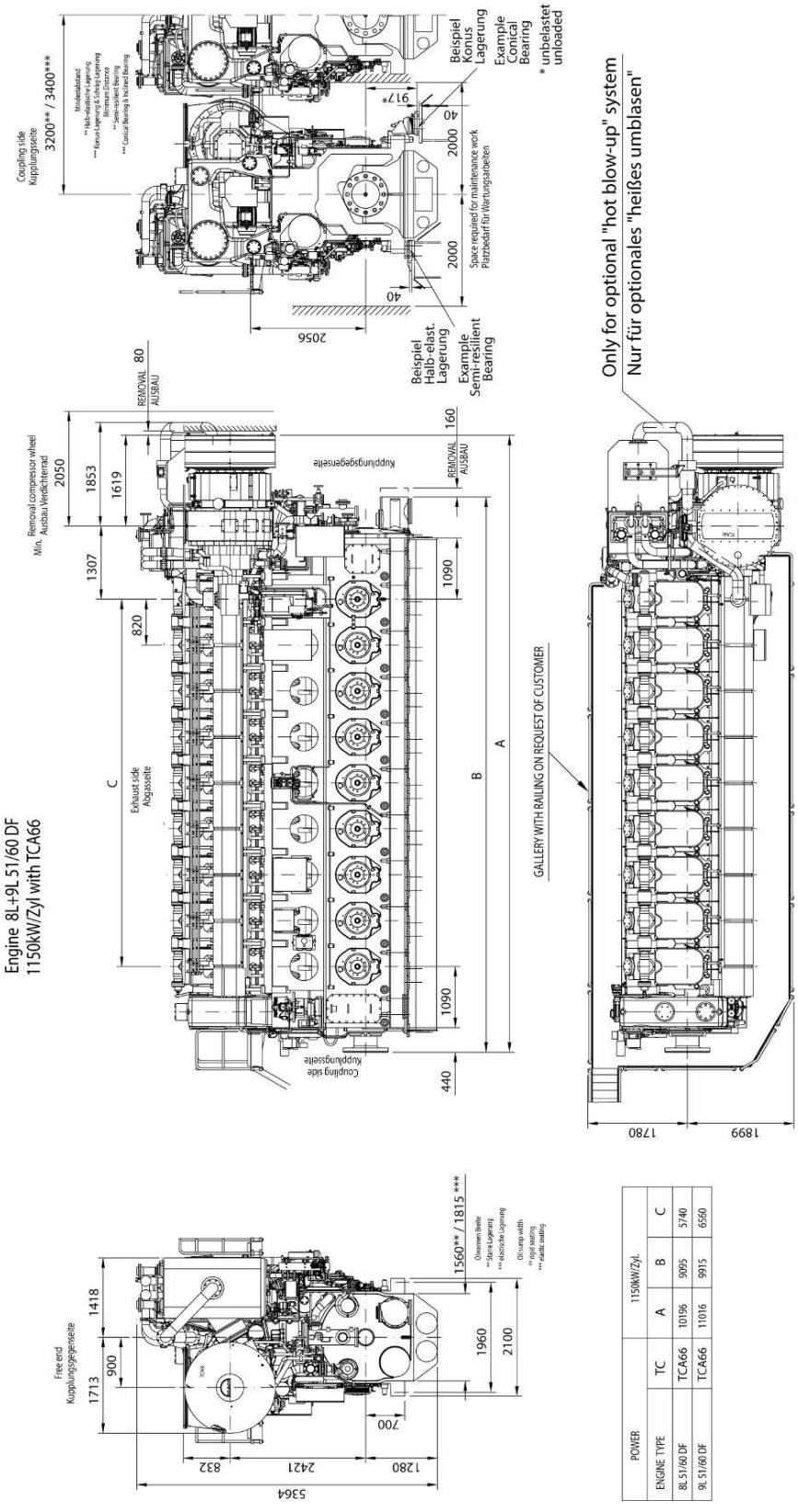
8L, 9L engine



POWER		1150kW/Zyl.		
ENGINE TYPE	TC	A	B	C
8L 51/60 DF	TCA66	10196	9095	5740
9L 51/60 DF	TCA66	11016	9915	6560

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Figure 166: Installation drawing 8L, 9L engine – Turbocharger on counter coupling side

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12V engine

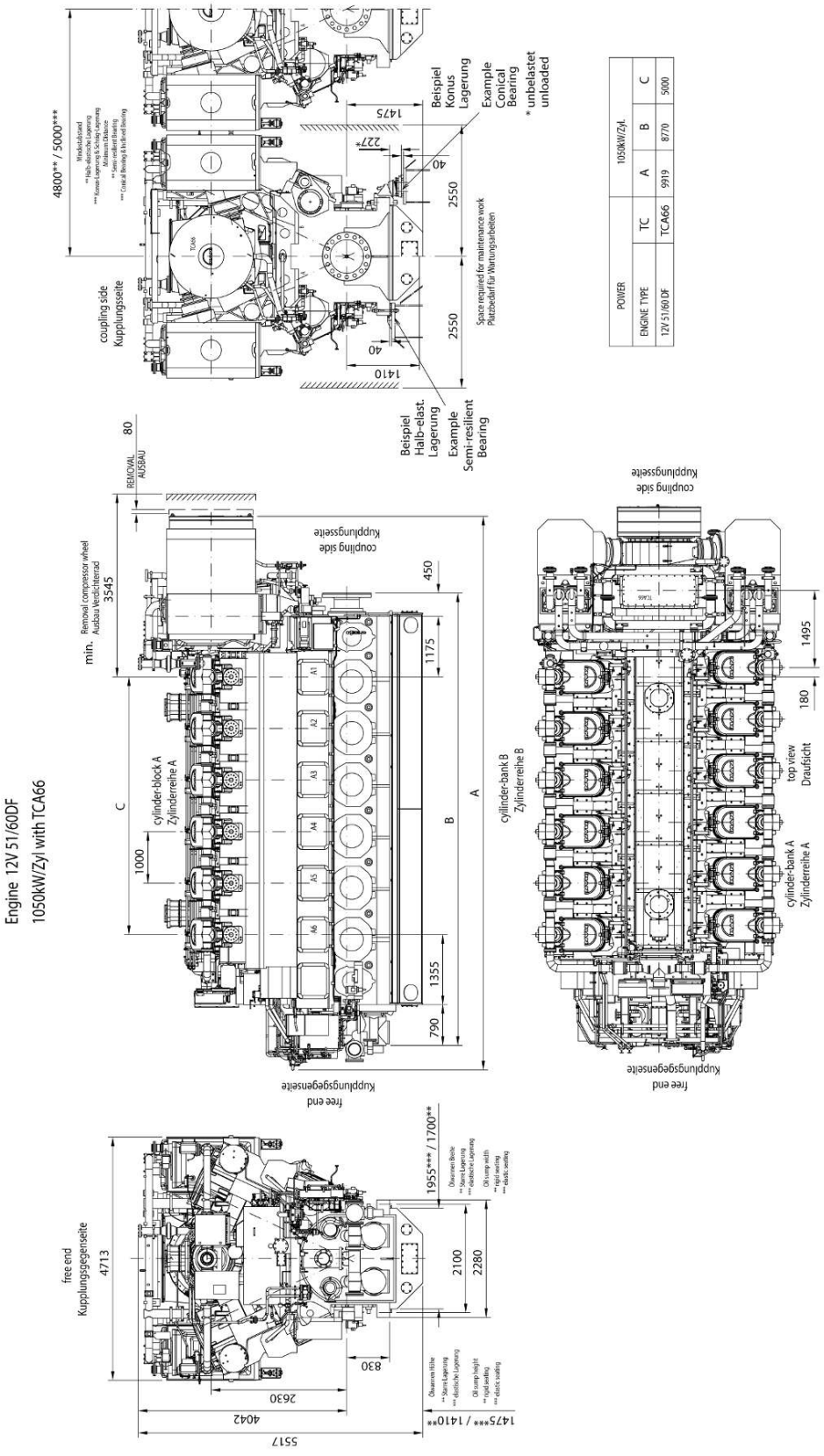

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Figure 167: Installation drawing 12V engine – Turbocharger on coupling side

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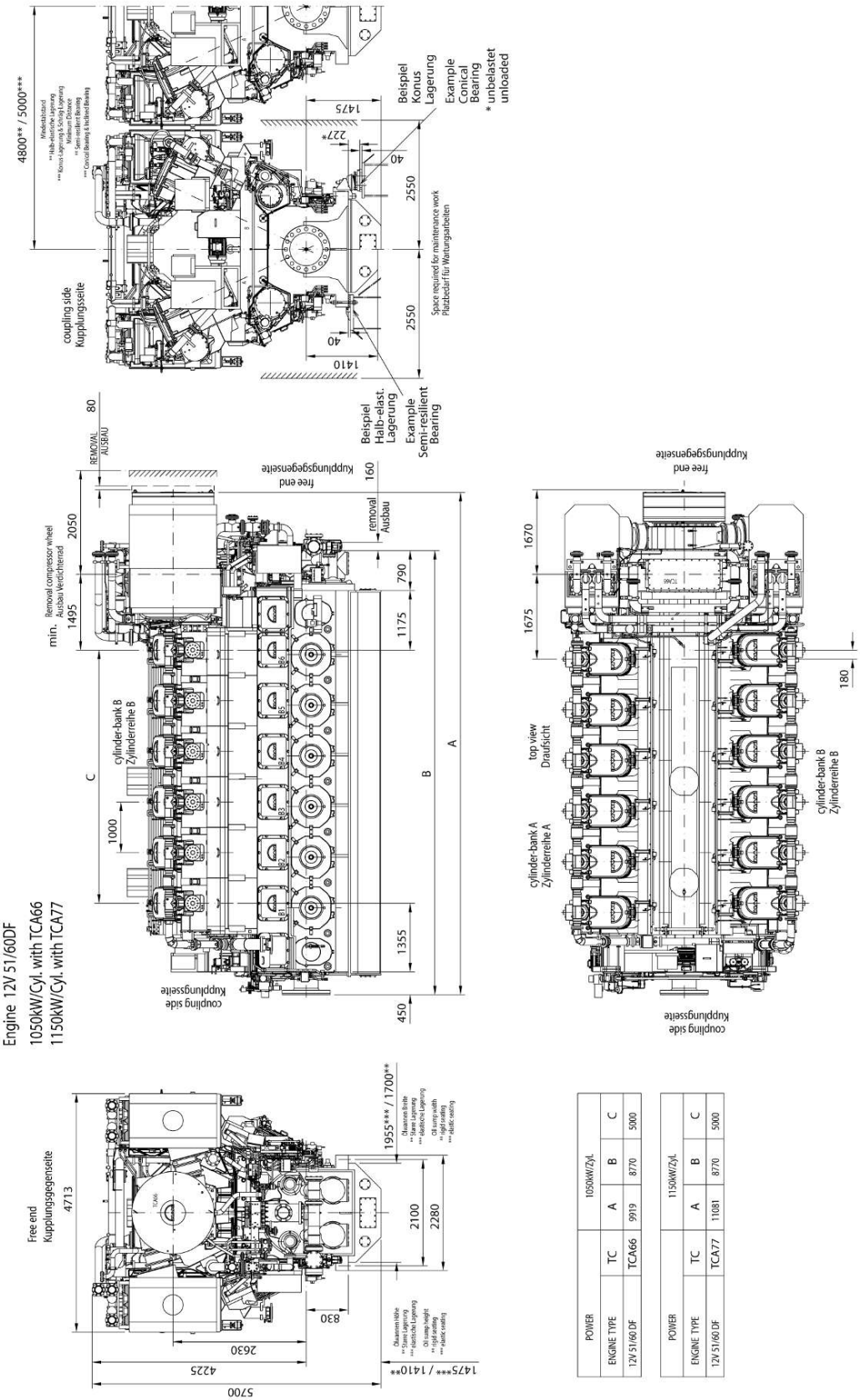
12V engine



POWER	1050kW/Cyl.		
ENGINE TYPE	TC	A	B	C
12V 51/60 DF	TCA66	9919	8770	5000

POWER	1150kW/Cyl.		
ENGINE TYPE	TC	A	B	C
12V 51/60 DF	TCA77	11081	8770	5000

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5 Engine room and application planning

5.14 Maintenance space and requirements

Figure 168: Installation drawing 12V engine – Turbocharger on counter coupling side

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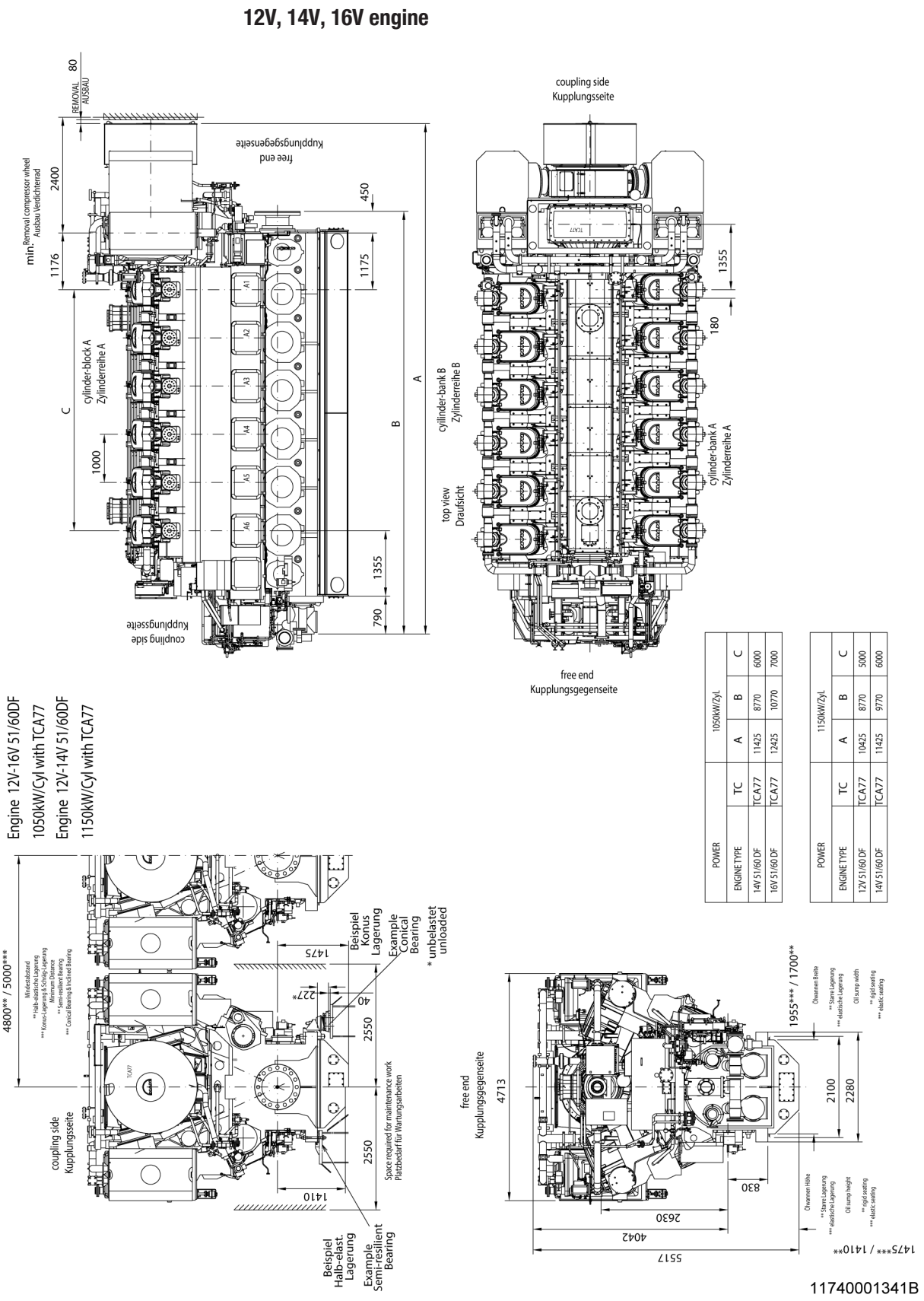
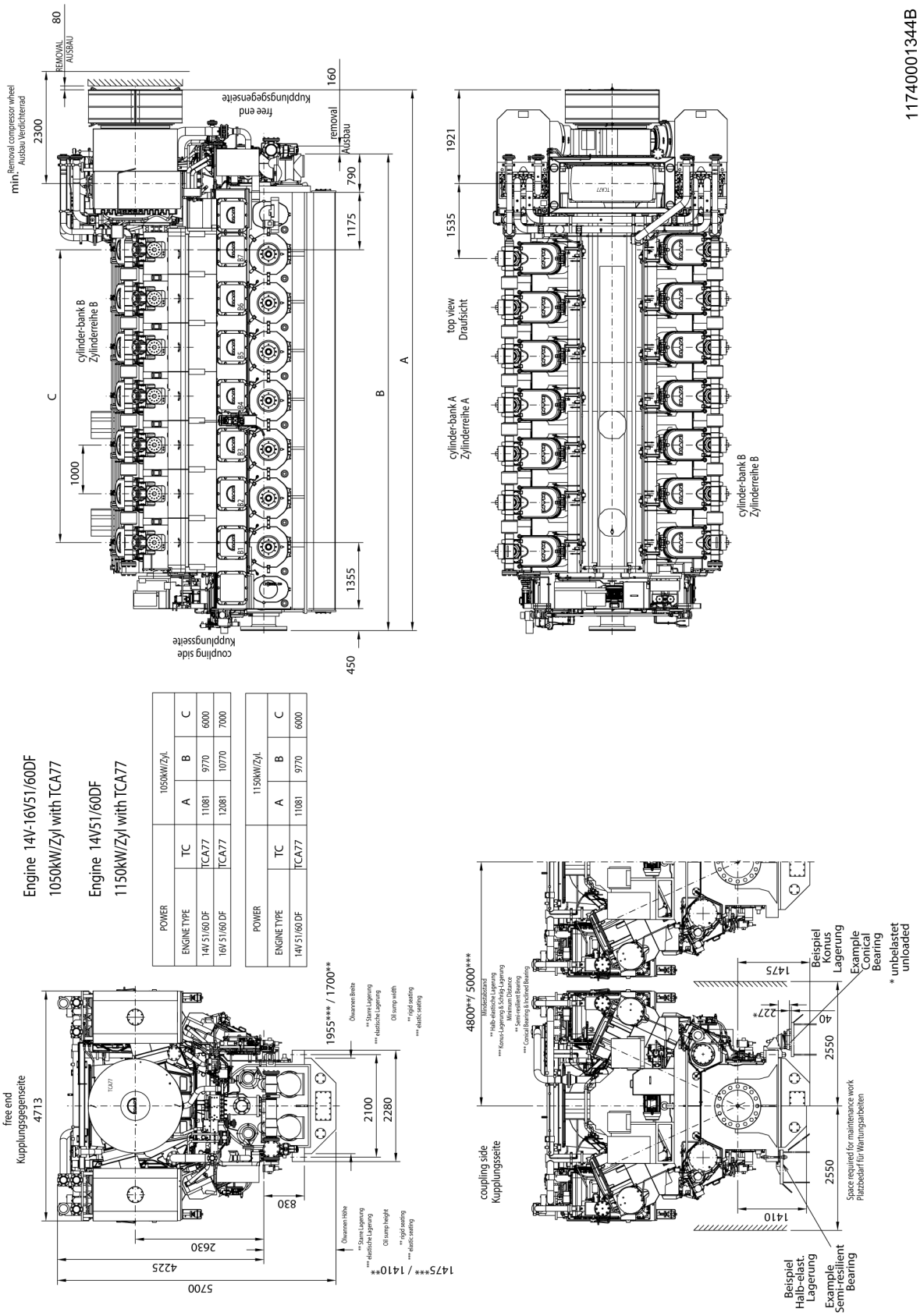


Figure 169: Installation drawing 12V, 14V, 16V engine – Turbocharger on coupling side

5 Engine room and application planning

5.14 Maintenance space and requirements

14V, 16V engine



Engine 14V-16V51/60DF
1050kW/Zyl with TCA77

POWER	1050kW/Zyl.			
ENGINE TYPE	TC	A	B	C
14V 51/60 DF	TCA77	11081	9770	6000
16V 51/60 DF	TCA77	12081	10770	7000

Engine 14V51/60DF
1150kW/Zyl with TCA77

POWER	1150kW/Zyl.			
ENGINE TYPE	TC	A	B	C
14V 51/60 DF	TCA77	11081	9770	6000

117400013448

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Figure 170: Installation drawing 14V, 16V engine – Turbocharger on counter coupling side

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5 Engine room and application planning

5.14 Maintenance space and requirements

16V engine

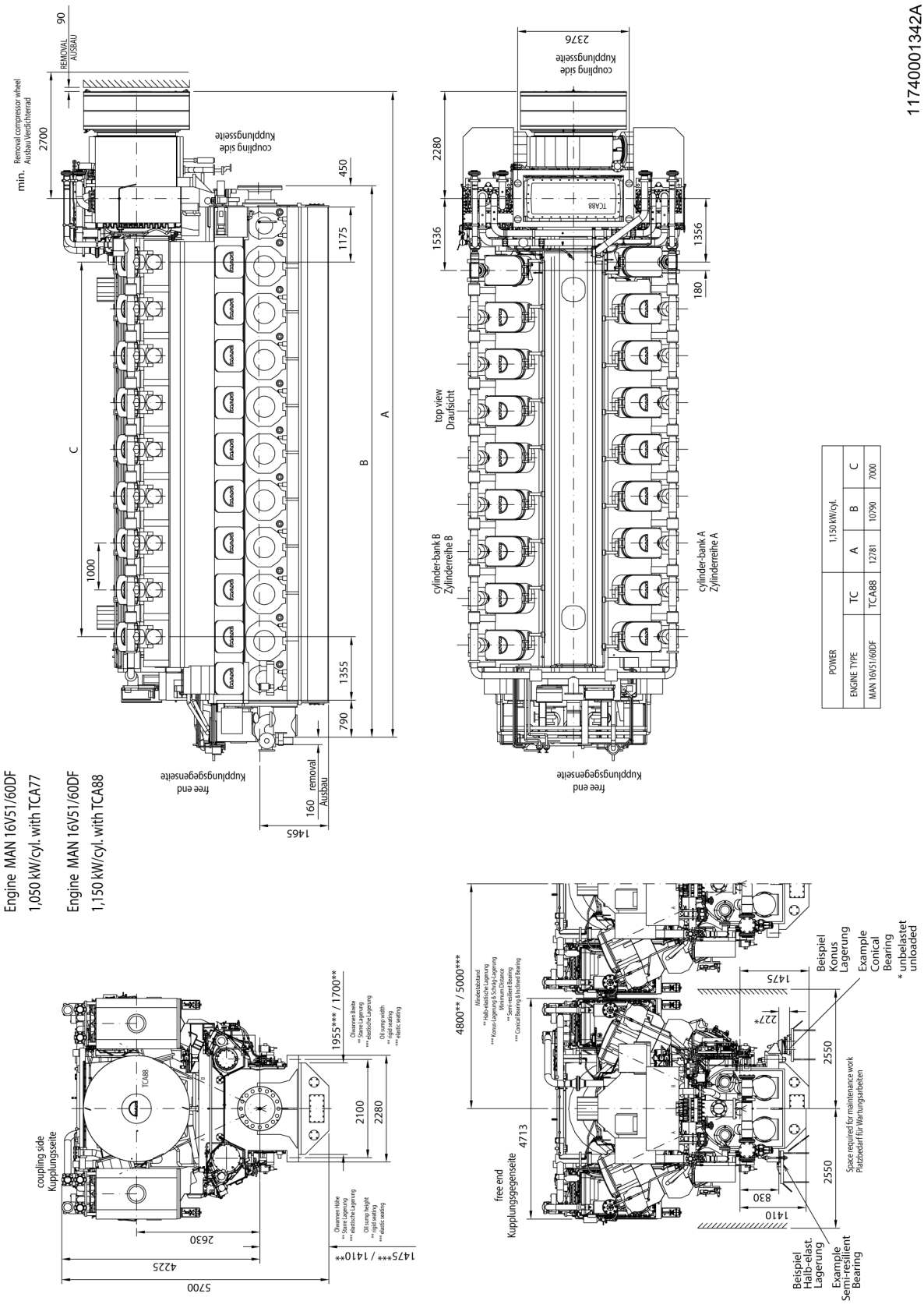

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Figure 171: Installation drawing 16V engine – Turbocharger on coupling side

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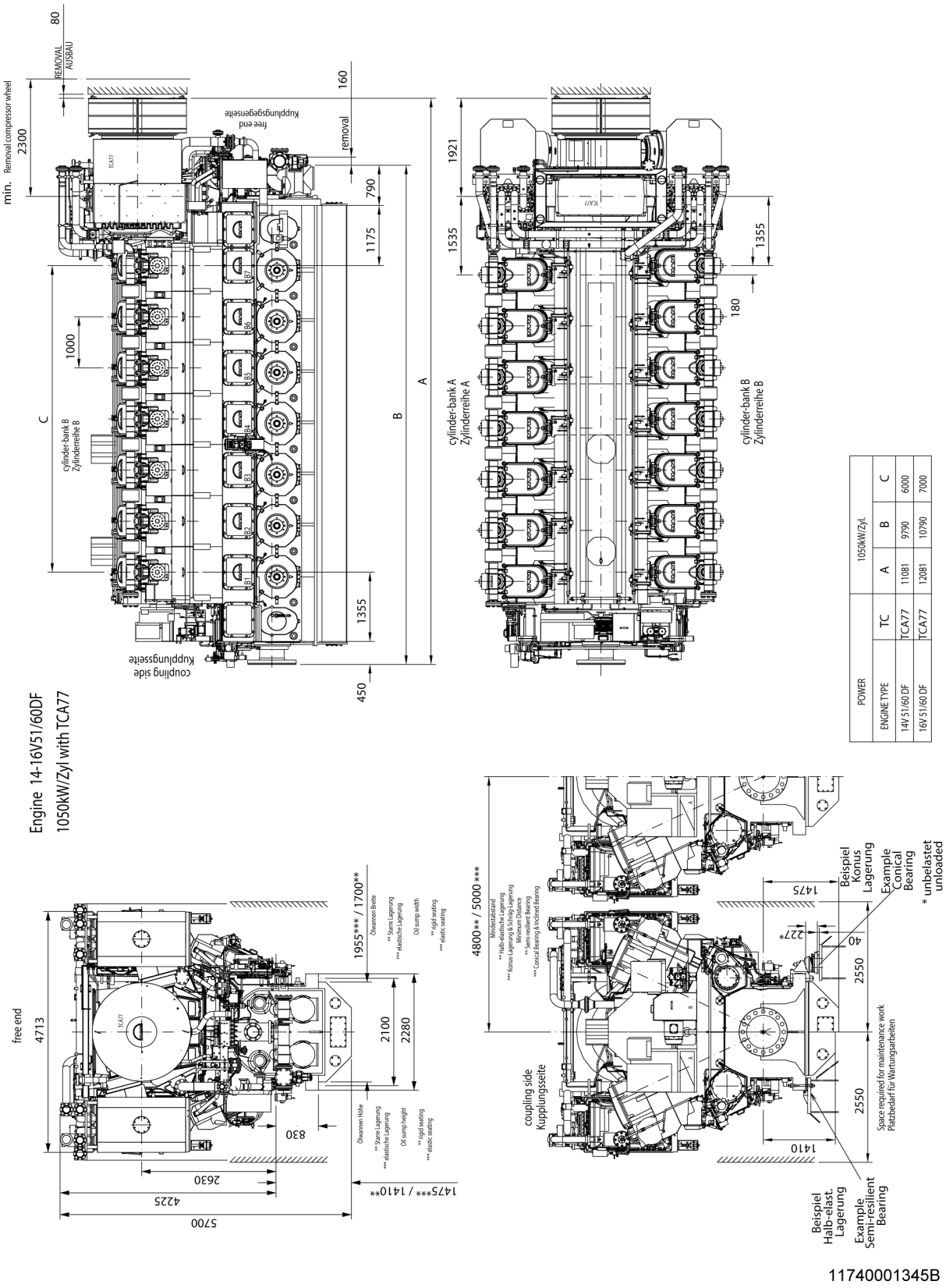
16V engine



POWER	1050kW/Zyl.			
ENGINE TYPE	TC	A	B	C
14V 51/60 DF	TCA77	11081	9790	6000
16V 51/60 DF	TCA77	12081	10790	7000

11740001345B

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Figure 172: Installation drawing 16V engine – Turbocharger on counter coupling side

5.14.3 Removal dimensions of piston and cylinder liner

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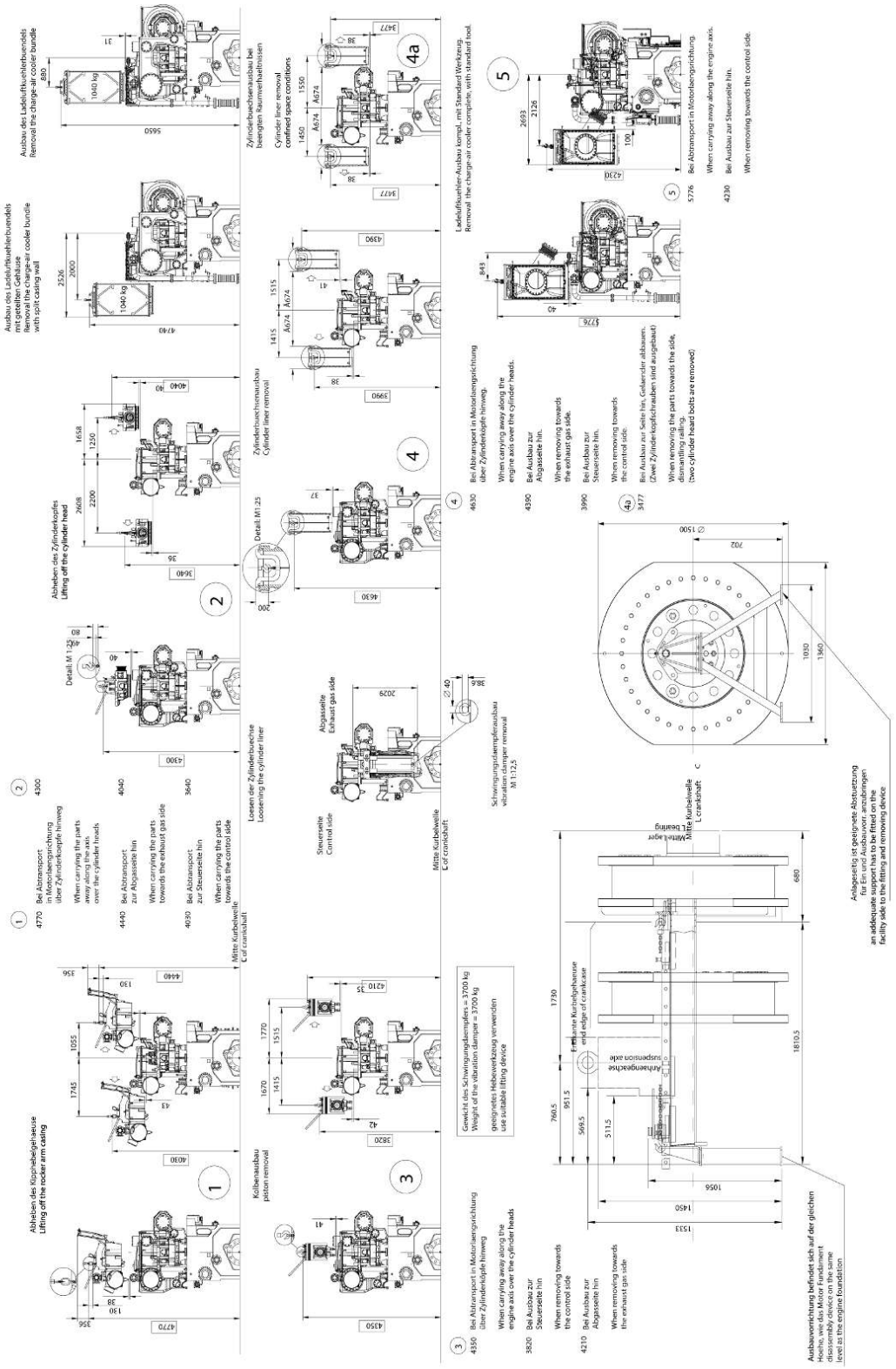

Figure 173: Removal dimensions of piston and cylinder liner – MAN L51/60DF

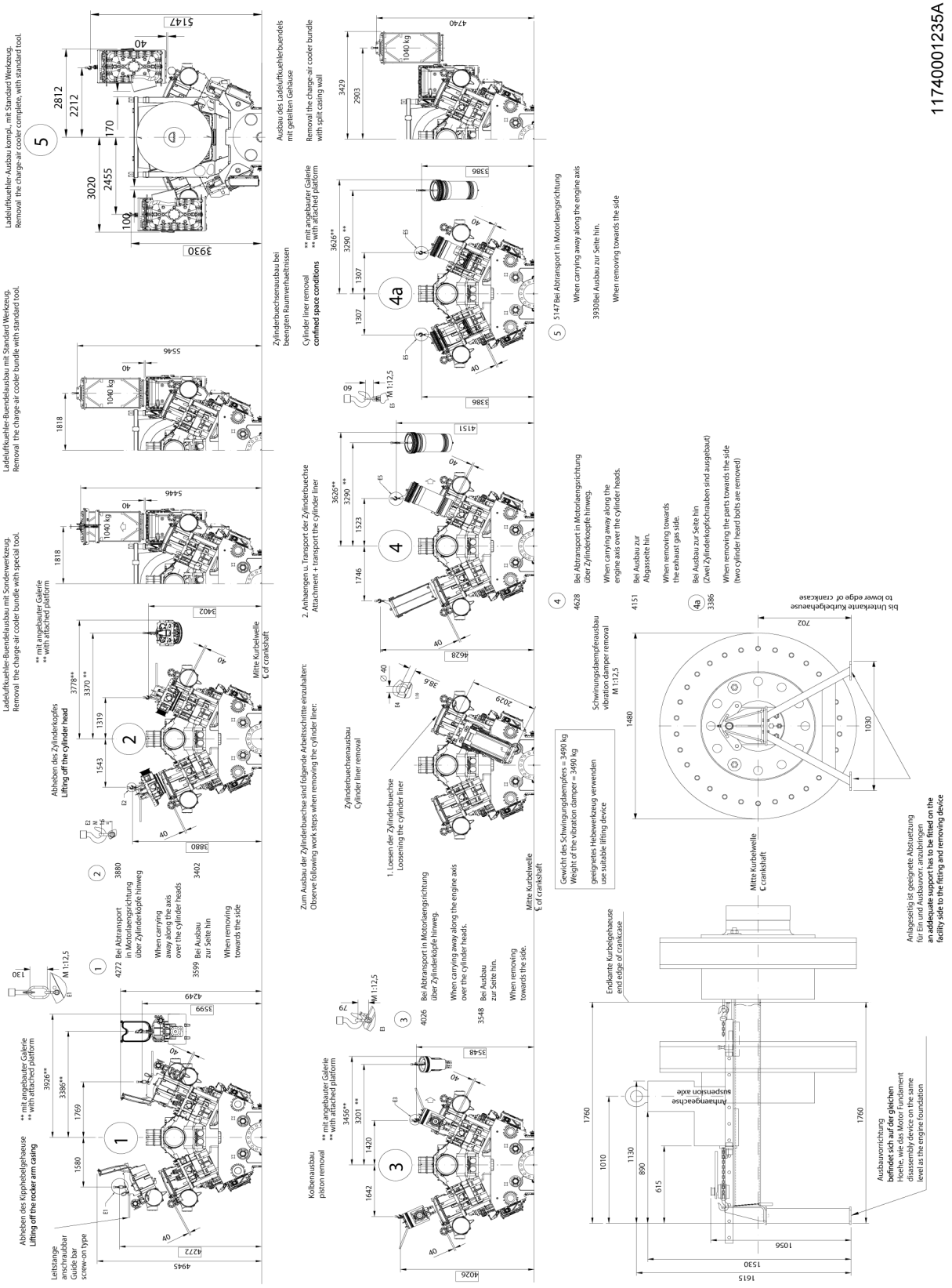

Figure 174: Removal dimensions of piston and cylinder liner – MAN V51/60DF

5.14.4 Engine arrangements

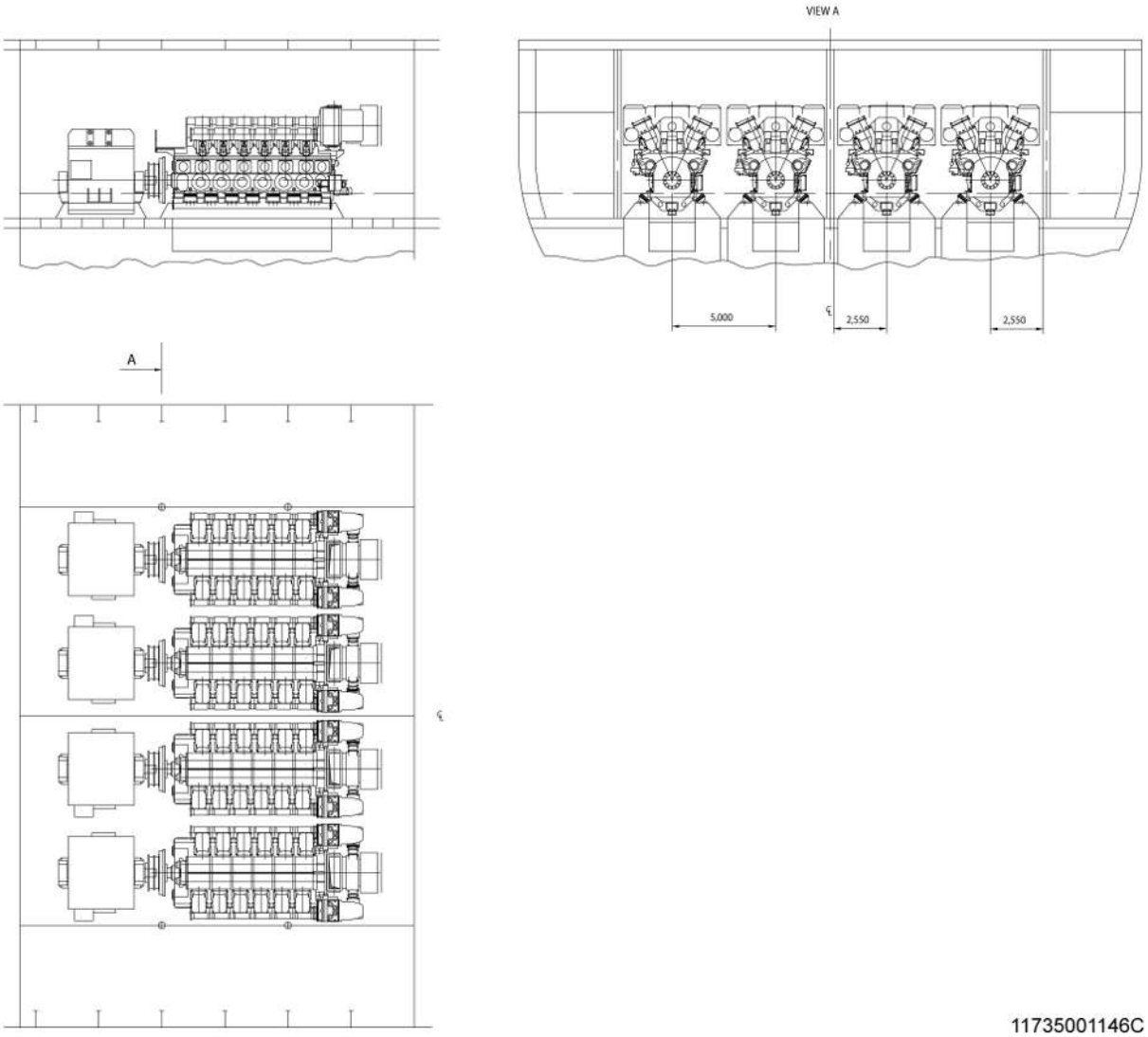

Figure 175: Example: Arrangement with engine 12V engine

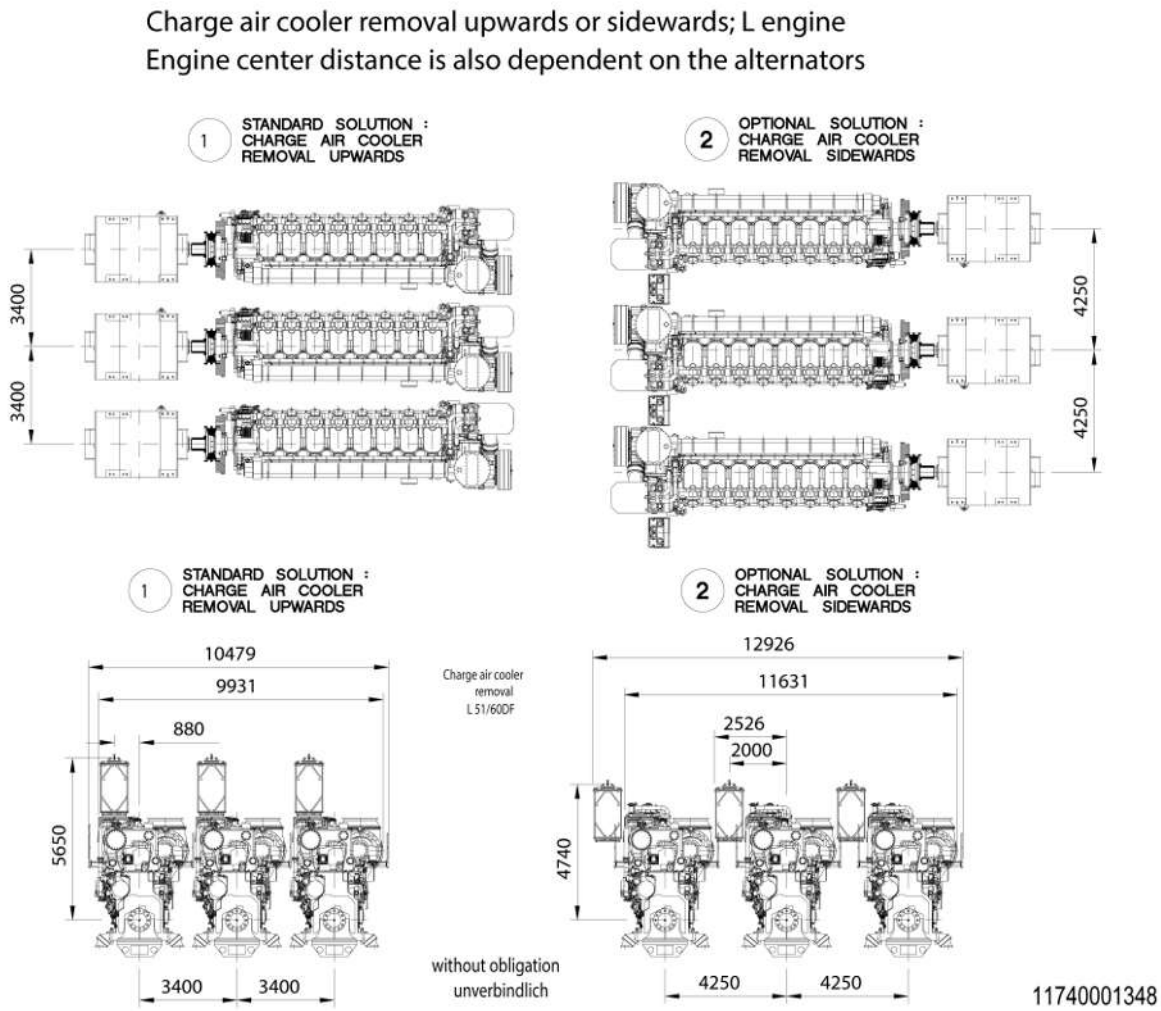

Figure 176: Charge air cooler removal upwards or sidewards; L engine

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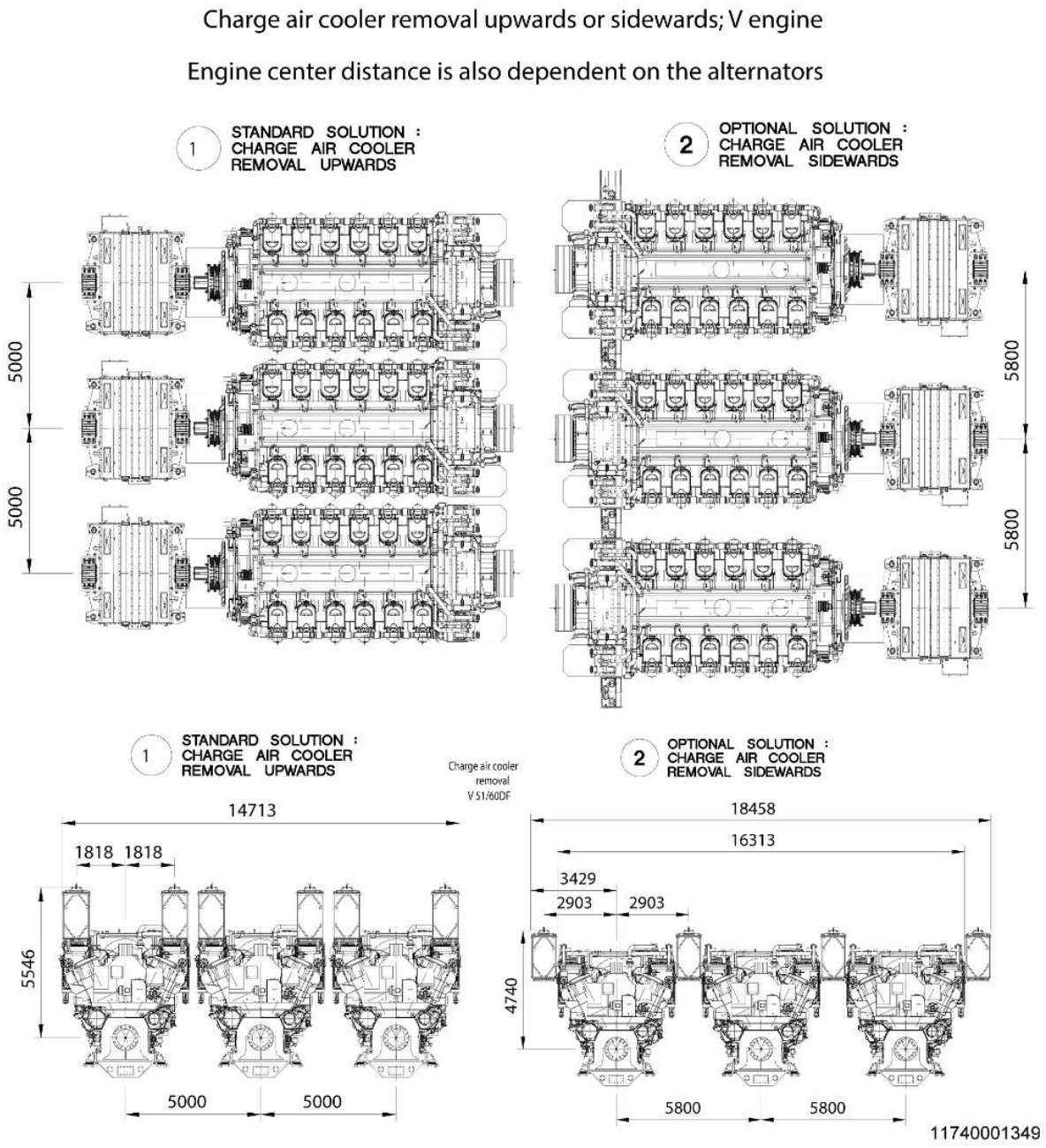

Figure 177: Charge air cooler removal upwards or sidewards; V engine

5.14.5 Lifting device

Lifting gear with varying lifting capacities are to be provided for servicing and repair work on the engine, turbocharger and charge air cooler.

Engine

Component weights

For servicing the engine an overhead traveling crane is required. The lifting capacity shall be sufficient to handle the heaviest component that has to be lifted during servicing of the engine and should foresee extra capacity e.g. to overcome the break loose torque while lifting cylinder heads. The overhead traveling crane can be chosen with the aid of the following table:

Components	Unit	Approximate weights
Cylinder head complete	kg	1,250
Piston with piston pin and connecting rod (for piston removal)		560
Cylinder liner		590
Charge air cooler		1,040

Table 221: Component weights

Crane arrangement

The rails for the crane are to be arranged in such a way that the crane can cover the whole of the engine beginning at the exhaust pipe.

The hook position must reach along the engine axis, past the centreline of the first and the last cylinder, so that valves can be dismantled and installed without pulling at an angle. Similarly, the crane must be able to reach the tie rod at the ends of the engine. In cramped conditions, eyelets must be welded under the deck above, to accommodate a lifting pulley.

The required crane capacity is to be determined by the crane supplier.

Crane design

It is necessary that:

- There is an arresting device for securing the crane while hoisting if operating in heavy seas
- There is a two-stage lifting speed

  Precision hoisting approximately = 0.5 m/min

  Normal hoisting approximately = 2 – 4 m/min

Places of storage

In planning the arrangement of the crane, a storage space must be provided in the engine room for the dismantled engine components which can be reached by the crane. It should be capable of holding two rocker arm casings, two cylinder covers and two pistons. If the cleaning and service work is to be carried out here, additional space for cleaning troughs and work surfaces should be planned.

Transport to the workshop

Grinding of valve cones and valve seats is carried out in the workshop or in a neighbouring room.

Transport rails and appropriate lifting tackle are to be provided for the further transport of the complete cylinder cover from the storage space to the workshop. For the necessary deck openings, see following figures and tables.

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Turbocharger

Section Turbocharger assignments, Page 31 shows which turbocharger type should be used for which engine variant.

Turbocharger dimensions

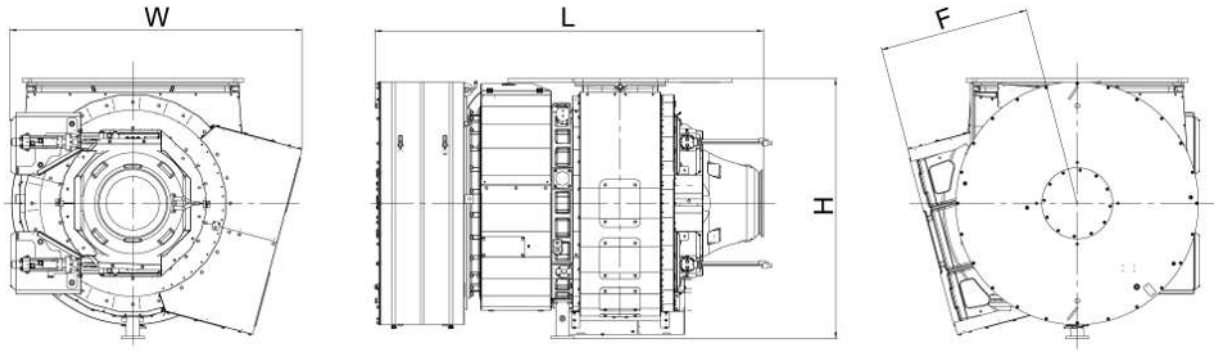

Figure 178: Exemplary illustration of TCA55

Turbocharger type	CS/CCS	L[1] [mm]	W[2] [mm]	H [mm]	F (casing foot) [mm]
TCA55	CS	2,193	1,651	1,469	850
TCA55	CCS	2,223	1,648	1,469	850
TCA66	CS	2,406	1,761	1,697	850
TCA66	CCS	2,406	1,823	1,697	850

[1] Valid for silencer. Values differ for suction pipe or casing.

[2] Valid for gas outlet casing 0°. For different mountig angles of the gas outlet casing refer to section Position of the outlet casing of the turbocharger, Page 432.

Table 222: Dimensions – TCA55, TCA66 on L engine

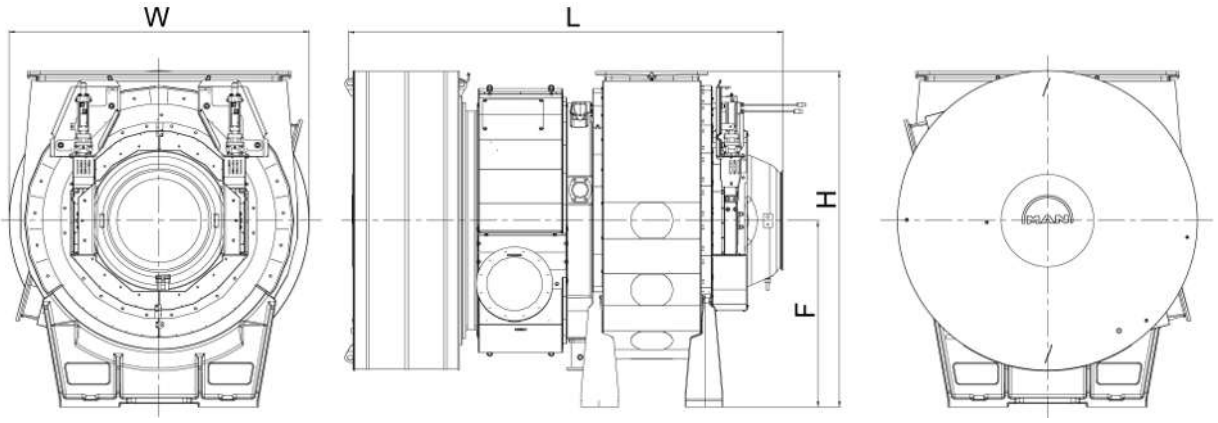

Figure 179: Exemplary illustration of TCA77

Turbocharger type	CS/CCS	L[1] [mm]	W [mm]	H [mm]	F (casing foot) [mm]
TCA66	CCS	2,654	1,625	1,658	850
TCA77	CCS	2,796	1,930	2,160	1,200

[1] Valid for silencer. Values differ for suction pipe or casing.

Table 223: Dimensions – TCA66, TCA77 on V engine

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Hoisting rail

A hoisting rail with a mobile trolley is to be provided over the centre of the turbocharger running parallel to its axis, into which a lifting tackle is suspended with the relevant lifting power for lifting the parts, which are mentioned in the table(s) below, to carry out the operations according to the maintenance schedule.

Turbocharger		TCA55	TCA66	TCA77
Turbine rotor	kg	139.1	233.6	366.6
Bearing case		582.7	944.9	1579.3
Compressor casing single socket		506.7	819.8	1,389.1
Compressor casing double socket		459.3	802.2	1,355.4
Gas admission casing axial		195.7	333.6	471.4
Space for removal of silencer	mm	70 + 100	80 + 100	80 + 100

Table 224: Hoisting rail of the axial turbocharger

Withdrawal space dimensions

The withdrawal space shown in section Removal dimensions, Page 467 and in the table(s) in paragraph Hoisting rail, Page 475 is required for separating the silencer from the turbocharger. The silencer must be shifted axially by this distance before it can be moved laterally.

In addition to this measure, another 100 mm are required for assembly clearance.

This is the minimum distance between silencer and bulkhead or tween-deck. We recommend to plan additional 300 – 400 mm as working space.

Make sure that the silencer can be removed either downwards or upwards or laterally and set aside, to make the turbocharger accessible for further servicing. Pipes must not be laid in these free spaces.

Fan shafts

The engine combustion air is to be supplied towards the intake silencer in a duct ending at a point 1.5 m away from the silencer inlet. If this duct impedes the maintenance operations, for instance the removal of the silencer, the end section of the duct must be removable. Suitable suspension lugs are to be provided on the deck and duct.

Gallery

If possible the ship deck should reach up to both sides of the turbocharger (clearance 50 mm) to obtain easy access for the maintenance personnel. Where deck levels are unfavourable, suspended galleries are to be provided.

Charge air cooler

For cleaning of the charge air cooler bundle, it must be possible to lift it vertically out of the cooler casing and lay it in a cleaning bath.

Exception MAN 32/40: The cooler bundle of this engine is drawn out at the end. Similarly, transport onto land must be possible.

For lifting and transportation of the bundle, a lifting rail is to be provided which runs in transverse or longitudinal direction to the engine (according to the available storage place), over the centreline of the charge air cooler, from which a trolley with hoisting tackle can be suspended.

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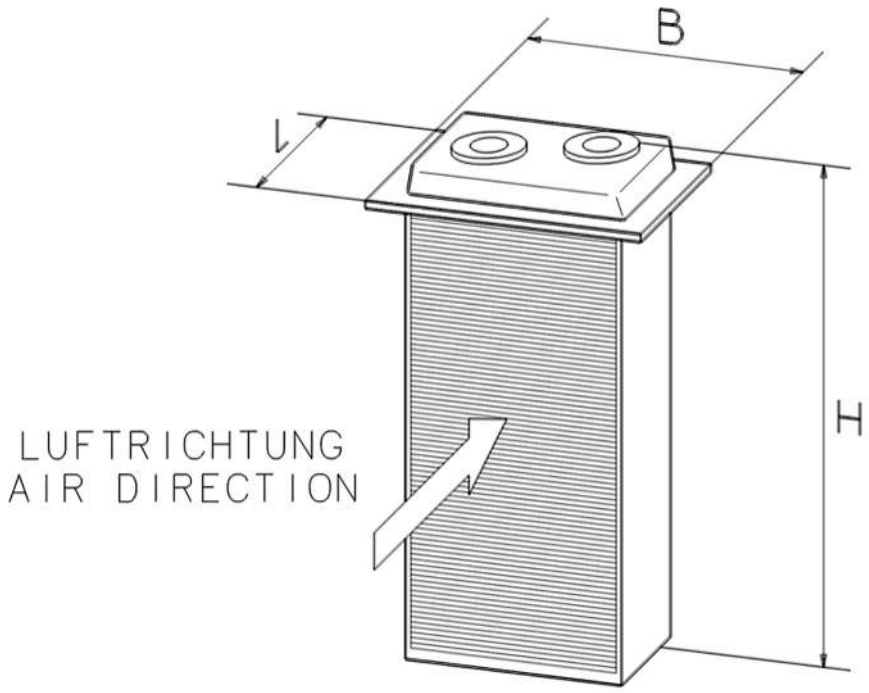

Figure 180: Air direction

Engine type	Weight	Length (L)	Width (B)	Height (H)
	kg	mm	mm	mm
L engine	1,000	730	1,052	1,904

Table 225: Weight and dimensions of charge air cooler bundle

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5.14.6 Space requirement for maintenance

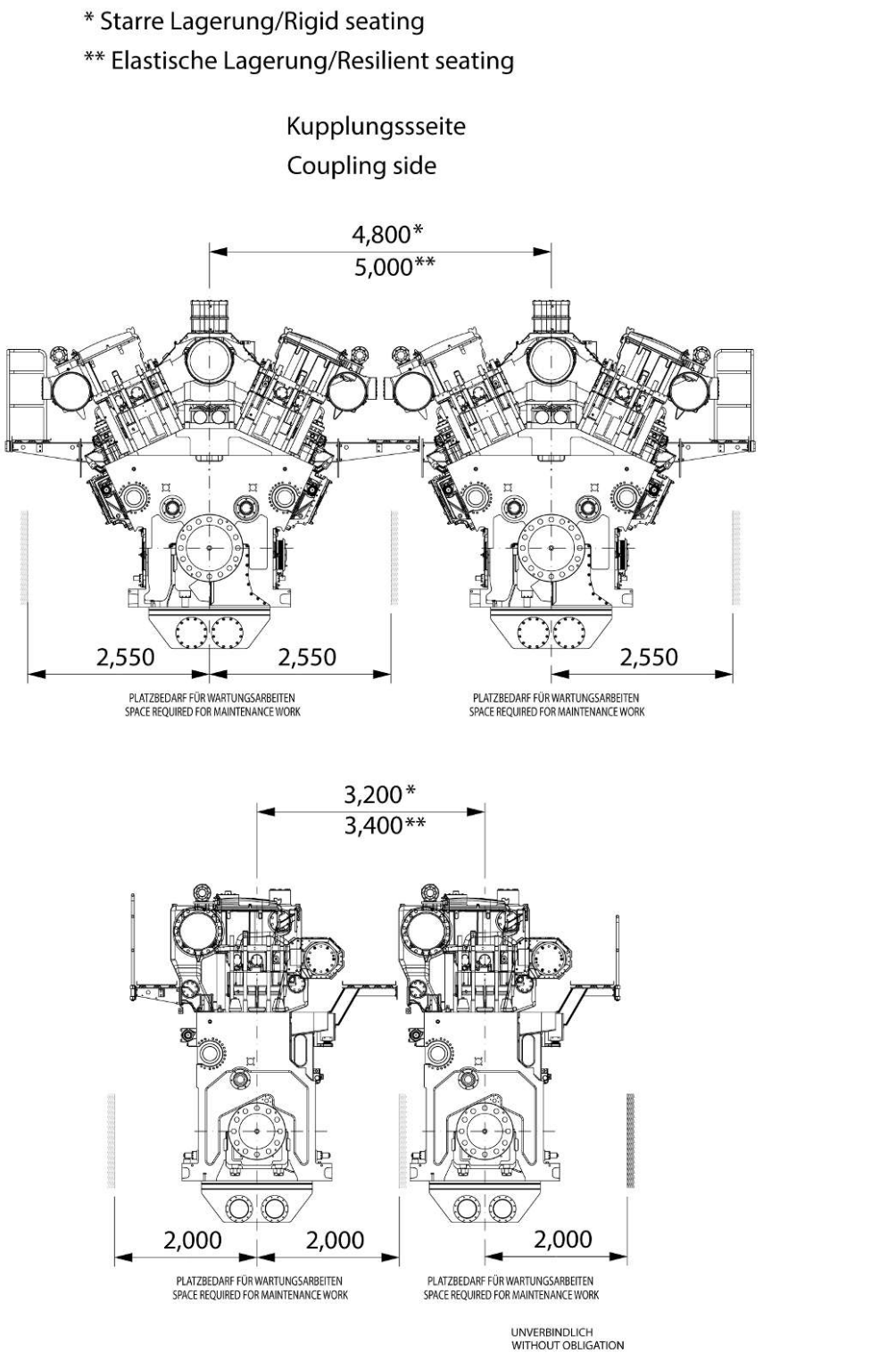

Figure 181: Space requirement for maintenance MAN 51/60DF

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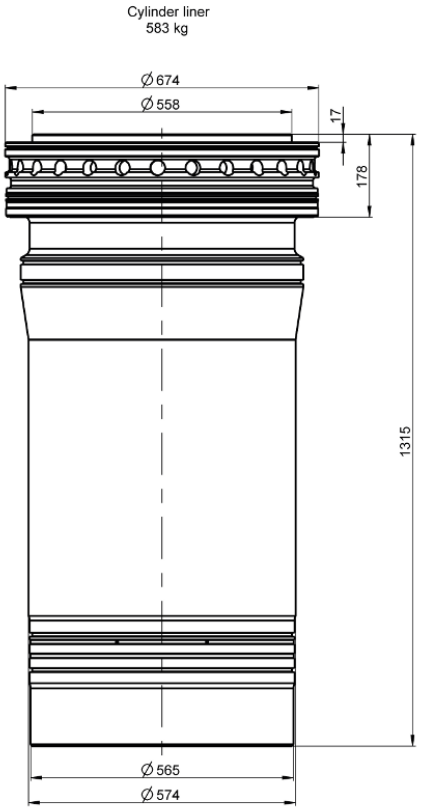
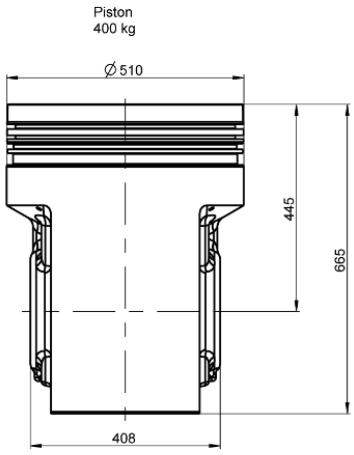
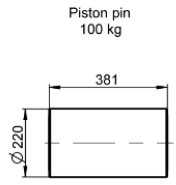
5.14.7 Major spare parts







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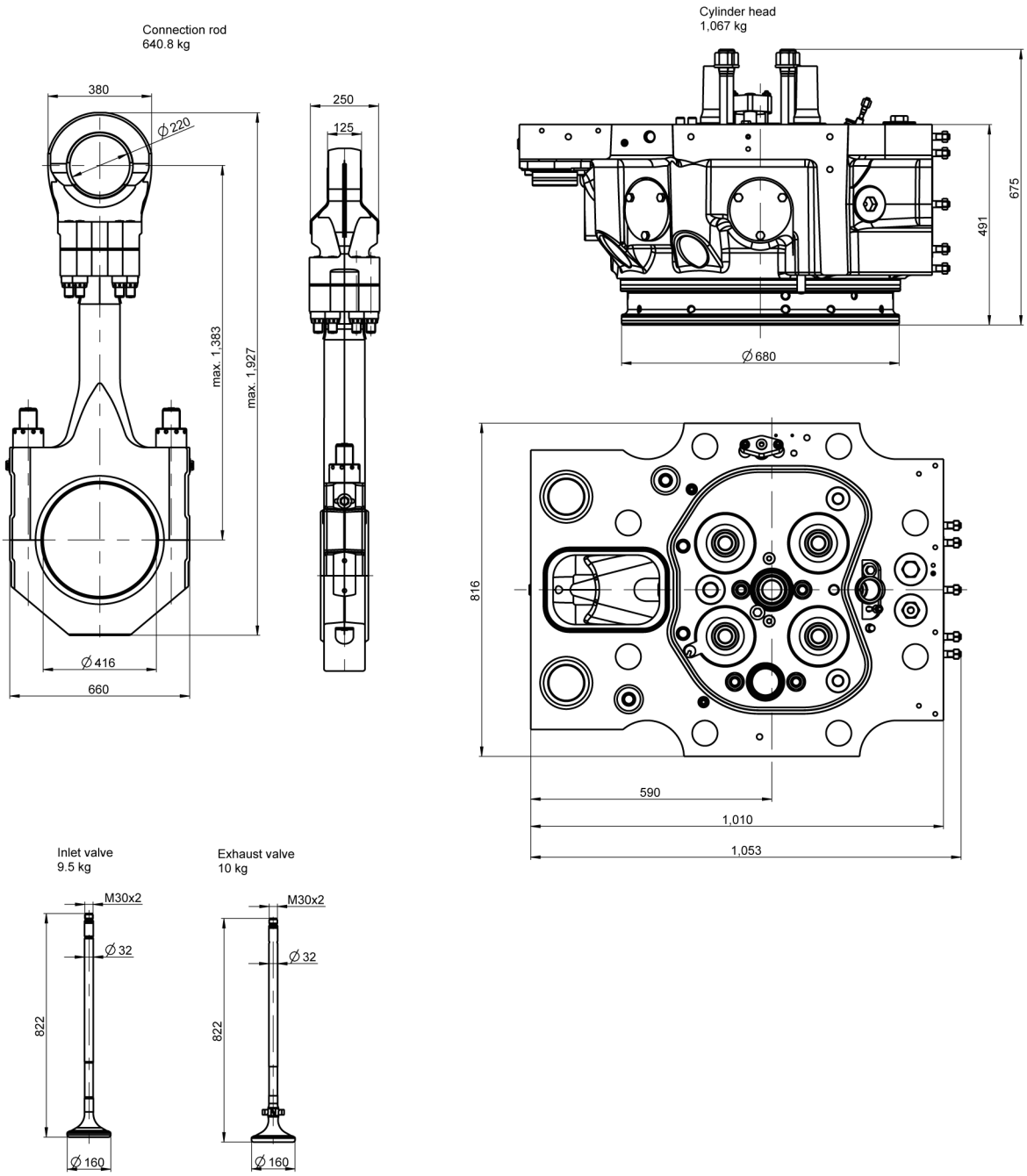

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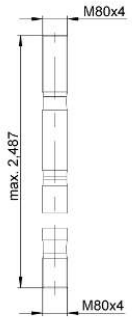
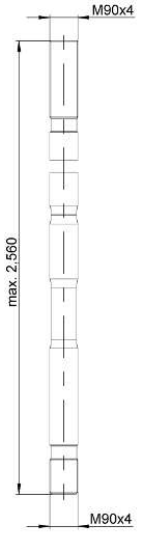
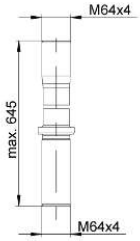
Tierod M80x4 only L engine
96 kg



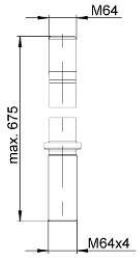
Tierod M90x4 only V engine
123 kg



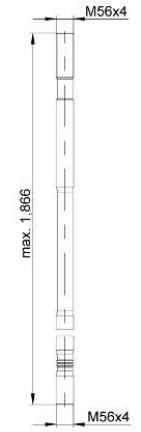
Stud screw only L engine
14.4 kg



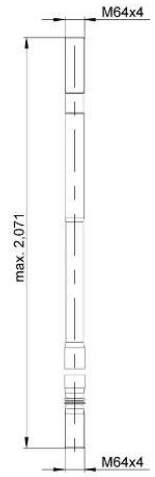
Stud screw only V engine
14.6 kg



Tierod only L engine
32 kg



Tierod only V engine
48.8 kg



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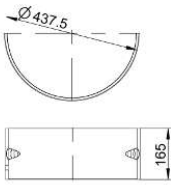
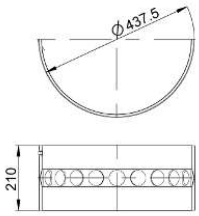
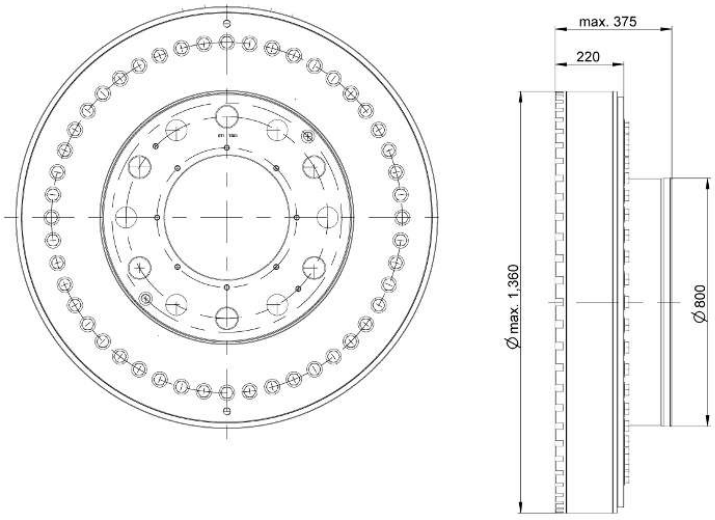
Crankshaft bearing shell only L engine
8 kg



Crankshaft bearing shell only V engine
8 kg



Vibration damper 6, 9L
2,277 kg



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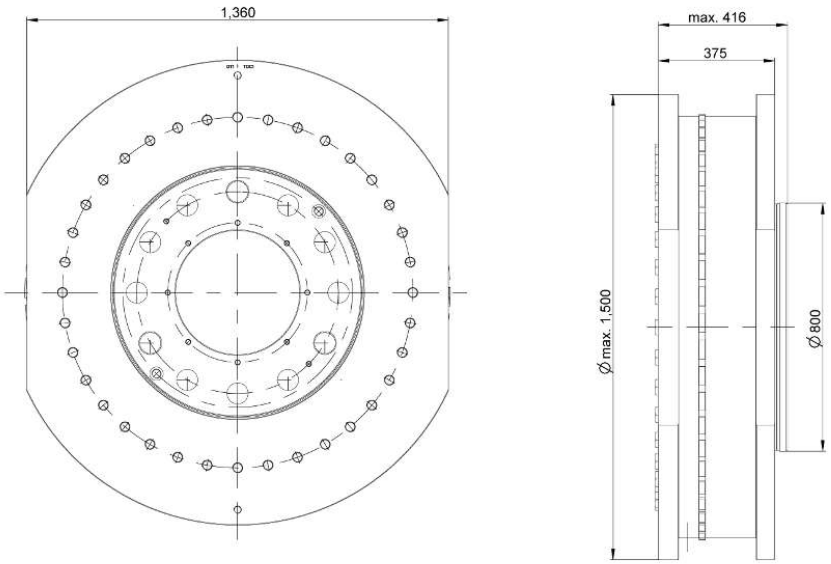
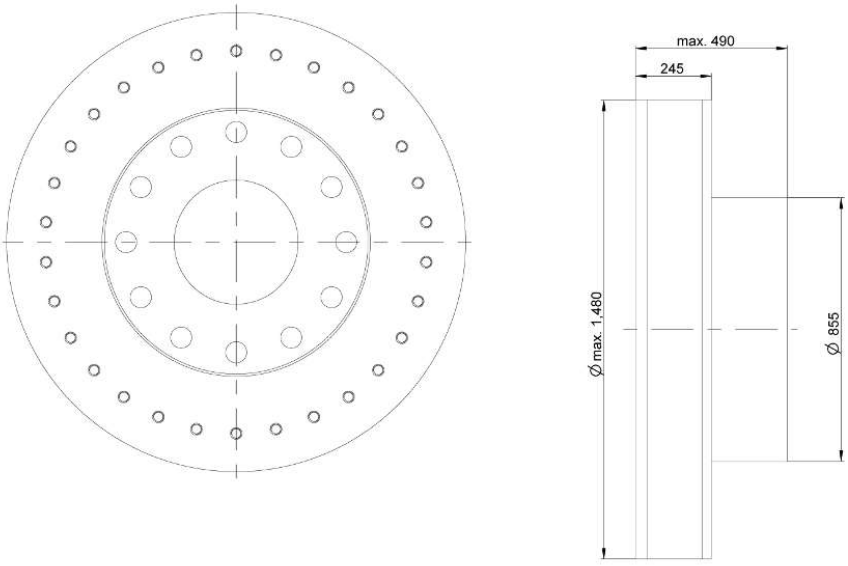
Vibration damper 7, 8L
3,616 kg



Vibration damper 12 – 18V
3,491 kg



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5.14.8 Mechanical propulsion system arrangement

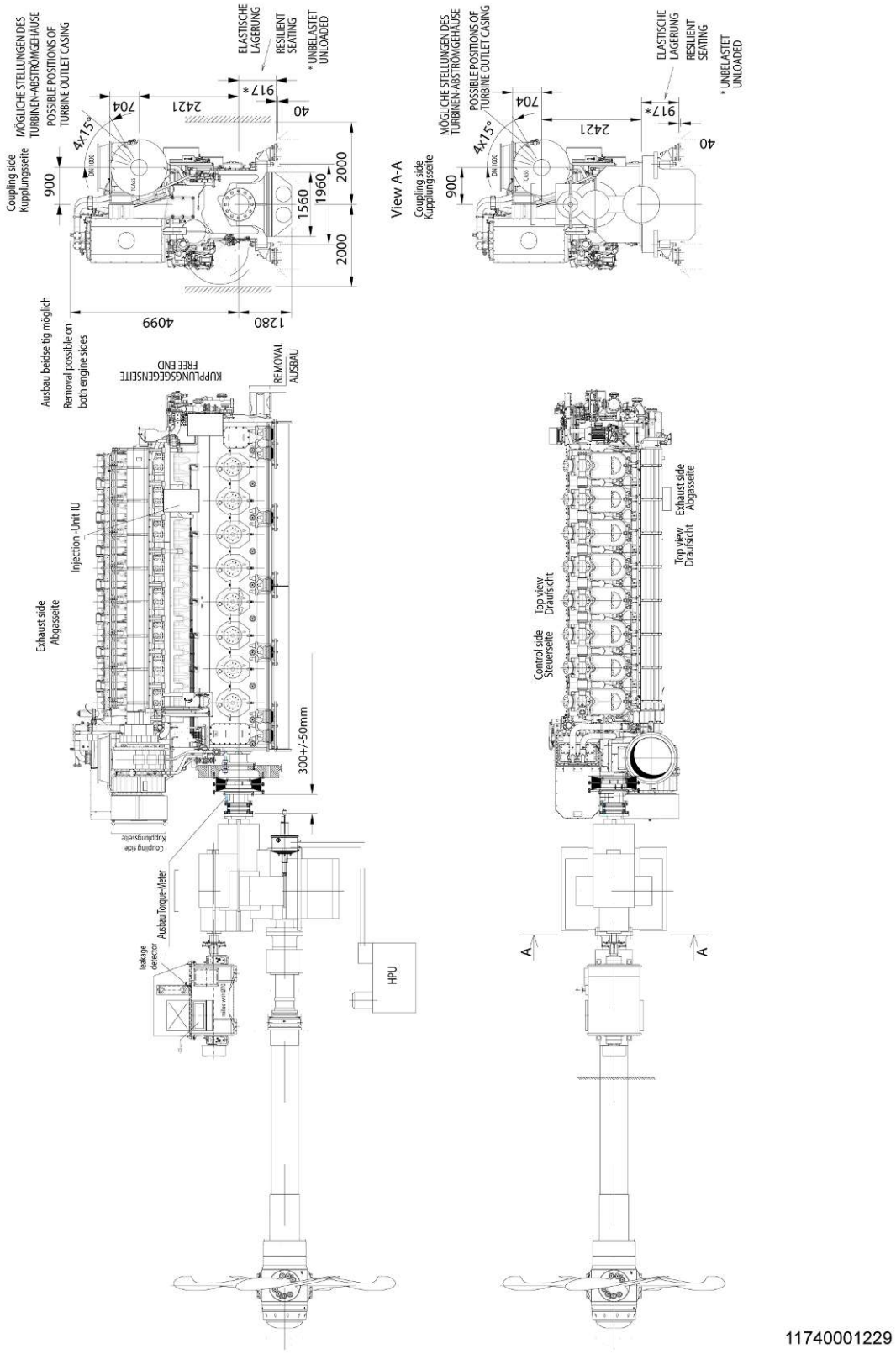

Figure 182: Example: Propulsion system arrangement MAN 8L51/60DF

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6 Propulsion packages

6.1 General

MAN Energy Solutions standard propulsion packages

The MAN Energy Solutions standard propulsion packages are optimised at 90 % MCR, 100 % rpm and 96.5 % of the ship speed. The propeller is calculated with the class notation "No Ice" and high skew propeller blade design. These propulsion packages are examples of different combinations of engines, gearboxes, propellers and shaft lines according to the design parameters above. Due to different and individual aft ship body designs and operational profiles your inquiry and order will be carefully reviewed and all given parameters will be considered in an individual calculation. The result of this calculation can differ from the standard propulsion packages by the assumption of e.g. a higher Ice Class or different design parameters.

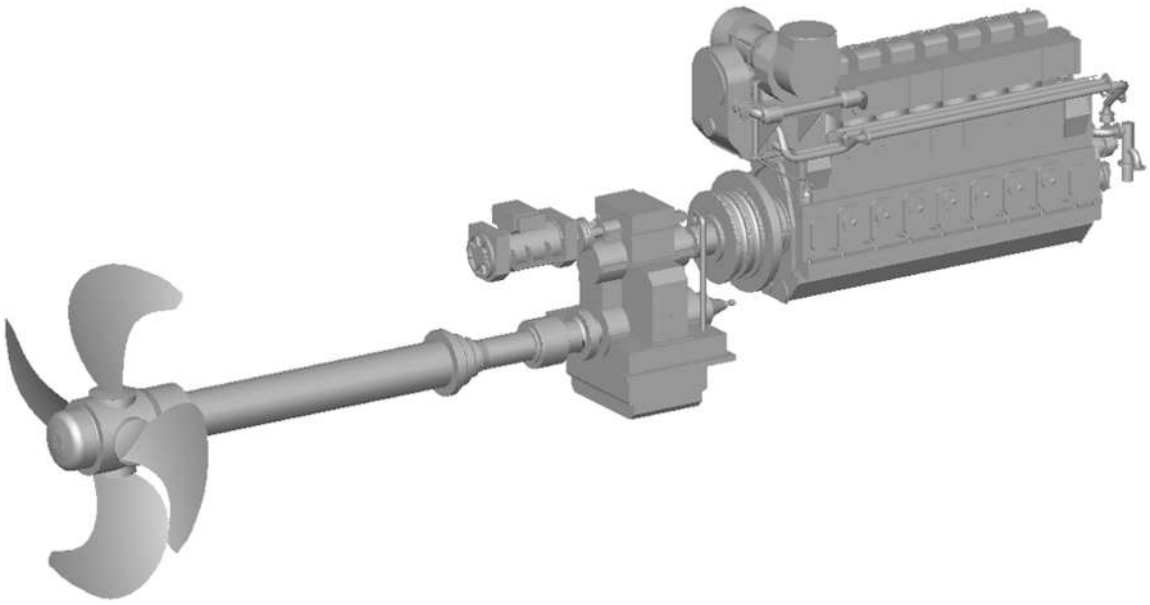

Figure 183: MAN Energy Solutions standard propulsion package with engine MAN 7L32/40 (example)

6.2 Propeller layout data

To find out which of our propeller fits you, fill in the propeller layout data sheet which you find here https://www.man-es.com/documentation-/project-guide-supplements and send it via e-mail to our sales department. The e-mail address is located under contacts on the web page.

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6.3 Propeller clearance

To reduce the emitted pressure impulses and vibrations from the propeller to the hull, MAN Energy Solutions recommends a minimum tip clearance see section Recommended configuration of foundation, Page 199.

For ships with slender aft body and favourable inflow conditions the lower values can be used whereas full after body and large variations in wake field causes the upper values to be used.

In twin-screw ships the blade tip may protrude below the base line.

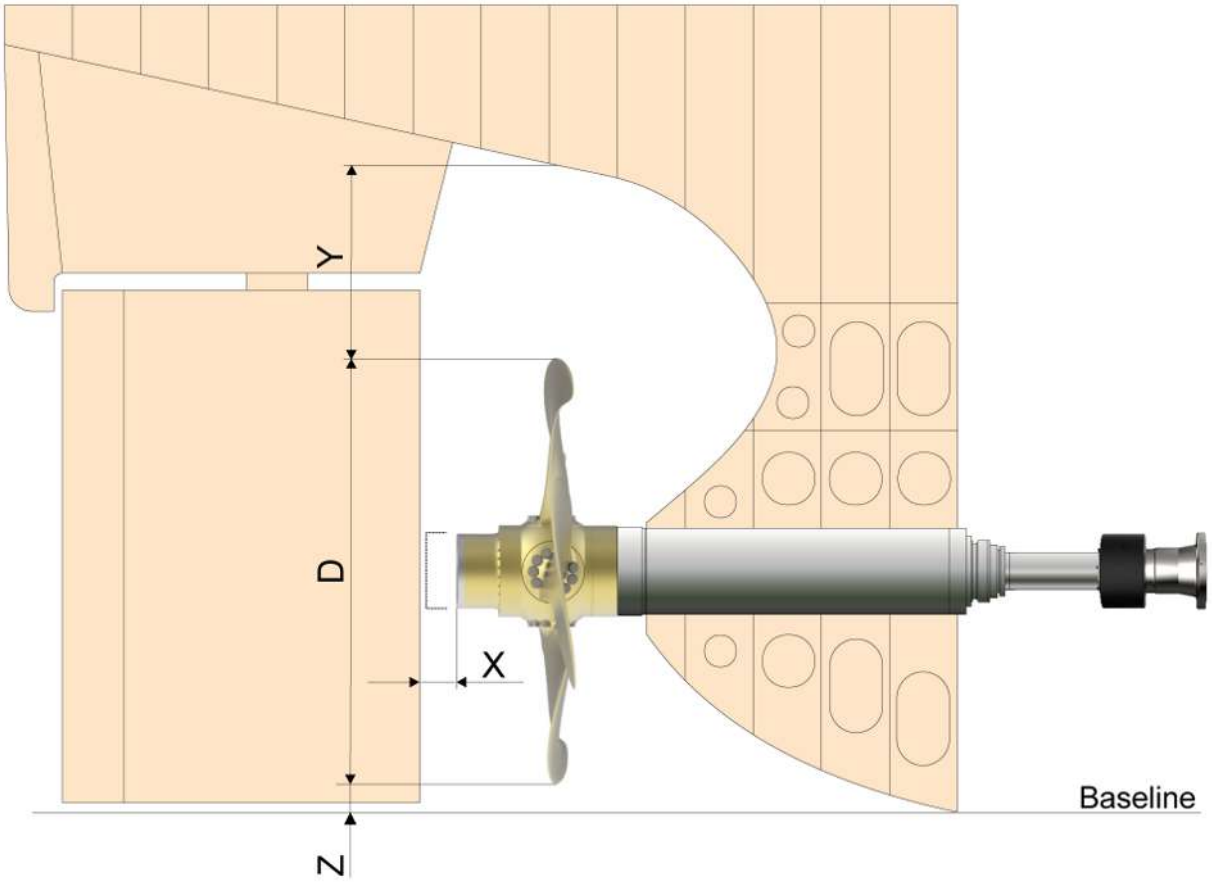

Figure 184: Recommended tip clearance

Hub	Dismantling of hub cylinder	High skew propeller	Non-skew propeller	Baseline clearance
	X min.	Y	Y	Z
Type	[mm]	[mm]	[mm]	[mm]
VBS1020	150	15 – 20 % of D	20 – 25 % of D	Minimum 50 – 100
VBS1100	160			
VBS1180	170			
VBS1260	175			
VBS1350	190			
VBS1450	200			
VBS1550	215			

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Hub	Dismantling of hub cylinder	High skew propeller	Non-skew propeller	Baseline clearance
	X min.	Y	Y	Z
Type	[mm]	[mm]	[mm]	[mm]
VBS1640	230			
VBS1730	395 [1]			
VBS1810	405			

[1] Dimension is not finally fixed.

6.4 Alphatronic 3000 propulsion control system

Alphatronic 3000 is MAN Energy Solutions´ propulsion control system for marine engines and propulsion system solutions. The following brief description is for controlling controllable pitch propeller (CPP) propulsion systems powered by four-stroke medium-speed engines with a standardised interface to the SaCoSone control and safety system.

Alphatronic 3000 provides:

- Safe control of the propulsion plant and reliable maneuvering of the ship.
- Economic operation thanks to optimised engine/propeller load control.
- Quick system response and efficient CPP maneuverability.
- User-friendly operator functions due to logic and ergonomic design of control panels, handles and displays.

The system offers three levels of propulsion control:

- Normal control with automatic load control.
- Backup control from bridge and engine control room.
- Independent telegraph system for communication from bridge to machinery space.

The Alphatronic 3000 system is based on a modular panel design concept to elegantly fit any ship console layout. Configurable touch screens in the propulsion control panels meet a wide range of customer specific functions.

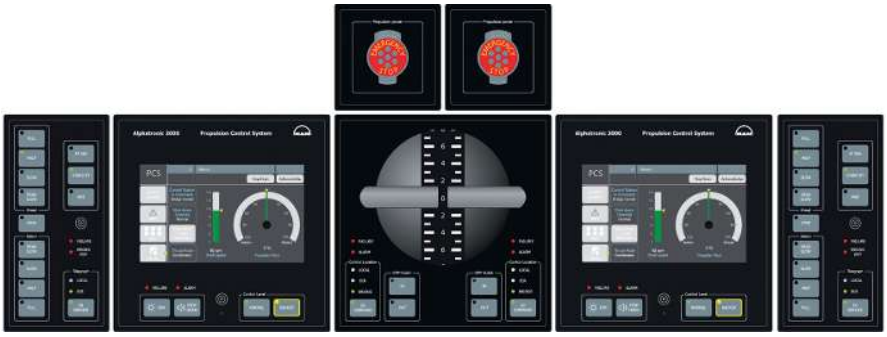

Figure 185: Control station layout for a twin CP propeller plant

A number of tailored features and functions can be provided by Alphatronic 3000 – as for example the speed pilot. The optional speed pilot feature is available with connection to the ship‘s GPS system for ‘speed over ground’ (SOG) input. The speed pilot optimises the voyage planning and operational speeds e.g. for pulling, steaming and convoy sailing – with fuel saving potentials of up to 4 %.

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For a more extensive description of the Alphatronic 3000 propulsion control system, functions, system architecture, interfaces, panels and displays, be referred to our 40-page product information "Alphatronic 3000" on this site:

https://marine.man-es.com/propeller-aft-ship/product-range/man-alpha-controlable-pitch-propeller---cpp

Figure 186: Manoeuvre handle panels

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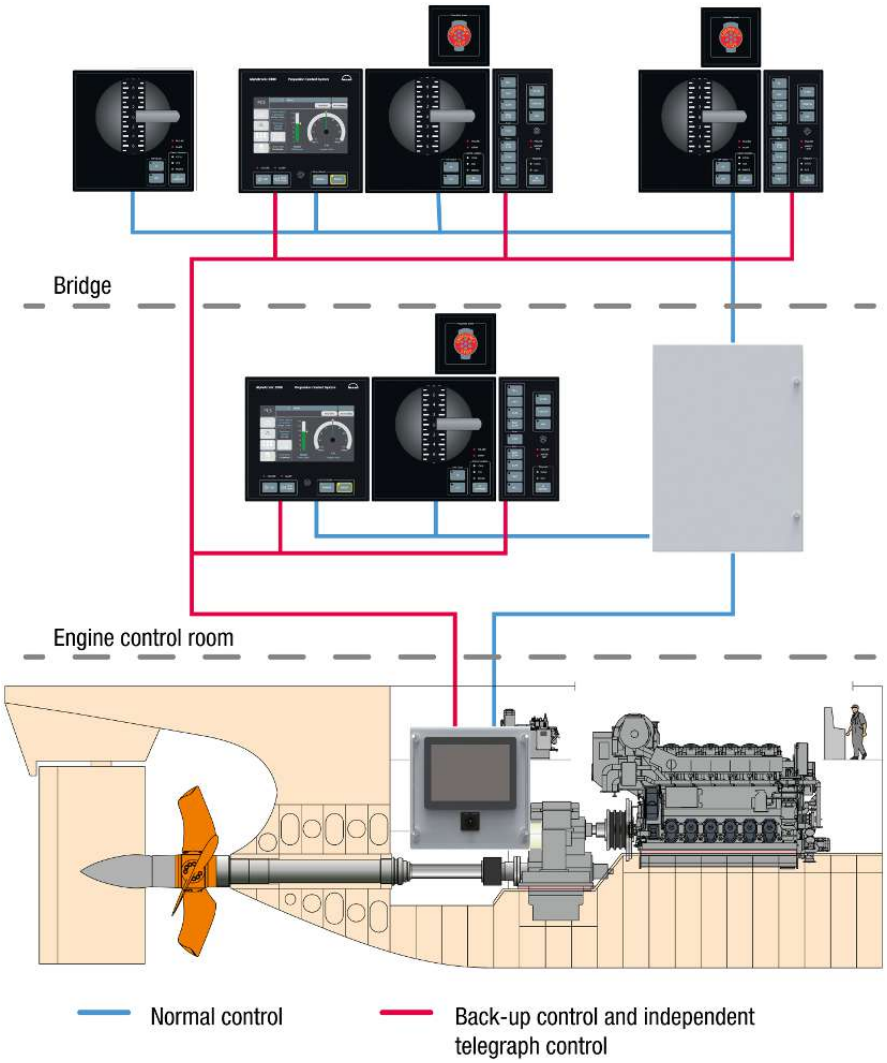

Figure 187: Simple control system architecture – Single CP propeller four-stroke propulsion example

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7 Electric propulsion plants

7.1 Advantages of electric propulsion

Due to different and individual types, purposes and operational profiles of electric propulsion driven vessels the design of an electric propulsion plant differs a lot and has to be evaluated case by case. All the following is for information purpose only and without obligation.

In general the advantages of electric propulsion can be summarized as follows:

- Lower fuel consumption and emissions due to the possibility to optimise the loading of diesel engines/GenSets. The GenSets in operation can run on high loads with high efficiency. This applies especially to vessels which have a large variation in power demand, for example for an offshore supply vessel.
- High reliability, due to multiple engine redundancy. Even if an engine/GenSet malfunctions, there will be sufficient power to operate the vessel safely. Reduced vulnerability to single point of failure providing the basis to fulfill high redundancy requirements.
- Reduced life cycle cost, resulting from lower operational and maintenance costs.
- Improved manoeuvrability and station-keeping ability, by deploying special propulsors such as azimuth thrusters or pods. Precise control of the electric propulsion motors controlled by frequency converters.
- Increased payload, as electric propulsion plants take less engine room space.
- More flexibility in location of diesel engine/GenSets and propulsors. The propulsors are supplied with electric power through cables. They do not need to be adjacent to the diesel engines/GenSets.
- Low propulsion noise and reduced vibrations. For example, a slow speed E-motor allows to avoid a gearbox and propulsors like pods keep most of the structure bore noise outside of the hull.
- Efficient performance and high motor torques, as the system can provide maximum torque also at slow propeller speeds, which gives advantages for example in icy conditions.

7.2 Losses in electric propulsion plants

An electric propulsion plant consists of standard electrical components. The following losses are typical:

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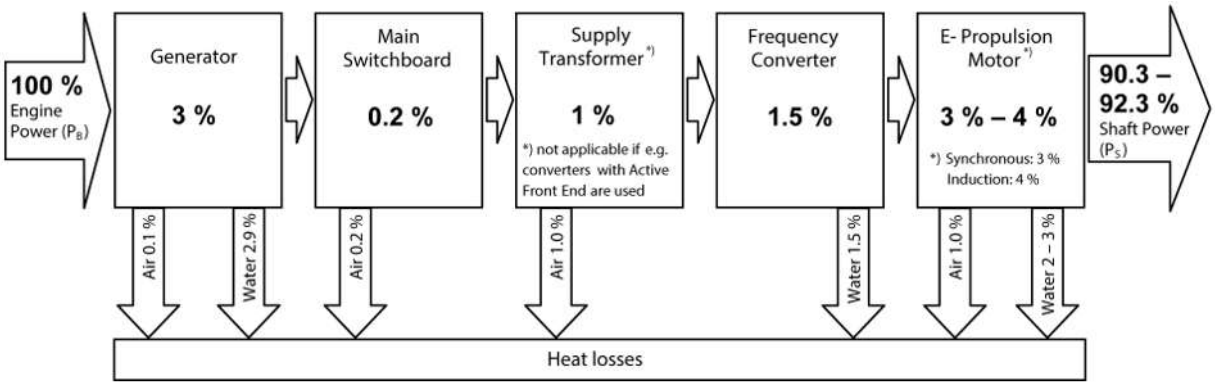
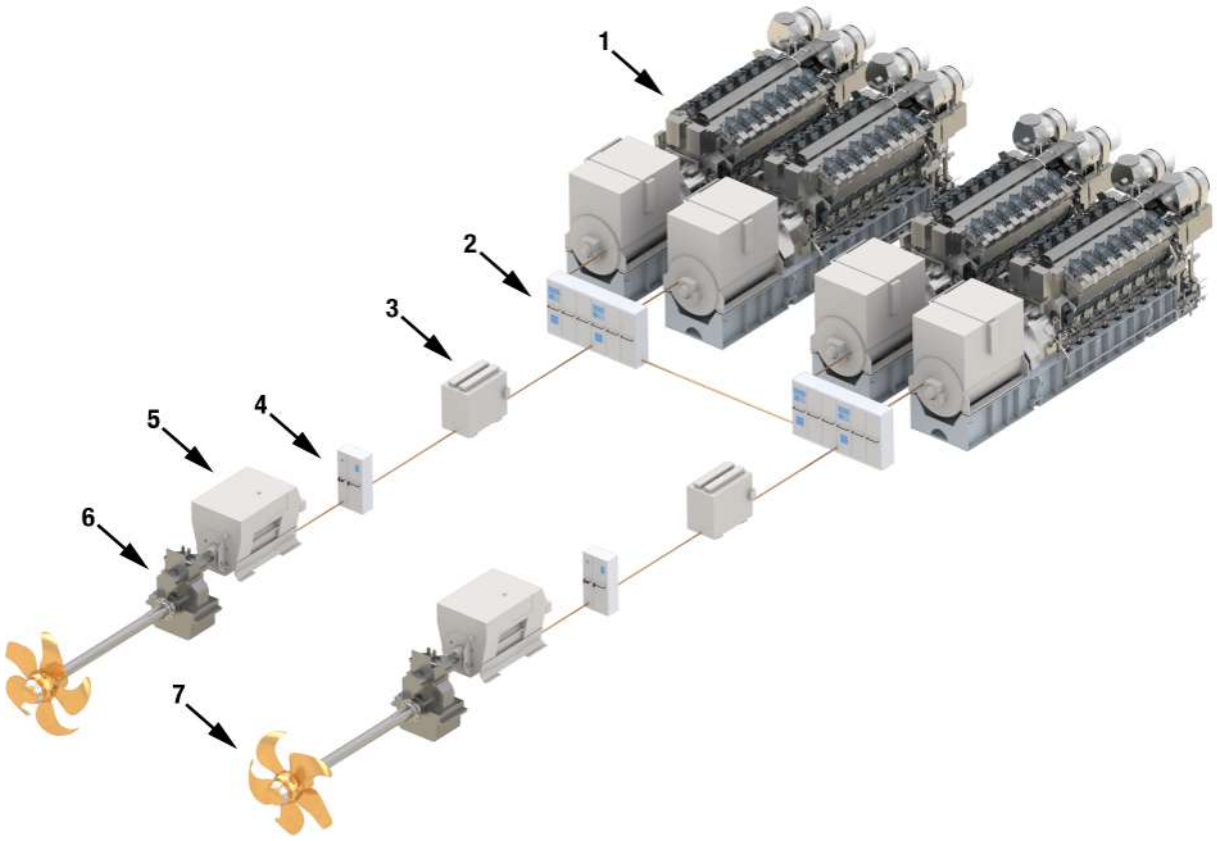


Figure 188: Typical losses of electric propulsion plants

7.3 Components of an electric propulsion plant



1 GenSets: Diesel engines and alternators
2 Main switchboards
3 Supply transformers: Dependent on the type of the converter. Not required in case of the use of frequency converters with six pulses, an active front end or a sinusoidal drive.
4 Frequency converters
5 Electric propulsion motors
6 Gearboxes (optional): Dependent on the speed of the E-propulsion motor
7 Propellers/propulsion

Figure 189: Example: Electric propulsion plant

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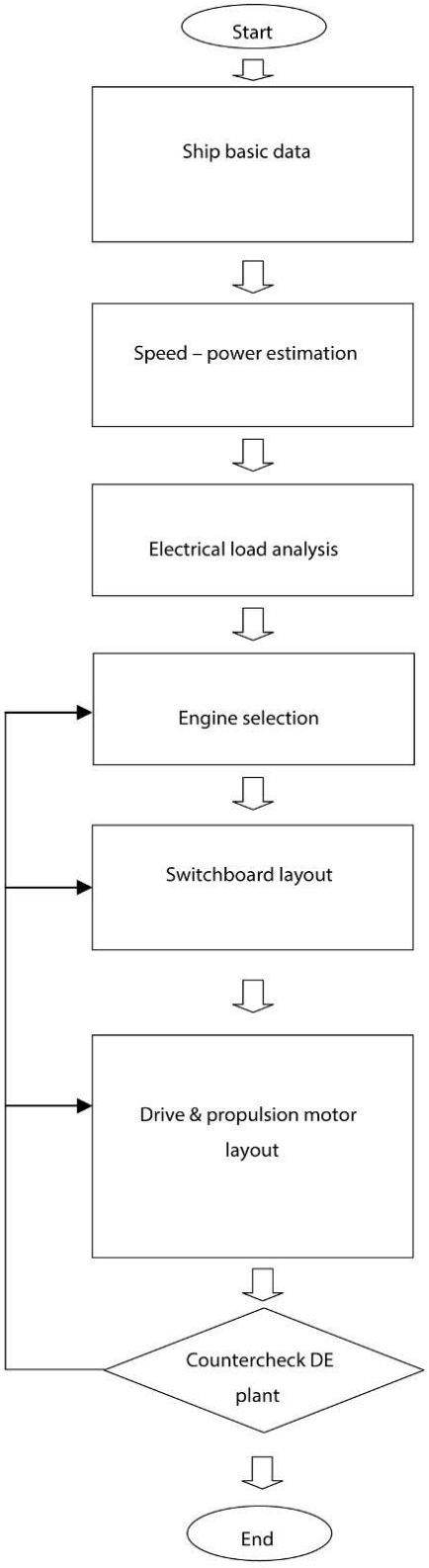
7.4 Electric propulsion plant design

Generic workflow how to design an electric propulsion plant:



- Type of vessel
- Propulsion type: Shaft line, thruster, pod, …
- Propeller type: FPP, CPP
- Operational profile
- Class notation: Propulsion redundancy, ice class, …

- Ship design points
- Propulsion power: At sea, maneuvering, at port, …
- Sea margin

- Electrical power: At sea, maneuvering, at port, …
- Efficiency of DE plant: Typically = 91 %
- Efficiency of alternators: Typically = 96 % – 97 %

- Number and type of engines/gensets: Installed power
- Max. allowed loading of engines: % of MCR
- Maintenance of engines: At sea operation, at port, …

- Frequency choice: 50/60 Hz
- Voltage choice: Low voltage, medium voltage
- Number of switchboard sections
- Alternator parameters: cos φ, xd"

- Selection of converter type: PWM, LCI, AFE, Sinusoidal, …
- Selection of pulse number: 6p, 12p, 24p
- Selection of supply transformer: Investigate transformer less configuration (e.g. "Active Front End")
- Selection of E-propulsion motor type and no. of windings
- THD mitigation method

- Check I_{sc}: "Increase voltage, optimize xd", …
- Check availability of reactive power: Change number/type of alternators, cos φ, …
- Check THD limits: Increase pulse number, add filters, …

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The requirements of a project will be considered in an application specific design, taking into account the technical and economical feasibility and later operation of the vessel. In order to provide you with appropriate data, fill the form "Diesel-electric propulsion plants: Questionnaire" you find here https://www.man-es.com/documentation-/project-guide-supplements and return it to your sales representative.

7.5 Engine selection

The engines for an electric propulsion plant have to be selected accordingly to the power demand at all the design points. For a concept evaluation the rating, the capability and the loading of engines can be calculated like this:

Example: Offshore supply vessel (at operation mode with the highest expected total load)

- Total propulsion power demand (at E-motor shaft) 10,000 kW (incl. sea margin)
- Max. electrical consumer load: 1,000 kW

No.	Item	Unit	
1.1	Shaft power on propulsion motors Electrical transmission efficiency	P_S [kW]	10,000 0.91
1.2	Engine brake power for propulsion	P_{B1} [kW]	10,989
2.1	Electric power for ship (E-load) Alternator efficiency	[kW]	1,000 0.965
2.2	Engine brake power for electric consumers	P_{B2} [kW]	1,036
2.3	Total engine brake power demand (= 1.2 + 2.2)	P_B [kW]	12,025
3.1	Diesel engine selection	Type	MAN 6L32/44CR
3.2	Rated power (MCR) running on MDO	[kW]	3,600
3.3	Number of engines	-	4
3.4	Total engine brake power installed	P_B [kW]	14,400
4.1	Loading of engines (= 2.3/3.4)	% of MCR	83.5
5.1	Check: Maximum permissible loading of engines	% of MCR	90.0

Table 226: Selection of the engines for an electric propulsion plant

For the detailed selection of the type and number of engines furthermore the operational profile of the vessel, the maintenance strategy of the engines and the boundary conditions given by the general arrangement have to be considered. For the optimal cylinder configuration of the engines often the power conditions in port are decisive.

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7.6 E-plant, switchboard and alternator design

The configuration and layout of an electric propulsion plant, the main switchboard and the alternators follows some basic design principles. For a concept evaluation the following items should be considered:

- A main switchboard which is divided in symmetrical sections is very reliable and redundancy requirements are easy to be met.

- An even number of GenSets/alternators ensures the symmetrical loading of the bus bar sections.
- Electric consumers should be arranged symmetrically on the bus bar sections.
- The switchboard design is mainly determined by the level of the short circuit currents which have to be withstand and by the breaking capacity of the circuit breakers (CB).
- The voltage choice for the main switchboard depends on several factors. On board of a vessel it is usually handier to use low voltage. Due to short circuit restrictions the following table can be used for voltage choice as a rule of thumb:

Total installed alternator power	Voltage	Breaking capacity of CB
< 10 – 12 MW (and: Single propulsion motor < 3.5 MW)	440 V	100 kA
< 13 – 15 MW (and: Single propulsion motor < 4.5 MW)	690 V	100 kA
< 48 MW	6,600 V	30 kA
< 130 MW	11,000 V	50 kA

Table 227: Rule of thumb for the voltage choice

- The design of the alternators and the electric plant always has to be balanced between voltage choice, availability of reactive power, short circuit level and permissible total harmonic distortion (THD).
- On the one hand side a small xd" of an alternator increases the short circuit current I_{sc}", which also increases the forces the switchboard has to withstand (F ~ I_{sc}" ^ 2). This may lead to the need of a higher voltage. On the other side a small xd" gives a lower THD but a higher weight and a bigger size of the alternator. As a rule of thumb a xd"=16 % is a good figure for low voltage alternators and a xd"=14 % is good for medium voltage alternators.
- For a rough estimation of the short circuit currents the following formulas can be used:

	Short circuit level [kA] (rough)	Legend
Alternators	$n * P_r / (\sqrt{3} * U_r * xd" * \cos \varphi_{Grid})$	n: No. of alternators connected P_r: Rated power of alternator [kWe] U_r: Rated voltage [V] xd": Subtransient reactance [%] cos φ: Power factor of the vessel´s network (typically = 0.9)
Motors	$n * 6 * P_r / (\sqrt{3} * U_r * xd" * \cos \varphi_{Motor})$	n: No. of motors (directly) connected P_r: Rated power of motor [kWe] U_r: Rated voltage [V] xd": Subtransient reactance [%] cos φ: Power factor of the motor (typically = 0.85 – 0.90 for an induction motor)

	Short circuit level [kA] (rough)	Legend
Converters	Frequency converters do not contribute to the I_{sc}"	-

Table 228: Formulas for a rough estimation of the short circuit currents

- The dimensioning of the cubicles in the main switchboard is usually done accordingly to the rated current for each incoming and outgoing panel. For a concept evaluation the following formulas can be used:

Type of switchboard cubicle	Rated current [kA]	Legend
Alternator incoming	$P_r / (\sqrt{3} * U_r * \cos \varphi_{Grid})$	P_r: Rated power of alternator [kWe] U_r: Rated voltage [V] cos φ: Power factor of the network (typically = 0.9)
Transformer outgoing	$S_r / (\sqrt{3} * U_r)$	S_r: Apparent power of transformer [kVA] U_r: Rated voltage [V]
Motor outgoing (induction motor controlled by a PWM-converter)	$P_r / (\sqrt{3} * U_r * \cos \varphi_{Converter} * \eta_{Motor} * \eta_{Converter})$	P_r: Rated power of motor [kWe] U_r: Rated voltage [V] cos φ: Power factor converter (typically = 0.95) η_{Motor}: Typically = 0.96 $\eta_{Converterr}$: Typically = 0.97
Motor outgoing (induction motor started: DoL, Y/△, soft-starter)	$P_r / (\sqrt{3} * U_r * \cos \varphi_{Motor} * \eta_{Motor})$	P_r: Rated power of motor [kWe] U_r: Rated voltage [V] cos φ: Power factor motor (typically = 0.85 – 0.90) η_{Motor}: Typically = 0.96

Table 229: Formulas to calculate the rated currents of switchboard panel

- The choice of the type of the E-motor depends on the application. Usually induction motors are used up to a power of 7 MW (η_{Motor}: Typically = 0.96). If it comes to applications above 7 MW per E-motor often synchronous machines are used. Also in applications with slow speed E-motors (without a reduction gearbox), for ice going or pod-driven vessels often synchronous E-motors (η_{Motor}: Typically = 0.97) are used.
- In plants with frequency converters based on VSI-technology (PWM type) the converter itself can deliver reactive power to the E-motor. So often a power factor cos φ = 0.9 is a good figure to design the alternator rating. Nevertheless there has to be sufficient reactive power for the ship consumers, so that a lack in reactive power does not lead to unnecessary starts of (stand-by) alternators.
- The harmonics can be improved (if necessary) by using supply transformers for the frequency converters with a 30 ° phase shift between the two secondary windings, which cancel the dominant 5^{th} and 7^{th} harmonic currents. Also an increase in the pulse number leads to lower THD. Using a 12-pulse configuration with a PWM type of converter the resulting harmonic distortion will normally be below the limits defined by the classification societies. When using a transformer less solution with a converter

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with an Active Front End (Sinusoidal input rectifier) or in a 6-pulse configuration usually THD-filters are necessary to mitigate the THD on the sub-distributions.

The final layout of the electrical plant and the components has always to be based on a detailed analysis and a calculation of the short circuit levels, the load flows and the THD levels as well as on an economical evaluation.

7.7 Over-torque capability

In electric propulsion plants, which are operating with a fix pitch propeller, the dimensioning of the electric propulsion motor has to be done accurately, in order to have sufficient propulsion power available. For dimensioning the electric motor it has to be investigated what amount of over-torque, which directly defines the motor´s cost, weight and space demand, is required to operate the propeller with sufficient power also in situations, where additional power is required (for example because of heavy weather or icy conditions).

Usually a constant power range of 5 % – 10 % is applied on the propulsion (Field weakening range), where constant E-motor power is available.

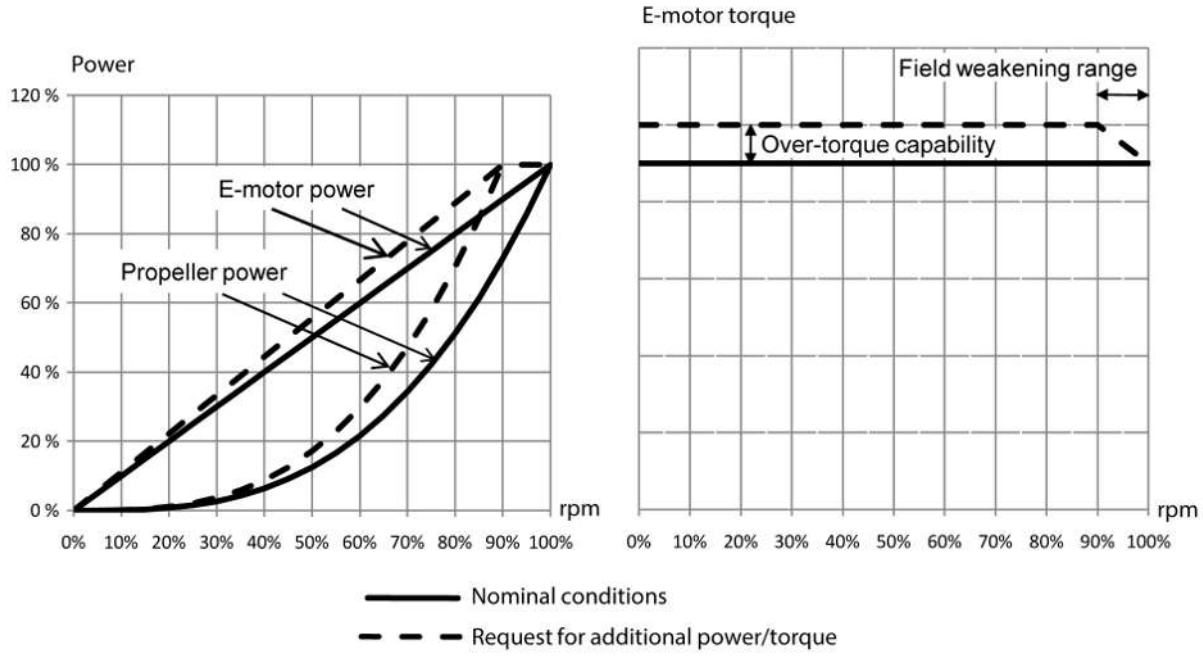

Figure 190: Example: Over-torque capability of an E-propulsion train for a FPP-driven vessel

7.8 Power management

Power management system

The following main functions are typical for a power management system (PMS):

- Automatic load dependent start/stop of GenSets/alternators
- Manual starting/stopping of GenSets/alternators
- Fault dependent start/stop of stand-by GenSets/alternators in cases of under-frequency and/or under-voltage
- Start of GenSets/alternators in case of a blackout (black-start capability)

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- Determining and selection of the starting/stopping sequence of GenSets/ alternators
- Start and supervise the automatic synchronization of alternators and bus tie breakers
- Balanced and unbalanced load application and sharing between GenSets/alternators. Often an emergency programme for quickest possible load acceptance is necessary
- Regulation of the network frequency (with static droop or constant frequency)
- Distribution of active load between alternators
- Distribution of reactive load between alternators
- Handling and blocking of heavy consumers
- Automatic load shedding
- Tripping of non-essential consumers
- Bus tie and breaker monitoring and control

All questions regarding the interfaces from/to the power management system have to be clarified with MAN Energy Solutions at an early project stage.

7.9 Example configurations of electric propulsion plants

Offshore Support Vessels

The term “Offshore Service & Supply Vessel” includes a large class of vessel types, such as platform supply vessels (PSV), anchor handling/tug/supply (AHTS), offshore construction vessel (OCV), diving support vessel (DSV), multipurpose vessel (MPV), etc.

Electric propulsion is the norm in ships which frequently require dynamic positioning and station keeping capability. Initially these vessels mainly used variable speed motor drives and fixed pitch propellers. Now they mostly deploy variable speed thrusters and they are also often equipped with hybrid propulsion systems.

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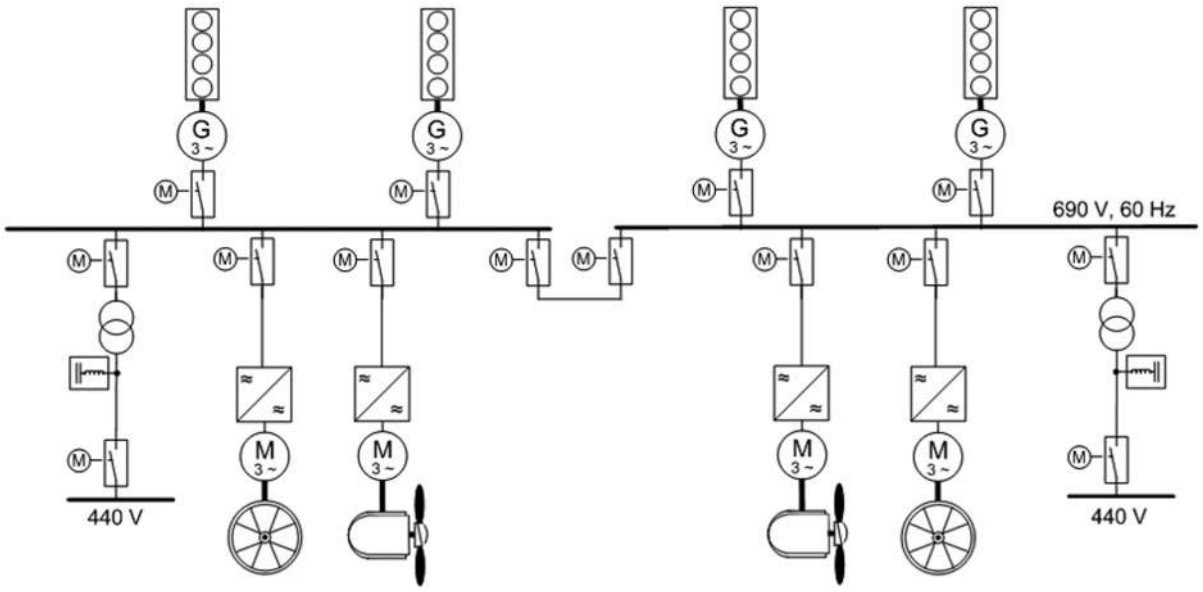

Figure 191: Example: Electric propulsion configuration of a PSV

In offshore applications often frequency converters with a 6-pulse configuration or with an active front end are used, which give specific benefits in the space consumption of the electric plant, as it is possible to get rid of the heavy and bulky supply transformers.

Type of converter/drive	Supply transformer	Type of E-motor	Pros & cons
6 pulse drive or active front end	-	Induction	+ Transformer less solution + Less space and weight – THD filters to be considered

Table 230: Main DE-components for offshore applications

LNG Carriers

A propulsion configuration with two E-motors (e.g. 600 rpm or 720 rpm) and a reduction gearbox (twin-in-single-out) is a typical configuration, which is used at LNG carriers where the installed alternator power is in the range of about 40 MW. The electric plant fulfils high redundancy requirements. Due to the high propulsion power, which is required and higher efficiencies, mainly synchronous E-motors are used.

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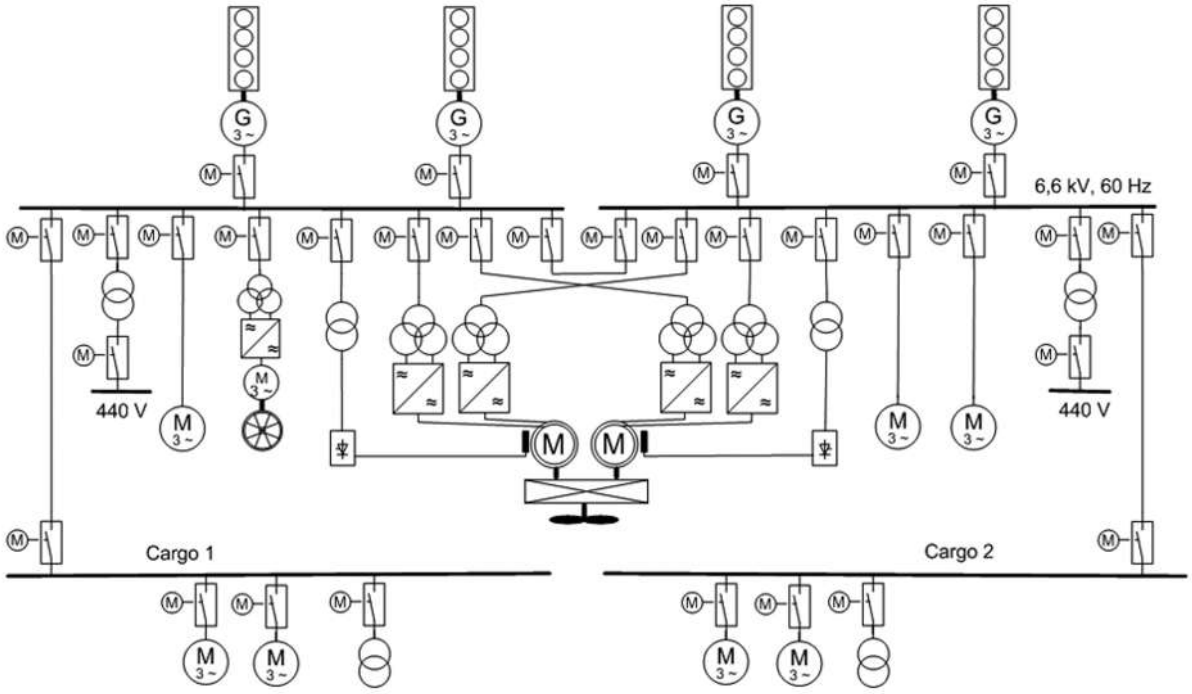

Figure 192: Example: Electric propulsion configuration of a LNG carrier with geared transmission, single screw and fixed pitch propeller

Type of converter/drive	Supply transformer	Type of E-motor	Pros & cons
VSI with PWM	24 pulse	Synchronous	+ High propulsion power + High drive & motor efficiency + Low harmonics – Complex E-plant configuration

Table 231: Main DE-components for a LNG carrier

For ice going carriers and tankers also podded propulsion is a robust solution, which has been applied in several vessels.

Cruise ships and ferries

Passenger vessels – cruise ships and ferries – are an important application field for electric propulsion. Safety and comfort are paramount. New regulations, as “Safe Return to Port”, require a high reliable and redundant electric propulsion plant and also onboard comfort is of high priority, allowing only low levels of noise and vibration from the ship´s machinery.

A typical electric propulsion plant is shown in the example below.

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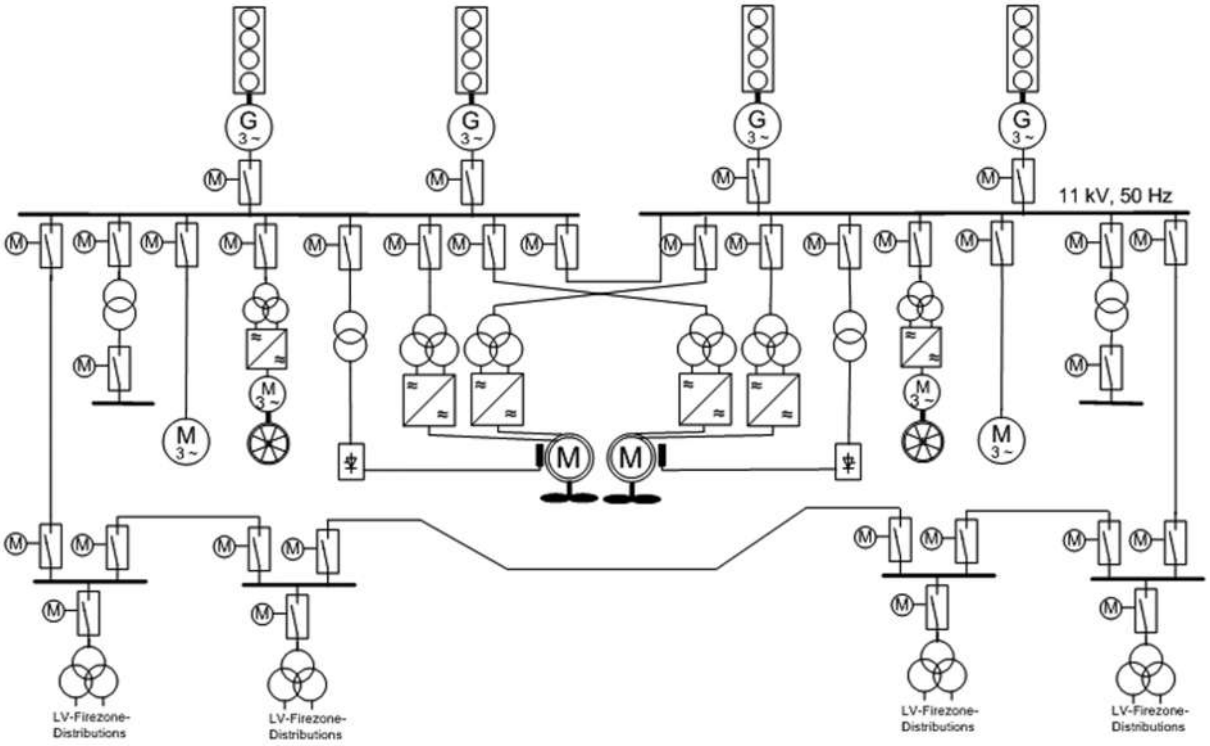

Figure 193: Example: Electric propulsion configuration of a cruise liner, twin screw, gear less

Type of converter/drive	Supply transformer	Type of E-motor	Pros & cons
VSI with PWM	24 pulse	Synchronous (e.g. slow speed 150 rpm)	+ Highly redundant & reliable + High drive & motor efficiency + Low noise & vibration – Complex E-plant configuration

Table 232: Main DE-components for a cruise liner

For cruise liners often also geared transmission is applied as well as pods.

For a RoPax ferry almost the same requirements are valid as for a cruise liner.

The figure below shows an electric propulsion plant with a “classical” configuration, consisting of E-motors (e.g. 1,200 rpm), geared transmission, frequency converters and supply transformers.

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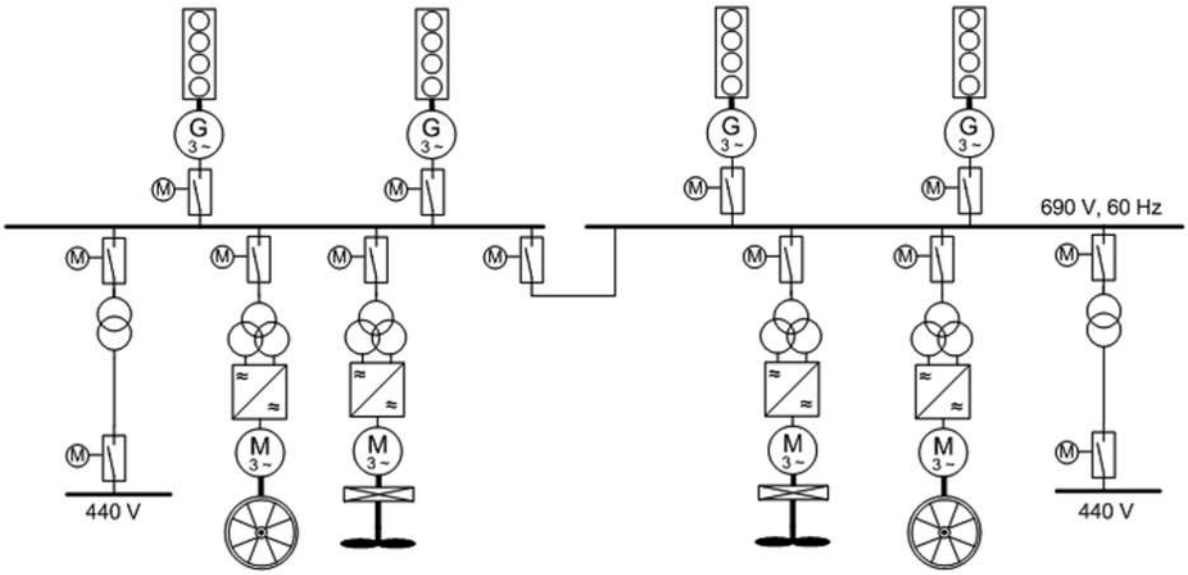

Figure 194: Example: Electric propulsion configuration of a RoPax ferry, twin screw, geared transmission

Type of converter/drive	Supply transformer	Type of E-motor	Pros & cons
VSI-type (with PWM technology)	12 pulse, two secondary windings, 30° phase shift	Induction	+ Robust & reliable technology + No seperate THD filters – More space & weight (compared to transformer less solution)

Table 233: Main DE-components for a RoPax ferry

Low loss applications

As MAN Energy Solutions works together with different suppliers for electric propulsion plants an optimal matched solution can be designed for each application, using the most efficient components from the market. The following example shows a low loss solution, patented by STADT AS (Norway).

In many cases a combination of an E-propulsion motor, running on two constants speeds (medium, high) and a controllable pitch propeller (CPP) gives a high reliable and compact solution.

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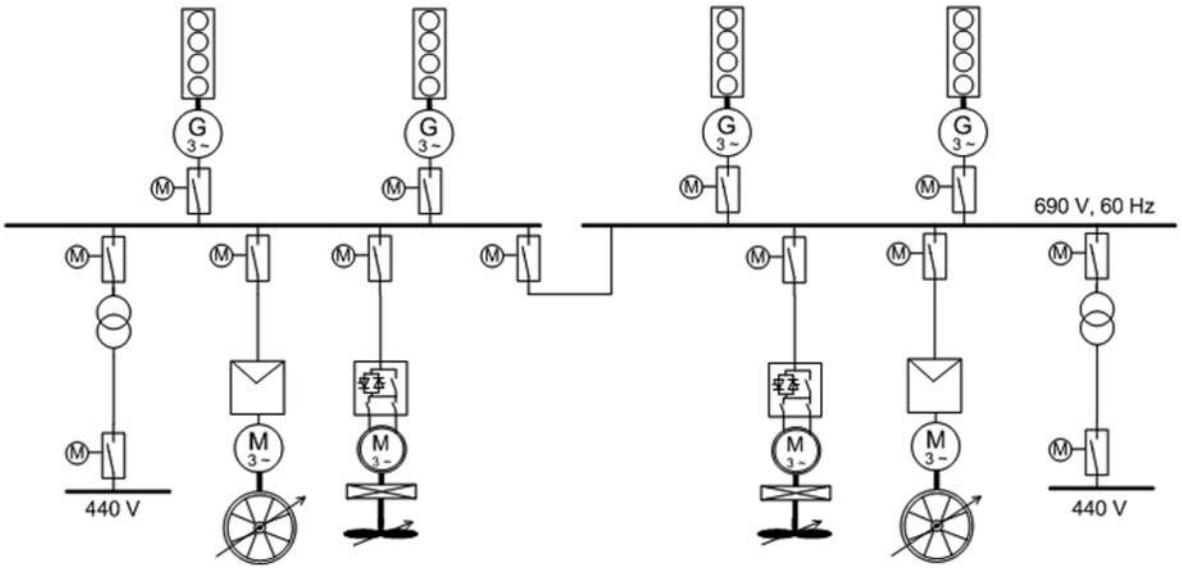

Figure 195: Example: Electric propulsion configuration of a RoRo, twin screw, geared transmission

Type of converter/drive	Supply transformer	Type of E-motor	Pros & cons
Sinusoidal drive (patented by STADT AS)	-	Induction (two speeds)	+ Highly reliable & compact + Very low losses + Transformer less solution + Low THD (no THD filters required) – Only applicable with a CP propeller

Table 234: Main DE-components of a low loss application (patented by STADT AS)

7.10 High-efficient electric propulsion plants with variable speed GenSets (EPROX-AC)

Variable speed power generation can provide significant fuel savings with electric propulsion when the operational profile of the vessel has a lot of variation in speed and power demand. Recent developments in the electric plant show solutions for adjusting the rotational speed of the main diesel engines, which allow the system frequency to vary within a range of 48 – 60 Hz. The main power system of the ship is specially engineered for variable frequency. Distribution for the auxiliary and hotel loads is provided by frequency convert especially when the total power demand is in range, where a medium voltage solution is required such a system concept can be applied.

The shift from constant speed to variable speed diesel engine operation results in a high fuel-oil efficiency at each system load. Such a solution enables a decoupled operation of diesel engines, propulsion drives and the consumers, as the power sources and hotel loads can be controlled and optimised independently.

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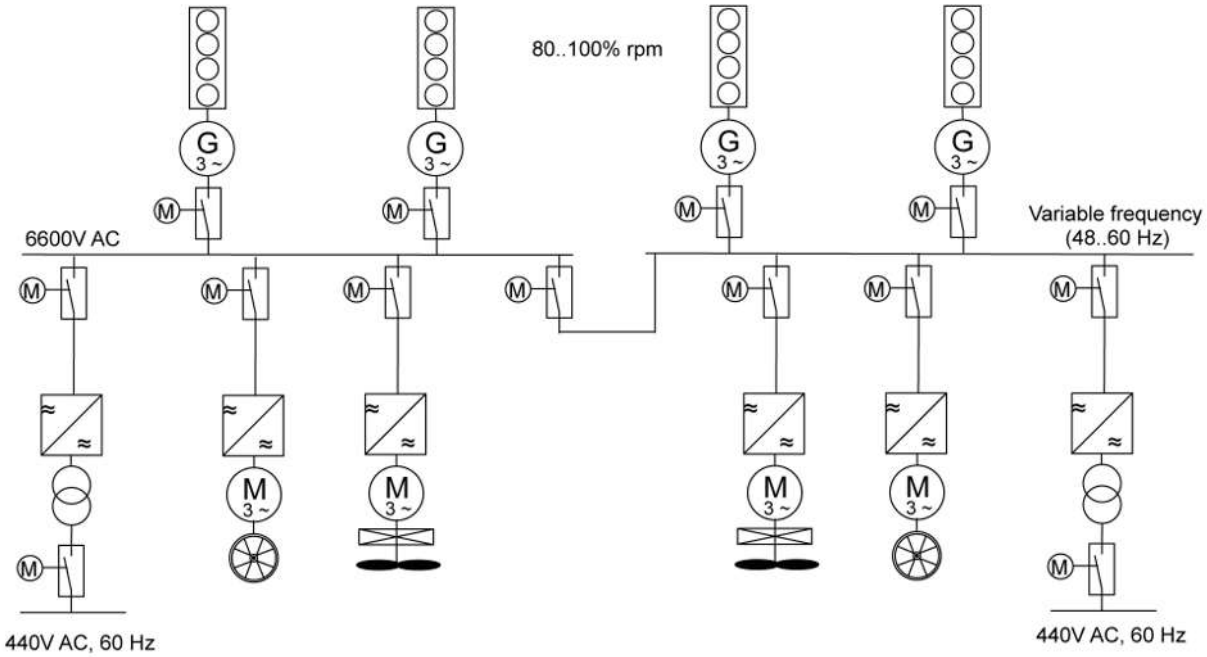

Figure 196: Example: High-efficient electric propulsion plant based on a main switchboard operated with variable frequency

Constant speed operation for the GenSets is no longer a constraint. When the main engines run at constant rpm, fuel efficiency is compromised. Utilising an enlarged engine operation map with a speed range of 80 % to 100 % paves the way to a high potential in fuel oil saving. According to the total system load all diesel engines can operate at a common floating speed set point.

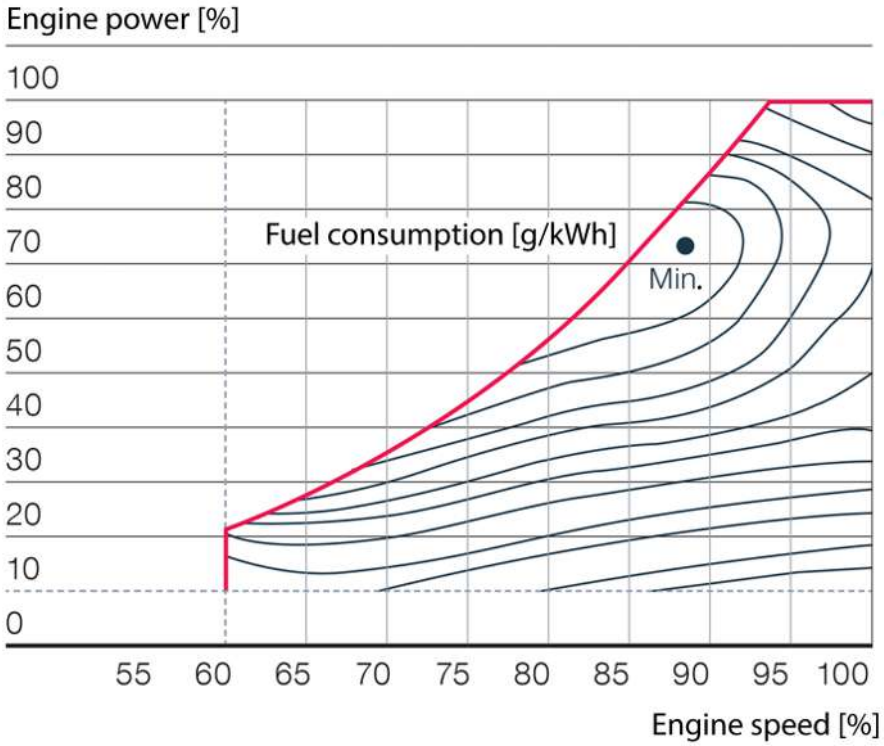

Figure 197: Typical SFOC map for a four-stroke medium-speed diesel engine (for illustration purpose only)

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The efficiency of the system can be even increased when energy storage devices, like batteries, are integrated. They can reduce the transient loads on the engines, improve the dynamic system response and the maneuverability of the propulsion system and absorb rapid power fluctuations from the vessel ´s grid. Fast load applications are removed from the engines and peak loads are shaved.

It is also beneficial to run the engines always on high loads, where the specific fuel oil consumption is lowest. This degree of freedom can be utilised and surplus power can charge the batteries. If less power is required, one engine can be shut down, with the remaining ones running still with a high loading, supported by power of the batteries.

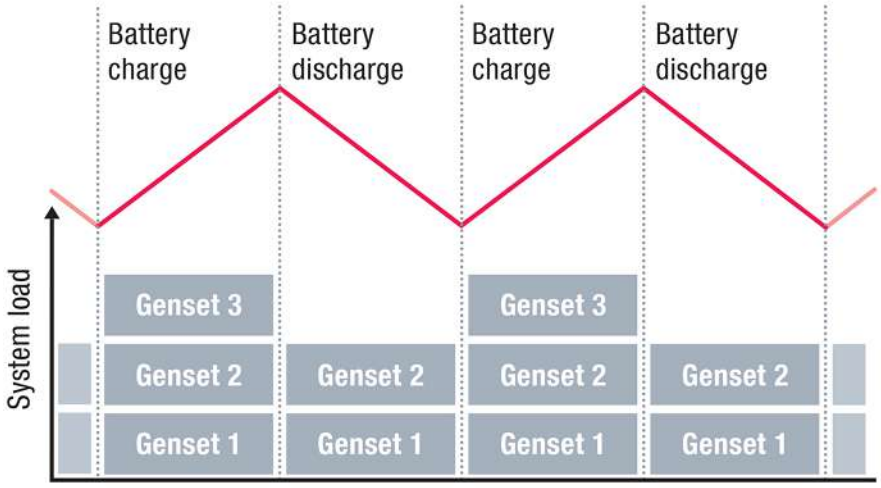

Figure 198: Batteries enable the diesel engines to operate at a high loading respectively with low specific fuel oil consumption

7.11 Fuel-saving hybrid propulsion system (HyProp ECO)

For many applications a hybrid propulsion system is a good choice, especially when flexibility, performance and efficiency are required. With HyProp ECO a system solution has been developed, which combines a diesel engine and an electric machine in a smart manner.

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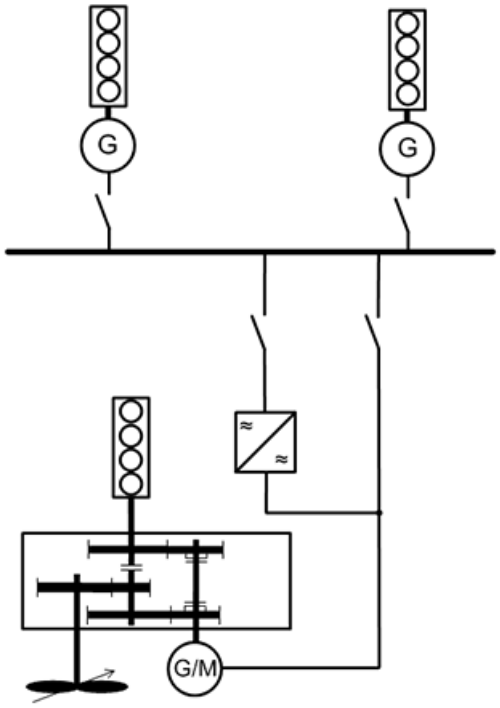

Figure 199: Principal layout of a HyProp ECO propulsion system

Beside the main diesel engine, the auxiliary GenSets, a 2-step reduction gearbox and the CP propeller a reversible electric machine, a frequency converter and a by-pass are the key components of the system. With this many operation modes can be achieved. When operating the system via the by-pass the normal PTO and PTI-boosting modes can be applied without any losses in the transmission line to/from the main switchboard. Utilising the frequency converter is done for two different purposes. Either it is used for starting-up the electric machine as emergency propulsion motor (PTH) in case the main engine is off. Usually the 2nd step in the gearbox is then used. Or the converter is of a bi-directional type and the propeller can be operated very efficiently at combinator mode with the PTO running in parallel with the auxiliary GenSets with a constant voltage and frequency towards the main switchboard. In this mode the converter can also be used for electric propulsion as variable speed drive for the propeller.

The major advantage of HyProp ECO is that costly components, like the frequency converter can be designed small. A typical figure for its size is 30 % of the installed alternator/motor power as for almost all modes, where the converter is involved, the required power is much lower compared to a design for pure PTO/PTI purposes. Therefore HyProp ECO combines lowest investment with optimised performance.

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8 Annex

8.1 Safety instructions and necessary safety measures

The following list of basic safety instructions, in combination with further engine documentation like user manual and working instructions, should ensure a safe handling of the engine. Due to variations between specific plants, this list does not claim to be complete and may vary with regard to project-specific requirements.

8.1.1 General

There are risks at the interfaces of the engine, which have to be eliminated or minimised in the context of integrating the engine into the plant system. Responsible for this is the legal person which is responsible for the integration of the engine.

Following prerequisites need to be fulfilled:

- Layout, calculation, design and execution of the plant have to be state of the art.
- All relevant classification rules, regulations and laws are considered, evaluated and are included in the system planning.
- The project-specific requirements of MAN Energy Solutions regarding the engine and its connection to the plant are implemented.
- In principle, the more stringent requirements of a specific document is applied if its relevance is given for the plant.

8.1.2 Safety equipment and measures provided by plant-side

- Proper execution of the work

  Generally, it is necessary to ensure that all work is properly done according to the task trained and qualified personnel.

  All tools and equipment must be provided to ensure adequate accesible and safe execution of works in all life cycles of the plant.

  Special attention must be paid to the execution of the electrical equipment. By selection of suitable specialised companies and personnel, it has to be ensured that a faulty feeding of media, electric voltage and electric currents will be avoided.
- Fire protection

  A fire protection concept for the plant needs to be executed. All from safety considerations resulting necessary measures must be implemented. The specific remaining risks, e.g. the escape of flammable media from leaking connections, must be considered.

  Generally, any ignition sources, such as smoking or open fire in the maintenance and protection area of the engine is prohibited.

  Smoke detection systems and fire alarm systems have to be installed and in operation.
- Electrical safety

  Standards and legislations for electrical safety have to be followed. Suitable measures must be taken to avoid electrical short circuit, lethal electric shocks and plant specific topics as static charging of the piping through the media flow itself.

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- Noise and vibration protection

  The noise emission of the engine must be considered early in the planning and design phase. A soundproofing or noise encapsulation could be necessary. The foundation must be suitable to withstand the engine vibration and torque fluctuations. The engine vibration may also have an impact on installations in the surrounding of the engine, as galleries for maintenance next to the engine. Vibrations act on the human body and may dependent on strength, frequency and duration harm health.

- Thermal hazards

  In workspaces and traffic areas hot surfaces must be isolated or covered, so that the surface temperatures comply with the limits by standards or legislations.

- Composition of the ground

  The ground, workspace, transport/traffic routes and storage areas have to be designed according to the physical and chemical characteristics of the excipients and supplies used in the plant.

  Safe work for maintenance and operational staff must always be possible.

- Adequate lighting

  Light sources for an adequate and sufficient lighting must be provided by plant-side. The current guidelines should be followed (100 Lux is recommended, see also DIN EN 1679-1).

- Working platforms/scaffolds

  For work on the engine working platforms/scaffolds must be provided and further safety precautions must be taken into consideration. Among other things, it must be possible to work secured by safety belts. Corresponding lifting points/devices have to be provided.

- Setting up storage areas

  Throughout the plant, suitable storage areas have to be determined for stabling of components and tools.

  It is important to ensure stability, carrying capacity and accessibility. The quality structure of the ground has to be considered (slip resistance, resistance against residual liquids of the stored components, consideration of the transport and traffic routes).

- Engine room ventilation

  An effective ventilation system has to be provided in the engine room to avoid endangering by contact or by inhalation of fluids, gases, vapours and dusts which could have harmful, toxic, corrosive and/or acid effects.

- Venting of crankcase and turbocharger

  The gases/vapours originating from crankcase and turbocharger are ignitable. It must be ensured that the gases/vapours will not be ignited by external sources. For multi-engine plants, each engine has to be ventilated separately. The engine ventilation of different engines must not be connected.

  In case of an installed suction system, it has to be ensured that it will not be stopped until at least 20 minutes after engine shutdown.

- Intake air filtering

  In case air intake is realised through piping and not by means of the turbocharger´s intake silencer, appropriate measures for air filtering must be provided. It must be ensured that particles exceeding 5 µm will be restrained by an air filtration system.

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- Quality of the intake air

  It has to be ensured that combustible media will not be sucked in by the engine.

  Intake air quality according to the section Specification of intake air (combustion air), Page 292 has to be guaranteed.

- Emergency stop system

  The emergency stop system requires special care during planning, realisation, commissioning and testing at site to avoid dangerous operating conditions. The assessment of the effects on other system components caused by an emergency stop of the engine must be carried out by plant-side.

- Fail-safe 24 V power supply

  Because engine control, alarm system and safety system are connected to a 24 V power supply this part of the plant has to be designed fail-safe to ensure a regular engine operation.

- Hazards by rotating parts/shafts

  Contact with rotating parts must be excluded by plant-side (e.g. free shaft end, flywheel, coupling).

- Safeguarding of the surrounding area of the flywheel

  The entire area of the flywheel has to be safeguarded by plant-side.

  Special care must be taken, inter alia, to prevent from: Ejection of parts, contact with moving machine parts and falling into the flywheel area.

- Securing of the engine´s turning gear

  The turning gear has to be equipped with an optical and acoustic warning device. When the turning gear is first activated, there has to be a certain delay between the emission of the warning device's signals and the start of the turning gear. The gear wheel of the turning gear has to be covered. The turning gear should be equipped with a remote control, allowing optimal positioning of the operator, overlooking the entire hazard area (a cable of approximately 20 m length is recommended). Unintentional engagement or start of the turning gear must be prevented reliably.

  It has to be prescribed in the form of a working instruction that:

  – The turning gear has to be operated by at least two persons.
  – The work area must be secured against unauthorised entry.
  – Only trained personnel is permissible to operate the turning gear.

- Securing of the starting air pipe

  To secure against unintentional restarting of the engine during maintenance work, a disconnection and depressurisation of the engine´s starting air system must be possible. A lockable starting air stop valve must be provided in the starting air pipe to the engine.

- Securing of the turbocharger rotor

  To secure against unintentional turning of the turbocharger rotor while maintenance work, it must be possible to prevent draught in the exhaust gas duct and, if necessary, to secure the rotor against rotation.

- Consideration of the blow-off zone of the crankcase cover´s relief valves

  During crankcase explosions, the resulting hot gases will be blown out of the crankcase through the relief valves. This must be considered in the overall planning.

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- Installation of flexible connections

  For installation of flexible connections follow strictly the information given in the planning and final documentation and the manufacturer manual.

  Flexible connections may be sensitive to corrosive media. For cleaning only adequate cleaning agents must be used (see manufacturer manual). Substances containing chlorine or other halogens are generally not permissible.

  Flexible connections have to be checked regularly and replaced after any damage or lifetime given in manufacturer manual.

- Connection of exhaust port of the turbocharger to the exhaust gas system of the plant

  The connection between the exhaust port of the turbocharger and the exhaust gas system of the plant has to be executed gas tight and must be equipped with a fire proof insulation.

  The surface temperature of the fire insulation must not exceed 220 °C.

  In workspaces and traffic areas, a suitable contact protection has to be provided whose surface temperature must not exceed 60 °C.

  The connection has to be equipped with compensators for longitudinal expansion and axis displacement in consideration of the occurring vibrations (the flange of the turbocharger reaches temperatures of up to 450 °C).

- Media systems

  The stated media system pressures must be complied. It must be possible to close off each plant-side media system from the engine and to depressurise these closed off pipings at the engine. Safety devices in case of system over pressure must be provided.

- Drainable supplies and excipients

  Supply system and excipient system must be drainable and must be secured against unintentional recommissioning (EN 1037). Sufficient ventilation at the filling, emptying and ventilation points must be ensured. The residual quantities which must be emptied have to be collected and disposed of properly.

- Spray guard has to be ensured for liquids possibly leaking from the flanges of the plant´s piping system. The emerging media must be drained off and collected safely.

- Charge air blow-off (if applied)

  The piping must be executed by plant-side and must be suitably isolated. In workspaces and traffic areas, a suitable contact protection has to be provided whose surface temperature must not exceed 60 °C.

  The compressed air is blown-off either outside the vessel or into the engine room. In both cases, installing a silencer after blow-off valve is recommended. If the blow-off valve is located upstream of the charge air cooler, air temperature can rise up to 200 °C. It is recommended to blow-off hot air outside the plant.

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- Signs
  - Following figure shows exemplarily the risks in the area of a combustion engine. This may vary slightly for the specific engine.

    This warning sign has to be mounted clearly visibly at the engine as well as at all entrances to the engine room.

Figure 200: Warning sign E11.48991-1108

  - Prohibited area signs.

    Depending on the application, it is possible that specific operating ranges of the engine must be prohibited.

    In these cases, the signs will be delivered together with the engine, which have to be mounted clearly visibly on places at the engine which allow intervention of the engine operation.

- Optical and acoustic warning device

  Communication in the engine room may be impaired by noise. Acoustic warning signals might not be heard. Therefore it is necessary to check where at the plant optical warning signals (e.g. flash lamp) should be provided.

  In any case, optical and acoustic warning devices are necessary while using the turning gear and while starting/stopping the engine.

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8.1.3 Provided by plant-side especially for gas-fueled engines

General

Definition of explosion zones within the plant must be provided by plant-side.

Note:
The engine is not designed for operation in hazardous areas. It has to be ensured by the ship's own systems, that the atmosphere of the engine room is monitored and in case of detecting a gas-containing atmosphere the engine will be stopped immediately.

Following safety equipment respectively safety measures must be provided by plant-side especially for gas-fueled engines

- Gas detectors in the engine room

In the engine room gas detectors for detection of gas leakages have to be installed. In case of a gas alarm triggered at a gas concentration widely below the lower explosion limit the engine has to be stopped and the power supply to the engines has to be switched off. The gas supply to the engine room must be immediately interrupted. Additionally it is necessary to switch off the power supply to all plant equipment, except the emergency equipment like engine room ventilation, gas alarm system, emergency lighting and devices etc. The emergency equipment has to be certified for application in explosion hazardous areas. It is necessary to connect the emergency equipment to an independent power supply in order to keep it in operation in case of a gas alarm.

To increase the availability of engine operation for dual fuel engines, it could be possible to switch the engine into the diesel mode at a very low gas concentration level. Dependent on the plant design it might be necessary to apply the same procedure for adjacent engines. In this case it is obligatory to shut off the gas supply to the engine room and to vent the gas piping in the engine room pressureless.

The leakage source shall be located and repaired by qualified staff using mobile gas detectors and special tools certified for using in explosion endangered areas.

- Earthing
  - Gas piping must be earthed in an appropriate manner.
  - The engine must be earthed in an appropriate manner.
- Explosion protection equipment at large volume exhaust system parts, e.g. exhaust silencer, exhaust gas boiler

  Due to the possibility that unburned gas penetrates the plant-side exhaust system parts, these must be equipped with explosion relief valves with integrated flame-arresters. The rupture discs must be monitored for example via wire break sensor. In case of bursting the engine has to be switched off.
- Deflagration protection of HT cooling water system, crankcase ventilation, gas valve unit

  Only in case of malfunctions in the engine´s combustion chamber area gas could be carry off to the high temperature cooling water circuit and would accumulate in the expansion tank. Therefore it is recommended to provide the high temperature cooling water system with deflagration protection.

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The crankcase ventilation pipe shall lead to a safe location outside the engine room, remote from any source of ignition. The end of the vent pipe has to be equipped with a flame arrester. The crankcase ventilation pipe may not be connected with any other ventilation pipes.

Note:

- In case of multi-engine plants the venting pipework has to be kept separately.
- All venting openings as well as open pipe ends are to be equipped with flame breakers and shall lead to a safe location outside the engine room remote from any source of ignition.
- Condensate trap overflows are to be connected via siphon to drain pipe.
- Specific requirements of the classification societies are to be strictly observed.

- The crankcase vent line must lead to the outside and must keep always sufficient distance to hot surfaces. The equipment installed in the crankcase venting line has to be classified for application in explosion hazardous areas.

  For more details see also project related documentation.

- Blower for venting the exhaust gas duct

  The exhaust system of gas/dual fuel engine installations needs to be ventilated after an engine stop or emergency shut down or prior to the engine start as well as maintenance. The exhaust system of gas engine installations in addition must also be ventilated during engine start. Therefore a suitable blower has to be provided, which blows in fresh air into the exhaust gas duct after turbocharger and compensator. The blower has to be classified for application in explosion hazardous areas (For more details see also project related documentation). Air demand (project-specific) for purging > 3 x exhaust system volume. The engine automation system provides an interface for the control of the exhaust blower.

- Absolutely safe and reliable gas shut-off device (gas blocking valve with automatic leak testing system and leakage line leading to the outside).

- Scavenging line with flame arrestors leading to the outside, so for maintenance the gas system can be kept free of gas, during commissioning the system can be vented and in case of emergency stop or switching to diesel-mode (dual fuel engine) existing gas can be blown out.

- Engine room ventilation

  An effective ventilation system has to be provided. The minimum air exchange rate shall be defined according to state of the art as required by European and/or local regulations. It might be necessary to design the engine room ventilation system explosion proof and to connect it to an independent power supply in order to keep it in operation in case of a gas alarm. To avoid the returning of exhaust air out of the ventilation outlets to the engine room, the ventilation outlets shall not be located near to the inlet/outlet openings of suction lines, exhaust gas ducts, gas venting lines or crankcase vent lines.

  Engine operation in a room without an effective ventilation or during the ventilation system is not available is strictly forbidden. This must be realised by the plant-side control systems or by other suitable measures (engine auto shut down respectively engine start blocking).

- Intake air

  The air intakes must be connected to ducts leading out of the engine room, if possible leading to the open air.

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The intakes of combustion air and the outlets of exhaust gas, crankcase and gas vent must be arranged in a way that a suction of exhaust gas, gas leakage as well as any other explosion endangered atmospheres will be avoided. The intake lines of different engines must not be connected together. Each engine must have its own intake ducts, completely separated from other engines.

- Lube oil system engine

  The lube oil can carry off gas into the lube oil system. Required measures must be taken according to Machinery Directive 2006/42/EG.

- HT cooling water system

  Only in case of malfunctions in the engine´s combustion chamber area gas could be carry off to the HT cooling water system and forms an explosion endangered atmosphere in the plant system.

Additional note:
All safety equipment has to be checked after installation/reinstallation and maintenance to ensure proper operation. This includes leakage tests, which shall be carried out according to the needs of each facility.

8.2 Programme for Factory Acceptance Test (FAT)

See overleaf!

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According to MAN ES instruction I0189EN

	Test points	Factory Acceptance Test (FAT)			
		Pre- Tests		Demonstration tests	
Diesel engines	100 %	-		60min*	
	110 %	-		M	
	85% (nominal continuous cruise power)**	-		M	
	Minimum speed at full constant torque - (mechanical pump drive only)	-		M	
	75%	M		-	
	50%	M		-	
	25%	M		-	
	Idle*** (only engines driving generators)	M		-	
Dual Fuel (DF-) engines		Gas mode	Diesel mode	Gas mode	Diesel mode
	100%	-	-	30min*	30min*
	110%	-	-	M	M
	85% (nominal continuous cruise power)**	-	-	M	M
	75%	M	M	-	-
	50%	M	M	-	-
	25%	-	M	-	-
	Idle*** (only engines driving generators)	-	M	-	-

* 2 readings have to be done at an interval of 30 min. On DF-engines only one reading in Diesel and one in Gas-mode.

** Replaces the 90% load point of classification rules.

M = Minimum 15 minutes and steady-state conditions reached (acc. Instruction conducting a measurement of MAN-ES)

Idle*** Nominal engine speed

Figure 201: Engine performance check – Table 1

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Content of MAN ES instruction I0189EN

1 Purpose

This instruction specifies tests and checks at Factory Acceptance Test (FAT) of marine engines produced by MAN ES. The following tests and checks are based on the rules and regulations of the classification societies as well as the ISO standard 3046 and 15550 in its version when this instruction has been published and have to be done to fulfill the requirements of a standard FAT.

2 Scope

This instruction is valid for all employees of companies and business units mentioned below, and is binding for all employees, which are affected by this instruction within the scope of their duties. The superior has to ensure that the employees know and observe the determinations of this instruction. It is valid for Marine application of Medium-, and high speed engines with nominal speed up to 1000rpm from MAN ES

Valid for the following companies: MAN Energy Solutions SE

Valid for the following locations: Augsburg; Frederikshavn; Saint-Nazaire; Aurangabad

Valid for the following business units: SBU E

Valid for the following departments: PEAA; PECFS; IN-ES; FR-EE

3 Terms and definitions

Term	Definition
FAT	Factory Acceptance Test
CPP	Controllable Pitch Propeller
FPP	Fixed Pitch Propeller
IACS	International Association of Classification Societies

4 Engine testing

The complete test of a medium speed engine on test bed needs several days. Therefore, the test is separated in two periods, the pre-test and the demonstration test. The pre-test is an internal test, which is partially to be done in presence of the classification society if required. The classification society has to be informed about pre-test date in time. The demonstration test is to prove the engine quality and observance of contracts in presence of classification society and customer. For each engine, the FAT procedure for the demonstration test has to be generated based on the tests and checks under point 4.2 of this instruction and contractual agreement with the customer. Additional test requests of classification society have to be considered in the FAT Procedure. Under the following points the standard scope of pre- test and demonstration test is specified.

4.1 Pre- Tests (internal tests)

The pre-test is divided in tests and checks to be done prior first engine start and checks during engine operation. Main focus is to detect and fix quality issues and adjusting the engine before starting the demonstration tests.

4.1.1 Main tests and checks before first engine fire

- check crank web deflection after alignment of the engine (results to be stated in FAT protocol)
- flushing of all media systems (lube oil, fuel oil, water)

Figure 202: Engine performance check – Part 1

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- visual check of engine including cam- and crank shaft housing (before first engine start and after running in procedure)
- test of alarm and safety system. The requirements of classification society and engine manufacturer specification (check of safety functions F0387EN) have to be fulfilled. The test results have to be stated in the FAT protocol.

4.1.2 Function- and performance check

Limits and requirements for each test can be found in project guides, rules of classification societies and internal specifications of MAN-ES. The internal testing of engines is generally specified in the guideline for engine testing of MAN ES (F0387EN). The following points represent the minimum of tests to be done to ensure the functionality and performance of the engine.

- Starting tests and calculation of start air consumption in gas and diesel operation. To be done at least once per shipset for each accordingly engine type. The results have to be stated in each FAT protocol for reference.
- Thermographic inspection (according working instruction W0202EN - Surface temperature examination on engines and components for marine application) of engine insulation for confirmation of SOLAS surface temperature requirements (only one engine per ship set). Results to be stated in FAT-protocol only by request of classification society
- Noise level and vibration measurement according ISO 10816-6 and manufacturer Guidelines (only by contractual agreement). Results have to be stated in FAT protocol
- Integration test on one engine per ship test. Test program is engine type specific and appointed before engine test. Results to be stated in FAT protocol
- Performance Check: The engine performance is to be tested on the load points listed in table 1. Further information can be found under point 4.3 of this instruction. Performance readings have to be stated in FAT protocol
- Governor test (load drop). Rapid load drop from full load to zero load. Results have to be stated in FAT protocol
- Check of engine attached flaps and valves (blow off, blow by, waste gate, jet assist)
- Visual check for leakages on fuel oil, lube oil and water systems on the engine (check of connecting flanges, screw connections and inspection drillings)
- Surge test of turbocharger on new engine configurations during the turbocharger-matching process. Results not to be stated in FAT protocol.

4.2 Demonstration tests

The demonstration test is a final quality approval of the engine before its delivery in presence of the classification and the customer by request. The demonstration test has to be done according to the engine specific FAT-procedure (demonstration test), which has to be sent out to customer and classification society one week prior demonstration test date. The scope of the FAT-procedure (demonstration test) is based on the following points:

1) One change over from diesel to gas operation and from gas to diesel operation at lowest and highest possible load (only on Dual Fuel engines)

2) One gas start (only on Dual fuel engines with gas start functionality)

3) Performance Check: The engine performance is to be tested on the load points listed in table 1. Further information can be found under point 4.3 of this instruction

4) Visual check for leakages on fuel oil, lube oil and water systems on the engine (check of connecting flanges, screw connections and inspection drillings)

5) Visual check of crank case, camshaft, rollers and gear drive after engine stop

6) Additional inspections only by contractual agreement with customer, request of classification society or by indication of irregularities of the operating values.

Figure 203: Engine performance check – Part 2

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7) Cylinder liner inspection (only engines ≥ 32cm liner diameter) of one cylinder liner per engine by borescope. In case of additional inspection (e.g. dismantling of piston) the borescope inspection is omitted. Evaluation of the cylinder liner surface according Q10.09121-3325 “Visual test Cylinder liner inspection after running-in”.

Optional tests and inspections only by request of classification society, contractual agreement with customer or by indication of irregularities.

4.3 Engine Performance Check

The engine performance check has to be done on all load points listed in table 1. Engines for generator application (incl. diesel-electric dredger), diesel-mechanic dredger appl. with propulsion function and CPP-application have to be tested on constant speed curve. Engines for diesel-mechanic propulsion application with fixed pitch propeller, mechanical pump drive only (dredger) and water jet application have to be tested on recommended FPP-curve of MAN-ES. Engines with ECOMAP function to be tested on standard full load map only. Additional mappings can be tested according contractual agreement with customer. Between pre-tests and demonstration tests minor adjustments for engine optimization are allowed. These adjustments must not lead to an excess of the engine specifications in any load point and have to be stated in the FAT-protocol. The performance parameter to be measured are specified in ISO 15550. For all measurement devices the calibration tolerances of this standard are to be fulfilled and the given calibration intervals of the specific manufacturers have to be considered. All performance data of the engines have to be within the specification limits (Documents: Quality criteria of engine) of the engine manufacturer. The performance readings have to be conducted according to the specified instructions of the engine manufacturer (Instruction I0158: Guideline for conducting a measurement at the test beds).

See Table 1: Engine Performance check

4.4 After engine operation

After finishing the test program, the engine has to be conserved according to working instruction W0176EN. The fuel delivery system has to be adjusted that overload power cannot be delivered on board on diesel-mechanic drives. Engines driving electrical generators have to be adjusted that 110% rated power is capable.

4.5 FAT - protocol

All operations, events, findings, measures and agreements have to be documented in one engine specific FAT protocol of MAN-ES. The final FAT-protocol has to be reviewed by the surveyor of the classification society and a copy has to be handed out to the customer. The FAT protocol contains the following informations:

- Cover sheet
- List of contents
- Signature sheet with representatives
- Operating record for all engine operation points
- Fuel oil analysis
- Line records for governor test or load jump tests
- Check of safety functions of engine and test bed before first start of the engine
- Visual inspection sheet with conducted checks and documentation of engine conditions
- Additional remarks
- Crank web deflection measurement results
- Software versions of engine control and safety system and emission relevant identification number
- Information about engine equipment

Figure 204: Engine performance check – Part 3

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- Starting test incl. air consumption calculation of reference engine
- NOx- emission trend line of reference engine
- Characteristic NOx emission Map (only for variable speed)
- Information about firing pressure indication measurement
- Operating data sheet to estimate the engine output
- Calibration reports of fuel oil and fuel gas measurement devices and power/torque measurement device

Figure 205: Engine performance check – Part 4

8.3 Engine running-in

Prerequisites

Engines require a running-in period in case one of the following conditions applies:

- When put into operation on site, if
  - after test run the pistons or bearings were dismantled for inspection or
  - the engine was partially or fully dismantled for transport.
- After fitting new drive train components, such as cylinder liners, pistons, piston rings, crankshaft bearings, big-end bearings and piston pin bearings.
- After the fitting of used bearing shells.
- After long-term low-load operation (> 500 operating hours).

Supplementary information

Operating Instructions

During the running-in procedure the unevenness of the piston-ring surfaces and cylinder contact surfaces is removed. The running-in period is completed once the first piston ring perfectly seals the combustion chamber. i.e. the first piston ring should show an evenly worn contact surface. If the engine is subjected to higher loads, prior to having been running-in, then the hot exhaust gases will pass between the piston rings and the contact surfaces of the cylinder. The oil film will be destroyed in such locations. The result is material damage (e.g. burn marks) on the contact surface of the piston rings and the cylinder liner. Later, this may result in increased engine wear and high lube oil consumption.

The time until the running-in procedure is completed is determined by the properties and quality of the surfaces of the cylinder liner, the quality of the fuel and lube oil, as well as by the load of the engine and speed. The running-in periods indicated in following figures may therefore only be regarded as approximate values.

Operating media

The running-in period may be carried out preferably using MGO (DMA, DFA) or MDO (DMB, DFB).

The fuel used must meet the quality standards see section Specification for engine supplies, Page 255 and the design of the fuel system.

For the running-in of gas four-stroke engines it is best to use the gas which is to be used later in operation.

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Dual fuel engines are run in using liquid fuel mode with the fuel intended as the pilot fuel.

Lube oil

The running-in lube oil must match the quality standards, with regard to the fuel quality.

Engine running-in

Checks

Inspections of the bearing temperature and crankcase must be conducted during the running-in period:

- The first inspection must take place after 10 minutes of operation at minimum speed.
- An inspection must take place after operation at full load respectively after operational output level has been reached.

The bearing temperatures (camshaft bearings, big-end and main bearings) must be determined in comparison with adjoining bearings. For this purpose an electrical sensor thermometer may be used as a measuring device.

At 85 % load and at 100 % load with nominal speed, the operating data (ignition pressures, exhaust gas temperatures, charge air pressures, etc.) must be measured and compared with the acceptance report.

Standard running-in programme

Dependent on the application the running-in programme can be derived from the figures in paragraph Diagram(s) of standard running-in, Page 521. During the entire running-in period, the engine output has to be within the marked output range. Critical speed ranges are thus avoided.

Running-in during commissioning on site

Most four-stroke engines are subjected to a test run at the manufacturer´s premises. As such, the engine has usually been run in. Nonetheless, after installation in the final location, another running-in period is required if the pistons or bearings were disassembled for inspection after the test run, or if the engine was partially or fully disassembled for transport.

Running-in after fitting new drive train components

If during revision work the cylinder liners, pistons, or piston rings are replaced, a new running-in period is required. A running-in period is also required if the piston rings are replaced in only one piston. The running-in period must be conducted according to following figures or according to the associated explanations.

The cylinder liner may be re-honed according to working instructions 050.05, if it is not replaced. A transportable honing machine may be requested from one of our service and support locations.

Running-in after refitting used or new bearing shells (crankshaft, connecting rod and piston pin bearings)

When used bearing shells are reused, or when new bearing shells are installed, these bearings have to be run in. The running-in period should be 3 to 5 hours under progressive loads, applied in stages. The instructions in the preceding text segments, particularly the ones regarding the "Inspections", and following figures must be observed.

Idling at higher speeds for long periods of operation should be avoided if at all possible.

Running-in after low-load operation

Continuous operation in the low-load range may result in substantial internal pollution of the engine. Residue from fuel and lube oil combustion may cause deposits on the top-land ring of the piston exposed to combustion, in the piston ring channels as well as in the inlet channels. Moreover, it is possible that the charge air and exhaust pipes, the charge air cooler, the turbocharger and the exhaust gas tank may be polluted with oil.

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Since the piston rings have adapted themselves to the cylinder liner according to the running load, increased wear resulting from quick acceleration and possibly with other engine trouble (leaking piston rings, piston wear) should be expected.

Therefore, after a longer period of low-load operation (≥ 500 hours of operation) a running-in period should be performed again, depending on the power, according to following figures.

Also for instruction see section Low-load operation, Page 65.

Note:
For further information, you may contact the MAN Energy Solutions customer service or the customer service of the licensee.

Diagrams of standard running-in

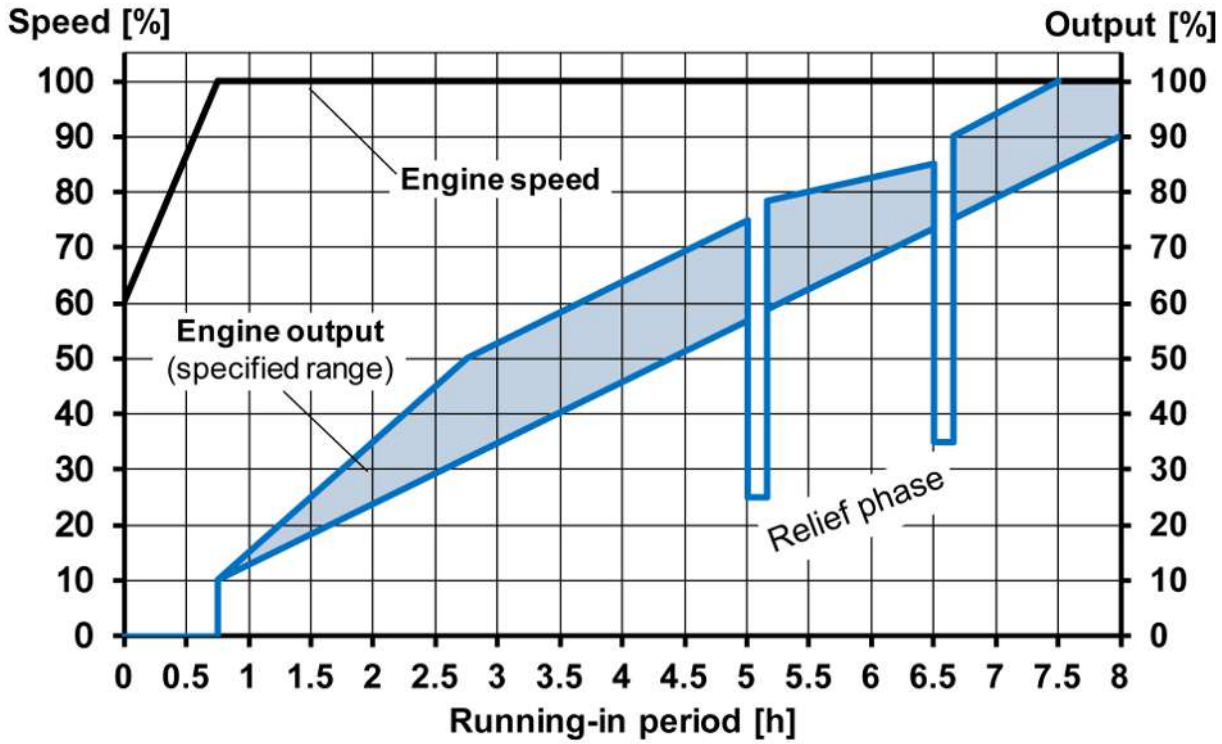

Figure 206: Standard running-in programme for engines operated with constant speed

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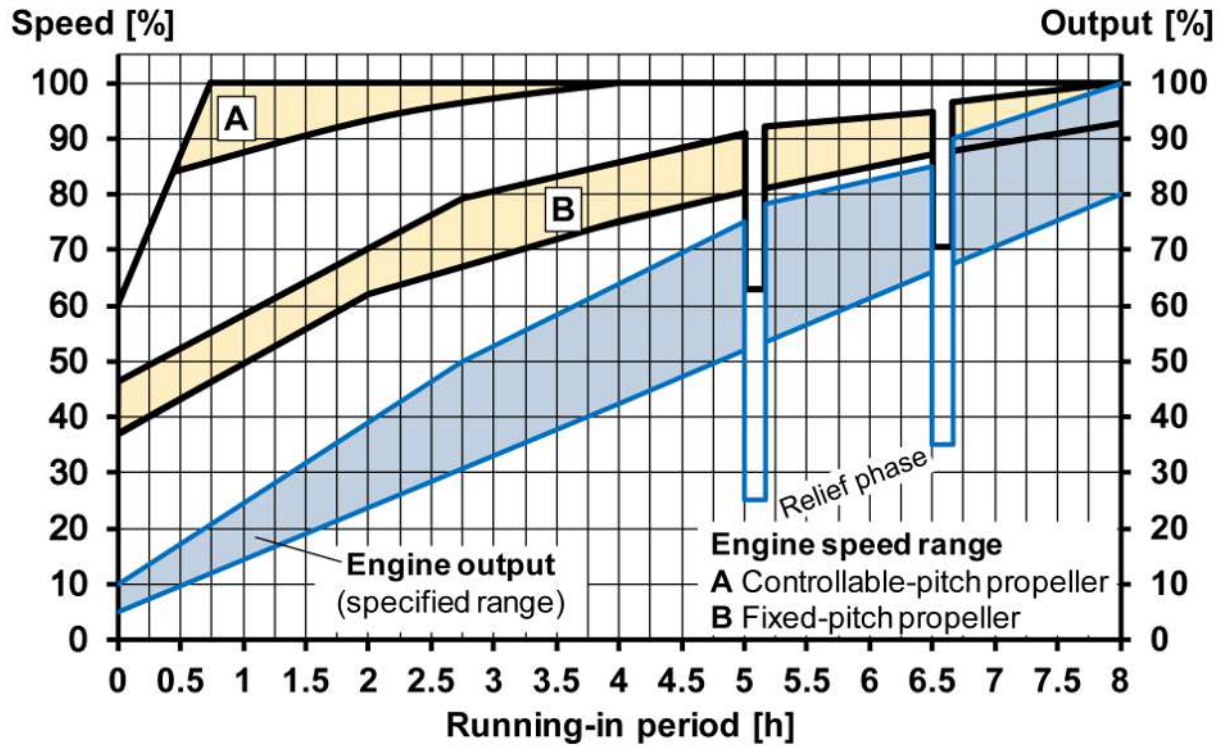

Figure 207: Standard running-in programme for marine engines (variable speed)

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8.4 Definitions

Auxiliary GenSet/auxiliary generator operation

A generator is driven by the engine, hereby the engine is operated at constant speed. The generator supplies the electrical power not for the main drive, but for supply systems of the vessel.

Load profile with focus between 40 % and 80 % load.

Engine´s certification for compliance with the NO_x limits according D2 Test cycle. See within section Engine ratings (output) for different applications, Page 43 if the engine is released for this kind of application and the corresponding available output $P_{Application}$.

Blackout

The classification societies define blackout on board ships as a loss of the main source of electrical power resulting in the main and auxiliary machinery to be out of operation and at the same time all necessary alternative energies (e.g. start air, battery electricity) for starting the engines are available.

Dead ship condition

The classification societies define dead ship condition as follows:

- The main propulsion plant, boilers and auxiliary machinery are not in operation due to the loss of the main source of electrical power.
- In restoring propulsion, the stored energy for starting the propulsion plant, the main source of electrical power and other essential auxiliary machinery is assumed not to be available.
- It is assumed that means are available to start the emergency generators at all times. These are used to restore the propulsion.

Designation of engine sides

- Coupling side, CS

  The coupling side is the main engine output side and is the side to which the propeller, the alternator or other working machine is coupled.
- Free engine end/counter coupling side, CCS

  The free engine end is the front face of the engine opposite the coupling side.

Designation of cylinders

The cylinders are numbered in sequence, from the coupling side, 1, 2, 3 etc. In V engines, looking on the coupling side, the left hand bank of cylinders is designated A, and the right hand bank is designated B. Accordingly, the cylinders are referred to as A1-A2-A3 or B1-B2-B3, etc.

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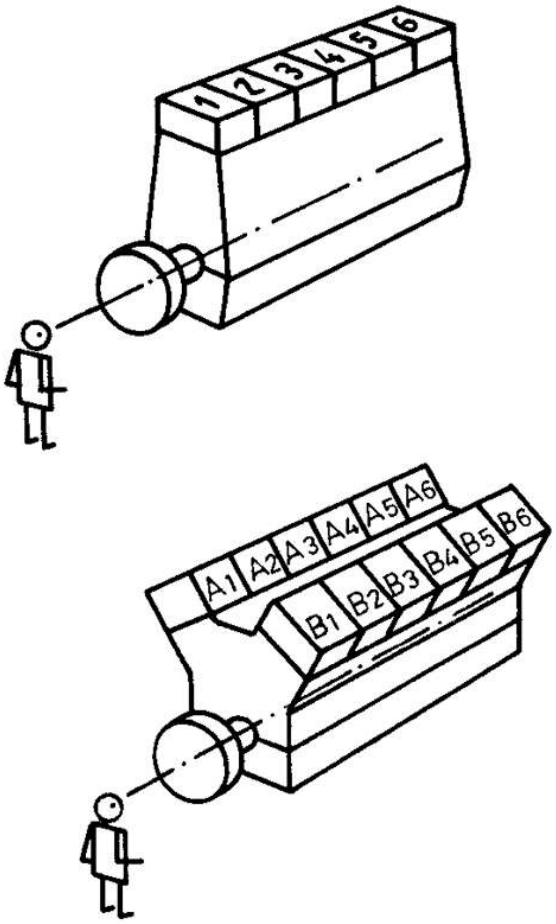

Figure 208: Designation of cylinders

Direction of rotation

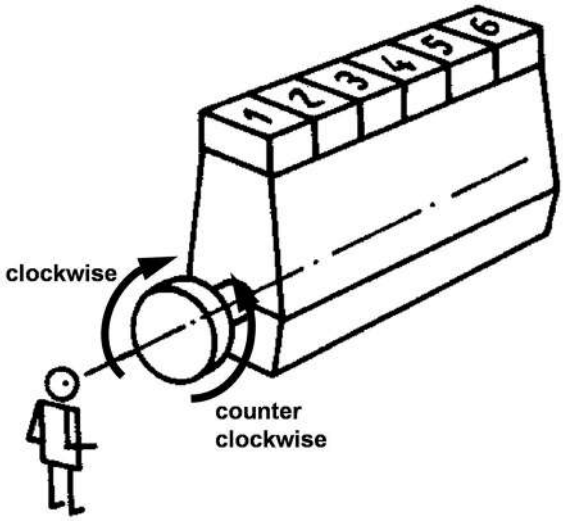

Figure 209: Designation: Direction of rotation seen from flywheel end

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Electric propulsion

The generator being driven by the engine supplies electrical power to drive an electric motor. The power of the electric motor is used to drive a controllable pitch or fixed pitch propeller, pods, thrusters, etc.

Load profile with focus between 80 % and 95 % load.

Engine´s certification for compliance with the NO_x limits according E2 Test cycle. See within section Engine ratings (output) for different applications, Page 43 if the engine is released for this kind of application and the corresponding available output $P_{Application}$.

GenSet

The term "GenSet" is used, if engine and electrical alternator are mounted together on a common base frame and form a single piece of equipment.

Gross calorific value (GCV)

This value supposes that the water of combustion is entirely condensed and that the heat contained in the water vapor is recovered.

Mechanical propulsion with controllable pitch propeller (CPP)

A propeller with adjustable blades is driven by the engine.

The CPP´s pitch can be adjusted to absorb all the power that the engine is capable of producing at nearly any rotational speed.

Load profile with focus between 80 % and 95 % load.

Engine´s certification for compliance with the NO_x limits according E2 Test cycle. See within section Engine ratings (output) for different applications, Page 43 if the engine is released for this kind of application and the corresponding available output $P_{Application}$.

Mechanical propulsion with fixed pitch propeller (FPP)

A fixed pitch propeller is driven by the engine. The FPP is always working very close to the theoretical propeller curve (power input ~ n^3). A higher torque in comparison to the CPP even at low rotational speed is present.

Load profile with focus between 80 % and 95 % load.

Engine´s certification for compliance with the NO_x limits according E3 Test cycle. See within section Engine ratings (output) for different applications, Page 43 if the engine is released for this kind of application and the corresponding available output $P_{Application}$.

Multi-engine propulsion plant

In a multi-engine propulsion plant at least two or more engines are available for propulsion.

Net calorific value (NCV)

This value supposes that the products of combustion contain the water vapor and that the heat in the water vapor is not recovered.

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Offshore application

Offshore construction and offshore drilling place high requirements regarding the engine´s acceleration and load application behaviour. Higher requirements exist also regarding the permissible engine´s inclination.

Due to the wide range of possible requirements such as flag state regulations, fire fighting items, redundancy, inclinations and dynamic positioning modes all project requirements need to be clarified at an early stage.

Output

- ISO standard output (as specified in DIN ISO 3046-1)

  Maximum continuous rating of the engine at nominal speed under ISO conditions, provided that maintenance is carried out as specified.
- Operating-standard-output (as specified in DIN ISO 3046-1)

  Maximum continuous rating of the engine at nominal speed taking in account the kind of application and the local ambient conditions, provided that maintenance is carried out as specified. For marine applications this is stated on the type plate of the engine.
- Fuel stop power (as specified in DIN ISO 3046-1)

  Fuel stop power defines the maximum rating of the engine theoretical possible, if the maximum possible fuel amount is used (blocking limit).
- Rated power (in accordance to rules of DNV)

  Maximum possible continuous power at rated speed and at defined ambient conditions, provided that maintenances carried out as specified.
- Output explanation

  Power of the engine at distinct speed and distinct torque.
- 100 % output

  100 % output is equal to the rated power only at rated speed. 100 % output of the engine can be reached at lower speed also if the torque is increased.
- Nominal output

  = rated power.
- MCR

  Maximum continuous rating.
- ECR

  Economic continuous rating = output of the engine with the lowest fuel consumption.

Overload power (at FAT or SAT/sea trial)

Only if required by rules of classification societies, it is admitted to operate the engine at 110 % of rated power for a maximum of 1 h in total as part of the FAT or SAT/sea trial and in addition a maximum of 1 h in total as part of the comissioning of the plant. Engine operation has to be done under supervision of trained MAN Energy Solutions personal.

Single-engine propulsion plant

In a single-engine propulsion plant only one single-engine is available for propulsion.

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Suction dredger application (mechanical drive of pumps)

For direct drive of a suction dredger pump by the engine via gear box the engine speed is directly influenced by the load on the suction pump.

The power demand of the dredge pump needs to be adapted to the operating range of the engine, particularly while start-up operation. Load profile with focus between 80 % and 100 % load.

Engine´s certification for compliance with the NO_x limits according C1 Test cycle. See within section Engine ratings (output) for different applications, Page 43 if the engine is released for this kind of application and the corresponding available output $P_{Application}$.

Water jet application

A marine propulsion system that creates a jet of water that propels the vessel. The water jet propulsion is always working close to the theoretical propeller curve (power input ~ n^3). With regard to its requirements the water jet propulsion is identical to the mechanical propulsion with FPP.

Load profile with focus between 80 % and 95 % load.

Engine´s certification for compliance with the NO_x limits according E3 Test cycle. See within section Engine ratings (output) for different applications, Page 43 if the engine is released for this kind of application and the corresponding available output $P_{Application}$.

Weight definitions for SCR

- Handling weight (reactor only):

  This is the "net weight" of the reactor without catalysts, relevant for transport, logistics, etc.
- Operational weight (with catalysts):

  That's the weight of the reactor in operation, that is equipped with a layer of catalyst and the second layer empty – as reserve.
- Maximum weight structurally:

  This is relevant for the static planning purposes maximum weight, that is equipped with two layers catalysts.

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8.5 Abbreviations

Abbreviation	Explanation
BN	Base number
CBM	Condition based maintenance
CCM	Crankcase monitoring system
CCS	Counter coupling side
CS	Coupling side
ECR	Economic continuous rating
EDS	Engine diagnostics system
GCV	Gross calorific value
GVU	Gas Valve Unit
HFO	Heavy fuel oil
HT CW	High temperature cooling water
LT CW	Low temperature cooling water
MCR	Maximum continuous rating
MDO	Marine diesel oil
MGO	Marine gas oil
MN	Methane number
NCV	Net calorific value
OMD	Oil mist detection
SaCoS	Safety and control system
SAT	Site acceptance test
SECA	Sulphur emission control area
SP	Sealed plunger
STC	Sequential turbocharging
TAN	Total acid number
TBO	Time between overhaul
TC	Turbocharger
TC	Temperature controller
ULSHFO	Ultra low sulphur heavy fuel oil

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8.6 Symbols

Note:
The symbols shown should only be seen as examples and can differ from the symbols in the diagrams.

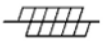
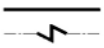
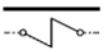
DESIGNATION	SYMBOL	CODE	DESIGNATION	SYMBOL	CODE
MAIN CIRCUIT (FLOW DIRECTION SHOWN)			FLEXIBLE PIPE CONNECTION		EH
SECONDARY CIRCUIT			EXPANSION BELLOWS (STEEL)		EB
CONTROL AIR PIPE, PULSE LINE, ELECTRICAL LINE			2/2 (TWO-PORT, TWO-POSITION) DIRECTIONAL CONTROL SOLENOID-OPERATED VALVE		
CAPILLARY TUBE (WITH THERMOSTATIC REGULATORS)			3/2 (THREE-PORT, TWO-POSITION) DIRECTIONAL CONTROL VALVE		
LAGGED PIPE			GENERAL SHUT-OFF VALVE		V
STEAM HEATED PIPE (HEAVY FUEL OIL OPERATION, MAIN CIRCUIT)			GATE VALVE		GV
ELECTRICALLY HEATED PIPE (HEAVY FUEL OIL OPERATION, MAIN CIRCUIT)			STRAIGHT-WAY VALVE		V

Figure 210: Symbols used in functional and pipeline diagrams 1

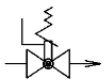
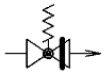
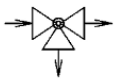
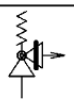
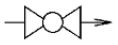
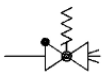
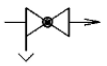
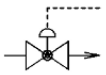
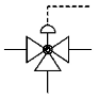
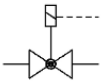
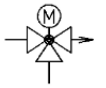
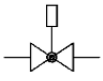
DESIGNATION	SYMBOL	CODE	DESIGNATION	SYMBOL	CODE
ANGLE VALVE		V	GENERAL SHUT-OFF VALVE, MOTOR DRIVEN		MOV
VALVE WITH MINIMUM FLOW		V	QUICK ACTING SHUT-OFF VALVE		V
GENERAL THREE-WAY VALVE		CK	SAFETY VALVE (STRAIGHT WAY)		PSV
THREE-WAY VALVE		V	SAFETY VALVE (ANGLE)		PSV
COCK		CK	BACK PRESSURE VALVE (STRAIGHT WAY)		BPV
SHUT-OFF VALVE WITH VENTILATION		V	BACK PRESSURE VALVE (ANGLE)		BPV
NON-RETURN VALVE		NRV	DIAPHRAGM VALVE (PNEUMATICALLY-OPERATED)		DV
NON-RETURN VALVE (CAN BE SHUT OFF)		NRV	DIAPHRAGM SHUTTLE VALVE		DV
ANGLE NON-RETURN VALVE		NRV	MOTORISED VALVE		MOV
SOLENOID-OPERATED VALVE		SOV	GENERAL THREE-WAY VALVE, MOTOR DRIVEN		MOV
SOLENOID-OPERATED VALVE (AUTOMATIC VENT)		SOV	MOTORISED SHUTTLE VALVE		MOV
GENERAL SHUT-OFF VALVE, GENERAL DRIVE		V	GENERAL SHUT-OFF VALVE WITH TRANSATORY MOTON VALVE		V

Figure 211: Symbols used in functional and pipeline diagrams 2

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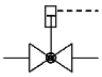
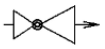
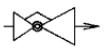
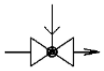
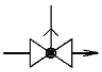
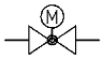
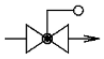
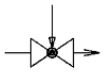
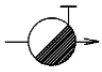
DESIGNATION	SYMBOL	CODE	DESIGNATION	SYMBOL	CODE
PISTON-OPERATED VALVE		POV	PRESSURE REDUCING VALVE		PCV
SHUTTLE VALVE WITH DRIVE PISTON		POV	PRESSURE CONTROL VALVE		PCV
SELF-CLOSING VALVE (STRAIGHT WAY)		QV	TEMPERATURE REGULATOR (SELF-ACTUATING DISTRIBUTION VALVE)		TCV
SELF-CLOSING VALVE (ANGLE)		QV	TEMPERATURE REGULATOR (SELF-ACTUIATING MIXING VALVE)		TCV
AUTOMATIC OPENING VALVE		V	TEMPERATURE CONTROL VALVE, ELECTRICALLY CONTROLLED		TCV
FLOAT VALVE		LOV	VANE HAND PUMP		P
SPRING-LOADED ADJUSTABLE PRESSURE LIMITING VALVE		PCV	WATER TRAP		TR
SAFETY VALVE		PSV	OIL TRAP		TR
FLOW-CONTROL VALVE, ADJUSTABLE			CONDENSATE TRAP, CAN BE SHUT OFF		TR
NON-RETURN VALVE, UNIDIRECTIONAL FLOW			ACCUMULATOR, GAS CYLINDER		
NON-RETURN VALVE WITH SPRING, NORMALLY CLOSED			OILER		
PILOTED NON-RETURN VALVE WITH SPRING, FLOW IN BOTH DIRECTIONS POSSIBLE DUE TO CONTROL PRESSURE			HEATING COIL (STEAM OR WATER)		H

Figure 212: Symbols used in functional and pipeline diagrams 3

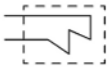
DESIGNATION	SYMBOL	CODE	DESIGNATION	SYMBOL	CODE
PLUG-IN HEATER (STEAM OR WATER)		H	SUCTION STRAINER		STR
PLUG-IN HEATER (ELECTRIC)		H	SUCTION BELL		SB
BACKFLOW PREVENTER (DISCOTYP)		NRV	FLAME TRAP		GS
NON-RETURN FLAP		NRF	VENT		
STRAINER		STR	VENT WITH FLAME TRAP		
FUNNEL (OPEN)		FU	VENT, OUTBOARD ABOVE DECK		
FUNNEL (CLOSED)		FU	VENT CAP		
OVERFLOW CHECK TANK, DISCHARGE FUNNEL		FU	Withdrawal point for fuel oil sample		

Figure 213: Symbols used in functional and pipeline diagrams 4

8.7 Preservation, packaging, storage

8.7.1 General

Introduction

Engines are internally and externally treated with preservation agent before delivery. The type of preservation and packaging must be adjusted to:

- Means of transport
- Type and period of storage

Improper storage may cause severe damage to the product.

Packaging and preservation of engine

The type of packaging depends on:

- The requirements imposed with transport and storage period
- Climatic and environmental effects
- Which preservative agents are used

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As standard, the preservation and packaging of an engine is designed for a storage period of 12 month and for sea transport.

Note:
The packaging must be protected against damage. It must only be removed when:

- A follow-up preservation is required
- The packaged material is to be used
- Shortly before operating the engine

The condition of the packaging must be checked regularly and repaired in case of damage. Especially a VCI packaging can only provide a proper corrosion protection if it is intact and completely closed.

In addition, the engine interiors are protected by vapour phase corrosion protection. Inner compartments must not be opened while transportation and storage. Otherwise, a re-preservation of the opened compartment will be required. The inner corrosion protection can remain inside the engine.

If bare metal surfaces get exposed for example, by disassembly of the coupling device, the unprotected metal must be treated with agent "f" according to the list of recommended anti-corrosion agents (https://www.man-es.com/documentation-/corrosion-protection).

This especially applies to the tie rod where the lifting device has been mounted.

In case of an installed intake air filter there is a steel plate cover or similar around the filter fleece, which has to be used during transportation and storage.

Note:
During storage and in case of a follow-up preservation the crankshaft must not be turned. If the crankshaft is turned, usually for the first time after preservation this will be done during commissioning, the preservation is partially removed. If the engine is to be stored again for a period thereafter, then adequate re-preservation is required.

Preservation and packaging of loose equipment

Unless stated otherwise in the customer specification, the preservation and packaging of loose equipment and engine parts which are dismantled for transport, must be carried out such that:

- The preservation and packaging of loose equipment and engine parts will not be damaged during transport
- The corrosion protection remains fully intact for at least 12 months when stored in a roofed dry room

Transport

Transport and packaging of the engine, loose equipment and engine parts must be coordinated.

After transportation, any damage to the corrosion protection and packaging must be rectified, and/or MAN Energy Solutions must be notified immediately.

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8.7.2 Storage location and duration

Storage location

Storage location of engine

As standard, the engine is packaged and preserved for outdoor storage.

The storage location must meet the following requirements:

- Engine is stored on firm and dry ground.
- Packaging material does not absorb any moisture from the ground.
- Engine is accessible for visual checks.

Storage location of loose equipment

Loose equipment must always be stored in a roofed dry room.

The storage location must meet the following requirements:

- Parts are protected against environmental effects and the elements.
- The room must be well ventilated.
- Parts are stored on firm and dry ground.
- Packaging material does not absorb any moisture from the ground.
- Parts cannot be damaged.
- Parts are accessible for visual inspection.
- An allocation of loose equipment to the order or requisition must be possible at all times.

Note:
Packaging made of or including VCI paper or VCI film must not be opened or must be closed immediately after opening.

Storage conditions

In general the following requirements must be met:

- Minimum ambient temperature: –10 °C
- Maximum ambient temperature: +60 °C

In case these conditions cannot be met, contact MAN Energy Solutions for clarification.

Storage period

The permissible storage period of 12 months must not be exceeded.

Before the maximum storage period is reached:

- Check the condition of the stored engine and loose equipment.
- Renew the preservation or install the engine or components at their intended location.

8.7.3 Follow-up preservation when preservation period is exceeded

A follow-up preservation must be performed before the maximum storage period has elapsed, i.e. generally after 12 months.

Request assistance by authorised personnel of MAN Energy Solutions.

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8.7.4 Removal of corrosion protection

Packaging, corrosion protection and silica gel must only be removed from the engine **immediately before commissioning** the engine in its installation location.

Remove outer protective layers, any foreign body from engine or component (VCI packs, blanking covers, etc.), check engine and components for damage and corrosion, perform corrective measures, if required.

The preservation agents sprayed inside the engine do not require any special attention. They will be washed off by engine oil during subsequent engine operation.

Contact MAN Energy Solutions if you have any questions.

8.8 Engine colour

Engine standard colour according RAL colour table is RAL 7040 Window grey. Other colours on request.

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